



**JOHN F. KENNEDY  
SPACE CENTER**

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Revision 1 - November 1, 1965

**ELECTRICAL REFERENCE HANDBOOK  
LAUNCH EQUIPMENT BRANCH  
ELECTRICAL SUPPORT EQUIPMENT  
LC - 39**

VOLUME I of II

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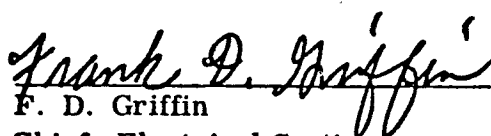
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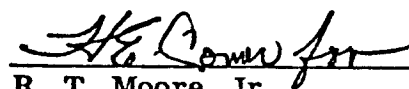
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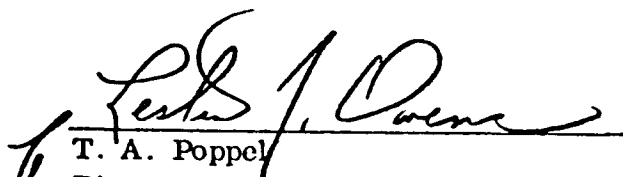
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APPROVAL  
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FOR  
LAUNCH EQUIPMENT BRANCH  
ELECTRICAL SUPPORT EQUIPMENT  
LC-39

  
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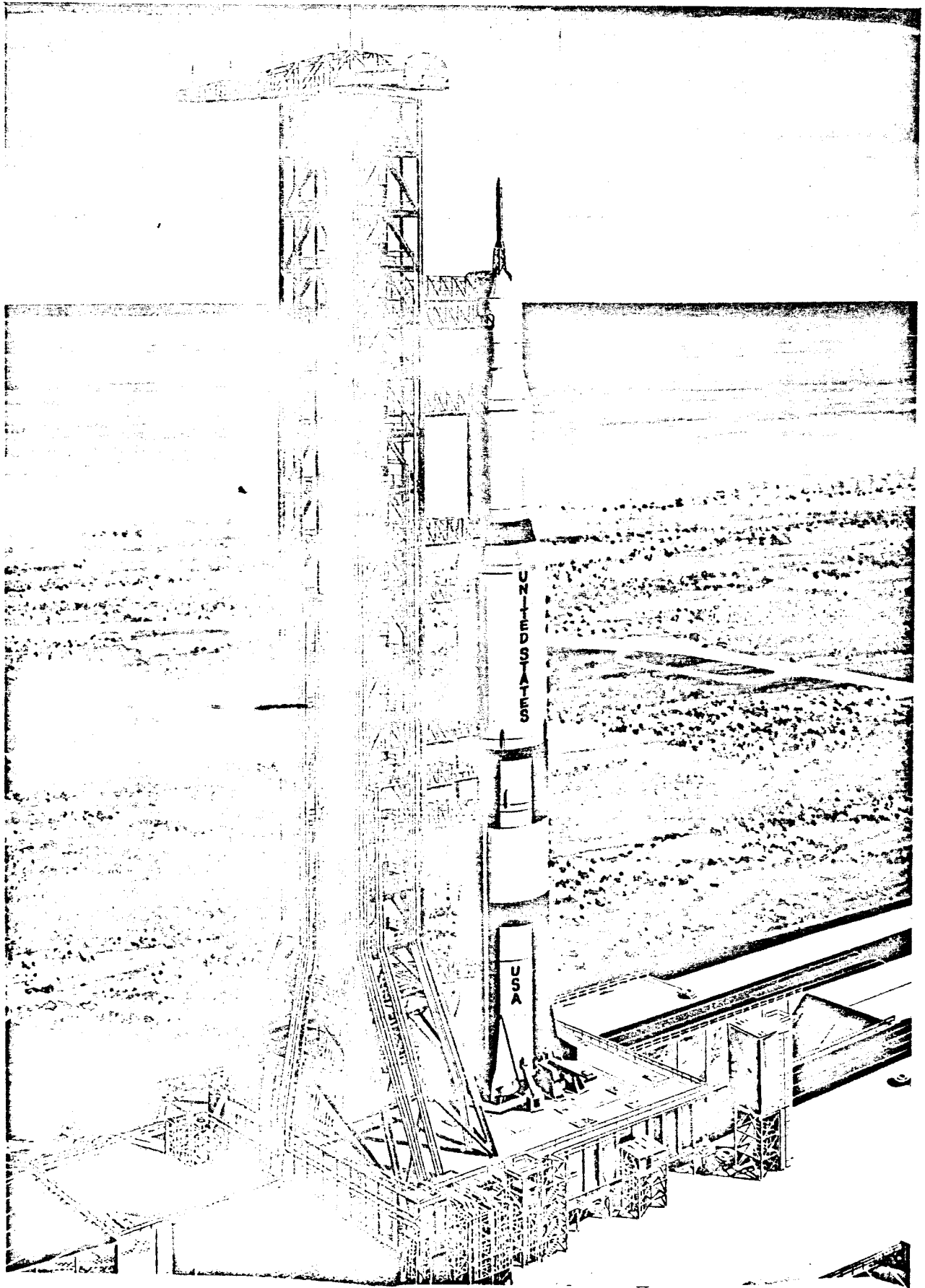
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SECTION I  
INTRODUCTION

1-1. SCOPE AND PURPOSE

This handbook describes that Electrical Support Equipment (ESE) which is the responsibility of the Launch Equipment Branch. The Launch Equipment Branch ESE comprises electrical subsystems which provide control, test, and monitoring of the service arms, command module access arm, tail service masts, pneumatic distribution panels, hydraulic charging and supply units, engine servicing platforms, Q-ball cover removal mechanism, and oxygen conditioning system. As the design of these systems is further refined, or as new systems are added to Launch Equipment Branch responsibility, this document will be revised and re-issued.

It should be noted that equipment also known as ESE is used in propellant loading, in certain vehicle-oriented systems on the Mobile Launcher, and in the integrated electrical networks for LC-39; no ESE except that named above is within the scope of this handbook.

In this document the ESE is described as an overall integrated system (section III), and again as individual panels and cabinets (section IV). It is also possible to determine the physical location of a particular item of Electrical Support Equipment installed in the Mobile Launcher or Launch Control Center. Detailed information about the equipments for which the ESE provides monitor and control, such as the tail service masts, may be obtained by reference to the document list, paragraph 1-2.

Section IV describes standard equipment such as panels, distributors, racks, test sets, and component modules employed throughout the ESE. These standard assemblies are the building blocks of present electrical ground networks and they should aid in the design of new systems.

Section V provides physical descriptions of the four ESE system test units. Their face panels and test functions are also given in section III in conjunction with the systems to be tested.

Reference material which applies to the entire ESE, such as the Mobile Launcher isometric drawing, the family tree of drawings, etc. is contained in section VI.

#### 1-2. RELATED DOCUMENTS.

The following documents offer detailed information on systems controlled and monitored by the ESE:

- a. Apollo Systems Description, Volume III  
NASA TM X-882
- b. Tail Service Mast Design Report  
TM-23-0-D  
Launch Equipment Branch  
Launch Support Equipment Engineering Division
- c. Saturn V Service Arms Preliminary Engineering Report  
TR-4-4-2-D  
Launch Equipment Branch  
Launch Support Equipment Engineering Division
- d. LC-39 L/UT GSE Installation  
SP-95-D  
Launch Equipment Branch  
Launch Support Equipment Engineering Division
- e. Saturn V Service Arms Functional Description  
Launch Equipment Branch  
Launch Support Equipment Engineering Division
- f. Standard Cables, Saturn I  
LTM-4-4
- g. Standard Electrical Components, GSE Saturn Vehicles  
LTM-4-11
- h. Data Booklet, Saturn V Service Arms and  
Related Equipment, Complex 39  
KSC SP-4-50-D  
Launch Equipment Branch  
Launch Support Equipment Engineering Division
- i. Information Handbook, Saturn V Vehicle Service  
Arms and Access Arm, Launch Complex 39  
SP-87-D  
Launch Equipment Branch  
Launch Support Equipment Engineering Division

SECTION II  
LAUNCH COMPLEX 39

2-1. GENERAL DESCRIPTION.

The Saturn/Apollo space vehicle (Saturn V) will be launched from NASA's Launch Complex 39. Owing to the complexity and size of the space vehicle and to the need for maximum reliability in the Apollo program, the launch procedure for LC-39 departs from previous ones as follows:

- a. Assembly and checkout of the space vehicle on its launcher, in the Vehicle Assembly Building (VAB) as shown in figure 2-1.
- b. Transfer of the assembled and checked space vehicle on its launcher to the launch pad, figures 2-2 and 2-3.
- c. Automatic checkout.
- d. Remote control of actual launch operations from a distant Launch Control Center (LCC).

As the Saturn V is assembled on its launcher in the VAB, all vehicle and launch support systems are remotely exercised and checked out from consoles in the Launch Control Center, which adjoins the VAB. After transfer to a launch pad the Mobile Launcher and Saturn V are again checked out from these consoles. Finally, all events at launch are sequenced by computers in the LCC and Mobile Launcher and are monitored from the LCC consoles.

One group of equipment thus controlled from the LCC is the Launch Equipment (see figure 3-6 ), which is utilized to mate the vehicle to the Mobile Launcher during installation, checkout, and launch operations. Among the Launch Equipment are the service arms, tail service masts, and their associated hydraulic, pneumatic, and electrical units. These systems are located aboard various levels of the Mobile Launcher tower and base.

Electrical control for them is provided by the Electrical Support Equipment (ESE), figures 2-4 and 2-5, which is the central topic of this handbook.

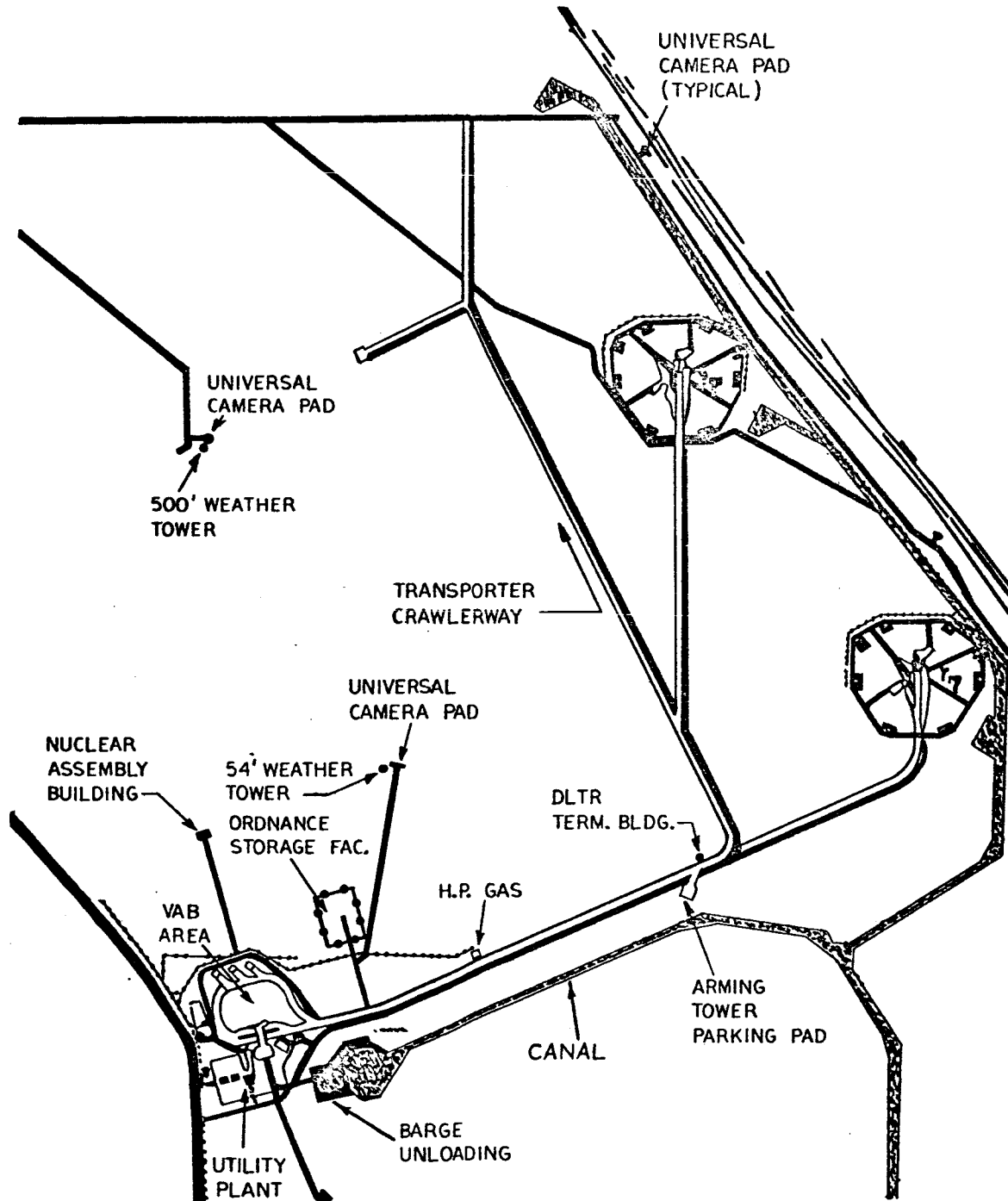


Figure 2-1 Launch Complex 39

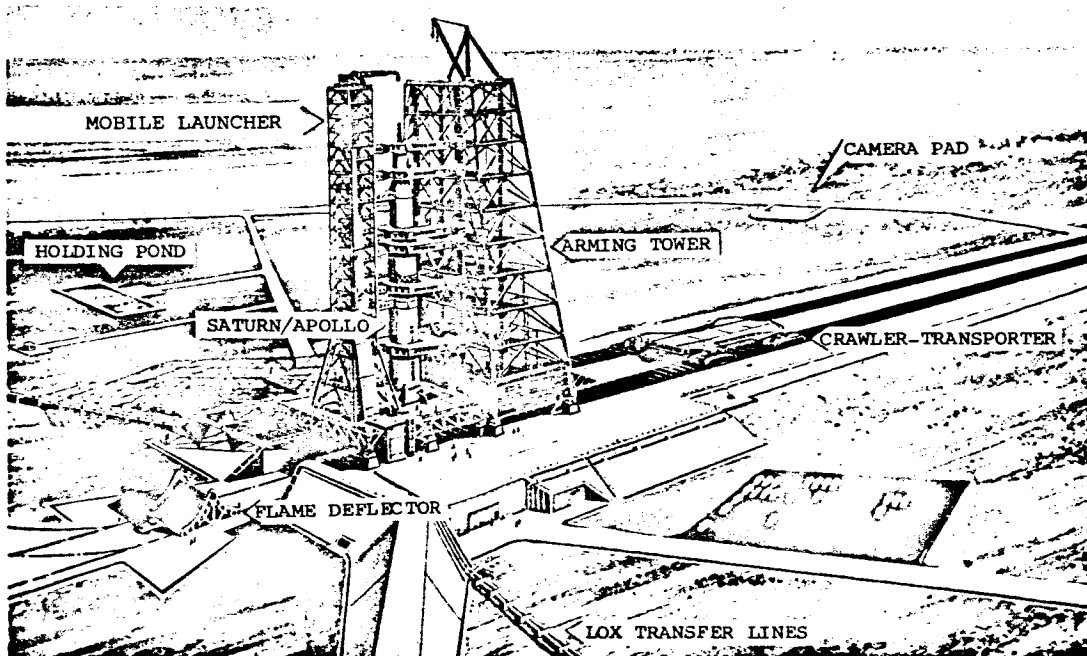
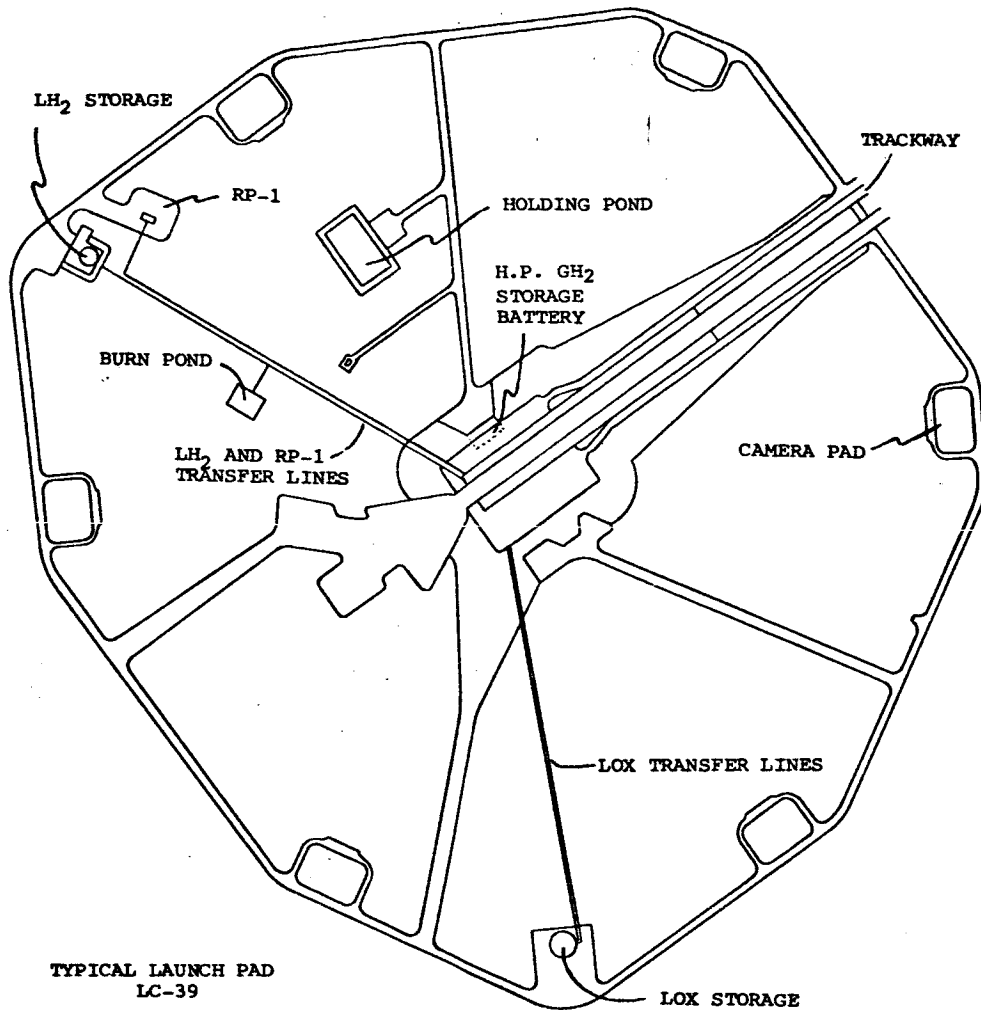


Figure 2-2 Typical Launch Pad, LC-39

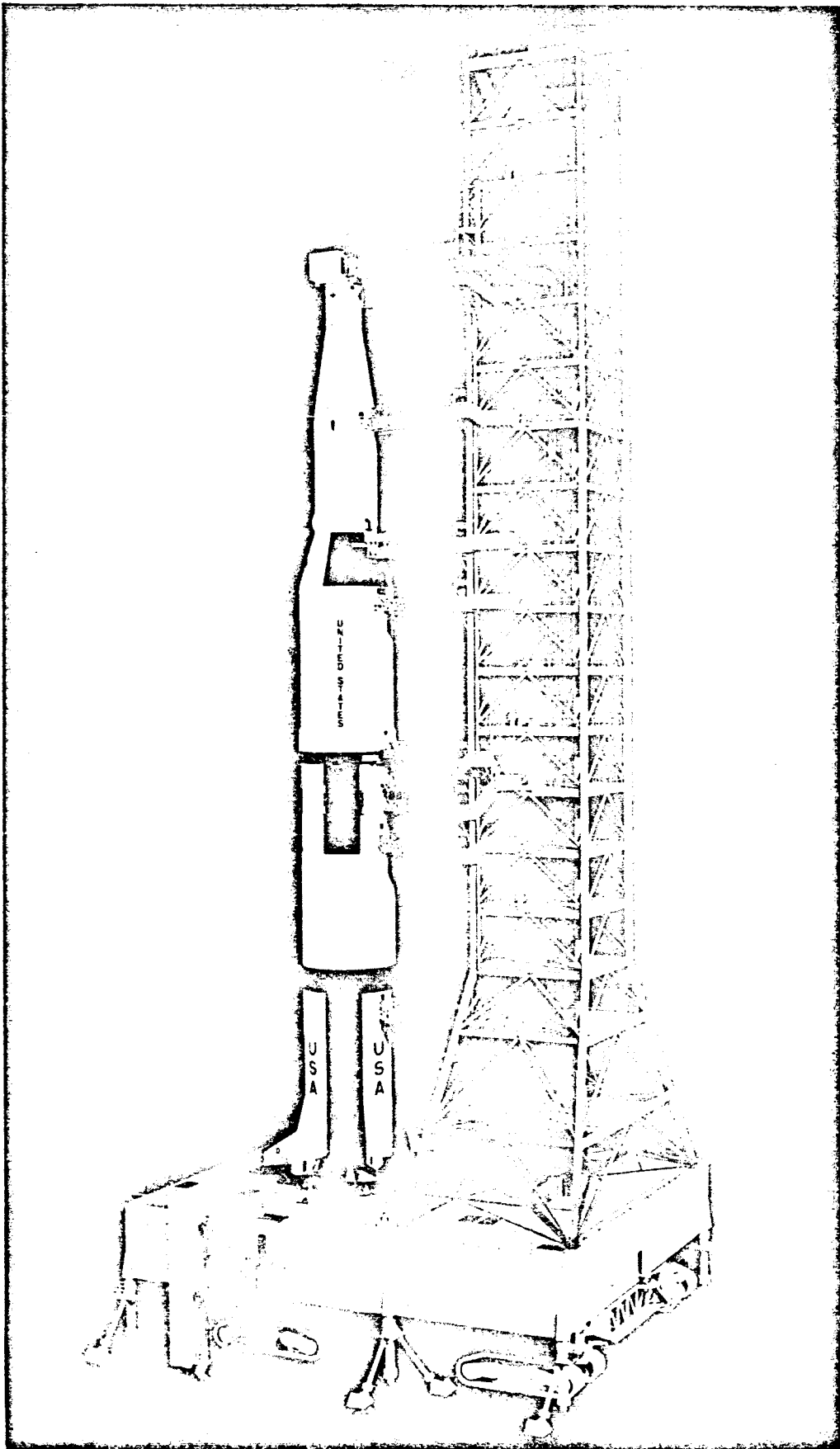


Figure 2-3 Mobile Launcher

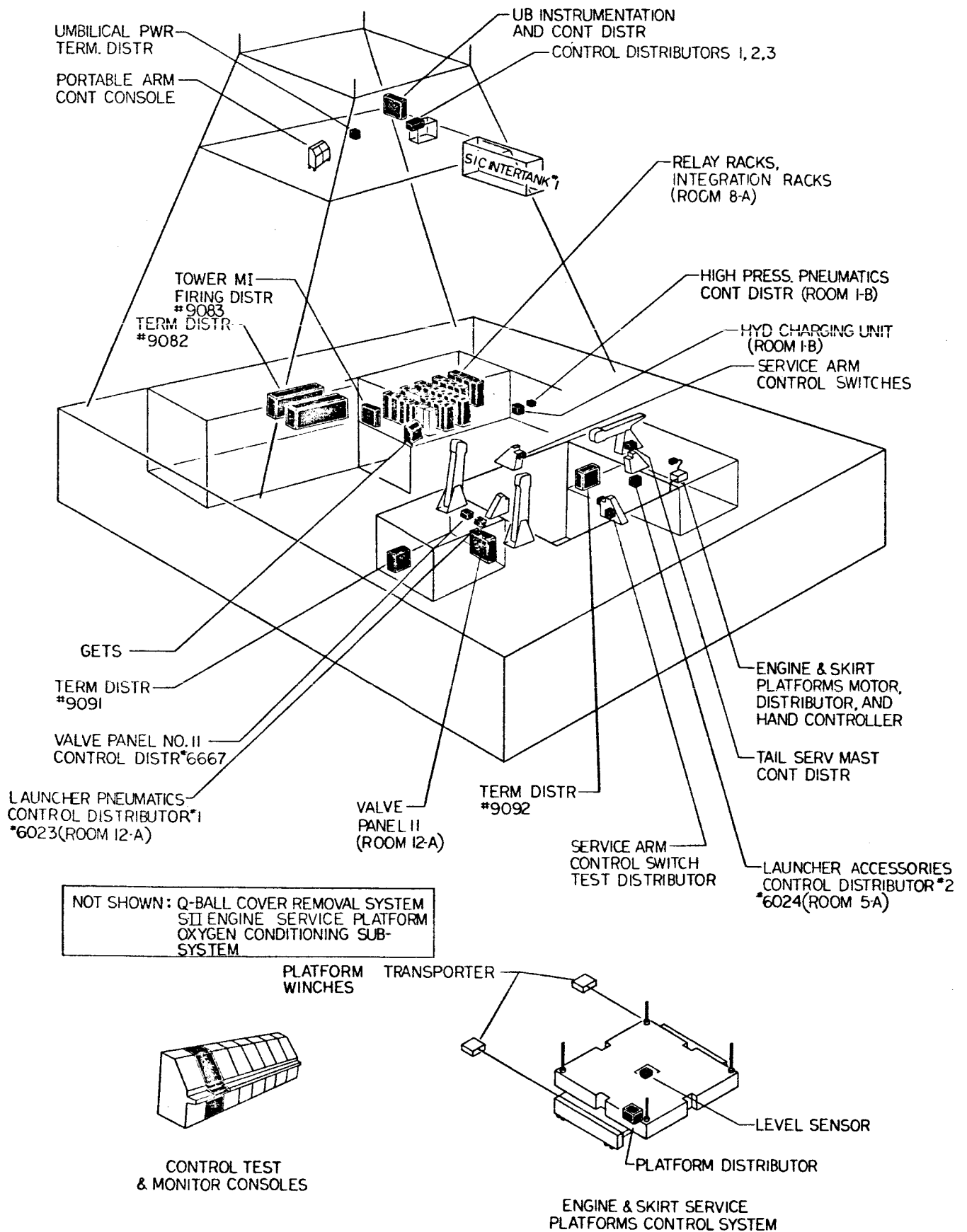


Figure 2-4 Electrical Support Equipment

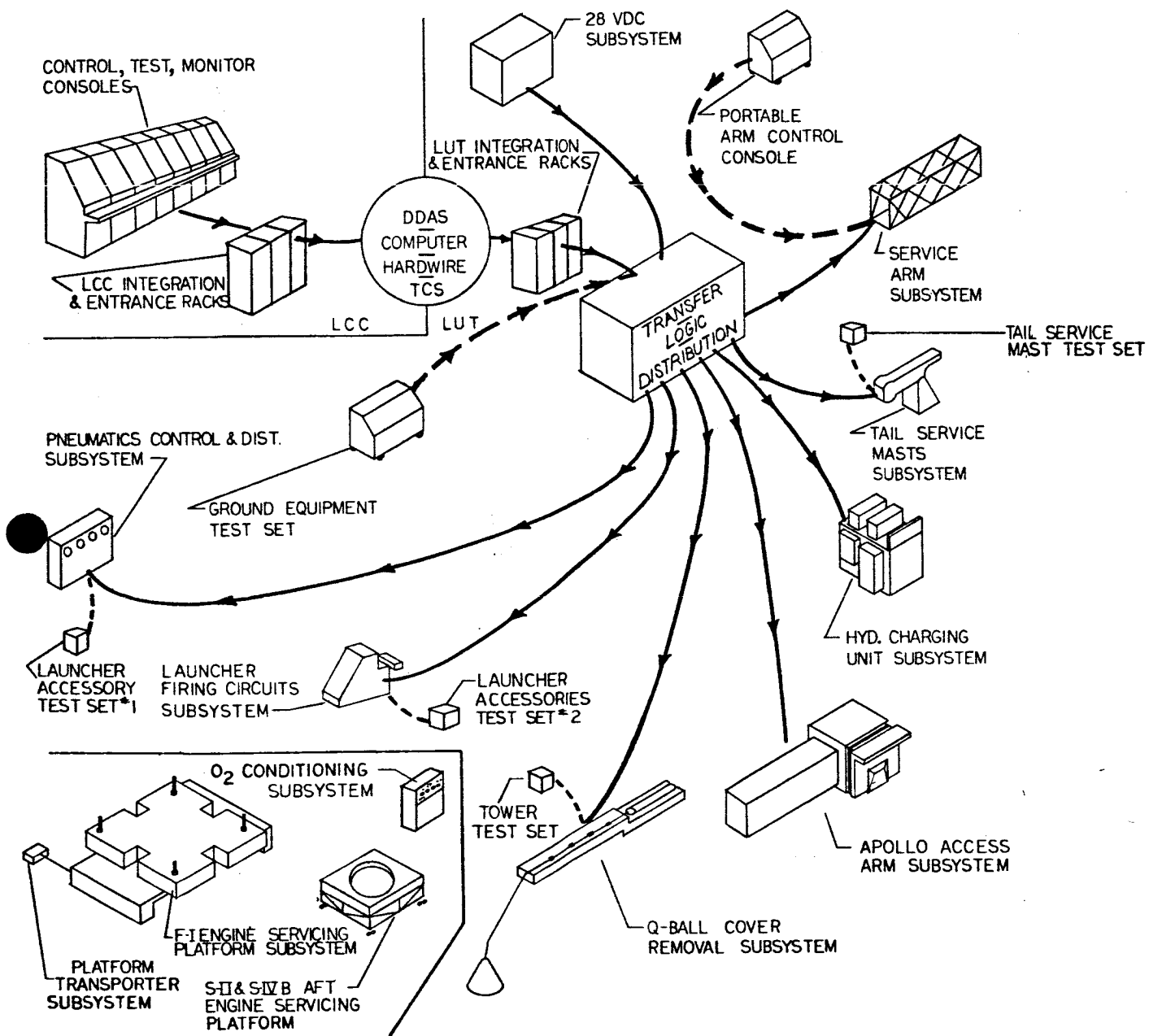


Figure 2-5 ESE Subsystems



SECTION III  
ELECTRICAL SUPPORT EQUIPMENT (ESE)

3-1. INTRODUCTION.

The tail service masts and tower service arms are moved by pneumatic and hydraulic actuators. Local hydraulic and pneumatic accumulators and lines for these actuators are charged for check-out operations in the VAB, and are normally reduced to standby pressure only during Mobile Launcher transit to the pad.

For prelaunch operations these actuators and their electrical controls are remotely checked, operated and monitored from ESE consoles in the Launch Control Center, using 28 volts dc primary power. During countdown and launch the service arms, the tail service masts, and their hydraulic/pneumatic actuators are controlled by the ESE consoles in the LCC, by the Mobile Launcher and LCC computer complex, and partly by ESE sequencing circuits in the Mobile Launcher.

The twelve subsystems comprising the Mobile Launcher ESE are shown in figure 2-5 and are individually delineated in paragraphs 3-2 through 3-58. Among these, only four subsystems are independent of the LCC. They are the S-IC Engine Servicing Platform, the Platform Transporter, the S-II/S-IVB Engine Servicing Platform, and the Oxygen Conditioning System.

The remaining nine electrical subsystems are primarily controlled from the Launch Control Center via the DDAS-Computer-Hardware-Terminal Countdown Sequencer (TCS) complex and the Transfer-Logic-Distribution Complex as in figure 2-5. The latter complex is comprised of relay racks which provide both interface separation and logic, and of distributors which provide a means for power and signal distribution.

The relay-rack portion of the Transfer-Logic-Distribution Complex also provides a central location for connecting the Launcher Ground Equipment Test

Sets (Launcher (GETS), which simulate either the LCC or any subsystem interfacing with the relay racks. The GETS (paragraph 5-2) can also test the relay racks themselves.

Another test unit shown in figure 2-5 is the Portable Arm Control Console. This unit allows an operator to check out or operate any service arm at the service arm level. See paragraph 5-12. In addition, test sets are provided to locally test launcher accessories and tail service masts.

In the following table some of the major LC-39 launch activities and events are given in chronological order. Those which make direct use of the Electrical Support Equipment are marked by an asterisk (\*).

#### VEHICLE ASSEMBLY BUILDING

L-58 to L-55 days	Offload S-IC and erect on mobile launcher.
L-55 to L-49 days	* Install S-IC engine servicing platform; connect ESE to LCC via data link.
L-53 to L-49 days	* Install S-II, S-IVB, and IU; make umbilical connections.
L-48 to L-29 days	Check telemetry, navigation, power transfer, launch vehicle systems, etc.
L-28 to L-17 days	Install spacecraft and test spacecraft systems; install light ordnance items.
L-16 to L-15 days	* Perform interface and compatibility tests, including vehicle-ESE; perform simulated flight test, check LCC-pad data link with simulators.

#### LAUNCH PAD

L-12 to L-10 days	* Crawler transports vehicle and mobile launcher to pad; mate launcher to ground connections; power on (all systems "quick look" to the LCC); position engine servicing platform.
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L-10 to L-4 days	* Position arming tower; perform RF tests, propellant loading tests, and limited electrical subsystem verification tests from LCC.
L-3 days	Perform simulated flight test.
L-2 to L-1 days	* Remove work platforms, test equipment, and arming tower; retract Q-ball cover.
L-0	Terminal countdown.
T-7 hours	Begin LOX and LH <sub>2</sub> loading.
T-4 hours	Astronauts ingress via Mobile Launcher.
T-105 min.	Conduct final systems tests.
T-187 sec	* Launch support equipment preparation complete and terminal countdown sequencer start.
T-60 sec	* Unlock & retract Apollo access arm.
T-30 sec	* Apollo access arm retracted and latched; unlock preflight service arms.
T-22 sec	* Retract S-IC intertank reconnect mechanism.
T-17 sec	* Retract S-IC Intertank & S-IC Forward service arms; unlock inflight service arms.
T-10 sec	* Pad water system on; tail service mast purge on
T-9 sec	* Launch support equipment ready for ignition
T-7 sec	* Ignition command
T-0 sec	* Launch commit - arm the lift off switches, release holddown arms.
T+0 sec	* Lift off, retract inflight service arms and tail service masts

3-2. SERVICE ARMS SUBSYSTEM.

3-3. Purpose of the Service Arms Subsystem. The Service Arms Subsystem provides system control, test, and monitoring for service arm operation.

3-4. Equipment for the Service Arms Subsystem. The Service Arms Subsystem employs the following equipment:

- a. Service Arms Nos. 1 through 8, figure 3-1
- b. Service Arm Control Distributors (Control Console No. 1)
  - #6062, #6064, Level 60
  - #6122, #6124, Level 120
  - #6142, #6144, Level 140
  - #6162, #6163, #6164, Level 160
  - #6202, #6203, #6204, Level 200
  - #6222, #6223, #6224, Level 220
  - #6262, #6263, #6264, #6265, Level 260
  - #6302, #6303, #6304, Level 300
- c. "J" Boxes (Control Console No. 2)
  - #6146, Level 140
  - #6206, Level 200
  - #6226, Level 220
  - #6266, Level 260
  - #6306, Level 300
- d. Service Arm Racks, room 8-A. See Racks 9, 10, 11, 12, 13, 14, 15, 16, 17, figure 6-2.
- e. Portable Arm Control Console.
- f. Control, Test, and Monitor Console, LCC.
- g. Integration Racks and Entrance Rack, room 8-A. See Racks 20, 21, and 22, figure 6-2.
- h. Terminal Distributor #9082, room 7-A.
- i. Power Racks, room 8-A. See Racks 5, 6, 7, 8, figure 6-2.
- j. UB Instrumentation & Control Distributors #9006, #9010, #9014, #9016, #9020, #9022, #9025, #9026, #9028 and #9030, figure 3-1.
- k. UB Power Distributors #9017, #9023, #9027, #9029, #9031, #9043, #9044, #9045, #9046.  
For locations see figure 3-44.
- l. Local Control Units #6067, #6127, #6147, #6170, #6207, #6227, #6267, #6307



3-5. Description of the Service Arms Subsystem. Control and display functions for each of the eight service arms are provided by the eight LCC Control, Test, and Monitor consoles (figures 3-4, typical) and the LCC Status Panel (figure 3-5). All ESE consoles are shown in figure 3-67. These consoles provide visual display for critical system parameters and electrical control of solenoid-operated valves and pressure switches associated with the service arms. The LCC consoles interface with the seven Relay Distributors (#6613 through #6618 and #6621) via the DDAS-Computer-Hardwire complex, as in figure 3-1. These relay racks provide control logic and act as transfer units for the Service Arm Subsystem as follows.

Manual Operation of switches on the LCC Console provides command signals which cause relays in the relay racks to be actuated. Relay operation in turn provides 28 volts dc via the terminal and control distributors, figure 3-2, to solenoid-operated valves, limit and pressure switches, and analog transducers associated with each tower level. These components are shown for a typical preflight arm in figure 3-3 and a typical inflight arm in figure 3-2. Resulting mechanical movements of the arm and its control mechanisms are monitored by switches and transducers which return feedback signals to the relay racks along a path parallel to the command signal. In the relay racks these feedback signals actuate additional relays which provide visual display at the LCC Control, Test, and Monitor Console. At the same time, some of these feedback signals provide a stimulus for automatic sequenced operation of the service arm.

#### NOTE

During launch conditions five of the eight service arms are retracted at vehicle liftoff. Initiating commands for retracting these five arms are provided by the Launch Equipment Firing Circuits Subsystem, paragraph 3-20.

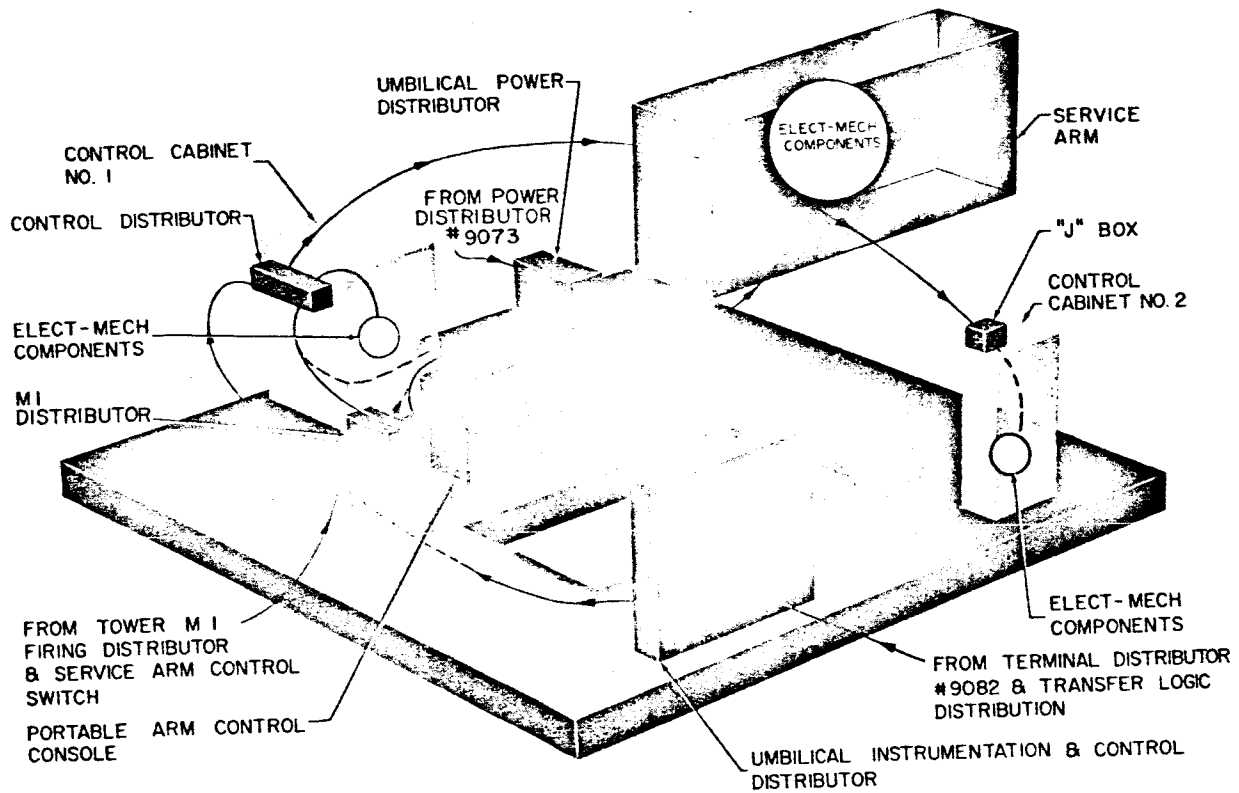


Figure 3-2 Typical Inflight Service Arm and Test Installation

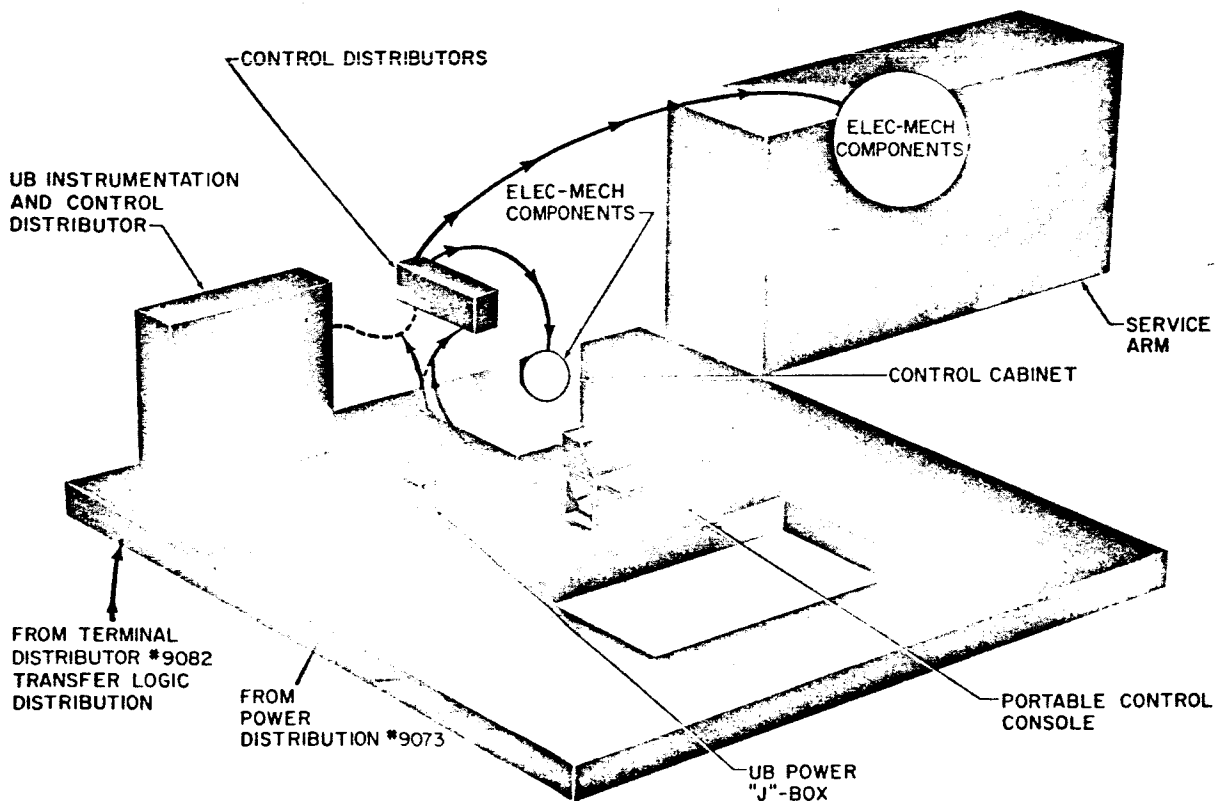


Figure 3-3 Typical Preflight Service Arm and Test Installation





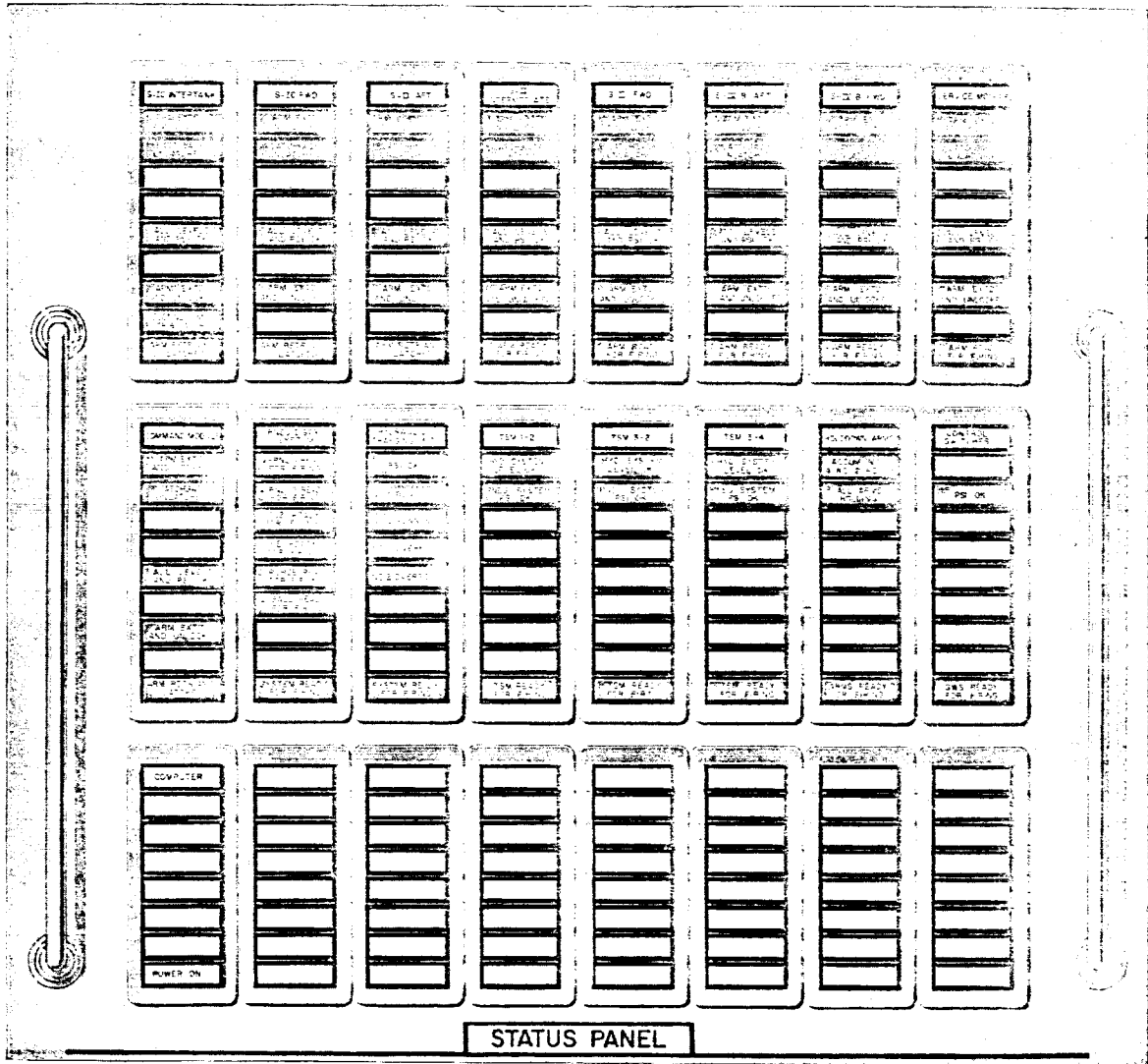


Figure 3-5 LCC Summation Panel D75M14440

3-6. Local Tests for the Service Arms Subsystem. In order to test and operate a service arm from the tower level with which it is associated the Portable Arm Control Console (figures 3-2 and 3-3) is employed. After this console is positioned on the tower level, special test cables are connected to appropriate distributors and electrical-mechanical components, in place of the system cables.

Once the Portable Arm Control Console is connected, the electrically controlled hydraulic and pneumatic systems can be charged and the condition of all pressure systems monitored. Upon completion of the charging sequence, the service arm can be operated as a complete system, or any of its three major sequential functions can be exercised individually. Provisions are made in the Portable Arm Control Console to allow the operation of each critical component to be monitored. See figures 3-7, 3-8 and 5-7.

The service and access arms can also be extended or retracted by use of the local control unit shown in figure 3-6. One such unit is stored in each Control Cabinet No. 1.

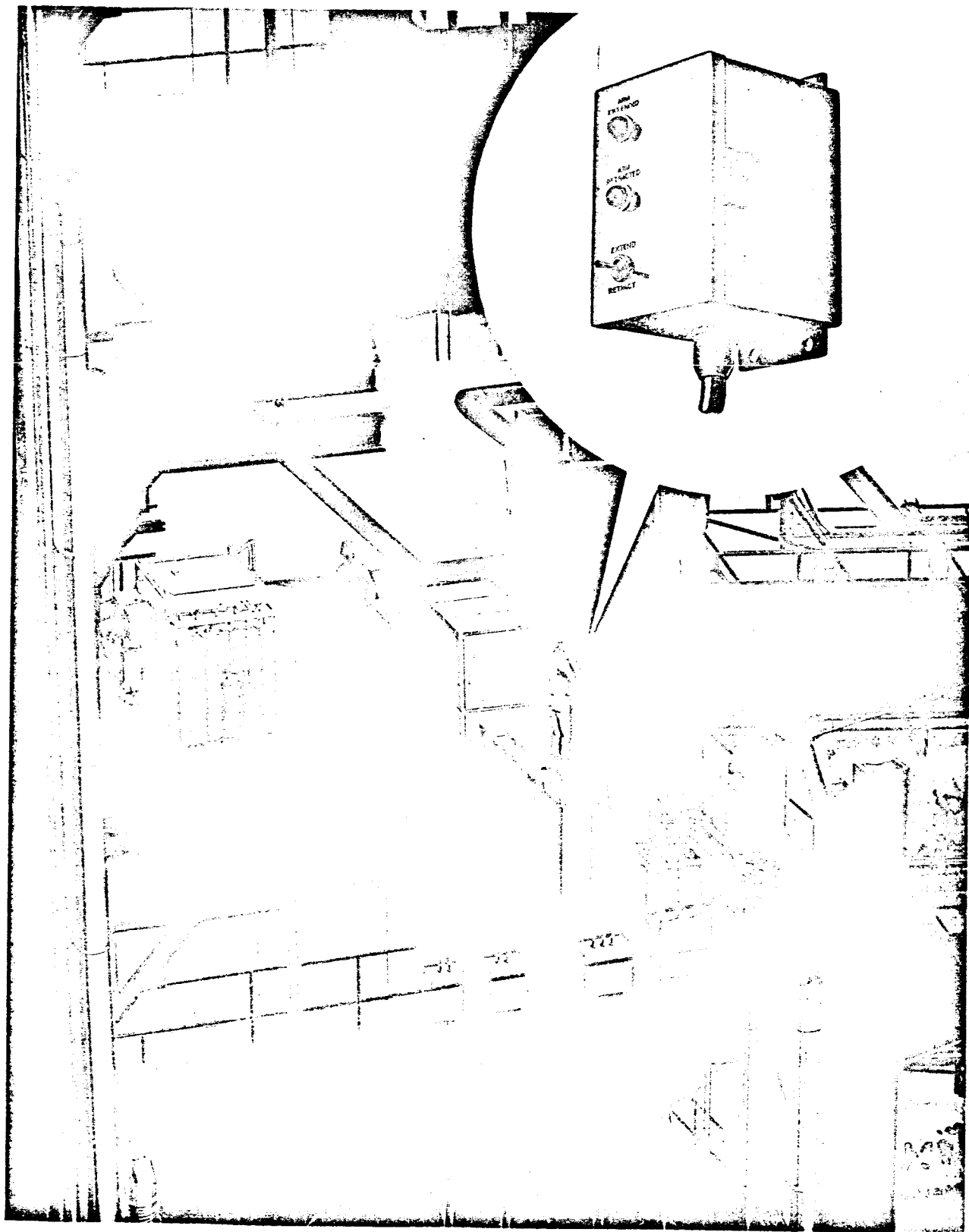


Figure 3-6 Local Control Unit - Service Arms

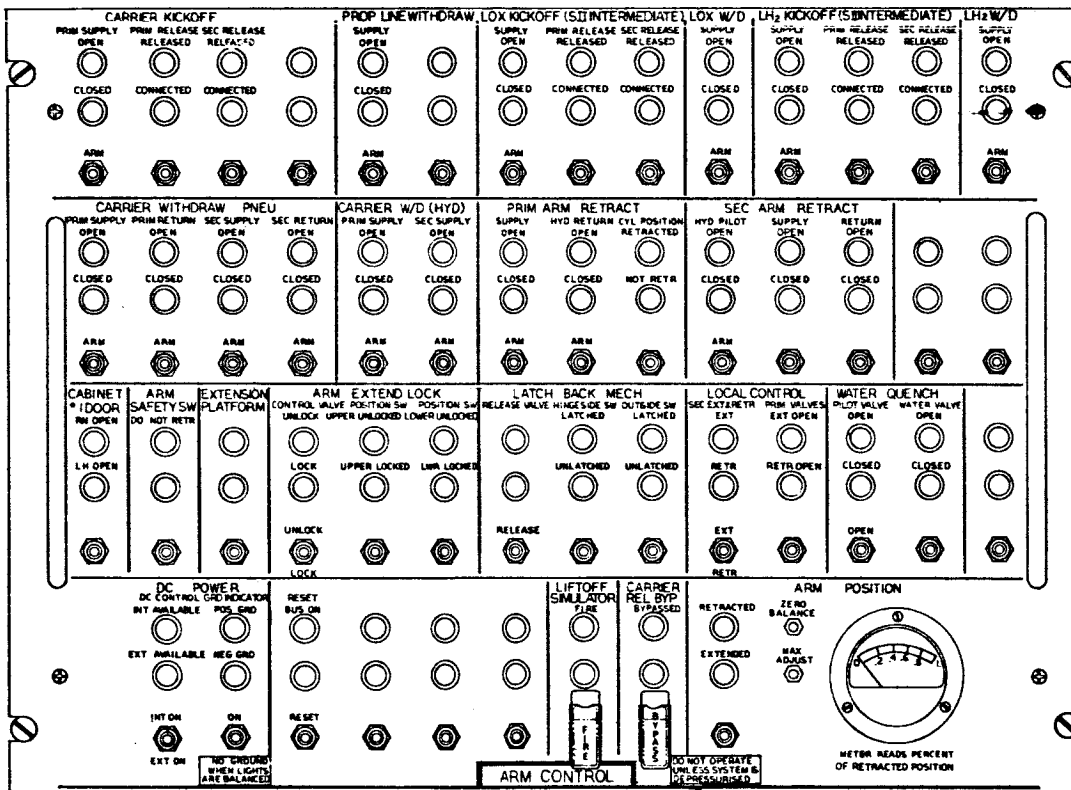


Figure 3-7 Arm Control Panel J75M07546-3, Portable Arm Console

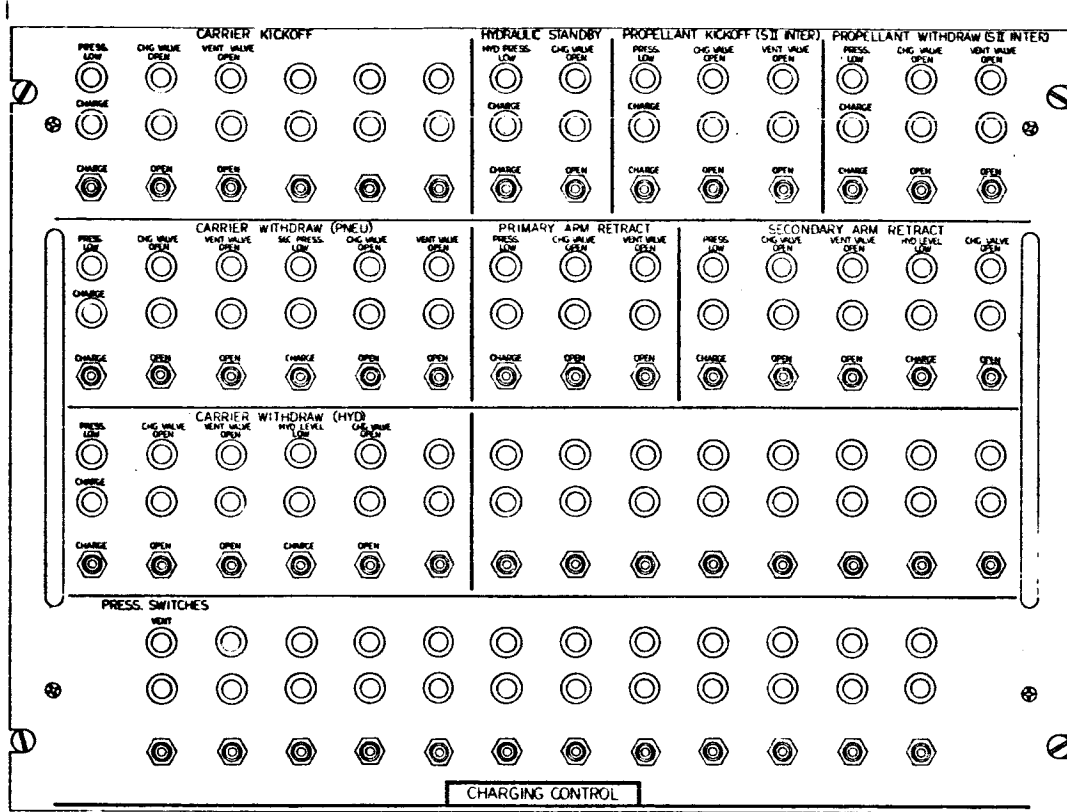


Figure 3-8 Charging Control Panel J75M07546-1, Portable Arm Console

3-7. APOLLO ACCESS ARM SUBSYSTEM.

3-8. Purpose of the Subsystem. The Apollo Access Arm Subsystem provides electrical control, test, and monitoring of the access arm and the hood positioning mechanism.

3-9. Equipment for the Subsystem. The Apollo Access Arm Subsystem employs the following equipment:

- a. Access Arm and associated electrical-mechanical components
- b. Hood positioning mechanism and associated electrical-mechanical components
- c. Apollo Access Arm Control Panel, LCC
- d. Relay Distributor #6619, Mobile Launcher room 8-A
- e. Power Rack, Mobile Launcher room 8-A
- f. Terminal Distributor #9082
- g. UB Instrumentation and Control Distributor #9032
- h. UB Power "J" Box #9047
- i. Control Cabinet No. 1 #6328
- j. Control Distributors #6322 and #6324
- k. Control Distributor #6357A55A2
- l. Portable Arm Control Console
- m. Local Control Unit #6339

3-10. Description of the Subsystem. The Apollo access arm provides personnel access, environmental controls, and checkout service outlets for the Apollo Command Module. During launch conditions the arm is controlled from the Apollo Access Arm Control panel in the LCC, figure 3-12. When the arm control system is manually given READY TO FIRE status from the panel, the access arm disconnects itself from the vehicle and retracts approximately one minute before lift off, upon command from the Terminal Countdown Sequencer.

The command signals flow through the DDAS-Computer-Hardware-TCS complex (figure 3-9) to Relay Distributor #6619 in the Mobile Launcher room 8-A. Relay closure provides 28 volt signals through Terminal Distributor #9082 to UB Instrumentation and Control Distributor #9032, at Mobile Launcher Level 320. The

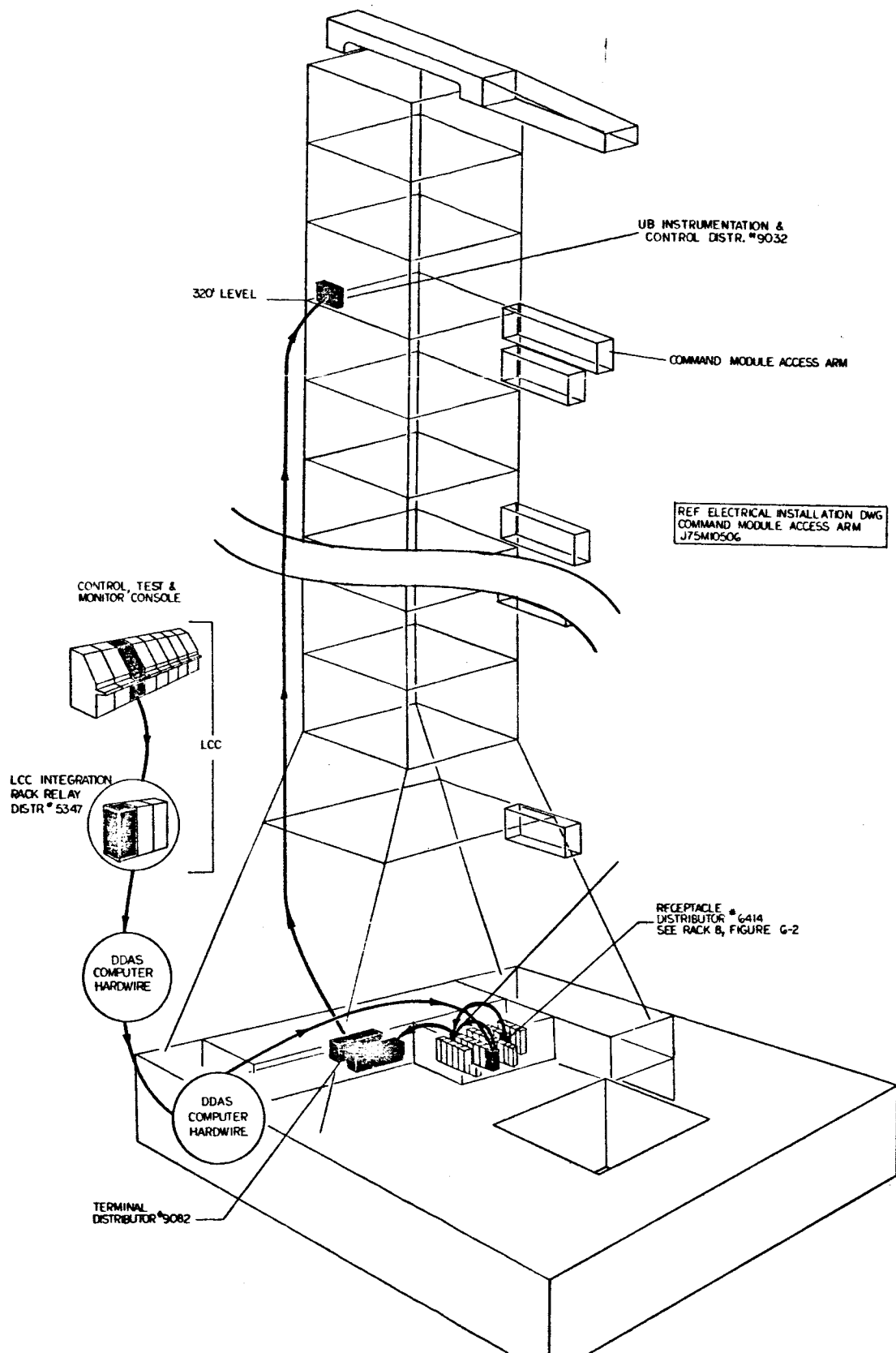


Figure 3-9 Apollo Access Arm Subsystem

control system then branches off into two operations, which are treated separately here.

The first operation, access hood positioning control, is illustrated in figure 3-10. At launch, command signals flow from the UB Instrumentation and Control Distributor #9032, through Control Distributor #6357A55A2, to various electrical-mechanical components in the vicinity of the access hood. These components control a pneumatic system which releases latching hooks, raises the hood, and retracts it several inches away from the Command Module.

The second operation, access arm control, is thereby automatically initiated. Signal flow as shown in figure 3-11 actuates electrical-mechanical components in the arm and in Control Cabinet No. 1. These components control the pneumatic and hydraulic system which swings the arm back and latches it to the tower.

During launch conditions or prelaunch testing, various specific mechanical movements of the access arm and hood positioning mechanism can be controlled and monitored from the LCC.

Thus at any time the arm can be extended and the access hood repositioned and latched to the space vehicle. See these functions in figure 3-12.

All major movements of the arm and its control elements are monitored by limit switches and analog transducers. Feedback signals from some of these components provide continuity for automatic operation. Others transmit monitor signals back to the LCC control panel.

3-11. Local Testing of the Access Arm. By use of the Portable Arm Control Console, the Apollo access arm can be locally tested and exercised. For this purpose the control console is outfitted with special overlay panels (figures 3-13 and 3-14) and is internally patchboard-programmed. It is then positioned on tower level 320 and connected to appropriate distributors and electrical-mechanical components, replacing system cables.

Thus the Portable Arm Control Console can charge all necessary



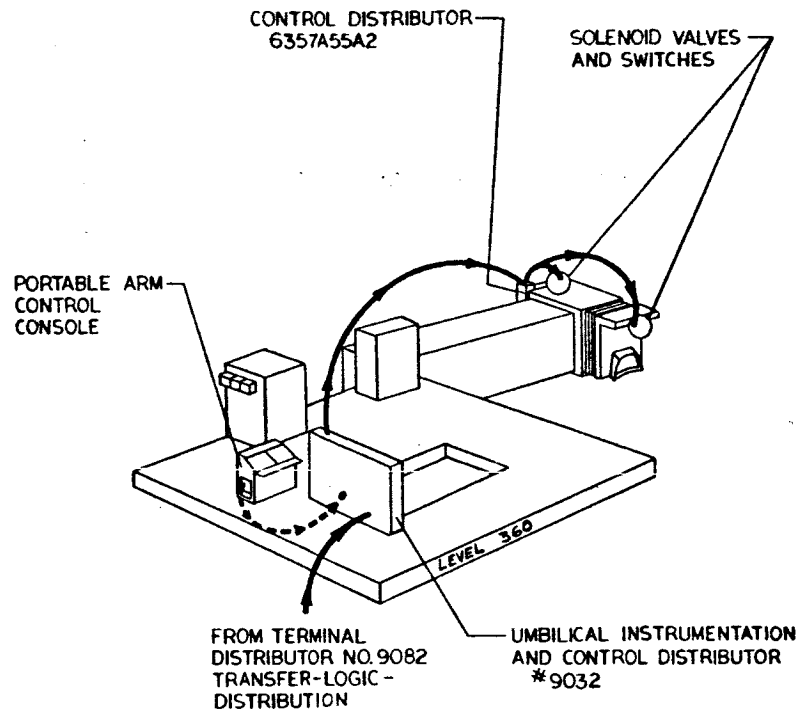


Figure 3-10 Apollo Access Arm Level (Access Hood Control)

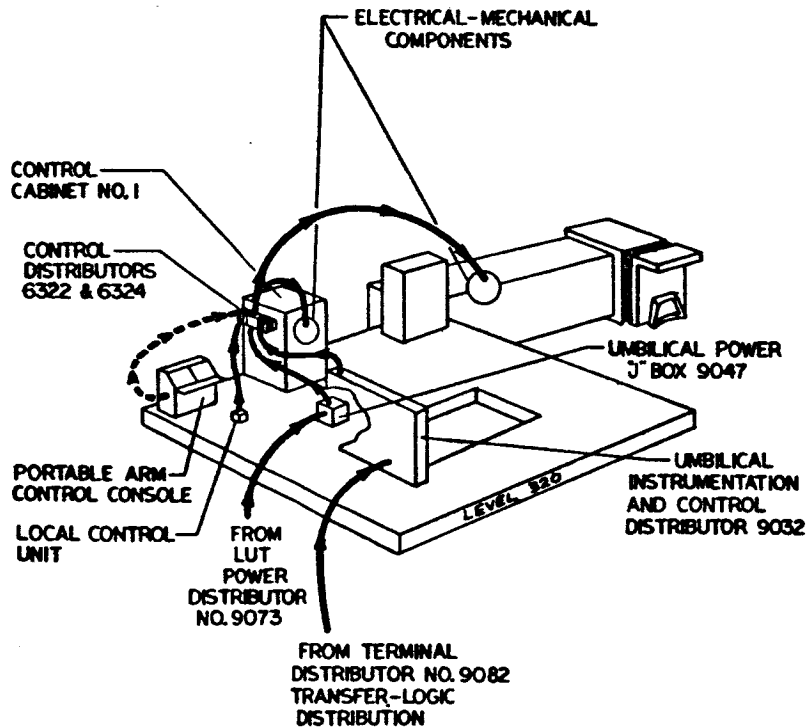


Figure 3-11 Apollo Access Arm Level (Arm Control)

hydraulic and pneumatic accumulators, and can fully exercise the access arm and hood positioning mechanism.

The arm itself can also be extended or retracted by Local Control Unit #6339, which is stored in Control Cabinet No. 1. A typical Local Control Unit is shown in figure 3-6.

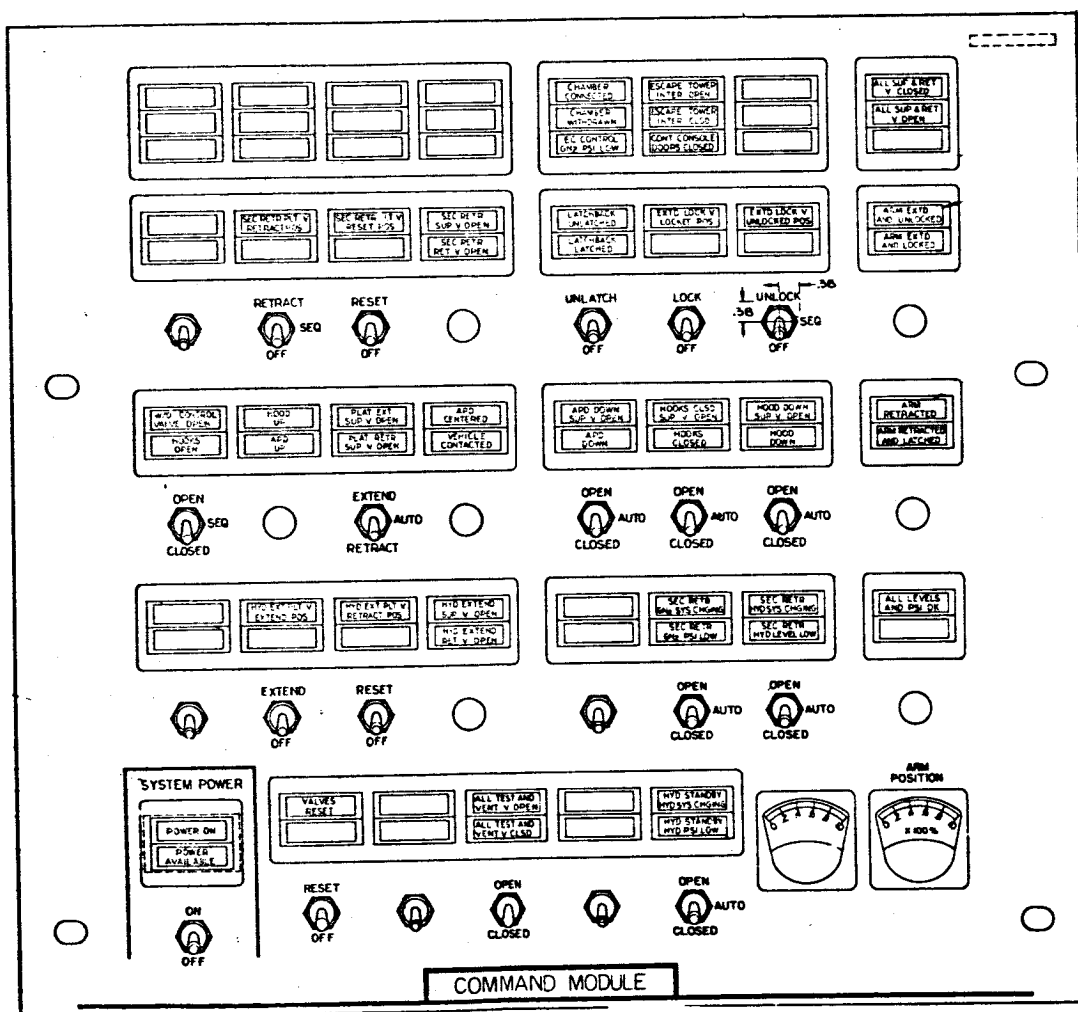


Figure 3-12 LCC Control Panel, Apollo Access Arm



3-12. OXYGEN CONDITIONING SUBSYSTEM.

13. Purpose of the Subsystem. The Oxygen Conditioning Subsystem provides pressurized gases to the GOX Module, figure 3-15, as follows:

a. Gaseous Oxygen (GOX) is supplied at 125 psi and 45° F, and is tapped for leak-testing the astronauts' suits, for cooling the interior of the Command Module, and for human consumption during pre-launch operations.

b. Gaseous helium and gaseous nitrogen are supplied at 125 psi for leak-testing the Command Module.

These pneumatic services are required only during testing and pre-launch checkout, and the Oxygen Conditioning Subsystem is shut down by tower technicians prior to launch.

3-14. Equipment for the Subsystem. The Oxygen Conditioning Subsystem employs the following equipment:

- a. Oxygen Conditioning Console #6326
- b. GOX Module #57A55A11
- c. Pneumatic lines and electrical cabling between these two assemblies.

3-15. Description of the Subsystem. The Oxygen Conditioning Subsystem is a locally controlled electrical-pneumatic system. It prepares oxygen at the Oxygen Conditioning Console #6326 (figure 3-16) and delivers it under controlled pressure and temperature to the GOX Module (figure 3-17).

The Oxygen Conditioning Console #6326 receives gaseous oxygen from storage bottles on level 300. It heats or cools the GOX as required (see controls, figure 3-16), and delivers it through a water-glycol jacketed line to the GOX Module. Electrical-mechanical components required to condition the oxygen include switches, solenoid valves, pressure switches, a pump, a cooler, and a heater. Switching and logic are provided by a relay distributor on the console. The Oxygen Conditioning Console has two separate banks of GOX controls, either of which can automatically switch to the other in case of malfunction.

Pneumatic valves and gauges are also provided on the console for supplying gaseous nitrogen and helium to the GOX Module.

The GOX Module contains manual valves, pressure gages, and indicating lamps as shown in figure 3-17. The entire Oxygen Conditioning Subsystem can be turned on from this Module.

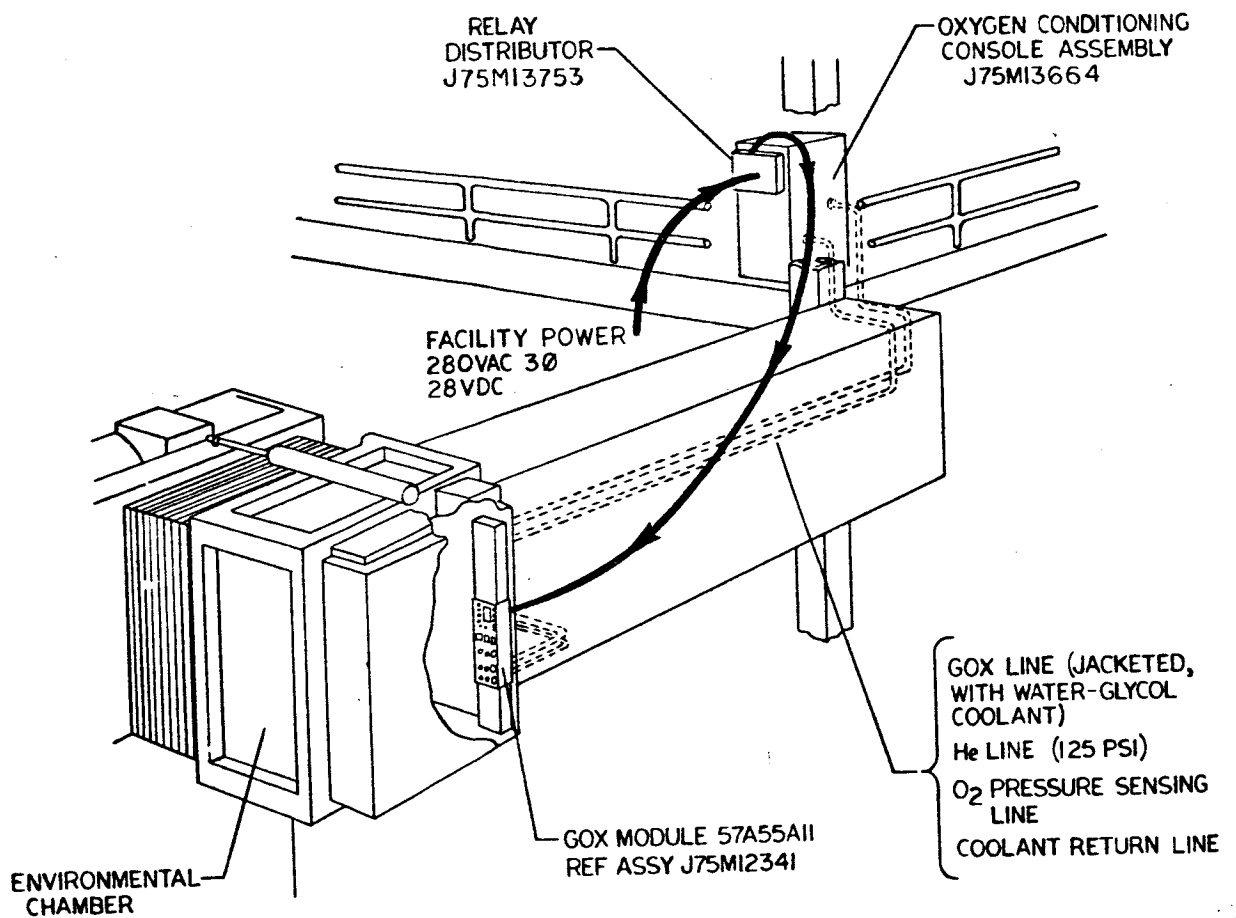


Figure 3-15 Oxygen Conditioning Subsystem



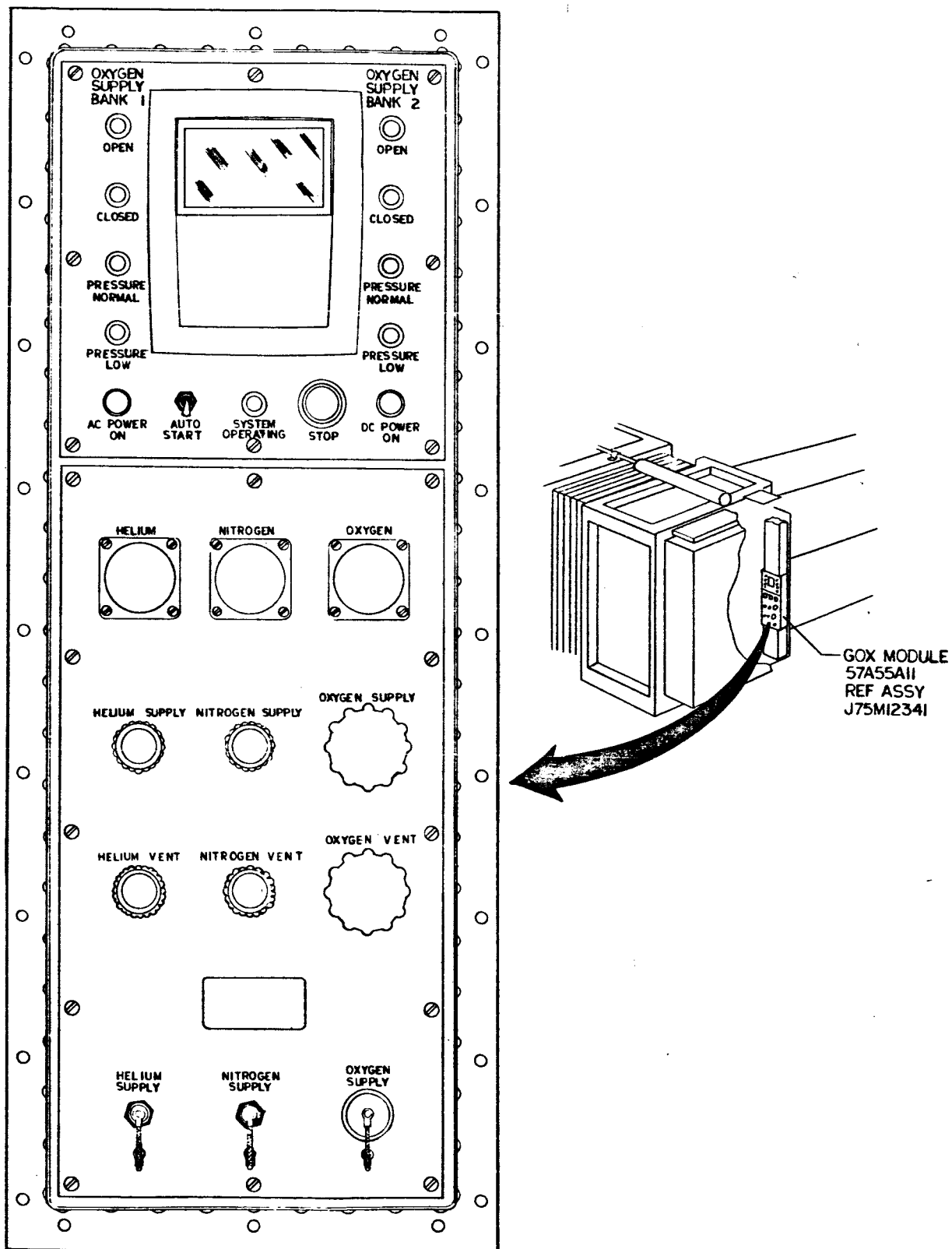


Figure 3-17 GOX Module J75M12341

3-16. HYDRAULIC CHARGING UNIT SUBSYSTEM.

17. Purpose of the Hydraulic Charging Unit Subsystem. The hydraulic charging unit furnishes high pressure hydraulic fluid to initially charge and periodically replenish the service and access arm control system hydraulic accumulators. The control system hydraulic accumulators supply the high pressure hydraulic fluid to the service and access arm hydraulic cylinders to allow umbilical carrier withdrawal or extension, and service/access arm retraction or extension. Periodic replenishment of the hydraulic accumulators occurs during in-transit and pre-launch operations to offset hydraulic fluid demands of the hydraulic mechanisms in tracking vehicle motion caused by wind loading, fuel operations, thermal bending or operation of the mobile launcher enroute from the VAB to the launch site.

3-18. Equipment for the Subsystem. The hydraulic charging unit subsystem is composed of the following equipment:

- a. Motor - Pump Units, #6426 and #6419.  
See figure 3-20.
- b. Motor Starter Units, #6424 and #6417.  
See figure 3-20.
- c. Control Distributors, #6423 and #6416.  
See figure 3-20.
- d. Local Control Panels, #6425 and #6418.  
See figure 3-20.
- e. Main Power Switches, #6415 and #6420.  
See figure 3-20.
- f. Pressure Transducers, #6423A2 and #6416A2.
- g. Pressure Switches, #6423A1 and #6416A1.
- h. Relay Rack, Room 8A, Mobile Launcher.  
See Rack 14, figure 6-2.
- i. LCC Control and Monitor Panel, #5419.  
See figure 3-19.



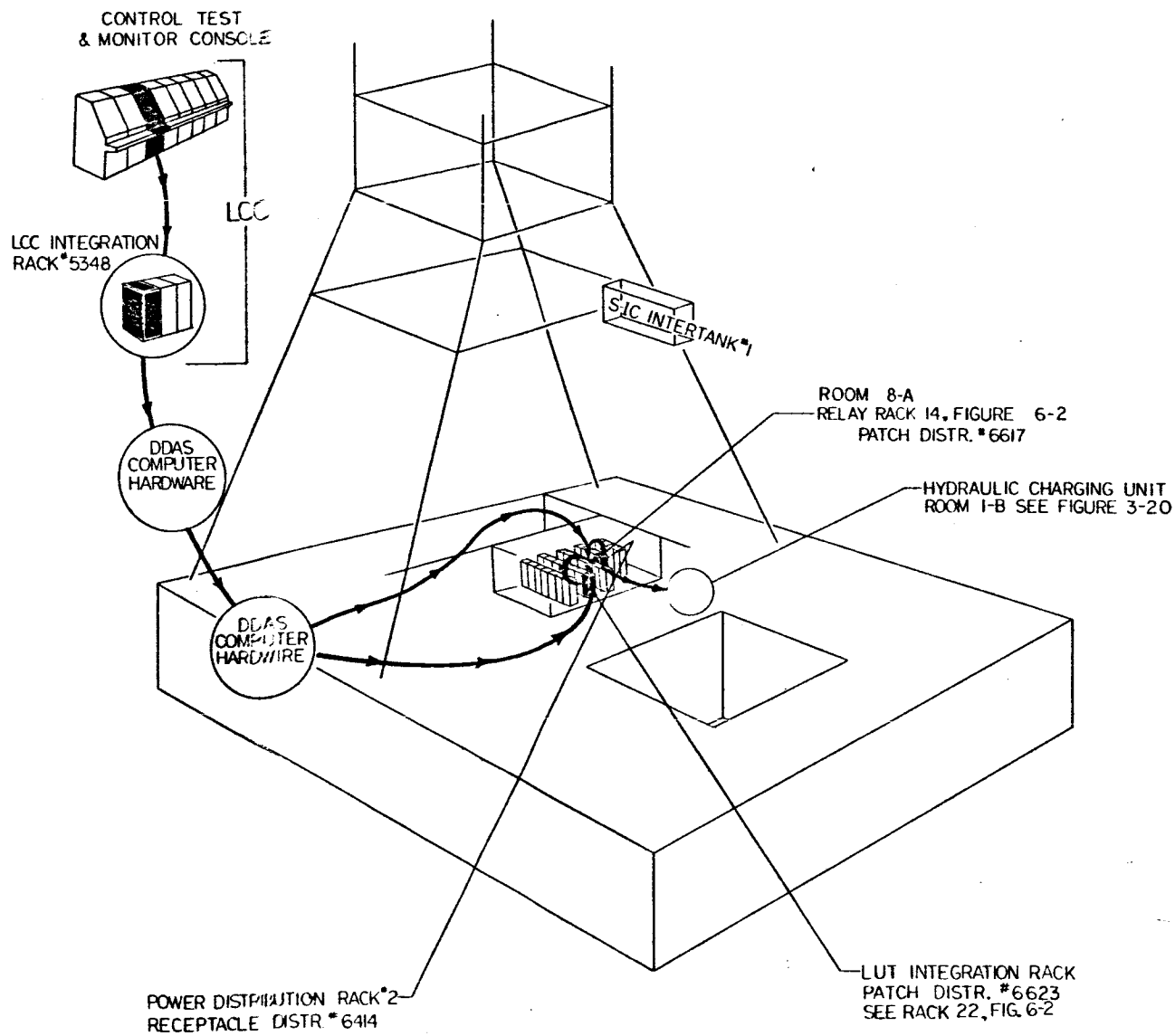


Figure 3-18 Hydraulic Charging Unit Subsystem

3-19. Description of the Subsystem. The hydraulic charging unit consists of a 500-gallon hydraulic fluid reservoir; two identical high pressure, low volume, parallel operated hydraulic pumping units; and separate, identical, electrical control systems.

The control systems for both motor-pump units employ separate motor starters and control distributors which may be controlled locally by two independent control panels (#6425 and #6418) or at the launch site by the LCC control panel (#5419). Both motor-pump units serve the same hydraulic reservoir.

During transit of the Mobile Launcher from the Vehicle Assembly Building (VAB) to the launch site, local control panels are utilized to control motor-pump operation. In order to reduce transients in the Crawler-Transporter power generator, internal circuitry prohibits simultaneous operation of both pumping units during in-transit operations of the Mobile Launcher.

Operation of the hydraulic charging unit at the launch site may be either through the use of the local control panels previously described or through use of the LCC control panel. Normal operation at the launch site employs the LCC panel.

The Control, Test, and Monitor Console housing the LCC control panel is electrically connected to the hydraulic charging unit control distributors by the Computer-Hardware complex and Relay Distributor #6617. Command functions are electrically connected by both hardwire and the computer; discrete monitor functions are electrically connected only through the computer. See figures 3-18 and 3-20.

The electrical control system of the hydraulic charging unit is designed to sense a loss of electrical power to the motor-pump due to internal or external causes, or a loss of hydraulic fluid pressure to the service/access arm hydraulic accumulators. Either occurrence will automatically place the alternate motor-pump on the line without manual supervision.

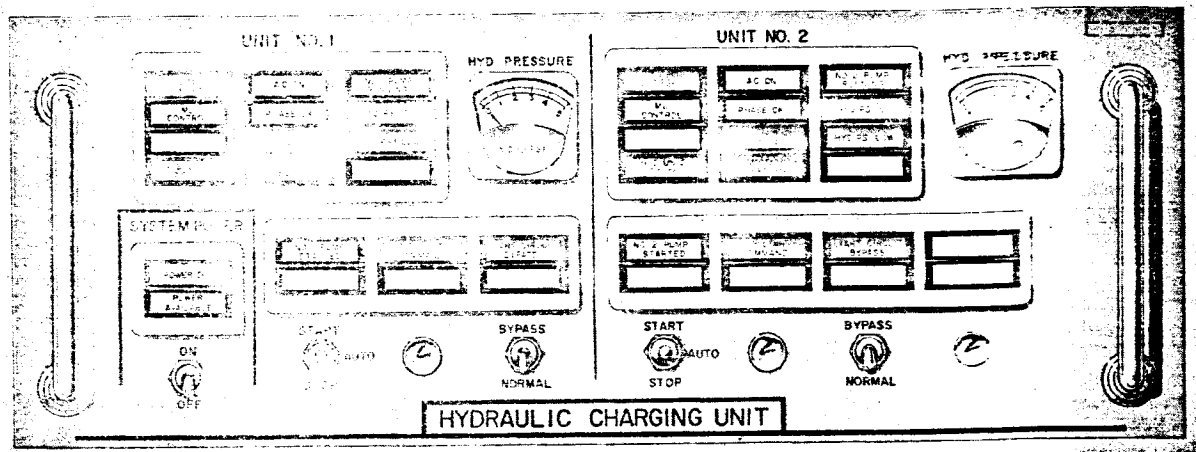


Figure 3-19 LCC Panel, Hydraulic Charging Unit

Hydraulic fluid pressure is indicated on the local and on the LCC control panels. Each panel contains electrical indicator lamps and meter circuits connected to the hydraulic supply line at the output of the main hydraulic pumps. Pressure transducers provide dynamic main-pump output pressures from 0 to 3000 psi, and fixed pressure switches provide discrete pressure indications (normal or abnormal pressure) at time of closure (2750 psi).

Should a hydraulic or electrical malfunction (with the exception of ac power loss) occur during critical launch countdown or mission abort, override capability of the start/control circuitry from the LCC control panel is provided. Use of the override capability will automatically cause a full-voltage start of the affected motor-pump, or, should the motor-pump be in a pump-running condition, continue operation of the affected pump until the override circuit is returned to the "normal" operating state.

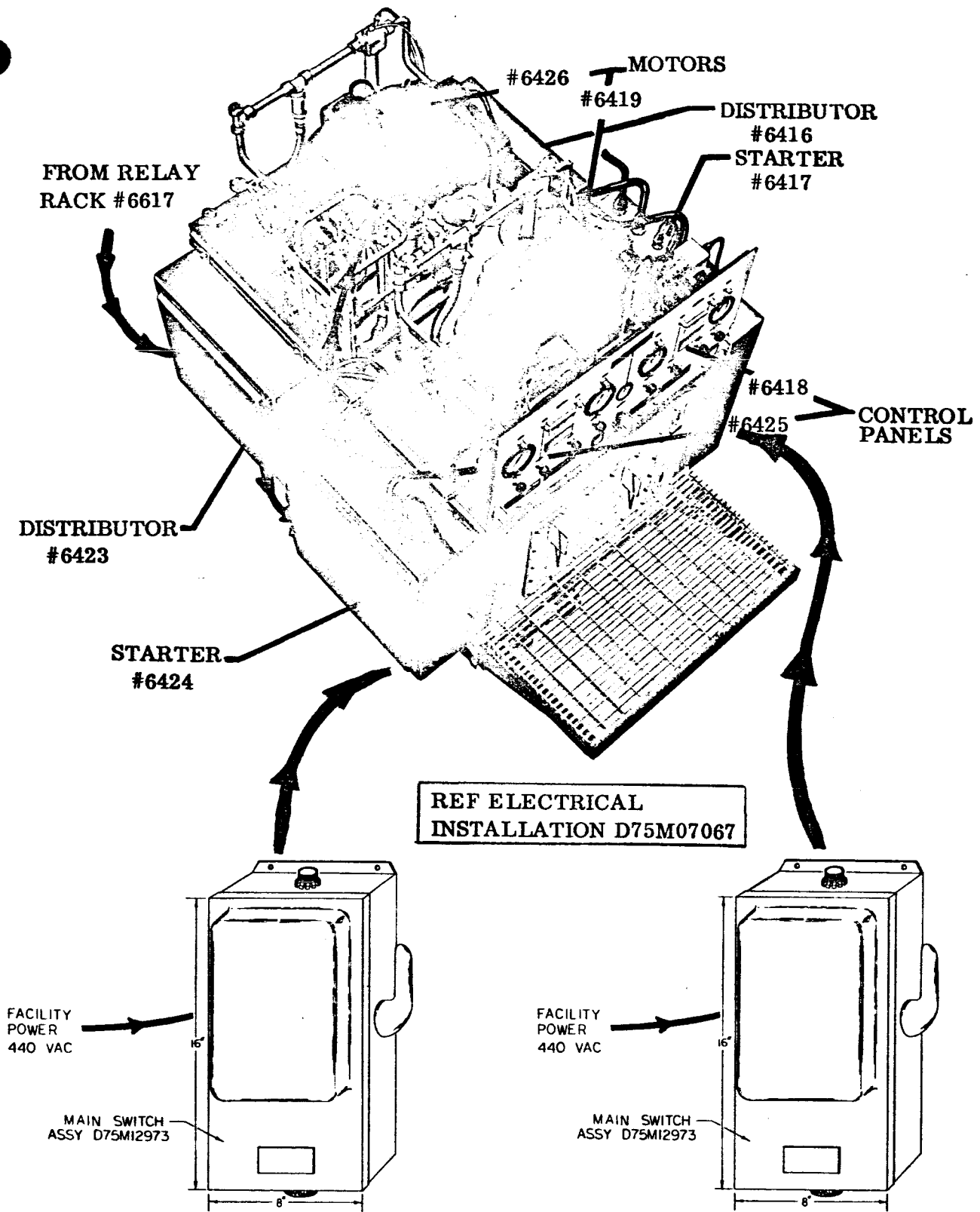


Figure 3-20 Hydraulic Charging Unit and Main Switches

3-20. LAUNCH EQUIPMENT FIRING CIRCUITS SUBSYSTEM.

3-21. Purpose of the Subsystem. The Launch Equipment Firing Circuits Subsystem initiates holddown arm release and inflight service arm retraction.

3-22. Equipment for the Subsystem. The Launch Equipment Firing Circuits Subsystem employs the following equipment:

- a. Holddown Arms and Purge Valves - LCC Panel #5424
- b. Relay Distributor #6651 (Rack 19, figure 6-2)
- c. Terminal Distributor #9082
- d. Terminal Distributor #9091
- e. Terminal Distributor #9092
- f. Control Distributor #6023
- g. Control Distributor #6024
- h. Holddown arms 1, 2, 3 and 4
- i. Relay Distributor #6615 (Rack 12, figure 6-2)
- j. Firing Battery #6631
- k. Firing Battery #6632
- l. Arming Panel #6661 (Rack 24, Figure 6-2)
- m. Arming Panel #6662 (Rack 24, figure 6-2)
- n. Liftoff switches (Holddown arms 2 & 4)
- o. Tower MI Firing Distributor #9083
- p. Service Arm Firing Distributors #9054, #9055, #9056, #9057, #9059

3-23. Description of the Holddown Arms Firing Circuit. The holddown arms firing circuit provides control, testing, and monitoring of the vehicle holddown arms. These functions are controlled from the LCC Firing Accessories Monitor and Test Panel, figure 3-21, or from the terminal countdown sequencer, and at launch, a signal is routed from the terminal countdown sequencer through relay distributor #6651 and then through electrical control distributor #6023 to two parallel solenoid-operated pneumatic valves. The operation of either of these valves allows high pressure GN<sub>2</sub> to release the holddown arms.

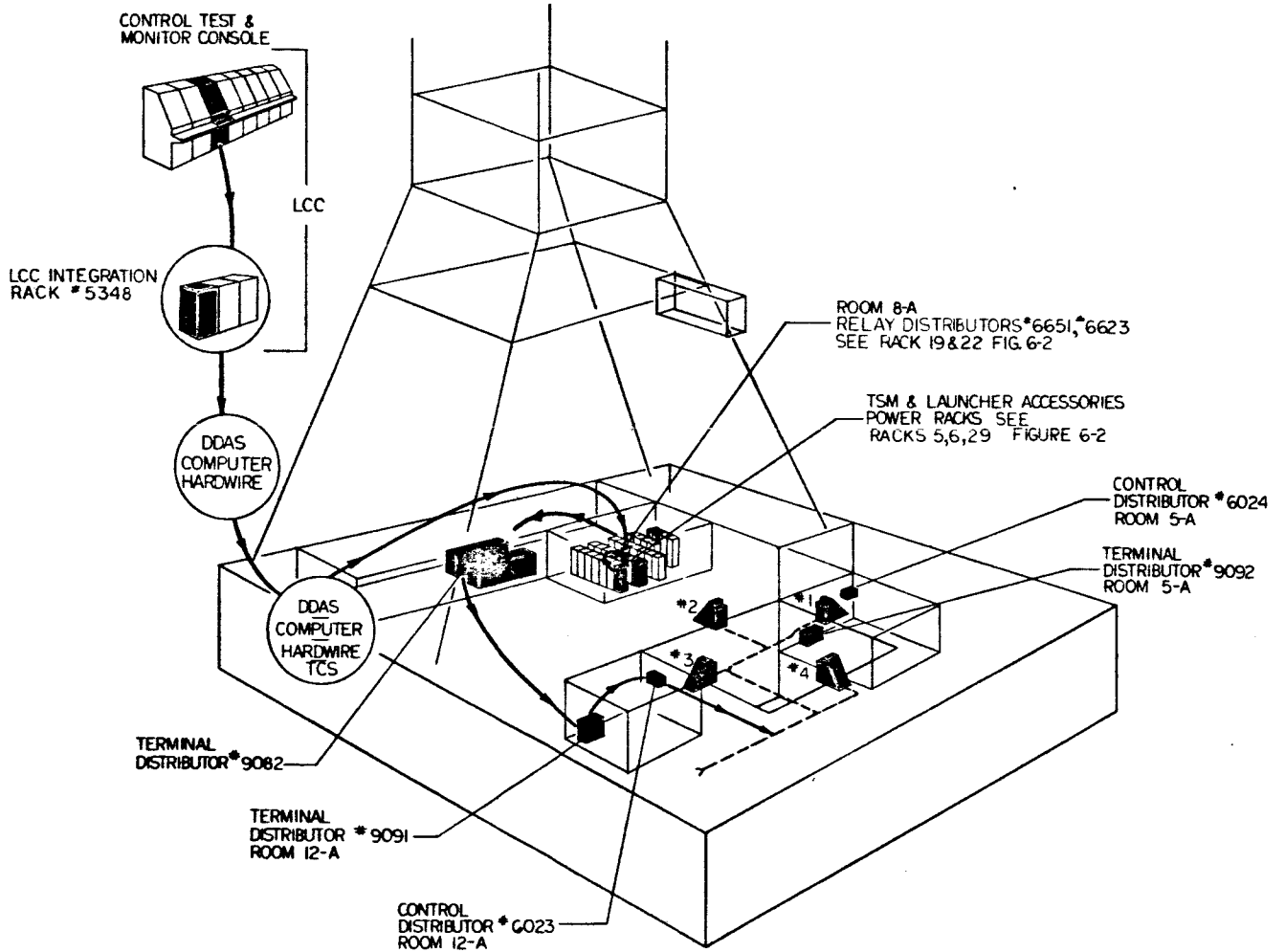
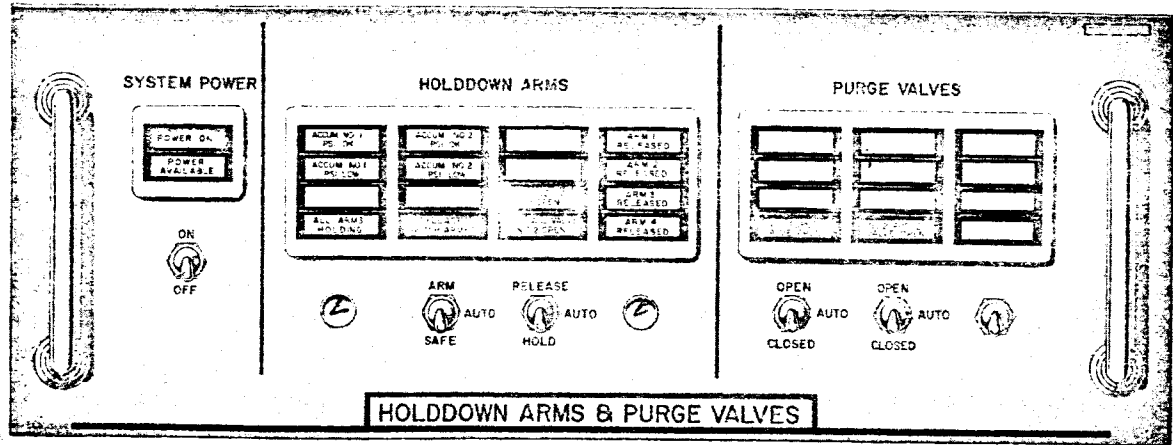


Figure 3-21 Holddown Arms Firing and Monitoring Circuits

When the holddown arms are fully retracted, limit switches return monitor signals to the LCC panel. The limit switches for holddown arms 1 and 4 are connected through distributors #6024 and #9092, and those for arms 2 and 3 are connected to distributors #6023 and #9091. The monitor signals flow back to the LCC through the DDAS-Computer-Hardware Complex.

While in the manual mode, the LCC can also provide various test commands as shown in figure 3-21.

3-24. Local Test, Holddown Arms Firing Circuit. The holddown arms can be exercised by use of Launcher Accessories Test Set No.2, figure 3-22. It is connected to Control Distributor #6023 in place of the power and system cables.

3-25. Description of the Service Arms Firing Circuit. The Service Arms Firing Circuit provides 28 volts dc and automatic sequencing for initiation of umbilical carrier kickoff, umbilical carrier withdrawal, and service arm retraction for the five inflight service arms. The circuit is illustrated in figures 3-23 and 3-24, and is described in the following paragraphs.

The Service Arm Firing Circuit is first armed by the terminal countdown sequencer, which sends a command to relays in Relay Distributor #6615. These relays arm the firing circuit by closing contactors in Arming Panels #6661 and #6662, figure 3-54. Contactor closure applies 28 volts dc from firing batteries #6631 and #6632 to the service arm control switches located in Holddown Arms 2 and 4, figure 3-24. This action provides 28 volts dc to two identical and redundant service arm control switch circuits.

When the vehicle begins rising, the holddown arm Primary Service Arm Control Switch closes, allowing 28 volts dc from the battery supply to be applied through the service arm control switches to the primary buses in the MI Firing Distributor #9083. Energizing these buses applies power to the primary buses in each of the five Service Arm MI Firing Distributors (paragraph 3-27) on the tower. Protected latching relays are also energized through the service arm control switches, allowing a parallel path around the switches to insure a maintained circuit during peak heat and vibration conditions.



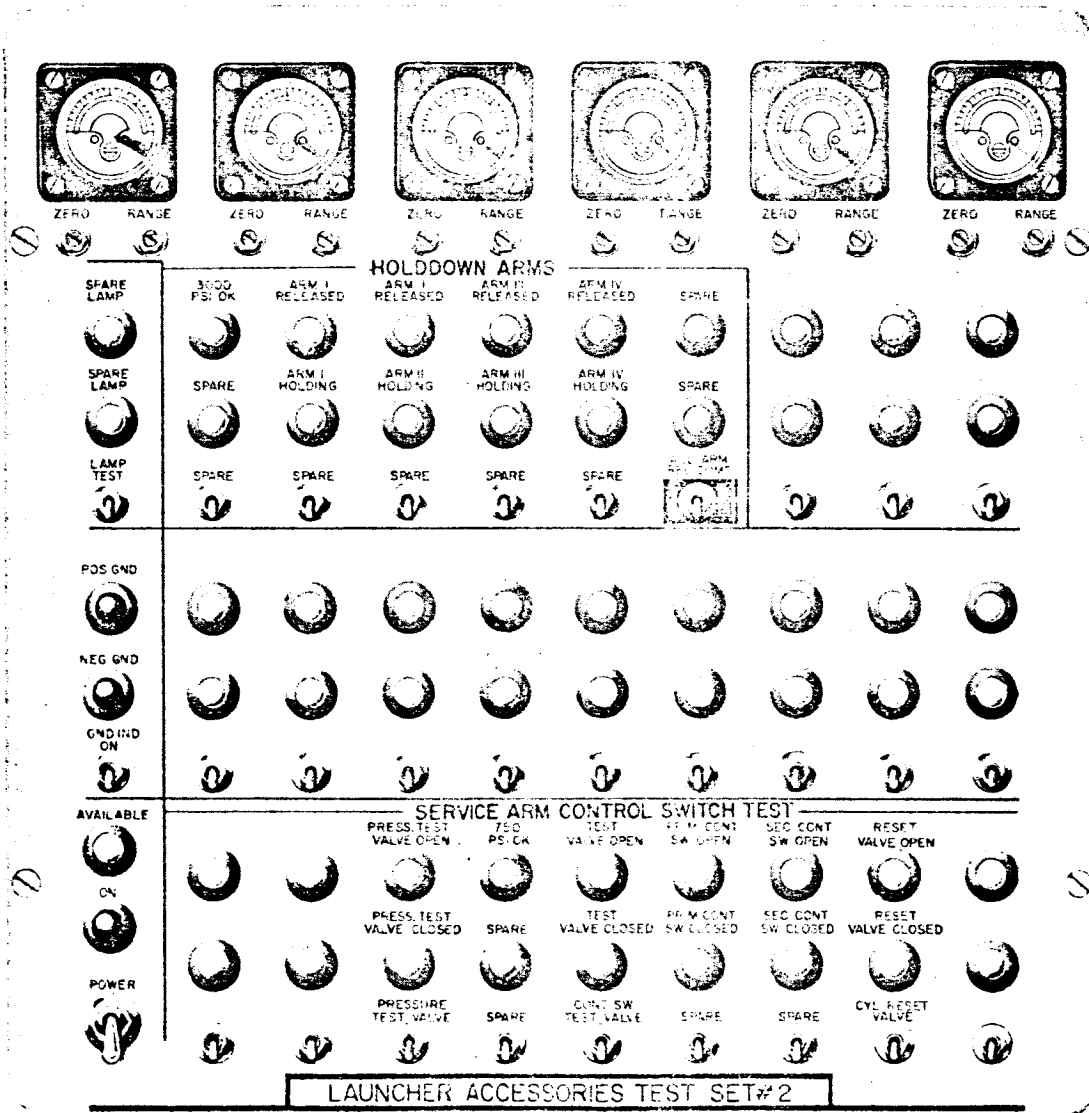
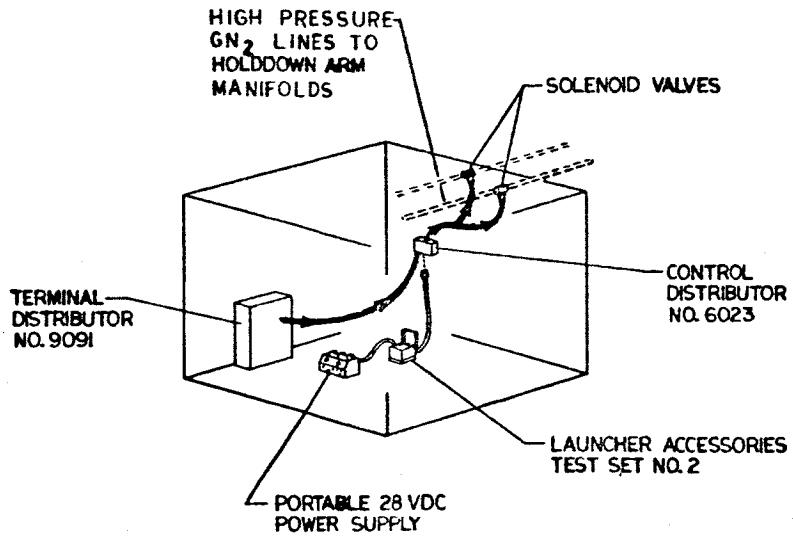


Figure 3-22 Launcher Accessories Test Set No. 2

Each MI Firing Distributor Bus supplies a signal which initiates umbilical carrier kickoff for the inflight arms. As the umbilical carrier separates, a release switch simultaneously initiates carrier withdrawal and arm retraction. Voltage for this last function is also supplied through the Service Arm MI Firing Distributor.

If the entire primary firing circuit functions properly, then carrier kickoff, carrier withdrawal, and arm retractions all occur before the secondary firing circuit is closed. Note, however, that the secondary firing circuit will be activated even if it is not needed.

The Secondary Service Arm Control Switch closes when the vehicle rises to 18 inches, providing excitation of a secondary firing circuit. This secondary circuit initiates carrier withdrawal and arm retraction directly, since carrier kickoff has already been effected mechanically. In all, two primary and two secondary circuits provide four possible actuations of the umbilical carrier and service arm at launch.

Functions relating to the five inflight arms not discussed here are functions of the Service Arm Subsystem, paragraph 3-2.

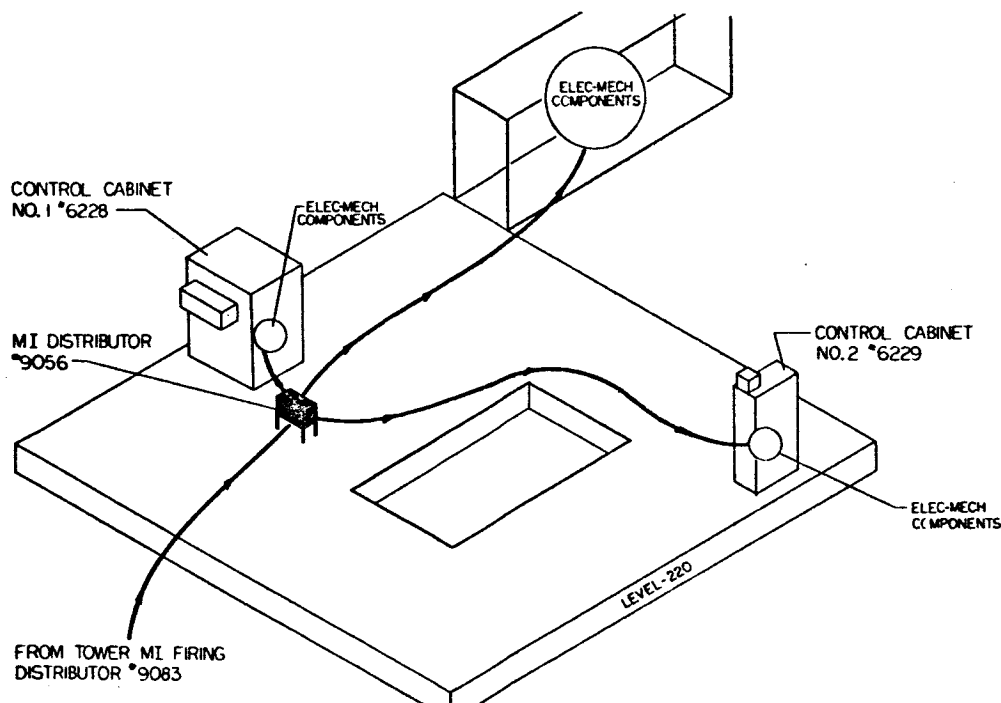


Figure 3-23 Typical Level, Service Arms Firing Circuit

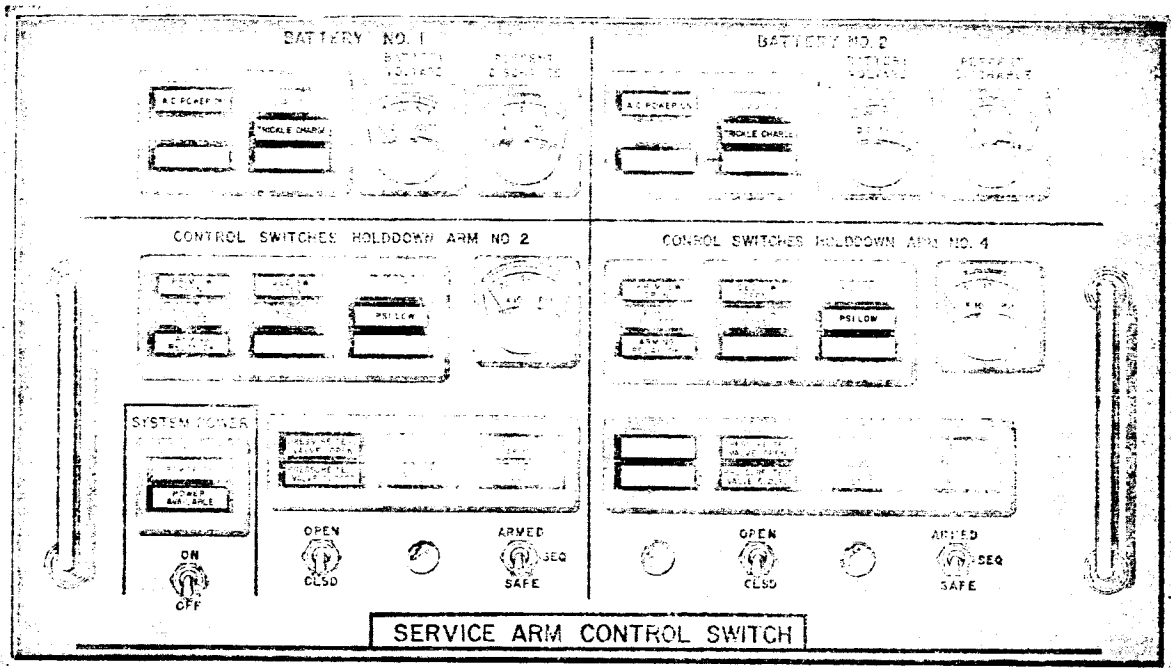


Figure 3-24A LCC Panel, Service Arms Firing Circuit

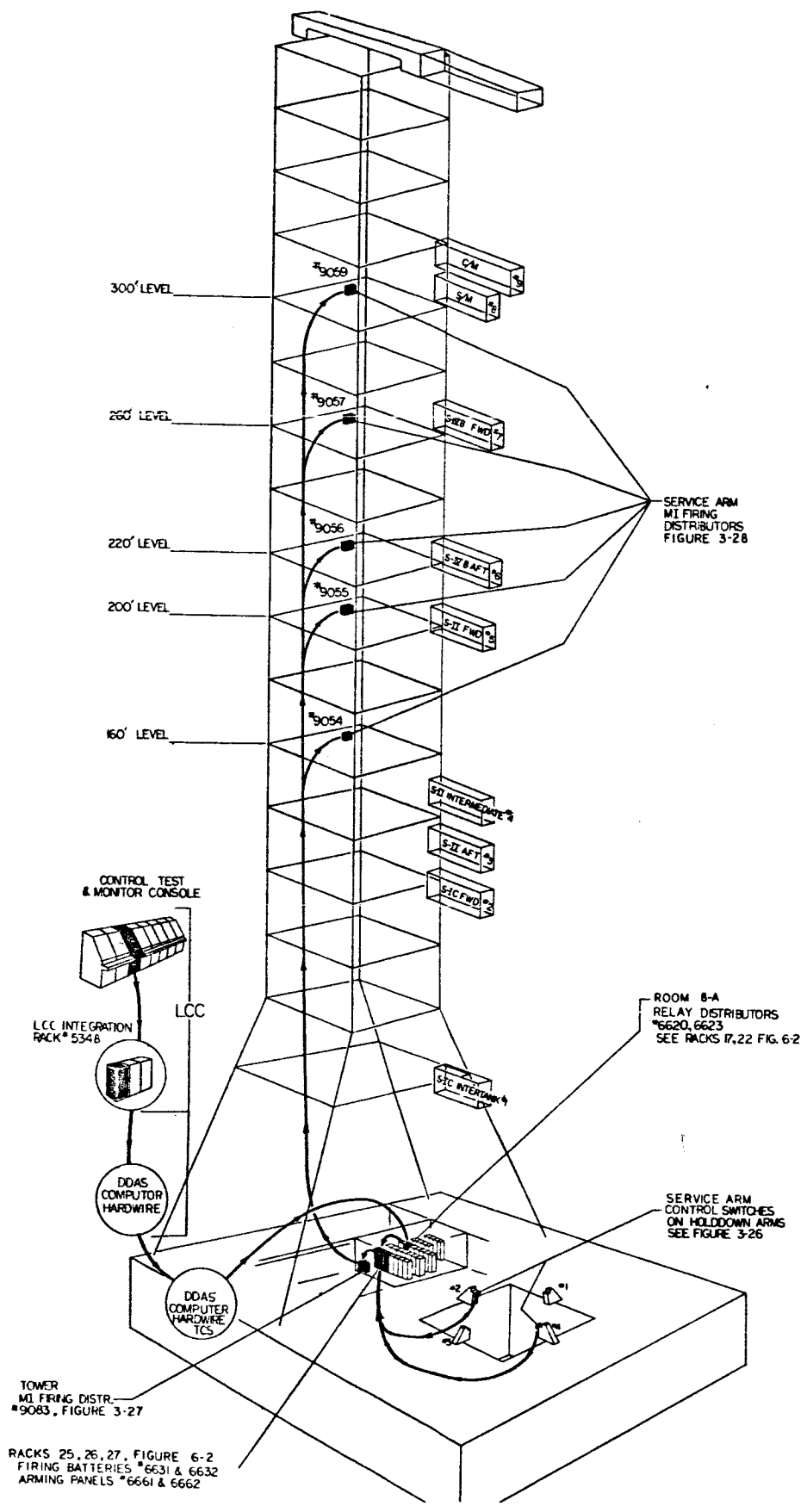


Figure 3-24B Service Arms Firing Circuit

3-26. Local Test of the Service Arm Firing Circuit. Within the service arm firing circuit, the only local test is that of the service arm control switches, figure 3-25. A pneumatic test mechanism is provided in the service arm control switch assembly which physically closes the switch for test. Pressure for the mechanism is supplied through a Pneumatic Box Assembly (figure 3-26), which is controlled through a standard eighteen-connector electrical distributor.

Test functions are applied to this distributor by the Launcher Pneumatics Test Set No. 2, a programming of the Universal Patchboard Test Set. This test set actuates the liftoff switch test mechanism and monitors pneumatic pressures and switch positions. Its test functions are outlined in figure 3-22.

3-27. MI Distributors and Cabling. The service arms firing circuit carries the umbilical release command through Tower MI Firing Distributor #9083 to the five service arm MI Firing Distributors. These are facility enclosures with distribution modules provided by the Launch Equipment Branch. The modules are shown in figures 3-27 and 3-28.

Owing to the critical nature of the service arm firing circuit, the firing distributors are interconnected by Mineral Insulated (MI) Cable. This is a copper-encased cable carrying one to seven conductors within compacted magnesium oxide insulation. It is highly resistant to heat, mechanical disturbances and RF radiation. Special bulkhead fittings are available for sealing its conductors against moisture and atmosphere.

Figure 3-29 shows a typical MI cable installation for the service arm firing circuit. MI cable also carries power to certain lighting J boxes and emergency J boxes aboard the service arms.

A typical termination of MI cable is shown in figure 3-30 and is described fully in procedure drawing A75M07450.

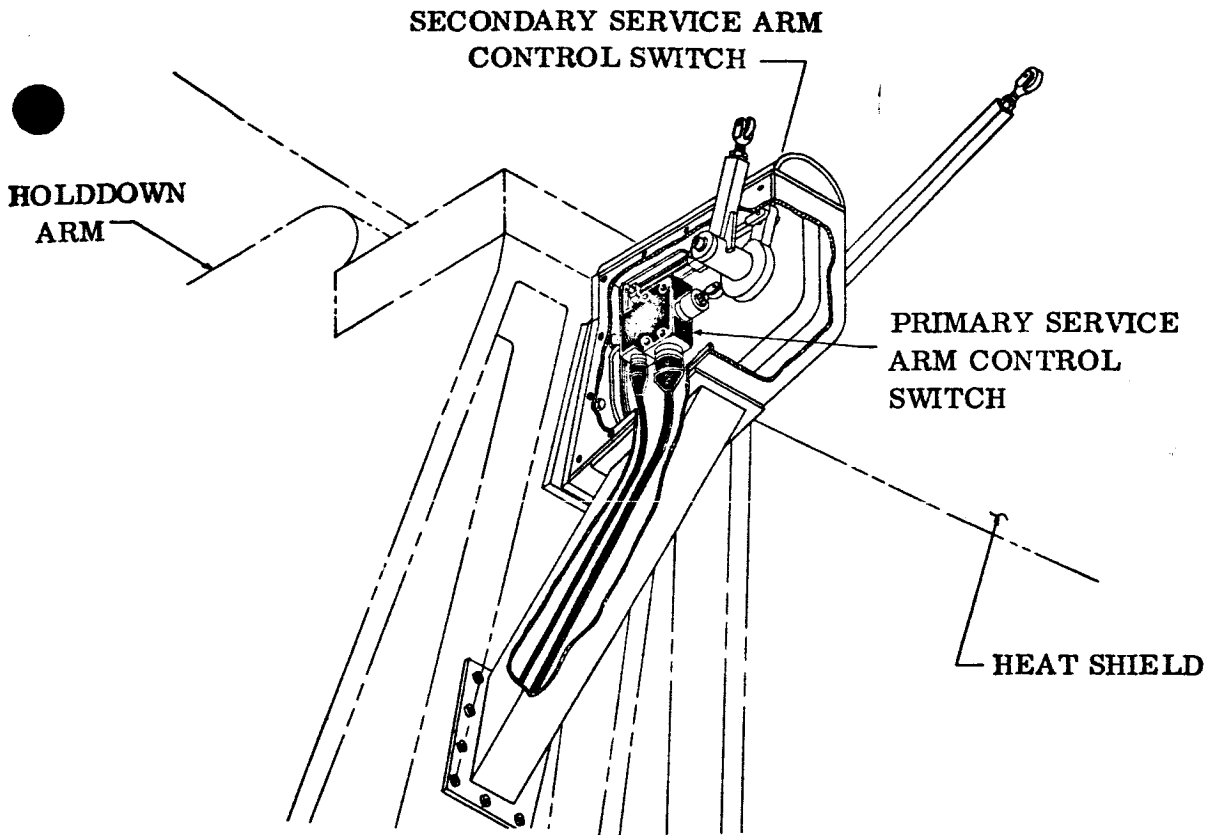


Figure 3-25 Typical Service Arm Control Switches

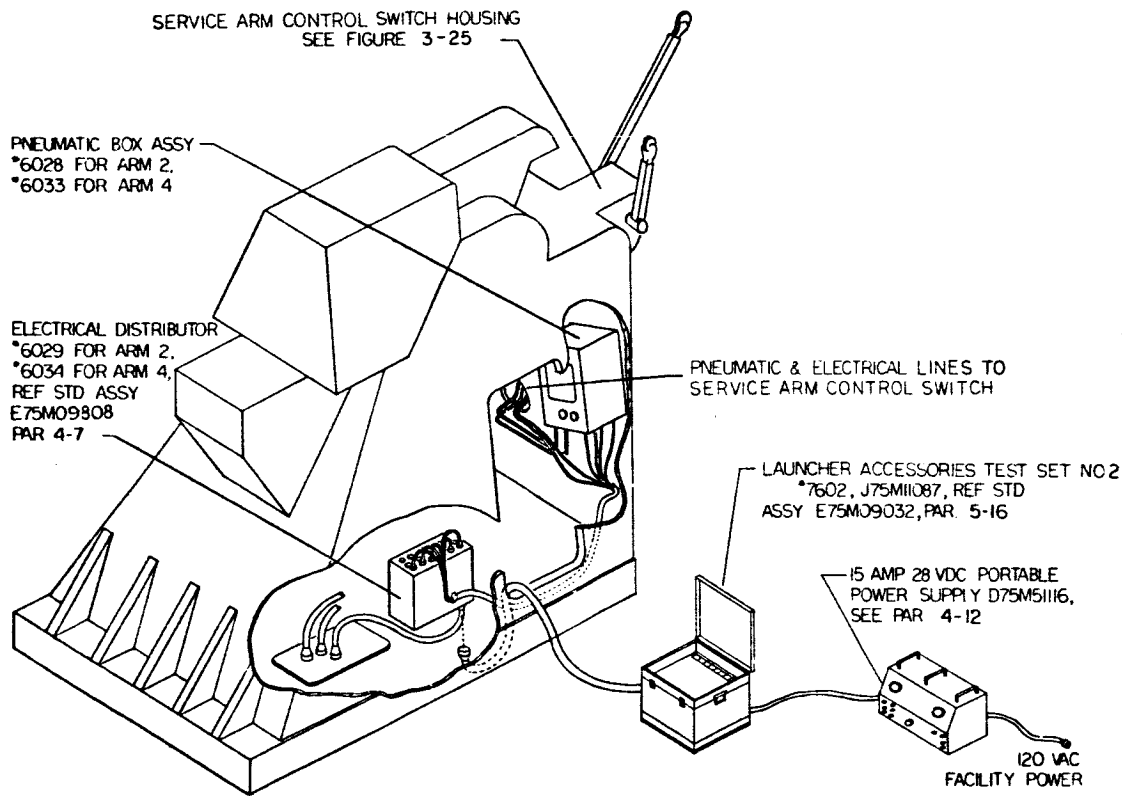
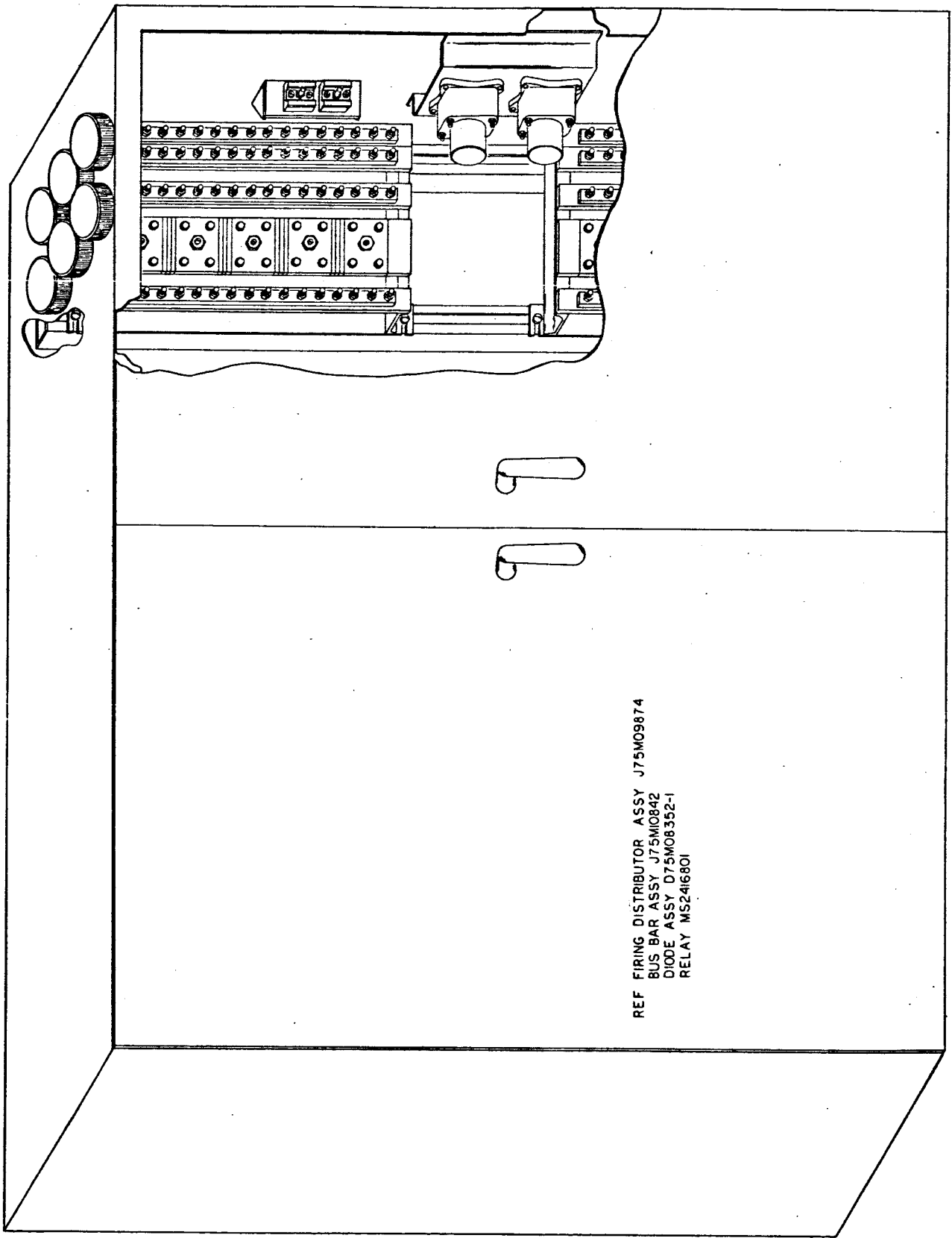
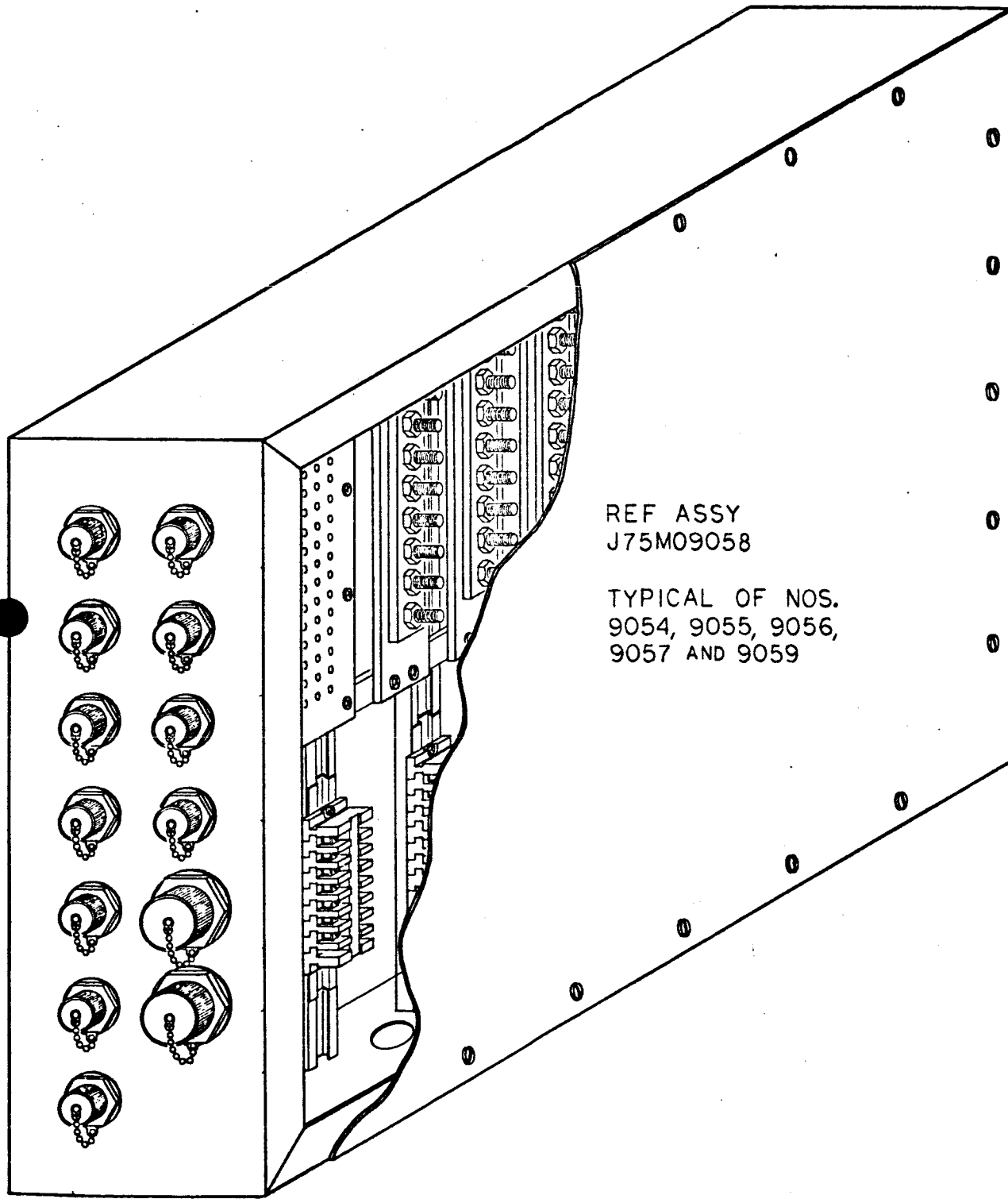


Figure 3-26 Typical Holddown Arm and Test Installation



REF FIRING DISTRIBUTOR ASSY J75M09874  
 BUS BAR ASSY J75M0842  
 DIODE ASSY D75M08352-1  
 RELAY MS2416801

Figure 3-27 Tower MI Firing Distributor #9083



REF ASSY  
J75M09058

TYPICAL OF NOS.  
9054, 9055, 9056,  
9057 AND 9059

Figure 3-28 Service Arm MI Firing Distributor (Typical)



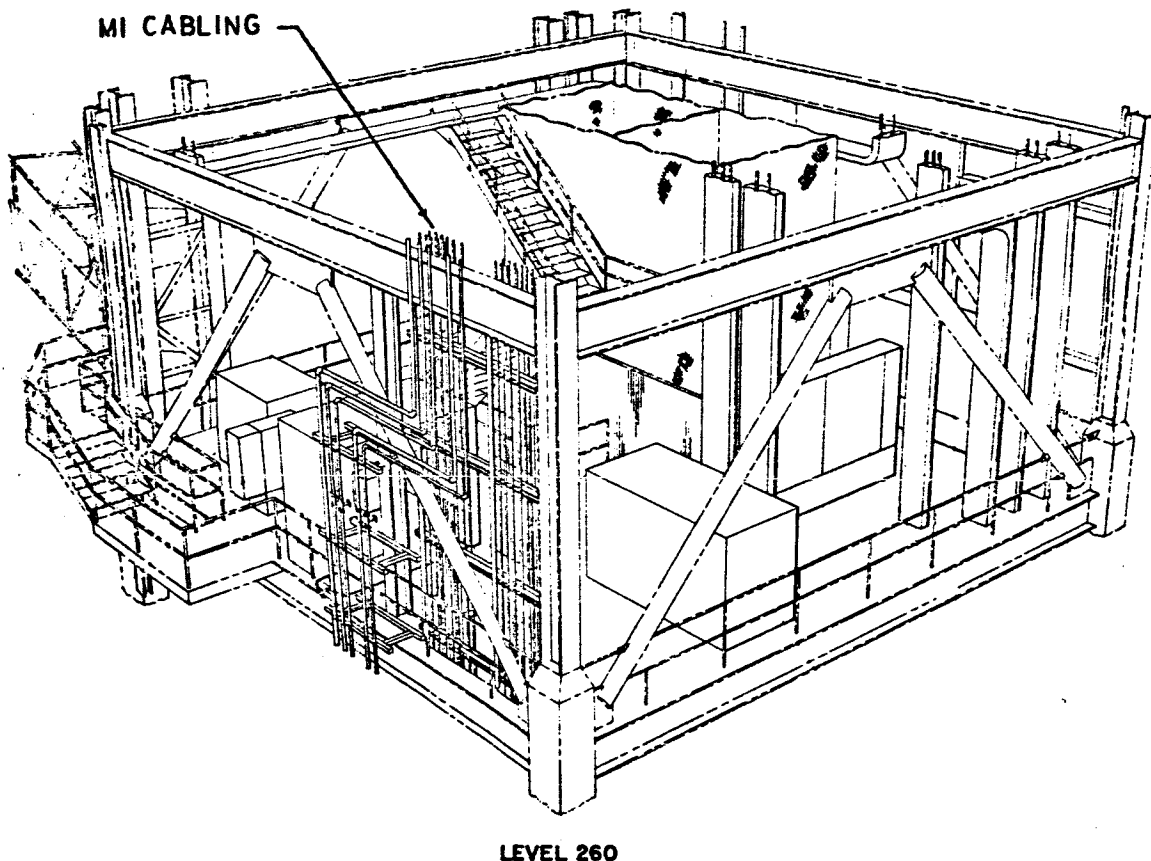


Figure 3-29 MI Cabling, Typical Mobile Launcher Level

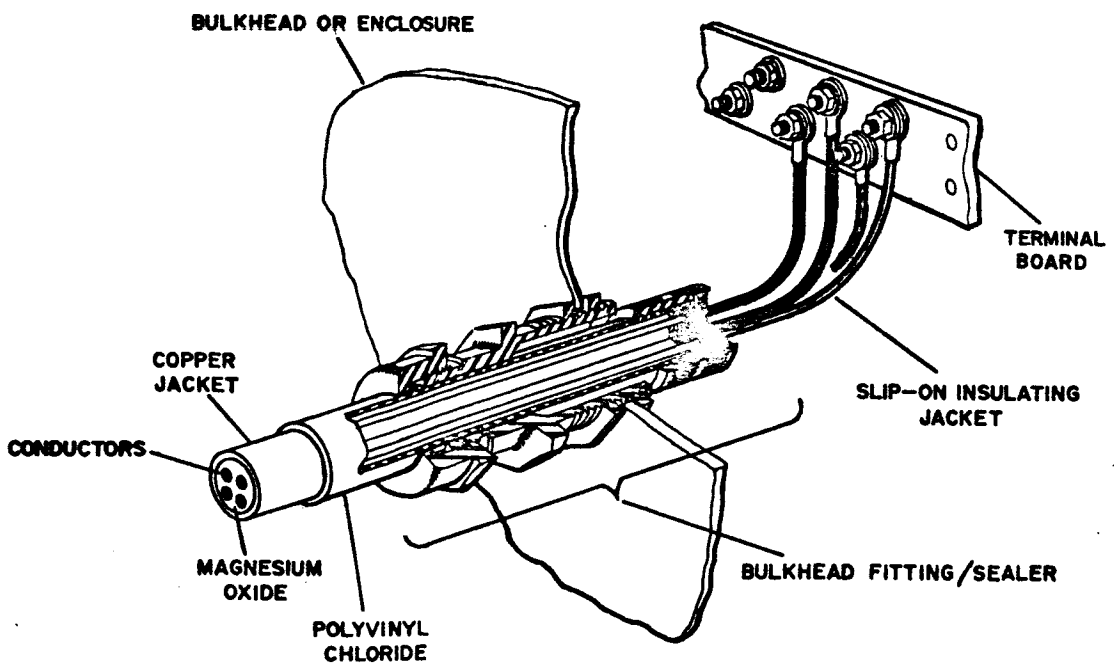


Figure 3-30 MI Cable, Typical Installation

3-28. TAIL SERVICE MASTS SUBSYSTEM.

3-29. Purpose of the Subsystem. This electrical subsystem provides system control, test, and monitor for tail service mast operation. The Tail Service Masts provide support for fuel, electrical and air conditioning lines which must enter the aft section of the S-IC vehicle.

3-30. Equipment for the Subsystem. The Tail Service Masts Subsystem uses the following equipment:

- a. TSM 1-2 #6006, 3-2 #6007, and 3-4 #6005, Level 0.
- b. TSM Control Distributor #6009, TSM 1-2
- c. TSM Control Distributor #6010, TSM 3-2
- d. TSM Control Distributor #6008, TSM 3-4
- e. Control Test and Monitor Consoles, LCC (TSM 1-2 #5420, TSM 3-2 #5421, TSM 3-4 #5422)
- f. TSM Portable Test Set #7601, Level 0
- g. Relay Racks, room 8-A. See Racks 18 and 19, figure 6-2
- h. Power Distribution Rack #6422, room 8-A. See Rack 29, figure 6-2
- i. Power Supply Rack #6421, room 8-A. See Rack 6, figure 6-2
- j. Standby Battery Rack #6408, room 8-A. See Rack 5, figure 6-2
- k. Terminal Distributor #9091, room 13-A
- l. Terminal Distributor #9092, room 5-A
- m. Terminal Distributor #9082, room 7-A
- n. Portable DC Power Supply, Level 0

3-31. Description of the Subsystem. The Tail Service Masts (TSM) are operated by pneumatically pressurized hydraulic fluid and are electrically controlled. Each mast has 31 electrical and electro-mechanical components and an electrical control distributor located within its structure. Component control and monitoring signals are transferred into 60-conductor cables via these control distributors and routed through an interface plate and facility distributors to two Launcher Accessory Relay Distributors (6650 and 6651) located in Room 8-A of the Mobile Launcher.

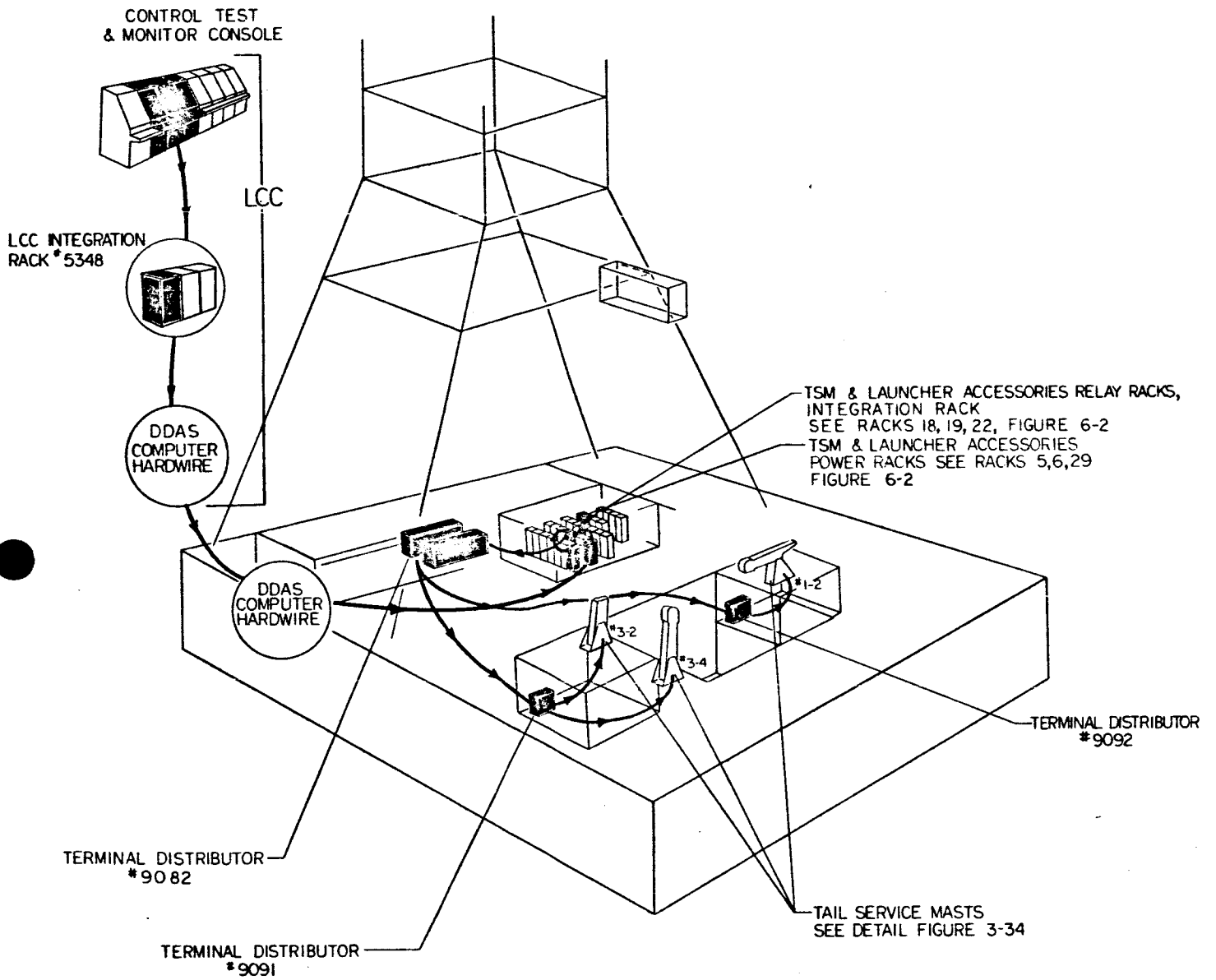


Figure 3-31 Tail Service Masts Subsystem

Relay closures in the relay distributors provide control logic and permit the transfer of 28 volts dc via the terminal and control distributors to limit and pressure switches and solenoid-operated valves within each TSM. Transducer excitation voltage (50 vdc) is transferred from the MSFC-Astrionics integration distributor through the relay distributor and the terminal and control distributors to the transducers. See figure 3-31.

The three control panels provide visual displays of all critical parameters for the TSM subsystem. These panels also provide controls for pneumatic and hydraulic charging and mast retract tests. The masts can be extended locally only by a control on the side of each mast or by a test set (a typical panel is illustrated in figure 3-32). Test and control functions may be programmed into the LCC Mobile Launcher computer, but all signals so programmed can be manually overridden from the control panel.

3-32. Local Testing of the Tail Service Masts. For local control and testing, a portable TSM Test Set (see figure 3-34) is employed. This is a special programming of the Universal Patchboard Test Set. The face panel of this test set is illustrated in figure 3-33. Two cables from the TSM Test Set replace cables from the base of the TSM Control Distributor. Power is supplied to the test set by a portable 28 volt dc supply, paragraph 4-12. By use of the test set, the masts may be individually retracted or extended and each electrical component within the masts may be controlled and monitored.

<b>LCC Control Pnl Dwg</b>	
<b>Tail Service Mast 1-2</b>	<b>E75M09638</b>
<b>Tail Service Mast 3-2</b>	<b>E75M09639</b>
<b>Tail Service Mast 3-4</b>	<b>E75M09637</b>

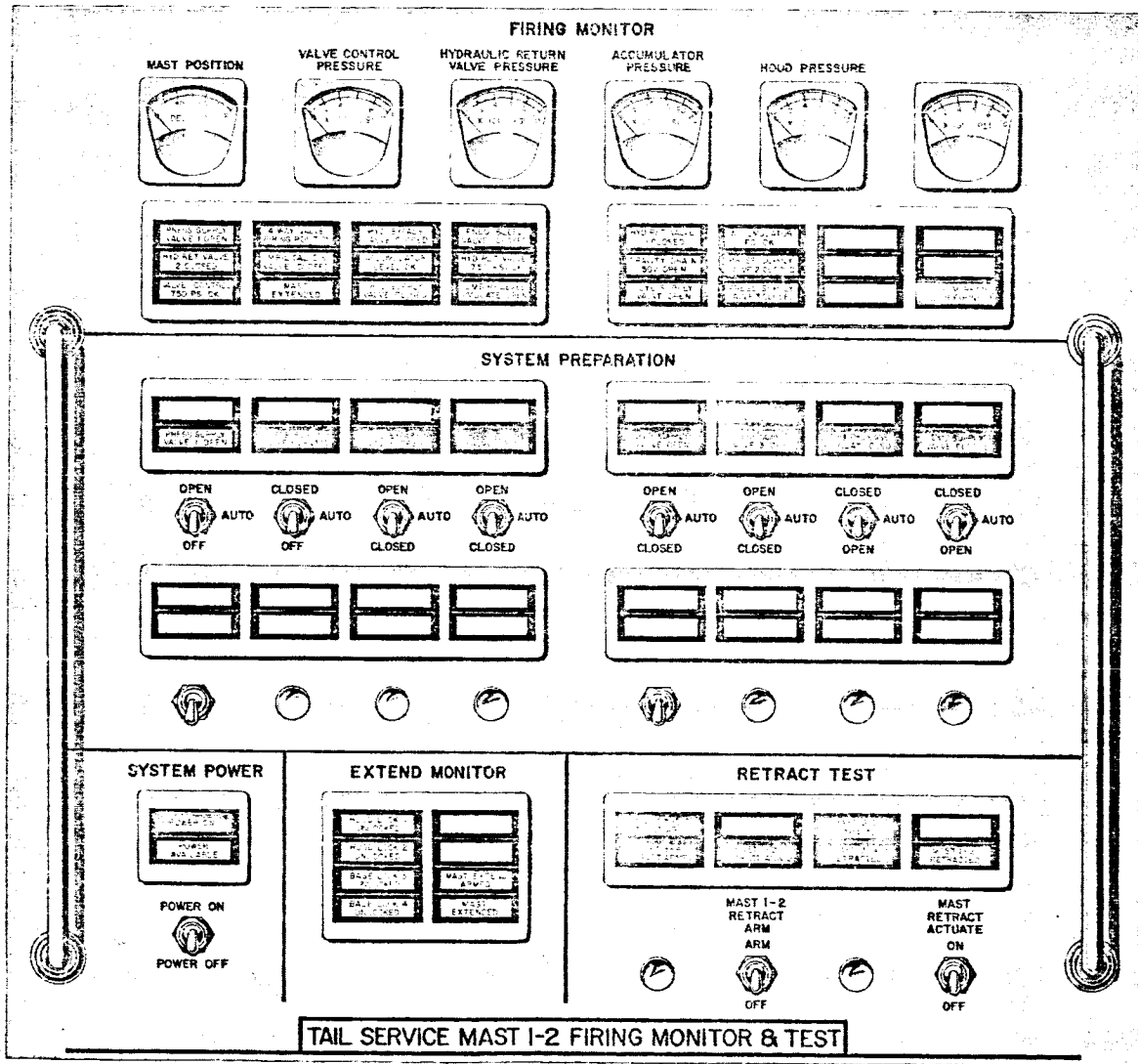


Figure 3-32 LCC Control Panel, Tail Service Mast

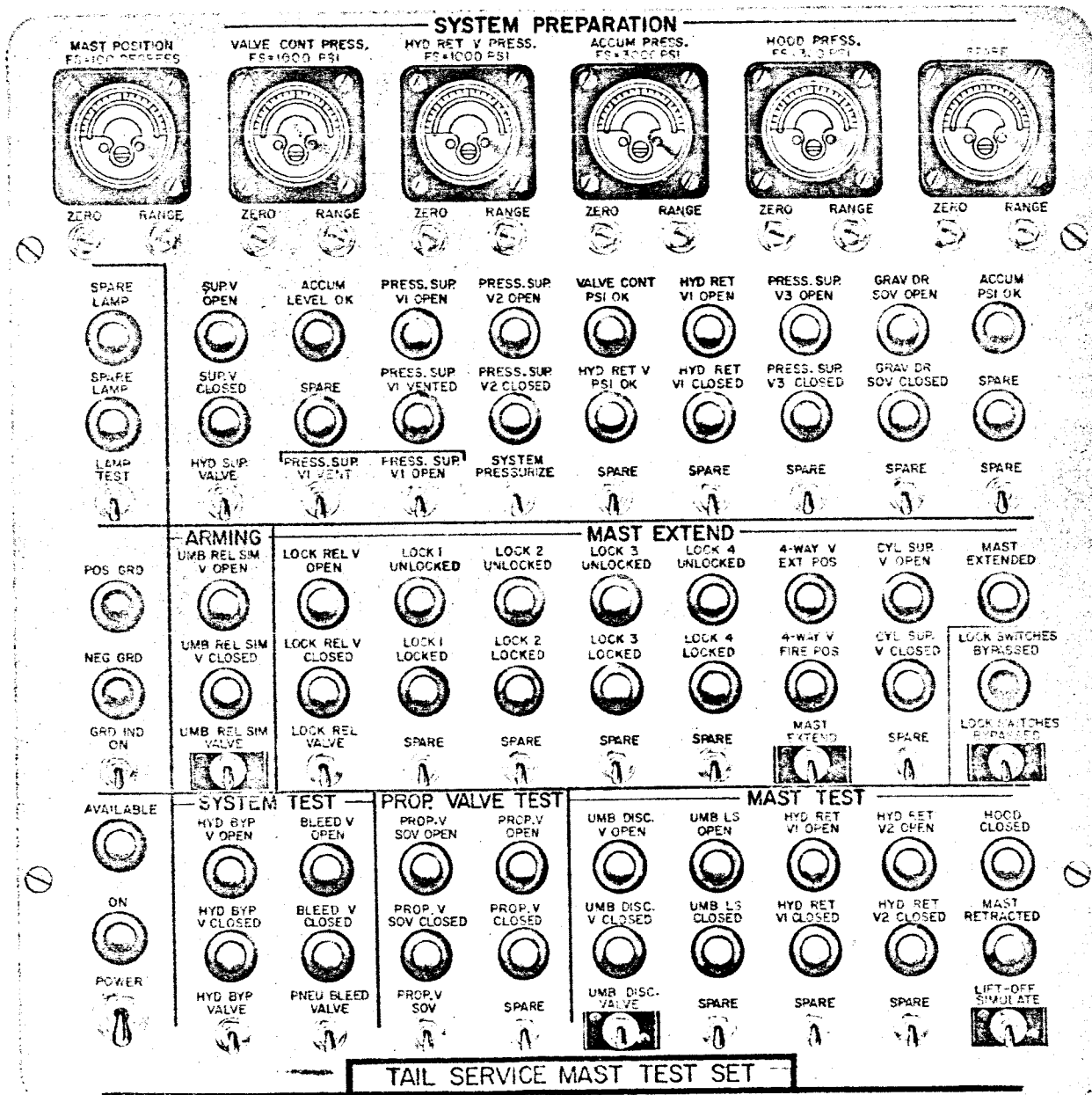


Figure 3-33 Face Panel, TSM Test Set E75M07658

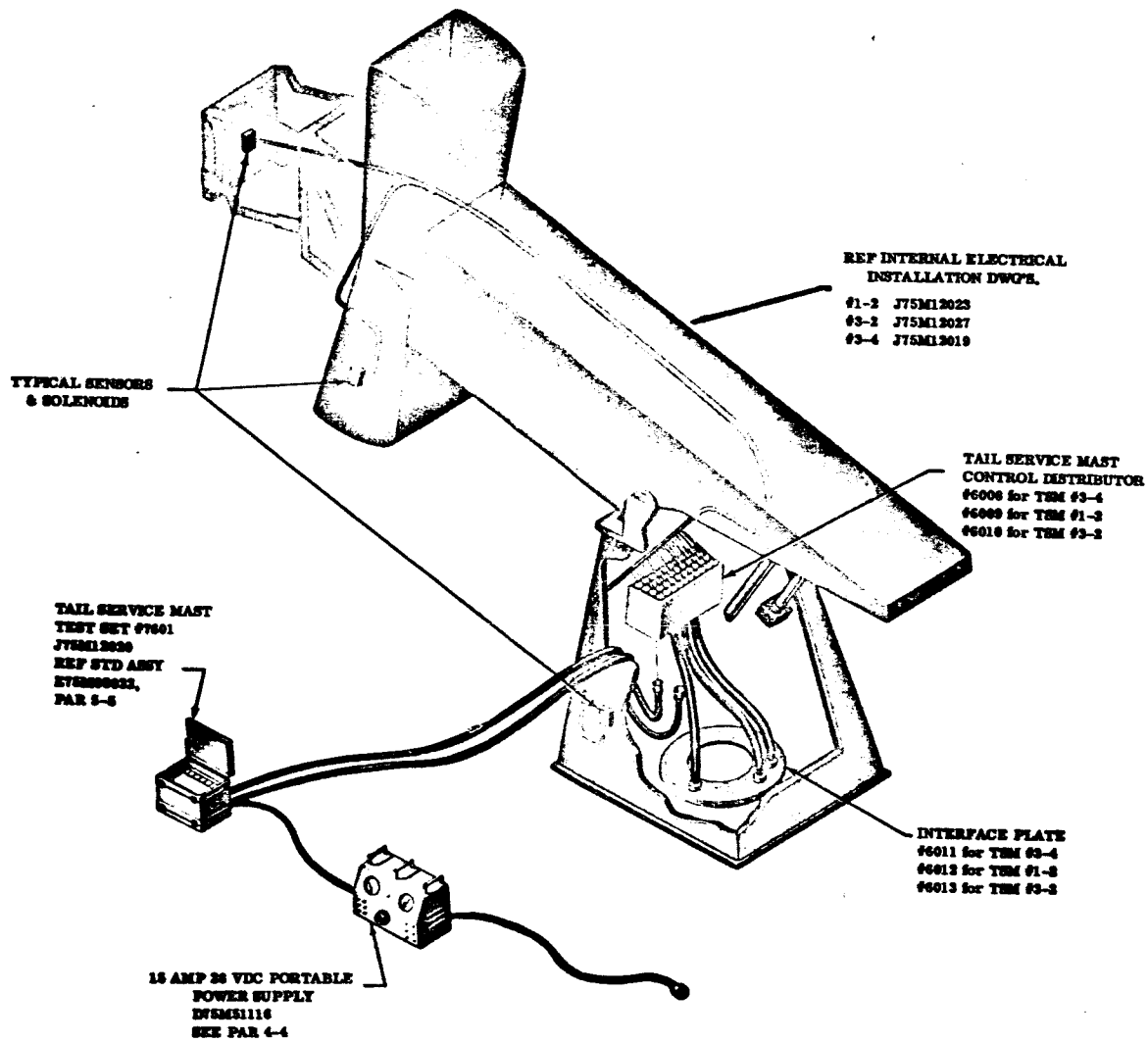


Figure 3-34 Typical Installation, TSM Test Set E75M12020

3-33. PNEUMATICS CONTROL AND DISTRIBUTION SUBSYSTEM.

3-34. Purpose of the Subsystem. The Mobile Launcher pneumatic electrical system provides electrical controls for operating valves and monitoring pressure and valve positions for the  $\text{GN}_2$  and helium high pressure valving and panels, Valve Panel No.11 and Valve Panel No.12, and also the purge valves. These panels and valves supply pneumatic pressure for the holddown arms, tail service masts, service arms, hydraulic checkout console, Q-ball, engine servicing and purge applications.

3-35. Equipment for the Subsystem. The following equipment is used by the Pneumatics Control and Distribution Subsystem:

- a. Helium Control Panel #6404, room 1-B
- b.  $\text{GN}_2$  Control Panel #6405, room 1-B
- c. High Pressure Pneumatics Control Distributor #6407, room -1B
- d. Valve Panel No.11 #6605, room 12-A
- e. Valve Panel No.11 Control Distributor #6667, room 12-A
- f. Holddown Arm Control Distributor #6023, room 12-A
- g. Valve Panel No.12 #6167, level 160
- h. Valve Panel No.12 Control Distributor #6165, level 160
- i. Relay Distributor #6651, Room 8-A. See rack 19, figure 6-2.
- j. UB Instrumentation and Control Distributor #9016, level 160.
- k. Terminal Distributor #9082, room 7-A
- l. Terminal Distributor #9091, room 12-A
- m. Terminal Distributor #9092, room 5-A
- n. Control, Test and Firing Monitor Console, LCC

3-36. Description of the Subsystem.  $\text{GN}_2$  and helium valves, valve positions, and pressures aboard the Mobile Launcher are controlled and monitored from the Pneumatic Distribution System Panel (figure 3-36). This panel interfaces with the Launcher Accessories Relay Distributor #6651 via the DDAS-Computer-Hardware complex, figure 3-35. The relay rack provides the required logic for control and monitoring functions, and acts as an interface transfer unit. Terminal and control distributors connect this relay rack to the valve panels and valves for each of the following areas:



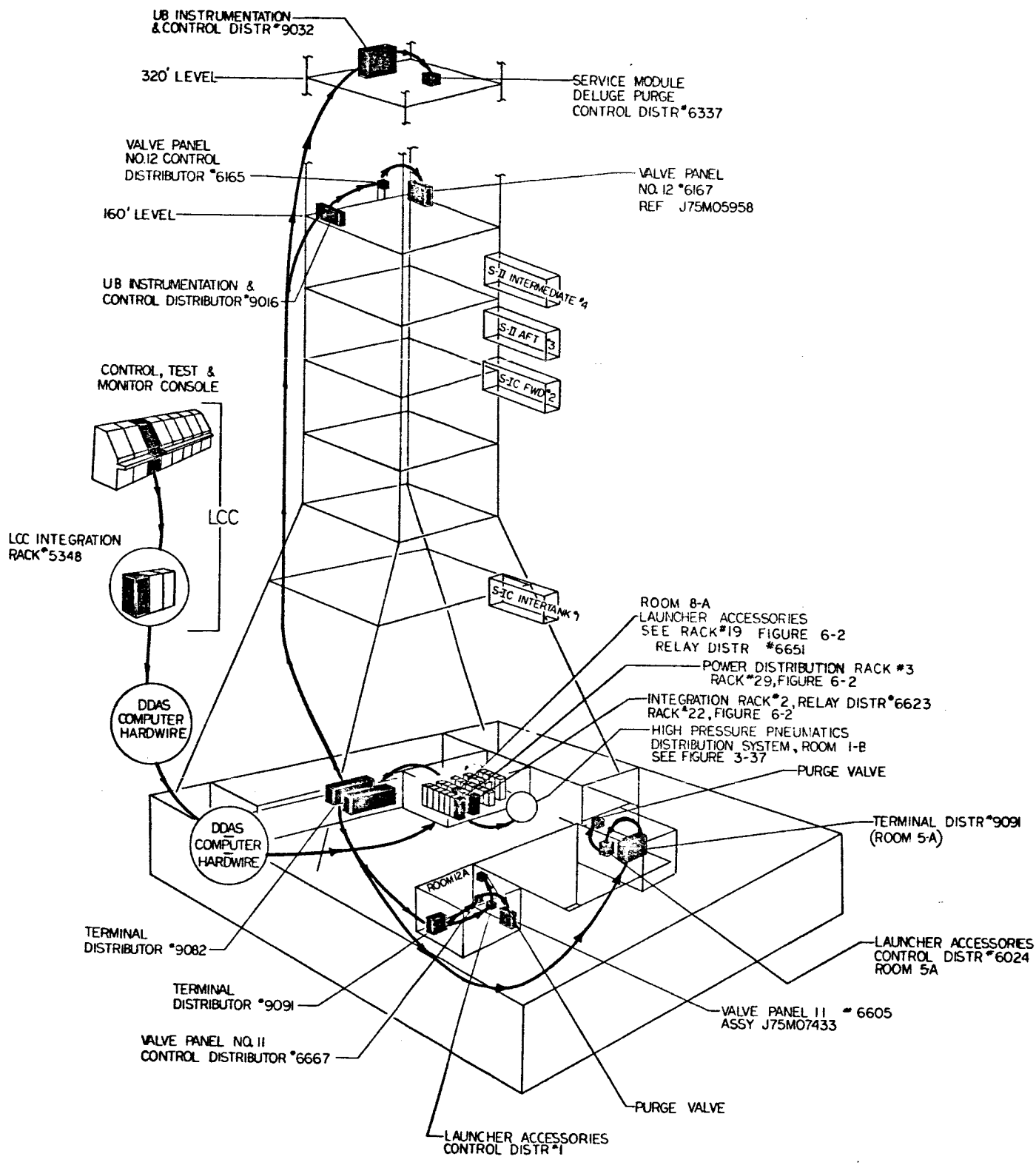


Figure 3-35 Pneumatics Control and Distribution Subsystem

- a. High Pressure Pneumatics System
- b. Valve Panel No. 11, J75M07433
- c. Valve Panel No. 12, J75M05958
- d. Deluge Purge Panel
- e. Launcher Purge Valves

The High Pressure Pneumatics System, figure 3-37, consists of a manifold and valving system for distributing GN<sub>2</sub> and helium to valve panels and consoles throughout the Mobile Launcher.

Valve Panel No. 11 provides:

- a. 3000 psi and 750 psi GN<sub>2</sub> for the Tail Service Masts and for checkout of the ground support hydraulics
- b. 1000 psi GN<sub>2</sub> for engine servicing and cleaning
- c. 1500 psi helium for the holddown arms pneumatics
- d. 50 psi GN<sub>2</sub> for hazardous purge requirements

Valve Panel No. 12 provides:

- a. 750 psi and 125 psi GN<sub>2</sub> for all service arm control pneumatics
- b. 50 psi GN<sub>2</sub> for hazardous purge

3-37. Tests for the Subsystem. Two different test units are used within the Pneumatics Control and Distribution Subsystem for local testing:

- a. Launcher Accessories Test Set No. 1 is a programming of the Universal Patchboard Test Set, figure 5-8. It measures pneumatic pressures, provides actuating test signals and monitors switch and valve positions for the High Pressure Pneumatics system (figure 3-37) and for Valve Panel No. 11 (figure 3-38). These functions are shown on the test set face panel, figure 3-39.

By plug swapping a test cable at the control distributor (#6407) and utilizing a portable power supply, the GN<sub>2</sub> and helium panels may be used for local control and monitoring.

- b. Tower Test Set E75M14137 is a programming of the Miniaturized Universal Test Set, figure 5-6. It measures pneumatic pressures, provides actuating signals, and monitors switch and valve positions for Valve Panel No. 12, figures 3-40 and 3-41. It also provides control and monitor functions for the Q-Ball Cover Removal Subsystem, figure 3-43.

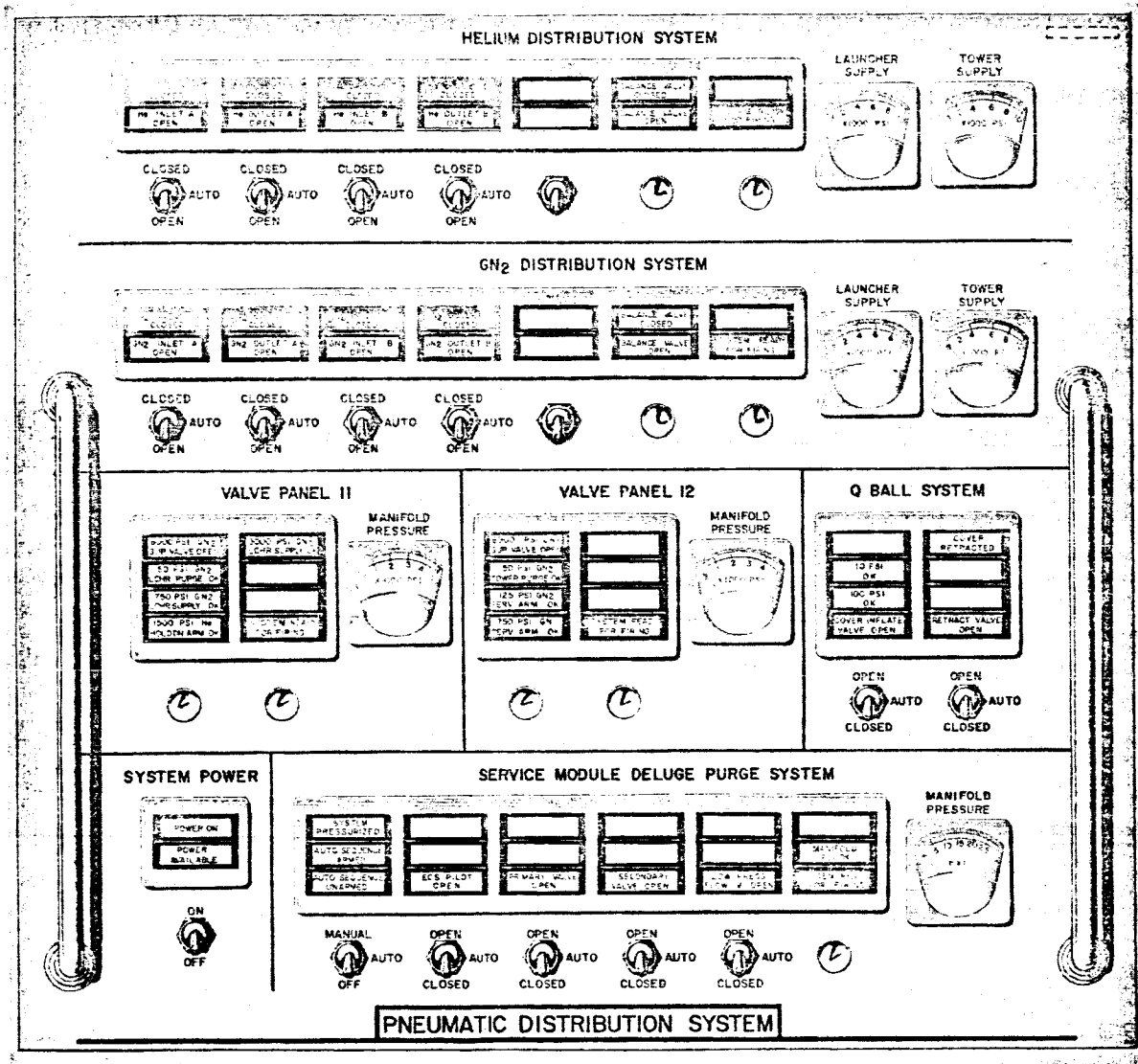
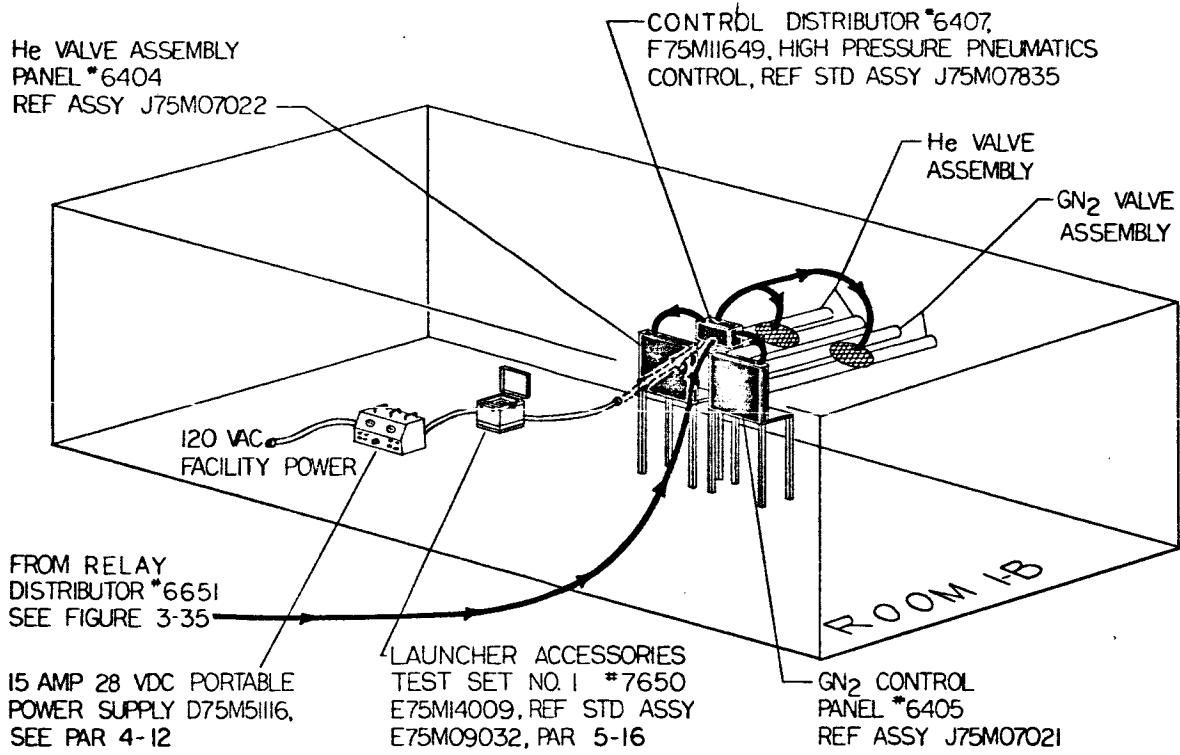
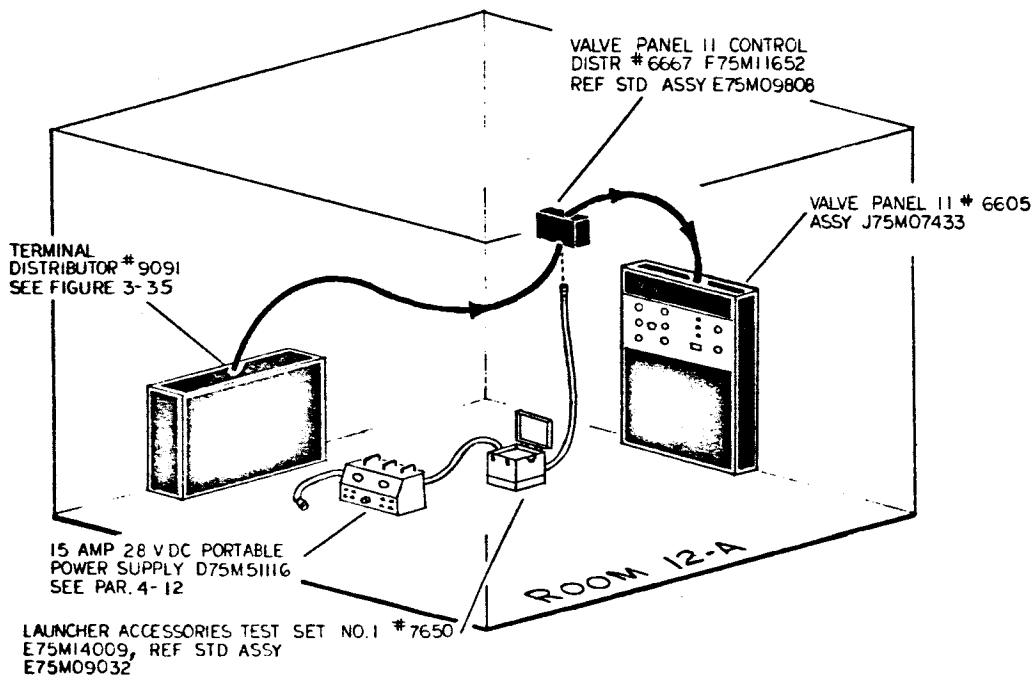


Figure 3-36 LCC Panel, Pneumatic Distribution System



**Figure 3-37 High Pressure Pneumatics System**



**Figure 3-38 Typical Test Installation, Valve Panel 11**

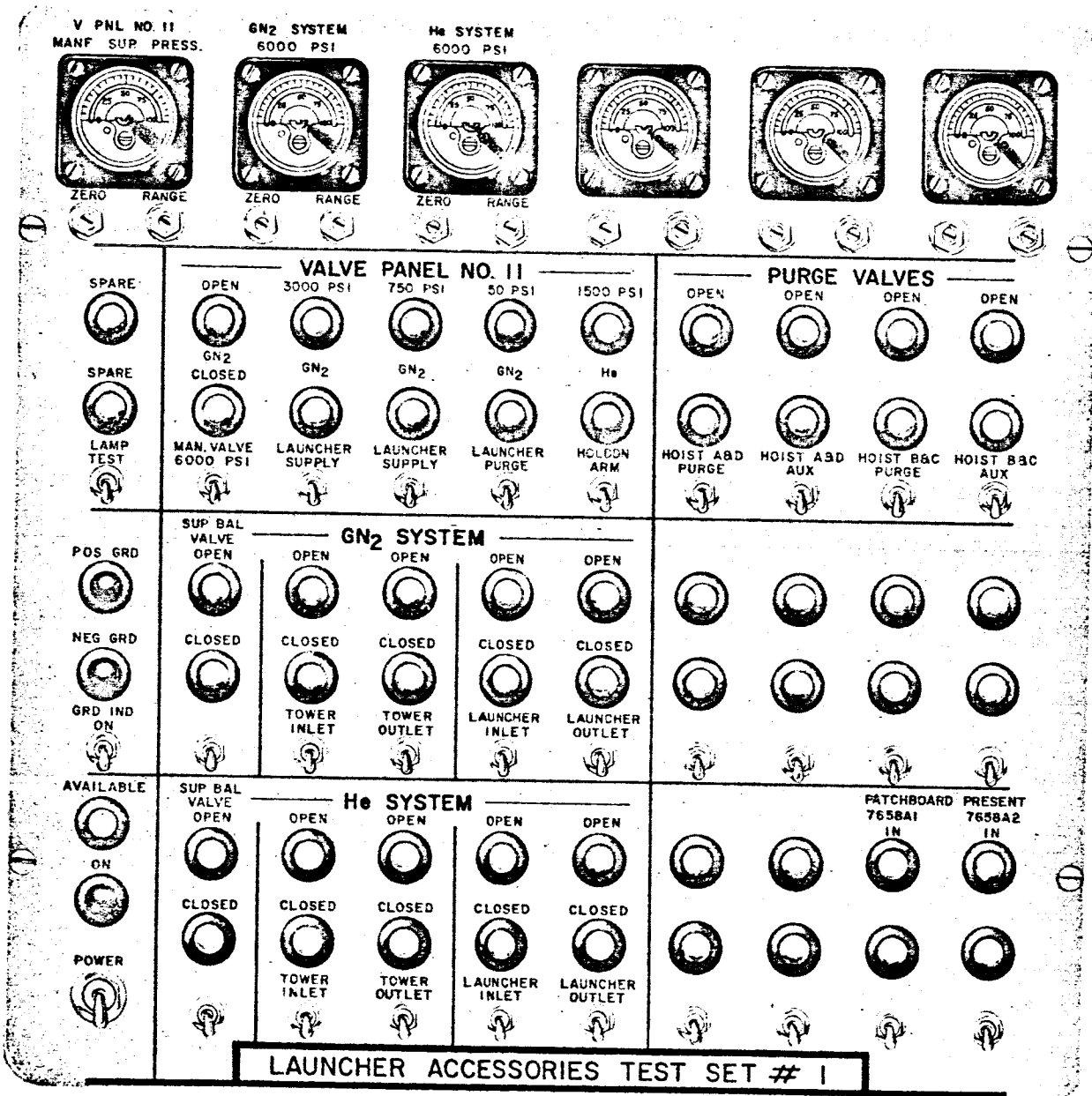


Figure 3-39 Face Panel, Launcher Accessories Test Set No. 1, E75M14009

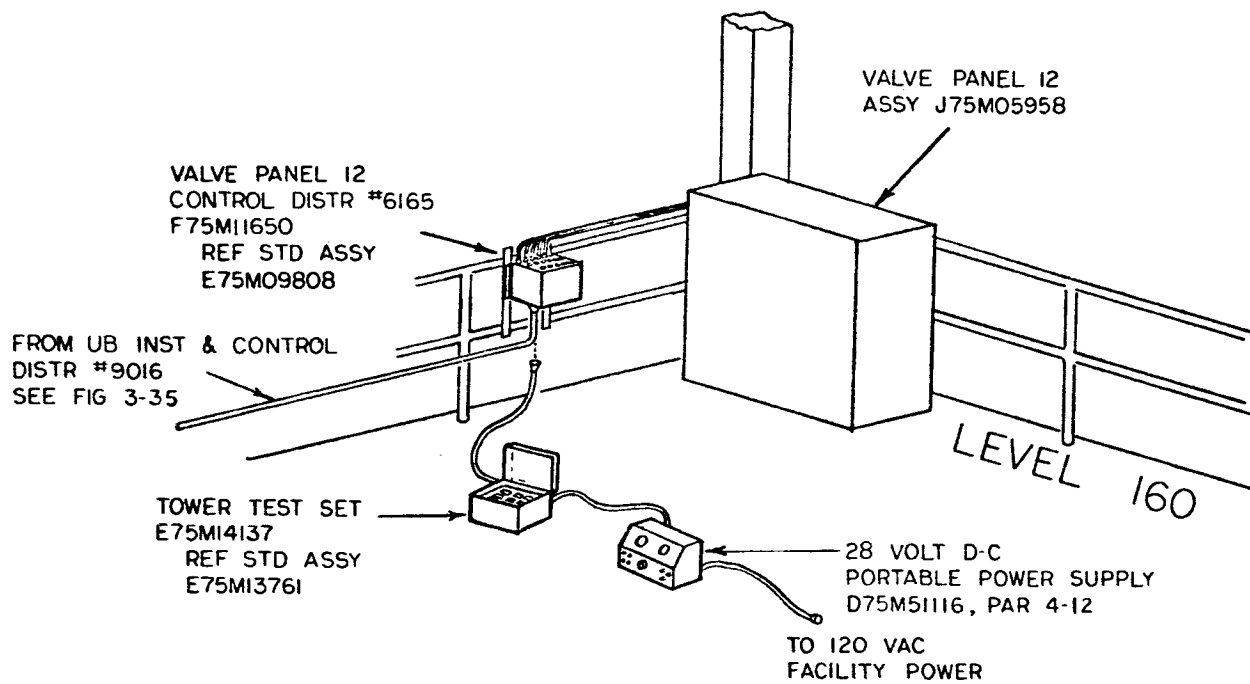


Figure 3-40 Typical Installation, Tower Test Set E75M14137

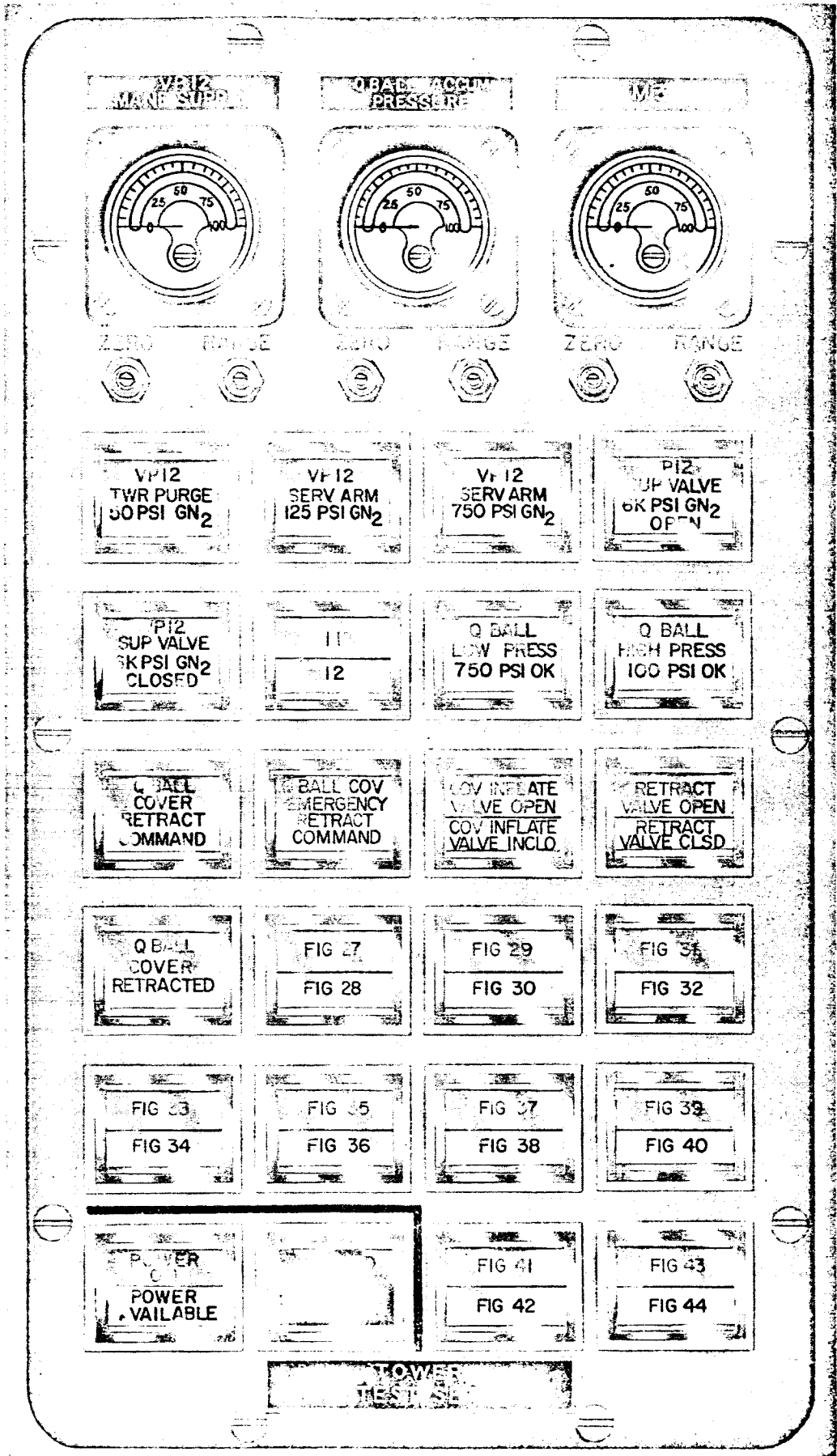


Figure 3-41 Face Panel, Tower Test Set E75M14137

3-38. Q-BALL COVER REMOVAL SUBSYSTEM

3-39. Purpose of the Subsystem. The Q-Ball Cover Removal Subsystem provides electrical control and monitoring of the Q-Ball cover mechanism.

3-40. Equipment for the Subsystem. The Q-Ball Cover Removal Subsystem employs the following equipment:

- a. Hoist Assembly (J75M15383)
- b. Pneumatic Assembly #6360 (J75M15388)
- c. Q-Ball Cover Control Distributor #6361 (F75M13155)
- d. UB Instrumentation and Control Distributor #9032
- e. Terminal Distributor #9082A32
- f. Relay Distributor #6651. See rack 19, figure 6-2.
- g. Panel Assembly Pneumatic Distribution System, LCC Panel, (D75M09674)

3-41. Description of the Subsystem. The Q-Ball Cover is a fiberglass cover for vehicle guidance components in the tip of the Saturn V Launch Escape System rocket. Prior to launch, this cover is ejected by pressurized GN<sub>2</sub> (10 psi). It then is retracted from the vehicle by a Hoist Assembly consisting of a lanyard, a pneumatic motor, and a weight and pulley system (see figure 3-43).

Pneumatic pressure for cover removal is supplied by the pneumatic assembly (#6360) located beside the electrical control distributor (#6361) on the 360-foot tower level. This pneumatic assembly is controlled remotely from the LCC Pneumatic Distribution System Panel, as in figure 3-42.

Several minutes before vehicle liftoff, a manual cover inflation switch on the Pneumatic Distribution System Panel (figure 3-36) is closed, providing a signal routed through the DDAS-Computer-Hardware Complex to Relay Distributor #6651 in room 8-A. Relay closure provides a 28 volt dc signal through distributor #9082A32



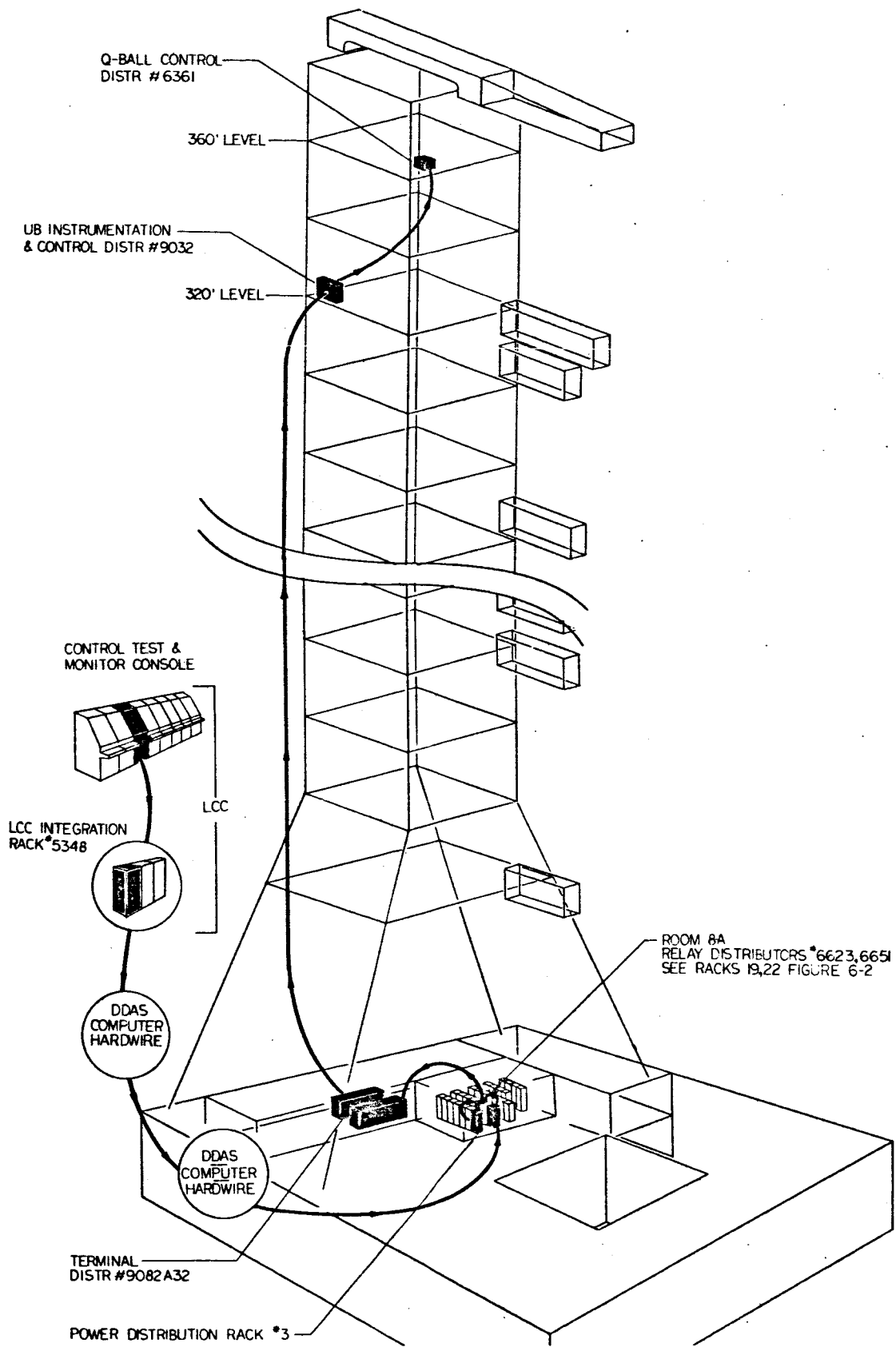


Figure 3-42 Q-Ball Cover Removal Subsystem

and #9032 to a cover inflation solenoid valve in the pneumatic assembly. The  $\text{GN}_2$  (10 psi) is thus supplied through pneumatic hose to a nylon bladder in the Q-Ball Cover. The bladder inflates and raises the Q-Ball Cover.

As the Q-Ball Cover falls, a steel weight in the gravity retract system actuates a motor-start limit switch and energizes a relay. One contact closure energizes another relay in series with the motor-stop limit switch and another seals in the motor-start limit switch. A contact from the relay in series with the motor-stop limit switch energizes the second solenoid valve. This retract valve opens and allows nitrogen at 125 psi to actuate the pneumatic motor and wind up the lanyard. When a mechanical stop on the lanyard reaches the hoist assembly, the motor-stop limit switch is actuated, closing the retract valve. As the cover retracts fully, a limit switch returns a signal to the LCC along a path parallel to the command signals. After the lower limit switch is activated, a Q-Ball Cover retract lamp on the LCC control panel indicates full retraction.

Panel indicator lamps are illuminated whenever the retract valve is open, the cover inflate valve is open, the low pressure reaches 10 psi in the Q-Ball pneumatic assembly  $\text{GN}_2$  accumulator, or the high pressure reaches 100 psi.

In case of emergency, a manual retract switch on the LCC pneumatic distribution system panel may be closed. This switch closure bypasses the relay logic and energizes the retract valve solenoid. The retract valve then opens, allowing 125 psi nitrogen to actuate the pneumatic motor, which in turn pulls a mechanical disconnect pin. This allows the Q-Ball Cover to open and be retracted by the counterweight.

3-42. Local Test. The Q-Ball Cover mechanism can be locally tested by use of Tower Test Set E75M14137. This test set is connected to Q-Ball Cover Control Distributor #6361 in place of the system cable, and provides control and monitor functions shown in figure 3-41.

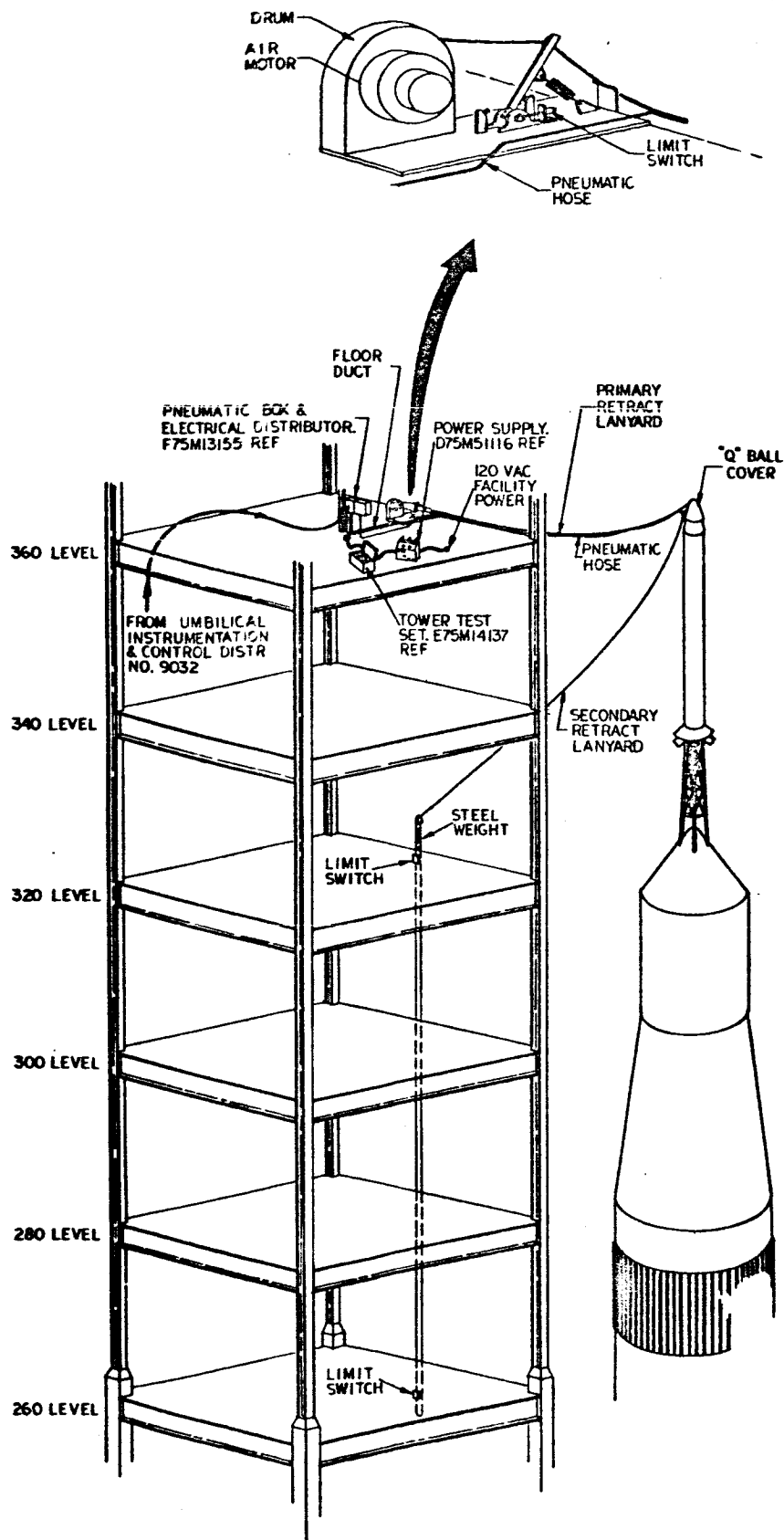


Figure 3-43 Q-Ball Cover Mechanism, Level 360

3-43. ESE POWER SUBSYSTEM.

3-44. Purpose of the Subsystem. This subsystem provides dc power to the ESE.

3-45. Equipment for the Subsystem. The ESE Power Subsystem uses the following equipment:

- a. Relay Distributor #6620 (power distribution only).
- b. Relay Distributors #6613 thru #6619, #6621, #6622, #6650, #6651 (Power distribution and other subsystems), figure 3-53.
- c. Battery Rack #1172, figure 3-52.
- d. Power Supply Rack #1162, figure 3-50.
- e. Power Distribution Racks, figure 3-51. See Racks 7, 8, 29, figure 6-2.
- f. UB Power Terminal Distributors #9042, #9044, #9054, #9017, #9023, #9027, #9029, #9046, and #9031 on various launcher levels as shown in figure 3-44.
- g. Terminal Distributor #9082, Room 7-A.
- h. Terminal Distributor #9073, Room 7-B.
- i. Control Distributors #6008, #6009, #6010, #6023, #6024, #6063, #6064, #6122, #6124, #6142, #6144, #6162, #6202, #6203, #6204, #6222, #6223, #6224, #6262, #6263, #6264, #6265, #6302 #6303, #6304, #6667 and #6407 on various launcher levels, figure 3-44.
- j. Power Control Console (Astrionics)

3-46. Description of the Subsystem. Power for ESE electrical equipment consists of (a) 28 volt monitor power, (b) 28 volt control power, and (c) power for DDAS and hardwire transmission. These three types of power are discussed separately below:

(a) 28 volt monitor power is applied to limit switches, pressure switches, and other components which monitor the state of the ESE subsystems. It is also applied to the ESE patch distributors where relay contacts monitor the state of the subsystems. Power in this category is generated in power supply 201-1162, and is distributed directly to the monitor switches and relay contacts without any command switching from a control panel. Monitor power for the service arms is shown in figure 3-44 and that for the service arms patch distributors in figure 3-45. Monitor power for the launcher networks and corresponding patch distributors is shown in figure 3-46.

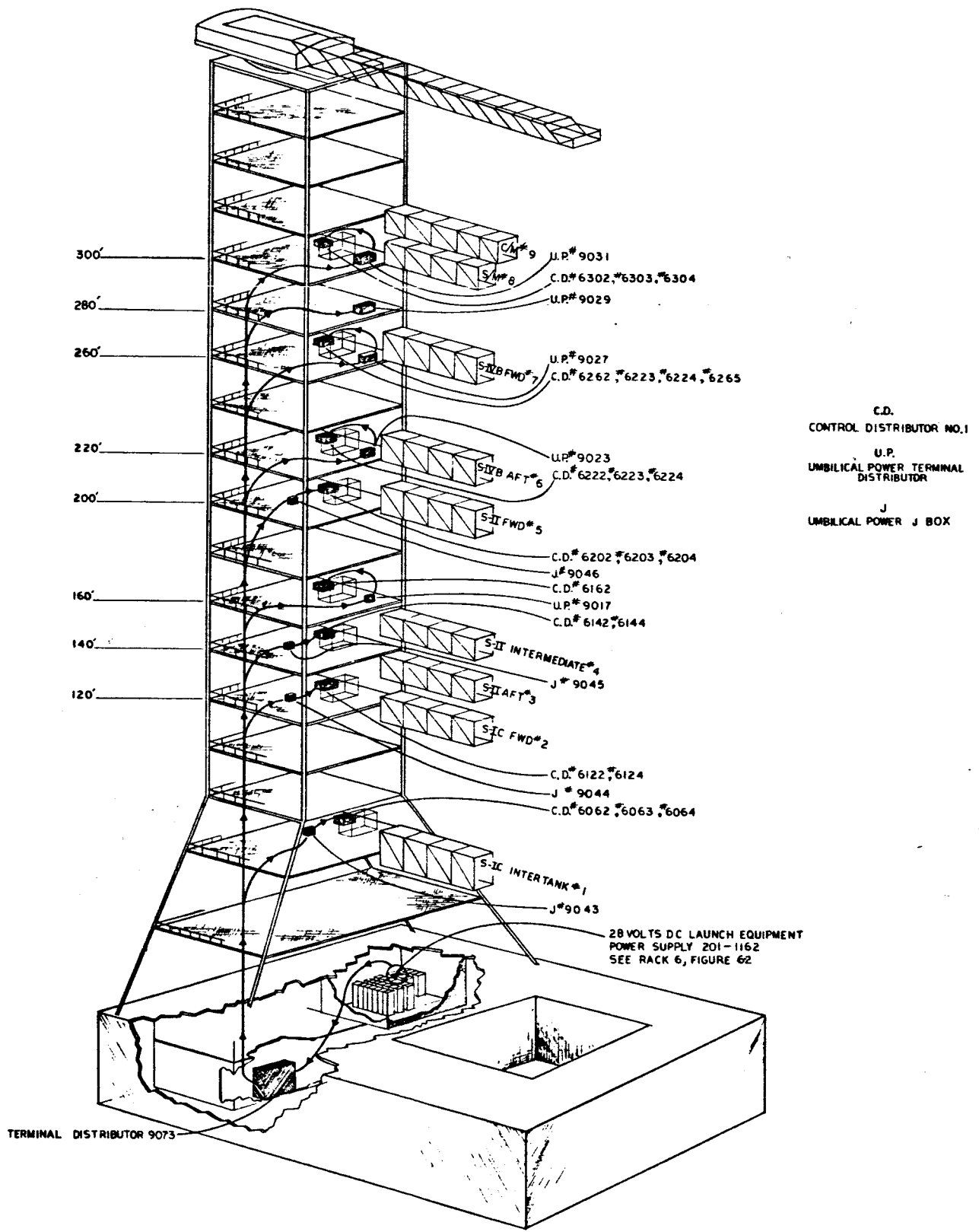


Figure 3-44 Monitor Power for Service Arms

(b) 28 volt control power is that power required for relays, solenoid-operated valves, and other components which effect actuation of the service arms, tail service masts, and launcher accessories. It is applied to buses in the patch distributors through power modules (contactors) which are controlled from the LCC control panels, as shown in figure 3-47. A similar control power is supplied to a bus in the MSFC-Astrionics signal conditioning equipment (figure 3-48) which returns the power through relay contact closures.

c. Power for DDAS and hardwire transmission as well as for the LCC control panels is supplied by Astrionics to buses in the integration racks and patch racks. See figure 3-49.

The source power for the system described above is supplied by MSFC and consists of a basic 500 amp power supply rack (#1162) and a standby battery rack (#1172). Operation of these racks is controlled from an LCC power control console also furnished by MSFC. See figure 3-55.

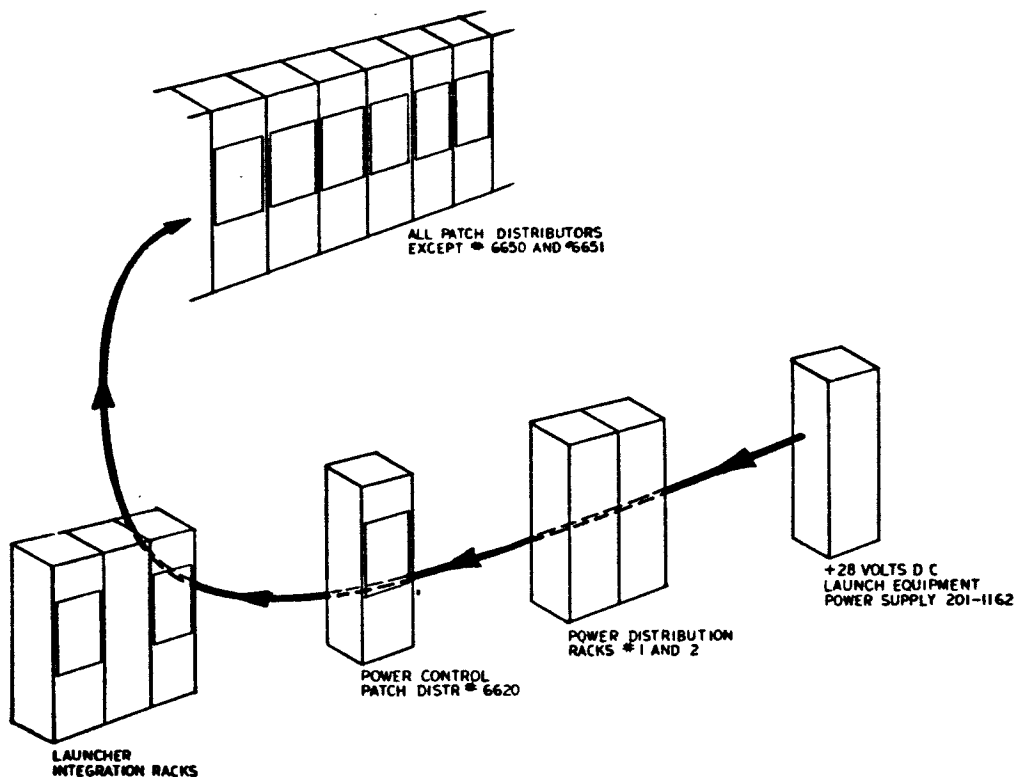


Figure 3-45 Monitor Power for Service Arms Patch Distributors

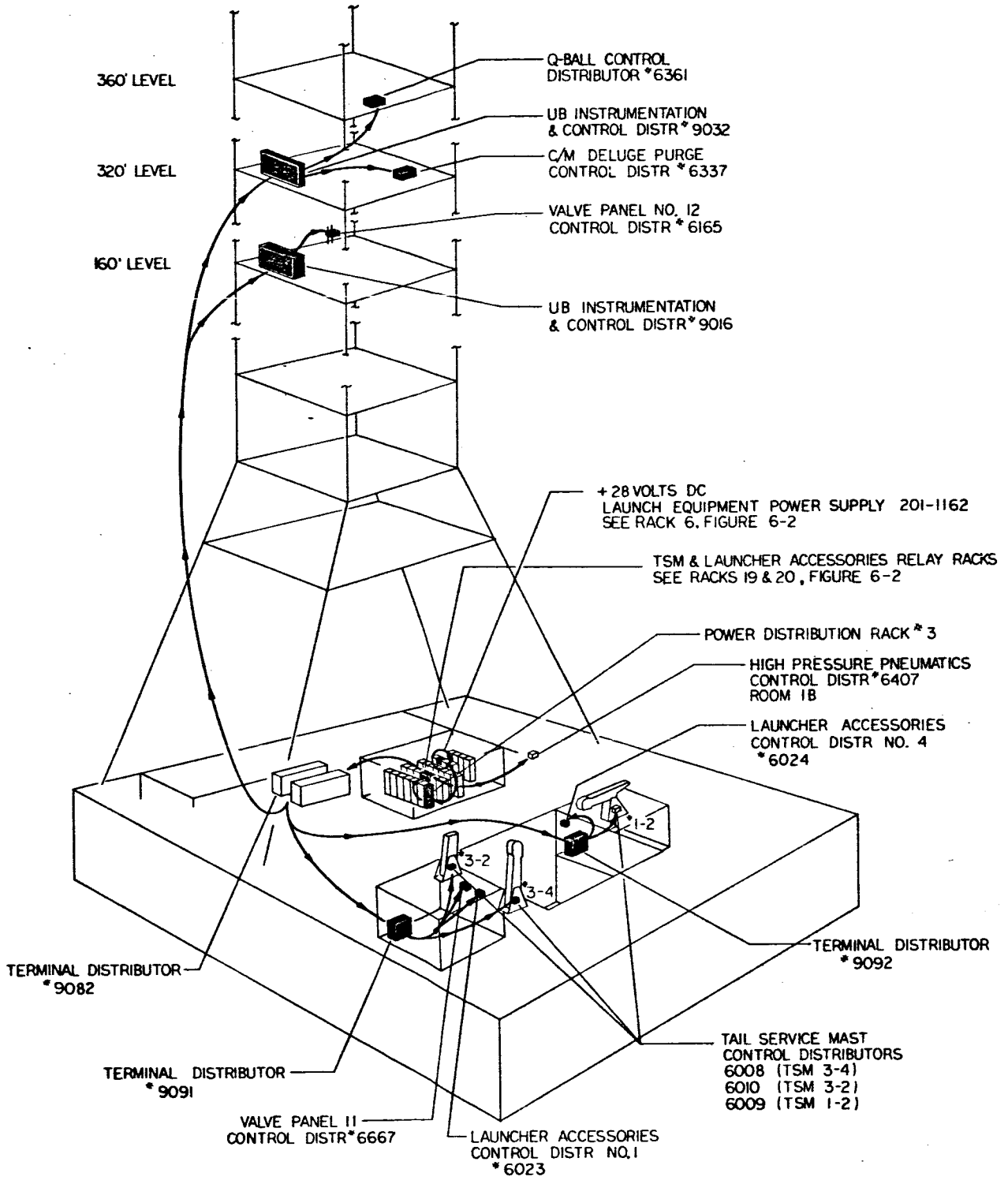


Figure 3-46 Monitor Power for Launcher Networks and Patch Distributors #6650 and #6651





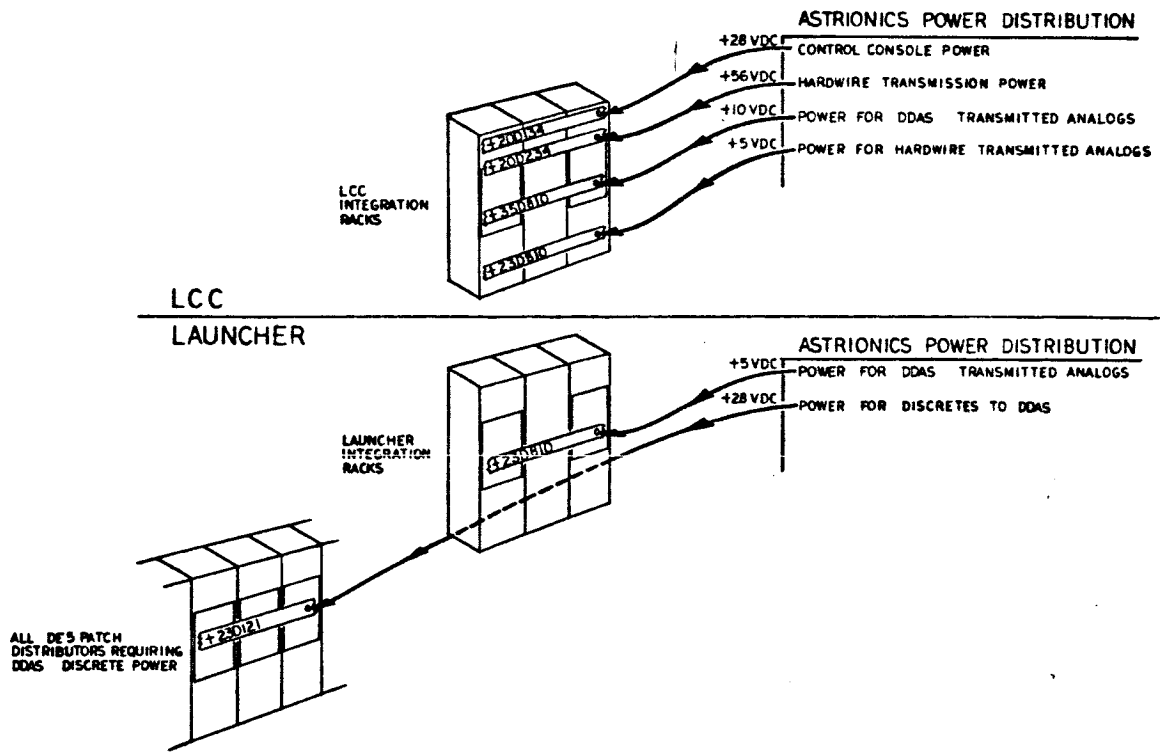


Figure 3-48 Control Power for Astrionics Signal Conditioning Equipment

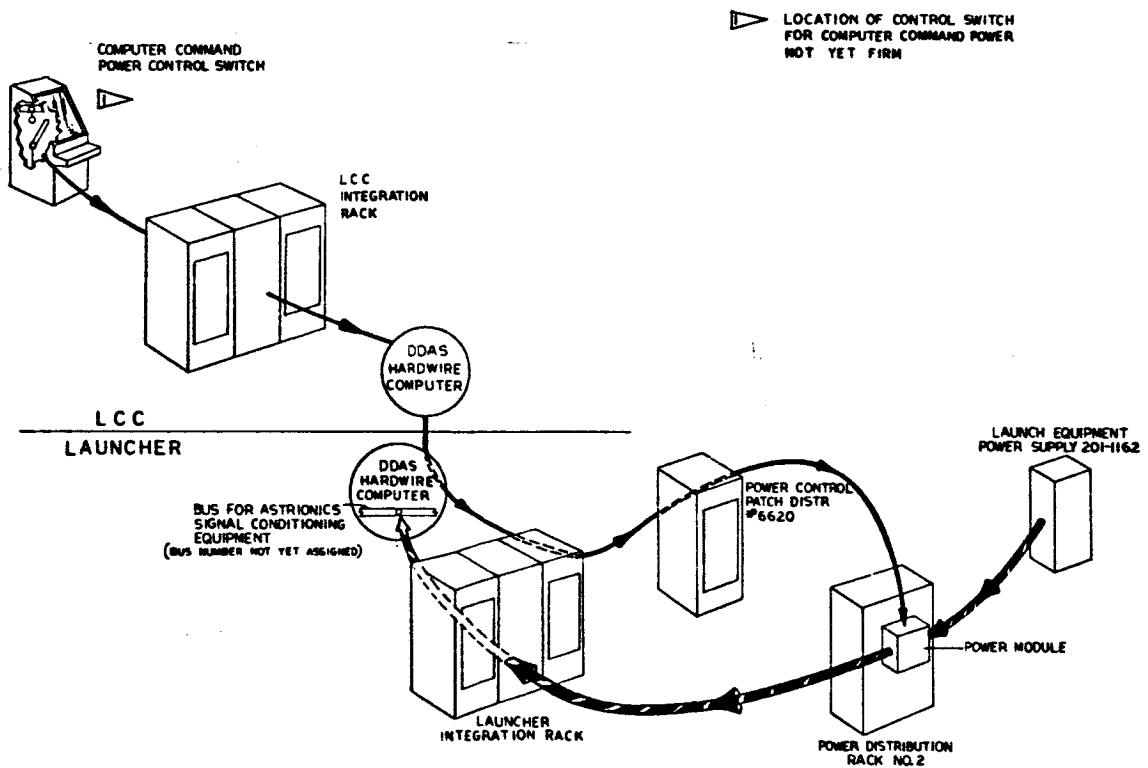


Figure 3-49 Power for Hardwire and DDAS Transmission

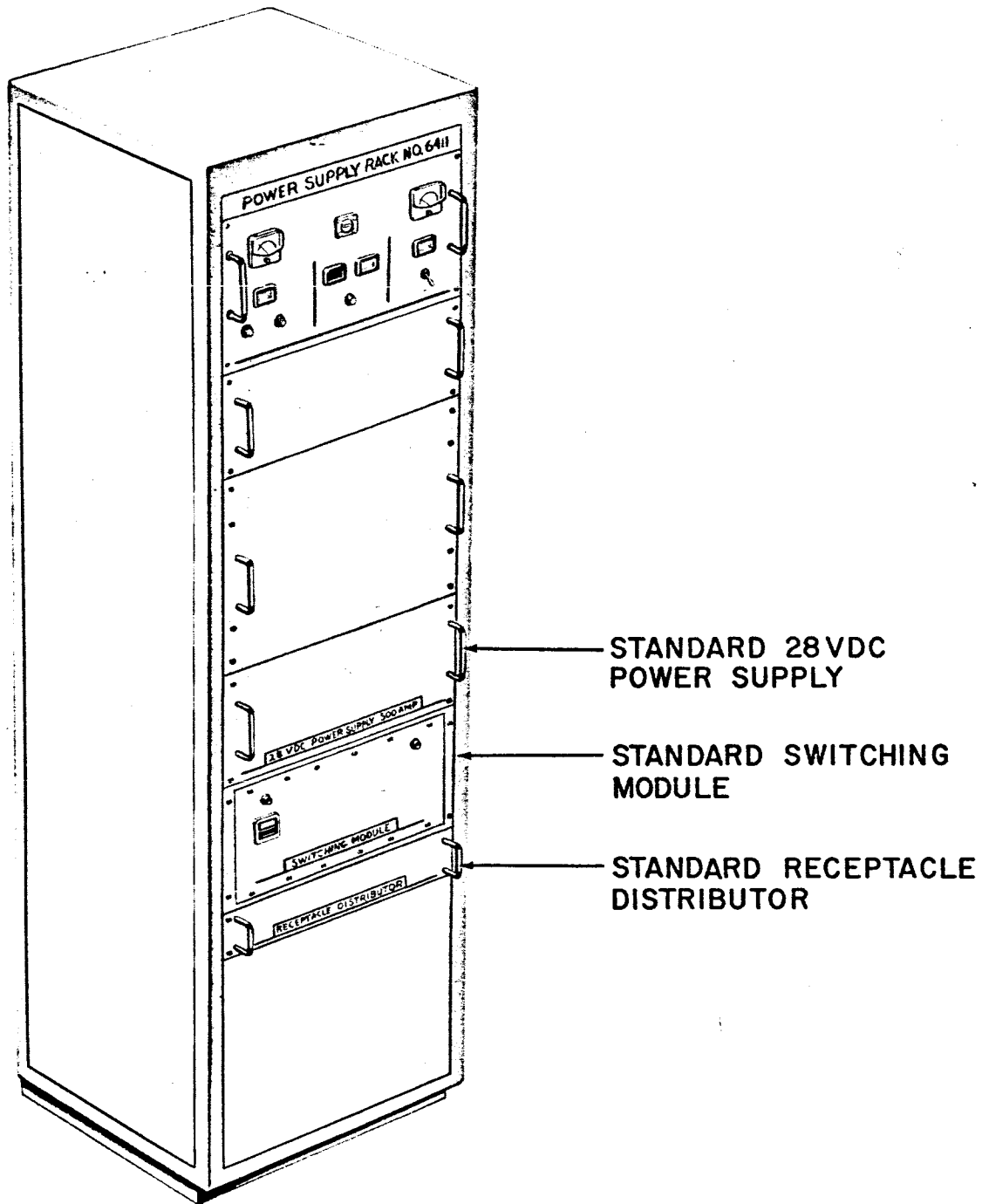


Figure 3-50 Power Supply Rack (Typical)

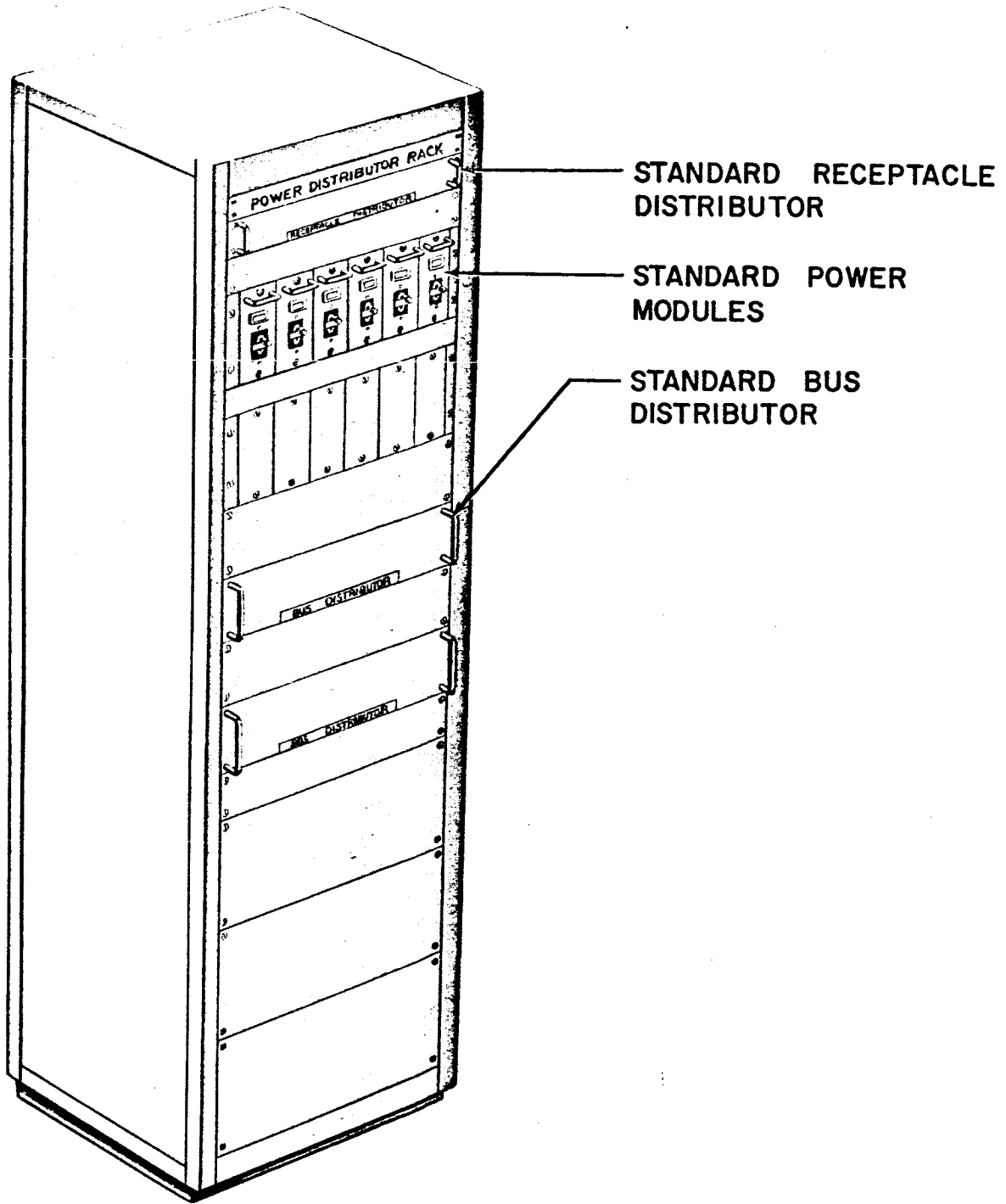


Figure 3-51 Power Distribution Rack (Typical)

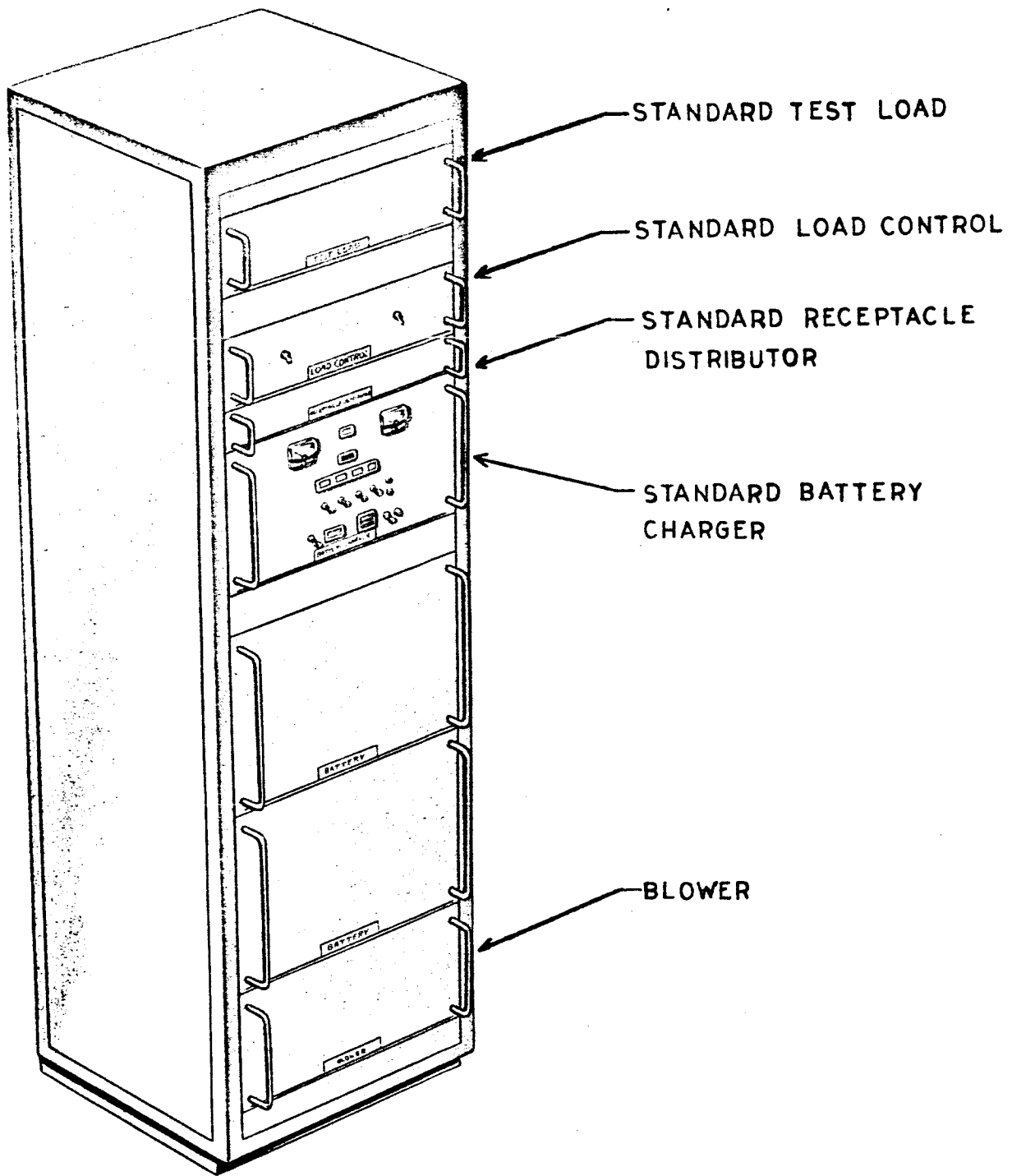


Figure 3-52 Battery Rack (Typical)

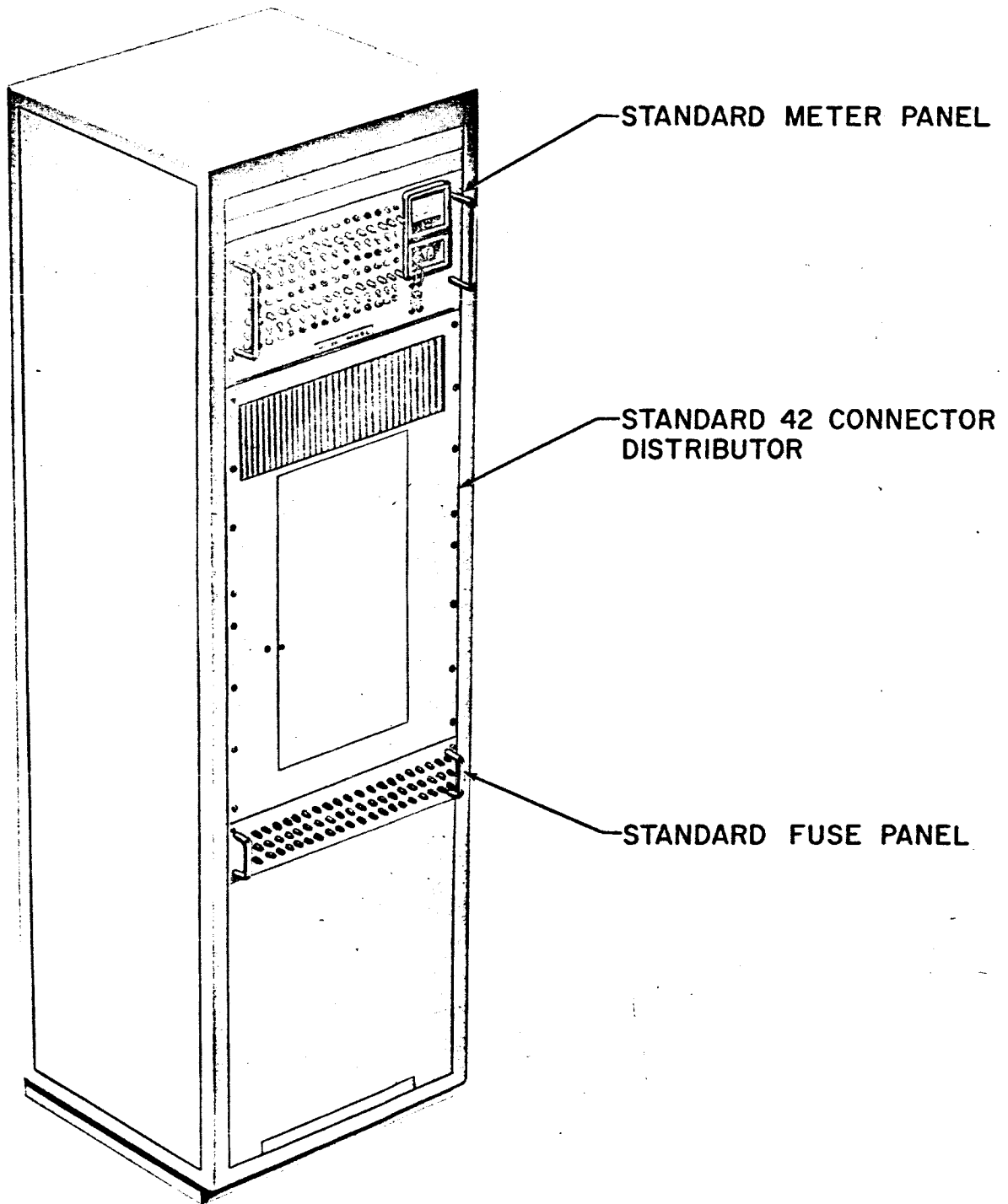


Figure 3-53 Relay Rack (Typical)

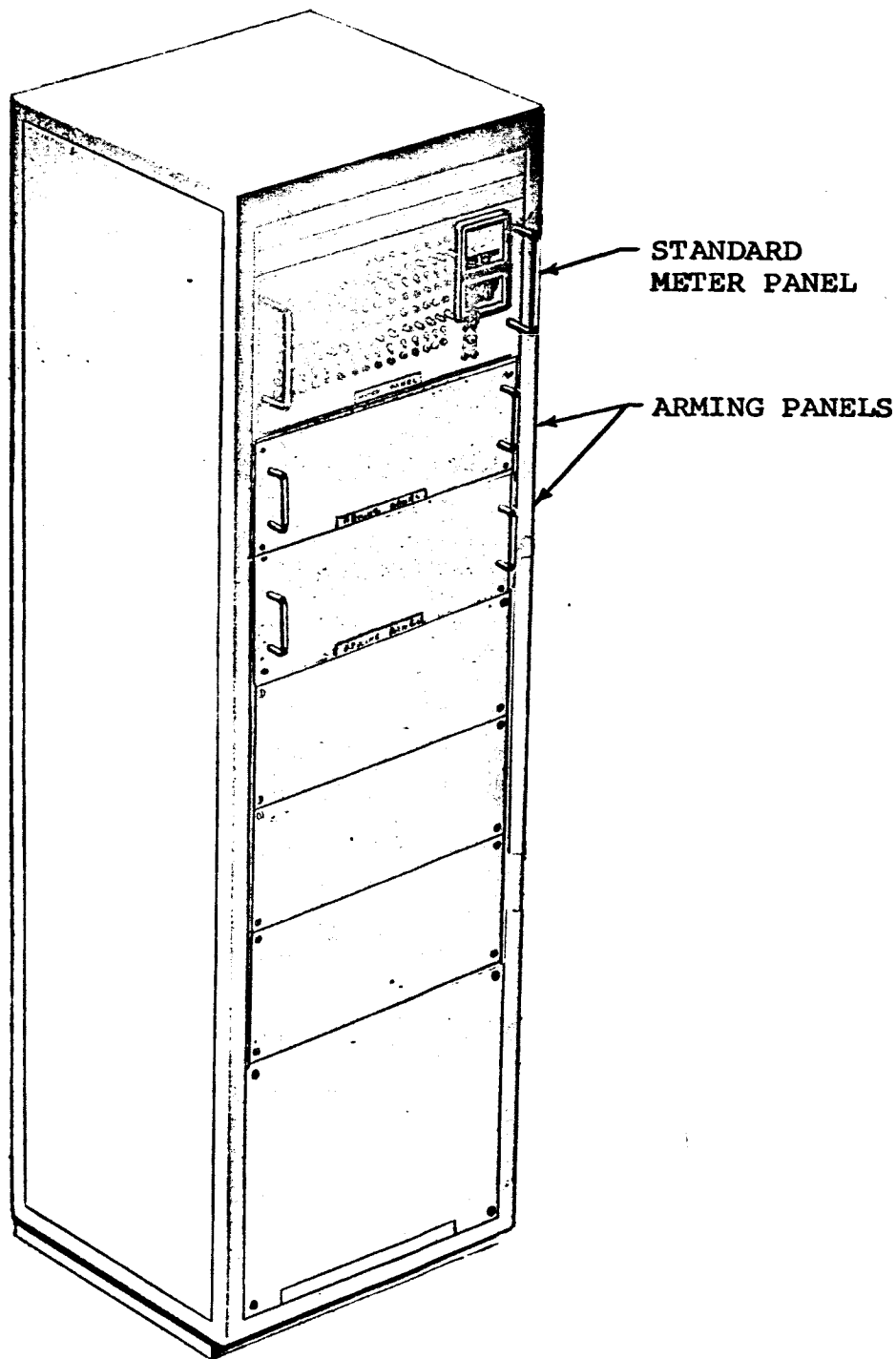


Figure 3-54 Arming Panel Rack (Typical)

TO BE PROVIDED

Figure 3-55 LCC Power Panel (Astrionics)

**3-47. S-IC ENGINE SERVICING PLATFORMS SUBSYSTEM.**

**3-48. Purpose of the Subsystem.** The S-IC Engine Servicing Platforms Subsystem controls the hoist assembly required to position any of the S-IC engine servicing platforms.

**3-49. Electrical Equipment for the Subsystem.** The S-IC Engine Servicing Platforms Subsystem uses the following equipment:

- a. Relay Distributor #6606, room 2-A
- b. Deck Distributor #6000, Level 0
- c. Platform Distributor #7201 and #5089, Servicing Platform
- d. Platform Controller #6000A1, Level 0
- e. Level Sensor #7200A3 and 5088A3 Servicing Platform
- f. Motor/Brake/Winch Assemblies #7853, #7852, #7851 and #7850, level 0
- g. Warning Buzzer #7200A4 and #5088A4
- h. Top Limit Switches #7200A8, A9, A12, A13 and #5088A8, A9, A12, A13
- i. Load Detector #7200A10, A11 and #5088A10, A11
- j. Interlock Distributor #7200A1, A2 and #5088A1, A2

**3-50. Description of the Subsystem.** Three separate platforms are required at the VAB and launch pad for servicing the Saturn S-IC stage:

- a. The Mobile Launcher Level Servicing Platform is a passive platform which provides servicing access to the S-IC stage at deck level 0.
- b. The S-IC Engine Servicing Platform (Pad) services the engine area while the Mobile Launcher is at the launch pad. This platform has level sensors, interlocks, and other control equipment not contained in the Mobile Launcher Level Servicing Platform, and offers greater load capacity.
- c. The S-IC Engine Servicing Platform (VAB) is used to service the engine area, install engine skirts or change an F-1 engine while the Mobile Launcher is in the VAB. It has the same control equipment as the S-IC Engine Servicing Platform (Pad), and still greater load capacity. See figure 3-57.



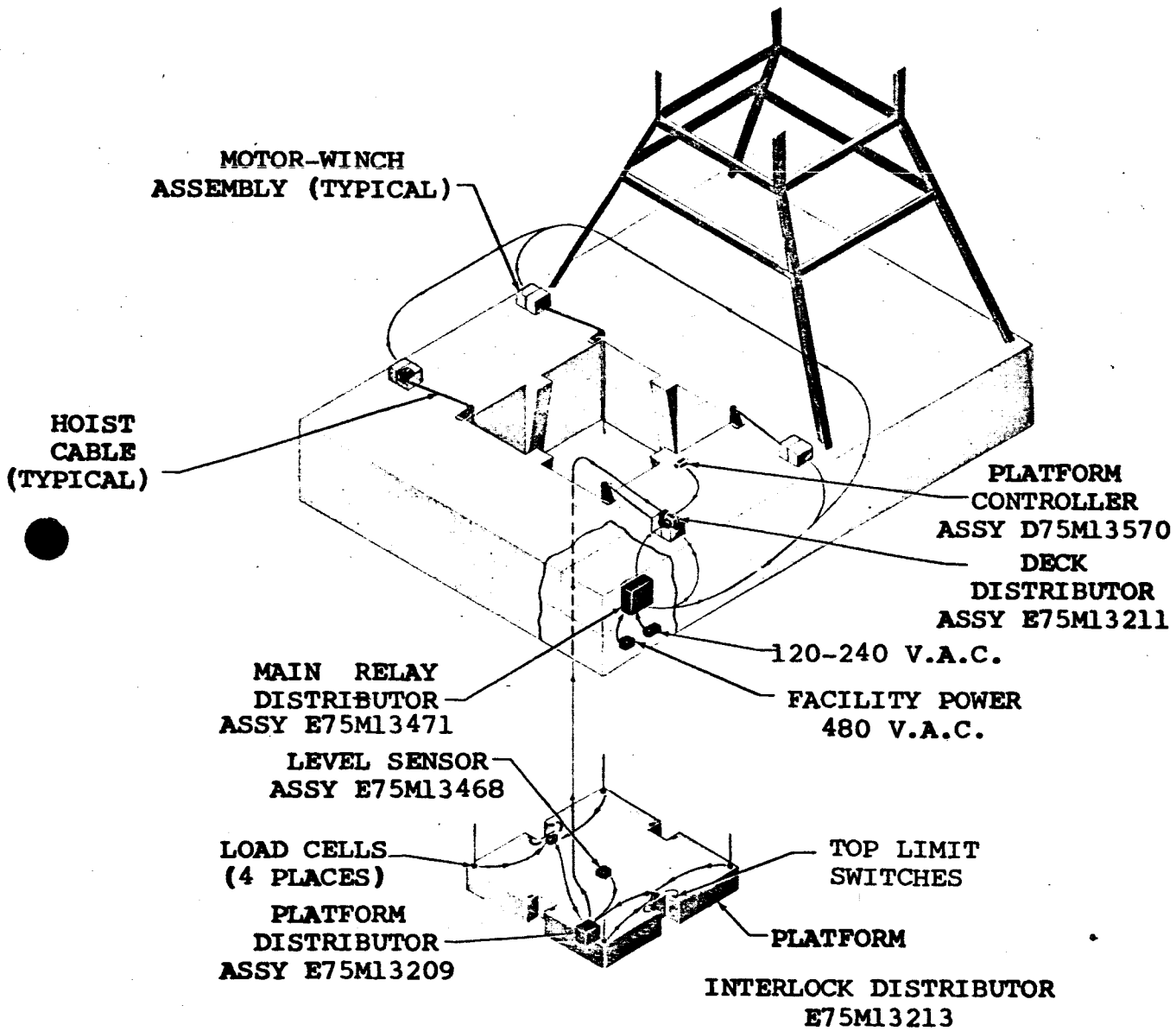


Figure 3-56 S-IC Engine Servicing Platforms Subsystem

These three platforms, represented by a single generalized platform in figure 3-56, are raised and lowered by a cable-hoist mechanism which is locally controlled by an operator on the Mobile Launcher deck. The electrical control system for this hoist is called the S-IC Engine Servicing Platforms Subsystem. See figure 3-58.

This subsystem normally employs four motor/winch assemblies operating simultaneously, giving it the capacity to lift a balanced platform loaded with one inboard F-1 engine. Step-control is achieved with two-speed motors.

Electrical power is also provided for a removable fifth motor/winch assembly. This assembly will provide additional hoist capacity for any platform corner raising an outboard F-1 engine.

The motor/winch assemblies are supported by several special equipments. These equipments and their functions are as follows:

- a. The Relay Distributor (room 2-A) provides control logic and power to the motors and control system. This distributor uses direct cabling from the distributor to the motors and control elements. It is shown in figure 3-59.
- b. The Deck Distributor (figure 3-60) located inside motor/winch housing "D" provides the means for connecting the platform controller to the system.
- c. The Platform Controller, which is connected to the system via the deck distributor, provides on-off, speed, and direction control of the motors. The motors may be controlled either individually or all together. See figures 3-61 and 3-62.
- d. The Platform Distributor routes interlock signals and distributes power to various platform locations.
- e. The Level Sensor (figure 3-60) detects tilt of the platform and sends this information to the platform controller via the platform and deck distributors. The information is monitored by the operator for manual corrections.
- f. A maximum-height interlock and cable-loading indicators are incorporated into the overall system. All exposed equipment is designed for protection against explosions and RFI.

Finally, it should be noted that the Platform Distributor, the Level Sensor, the maximum-height interlock, and the cable-loading indicators are not used on the Mobile Launcher Level Servicing Platform. This platform is positioned by an operator using the platform controller without signals from the platform itself.

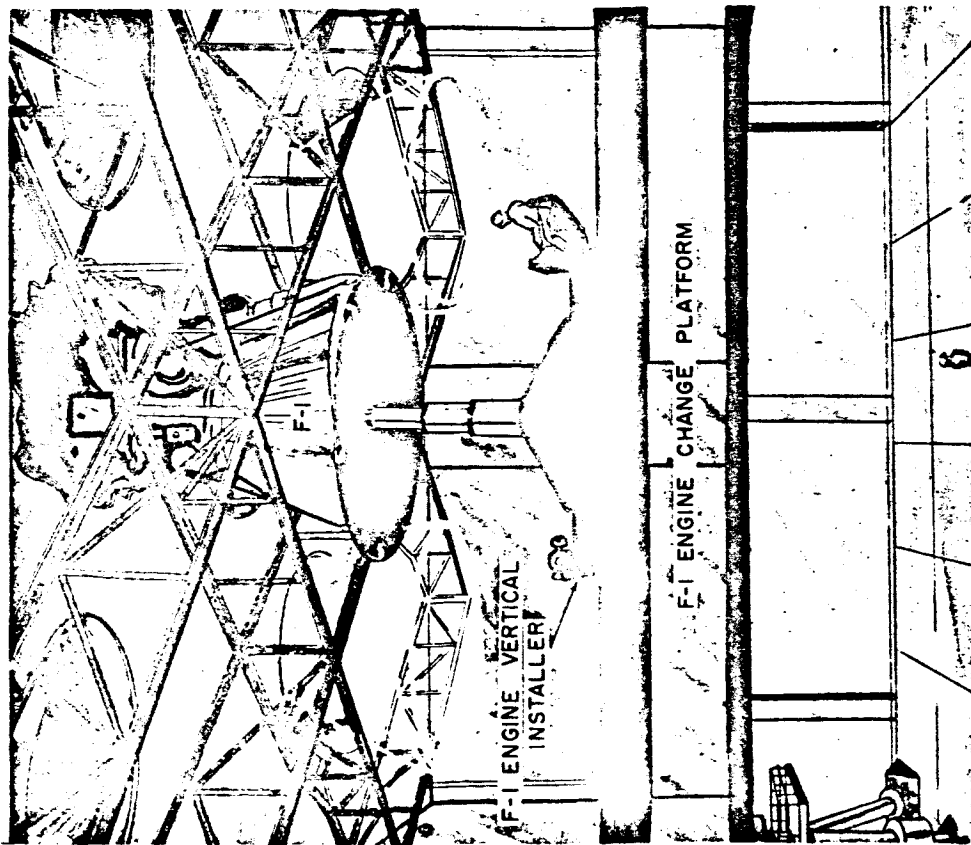


Figure 3-57 F-1 Engine Change

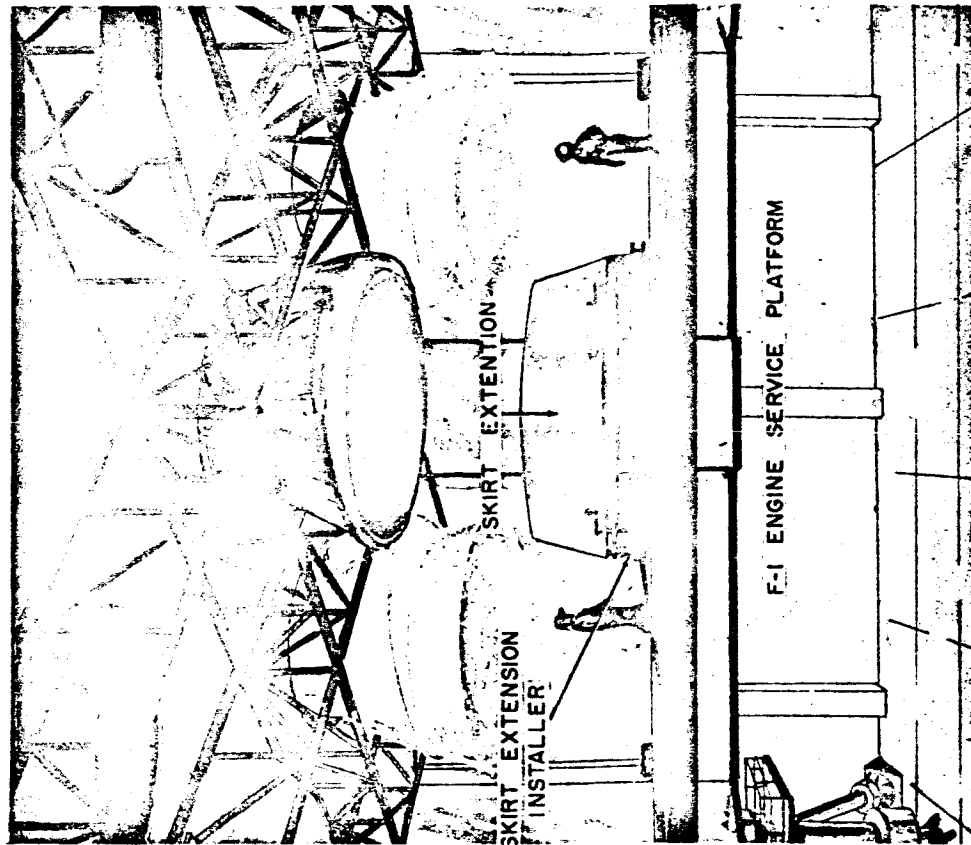


Figure 3-58 F-1 Engine Skirt Installation

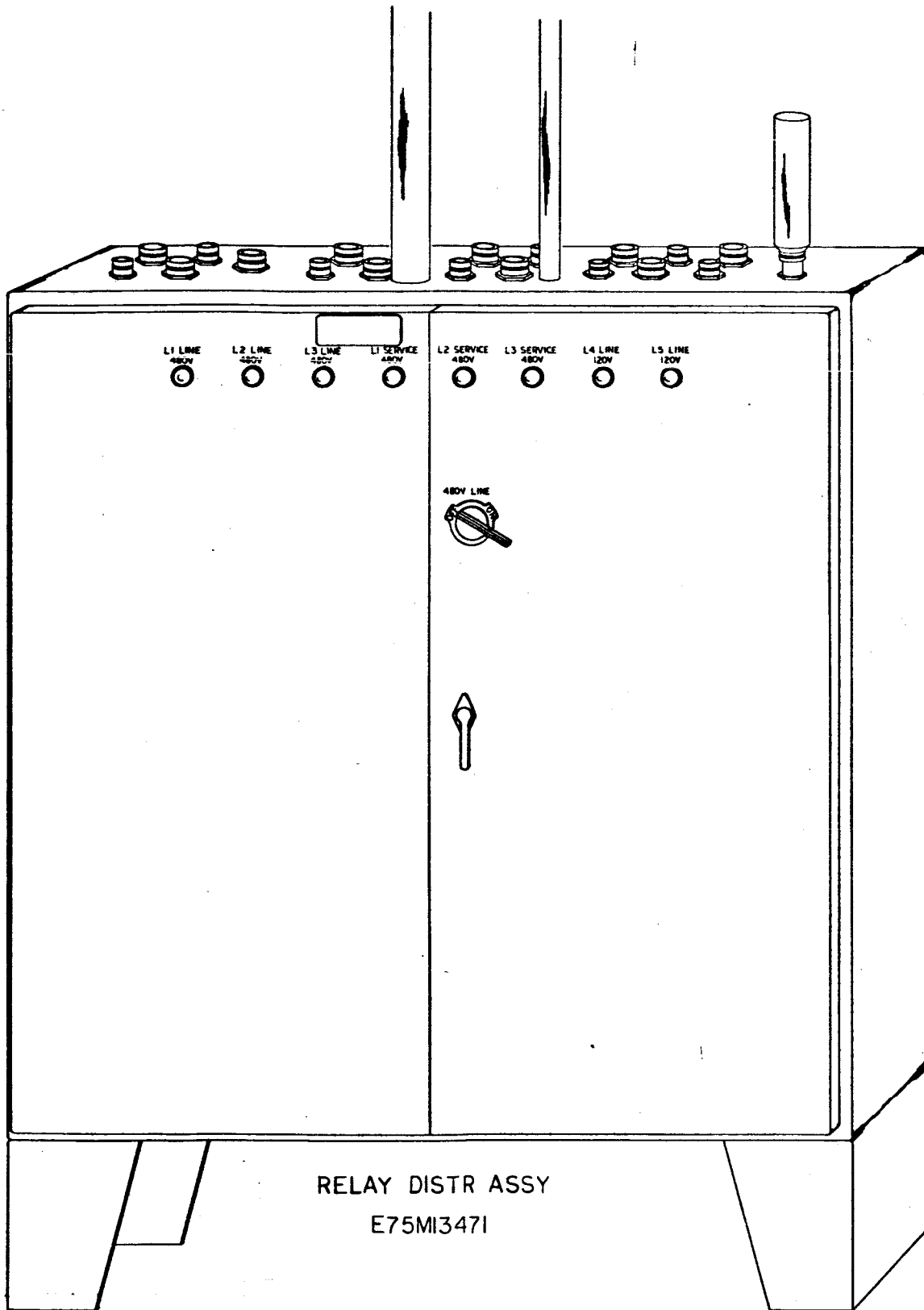


Figure 3-59 Relay Distributor, S-IC Engine Platforms Subsystem

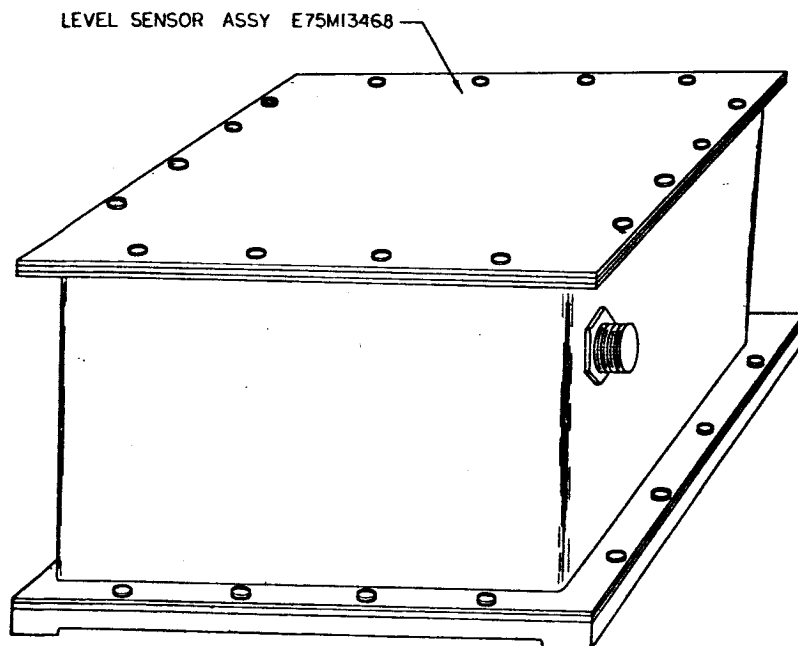
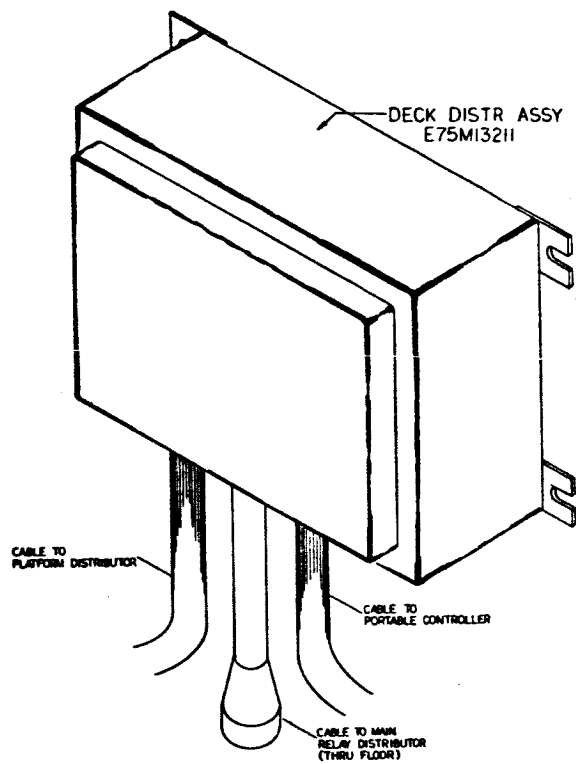


Figure 3-60 Electrical Equipment, S-IC Engine Platforms Subsystem

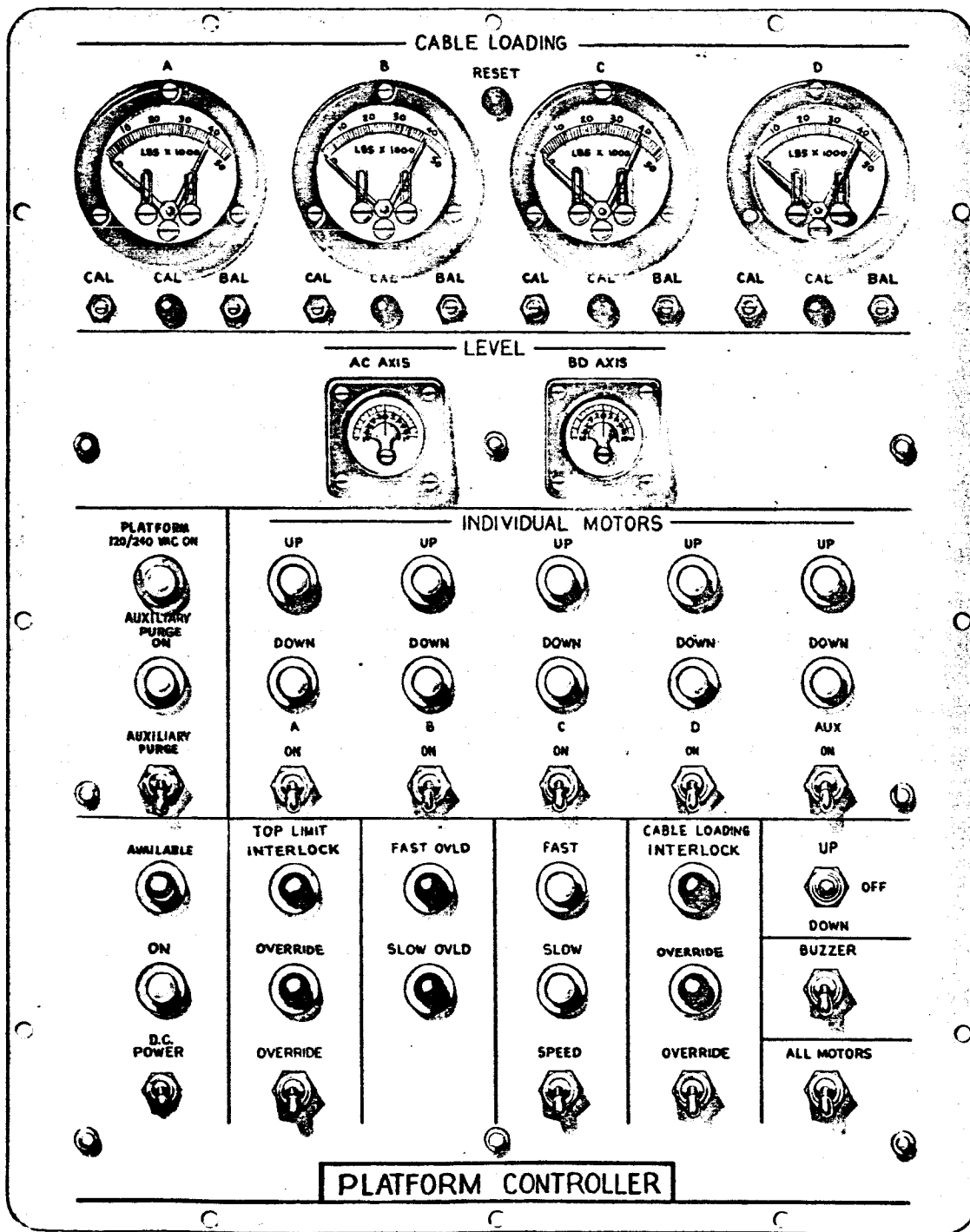


Figure 3-61 S-IC Engine Servicing Platform Controller E75M13570

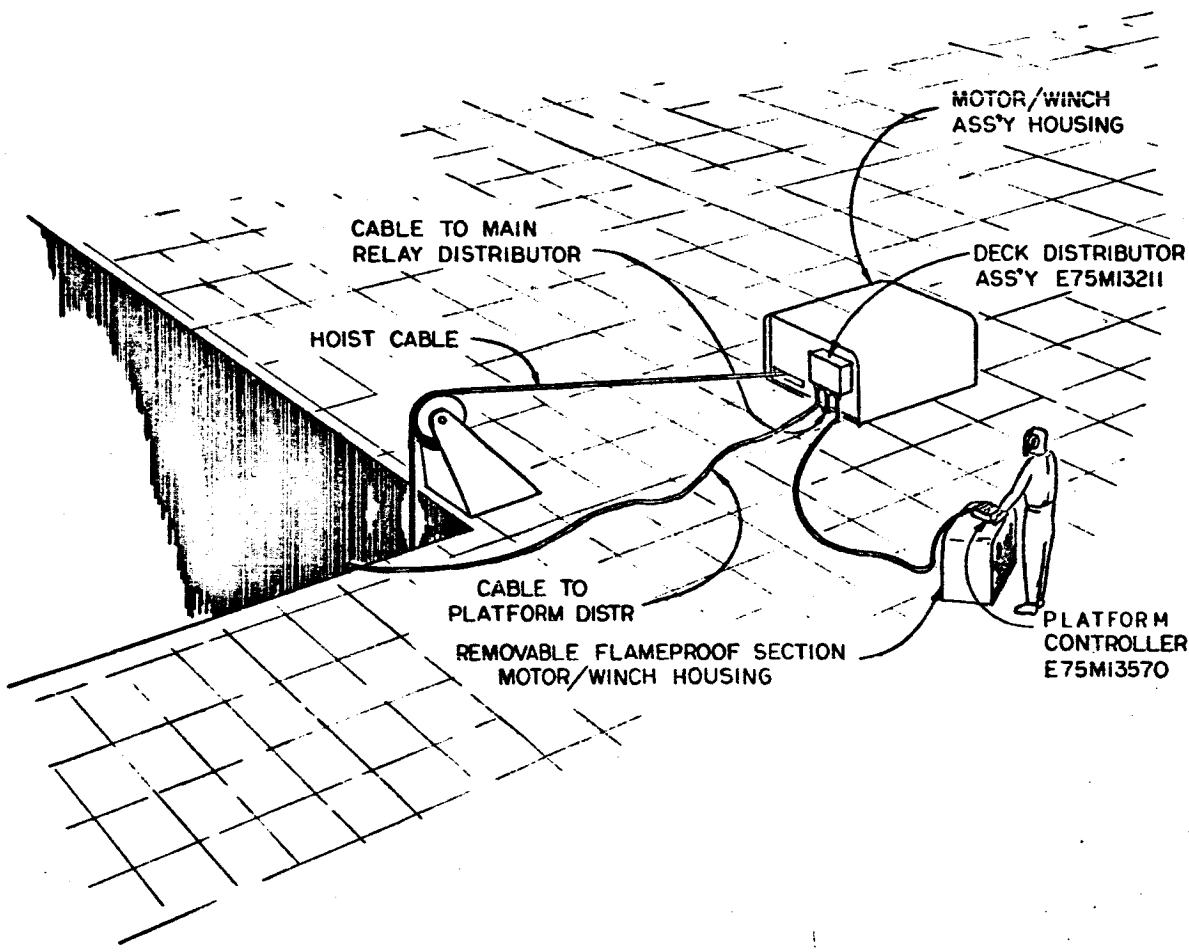


Figure 3-62 Typical Installation, Platform Controller

3-51. PLATFORM TRANSPORTER SUBSYSTEM.

3-52. Purpose of the Platform Transporter Subsystem. This equipment controls movement of the F-1 engine servicing platform transporter on the launch pad.

3-53. Electrical Equipment Employed. The following electrical equipment is included in the transporter subsystem:

- a. Electric Motors (4)
- b. Distribution Boxes (4)
- c. Portable Controllers (2)

3-54. Description of the Subsystem. The Platform Transporter is a rail-borne carriage which positions the F-1 Engine Servicing Platform under the Mobile Launcher. It rides just above the launch pad flame trench.

The transporter is pulled underneath the Mobile Launcher by a pair of winch assemblies (figure 3-63) at the north end of the platform transporter rails. These two winches normally operate simultaneously, with a balanced load. Load balance is achieved by use of a single cable which is strung continuously between the winches, through pulleys on the transporter. Both winches are controlled from a single portable controller.

A similar pair of winches at the south end of the rails withdraws the transporter from the Mobile Launcher.

A typical winch assembly is shown in figure 3-64. It is driven by a 3 phase, 480 volt, induction motor which is totally enclosed and fan-cooled.

The motor is controlled through a Nema type 4 distribution box equipped with a motor circuit switch, a starter contactor, a transformer for the contactor, and three overload elements.

The motor circuit switch is for quick make and break operation. It is manually operated by a handle on the front of the distributor.

The starter contactor is a Nema type 2 contactor for carrying 480 volt, 60 cycle power. Its magnetic coil is operated by 120 volts, 60 cycles and is readily accessible for quick change. Power for the coil is provided by a stepdown transformer mounted inside the contactor enclosure.



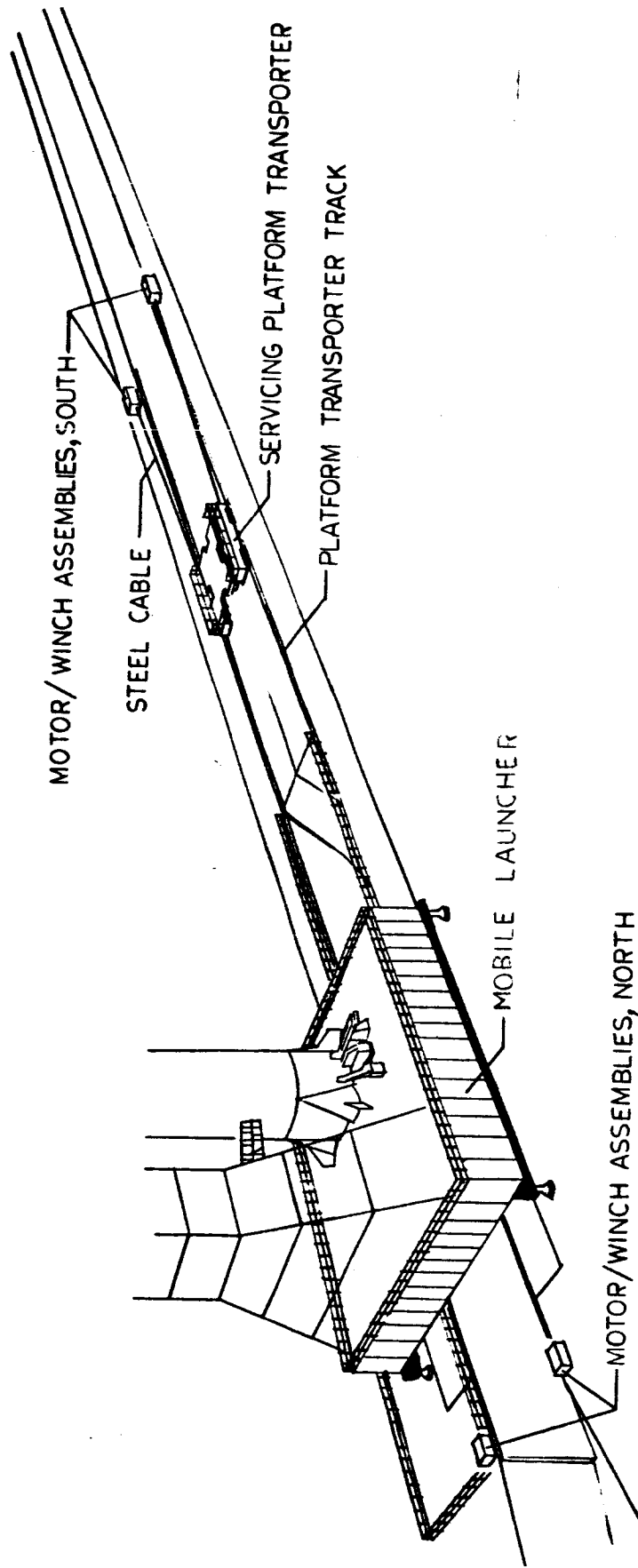


Figure 3-63 Servicing Platform Transporter Subsystem.

Three overload elements in the distributor provide protection for the 3-phase motor. An overload on any line to the motor opens the starter contactor, thereby stopping the motor.

Control for the distributor box is provided by a portable controller, figure 3-64, which is capable of operating either of two winches individually or both simultaneously. One controller is used in conjunction with the north winches and another with the south winches. Each controller is encased in heavy-duty water-tight stainless steel and weighs about ten pounds. It has magnetic reed-type push-button switch and two hand-operated selector switches equipped with watertight caps. Each controller operates under 120 volt, 60 cps power with a minimum contact rating of 2.0 amperes.

On launch pad A, the control cable interconnecting the two north-end motor winch assemblies is suspended over the flame trench by a steel rope. On future launch pads this control cable is carried in conduit through the concrete pad structure. All other control cables and all power lines are permanently installed in embedded conduit.

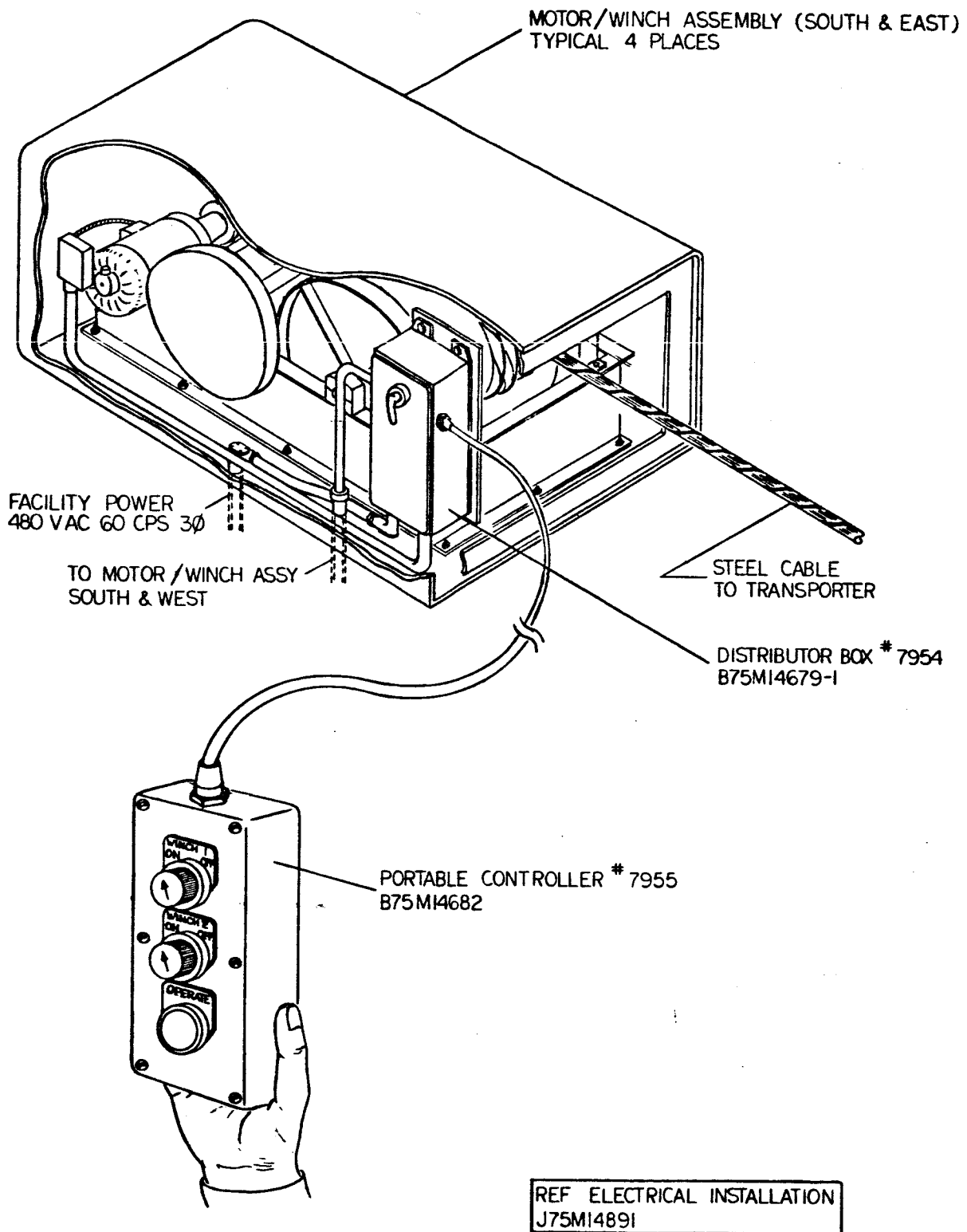


Figure 3-64 Motor/Winch Assembly, Platform Transporter Subsystem

3-55. S-II AND S-IVB ENGINE SERVICING PLATFORM SUBSYSTEM.

3-56. Purpose of the Subsystem. This subsystem provides electrical control for raising and lowering the S-II and S-IVB engine servicing platform.

3-57. Electrical Equipment for the Subsystem. The S-II and S-IVB Engine Servicing Platform Subsystem includes the following equipment:

- a. Motor/winch assemblies 7204A1, A2, A3 and A4.
- b. Electrical junction boxes 7205, 7206, 7207, and 7208.
- c. Distribution box 7204.
- d. Hand control unit 7204A5.

3-58. Description of the Subsystem. In the Vehicle Assembly Building, a special transporter (figure 3-65) is provided for Saturn V stages S-II and S-IVB. The vehicle stage is positioned on this transporter over a circular cutout. Personnel then gain access to the stage engines by means of a vertically movable platform, which is controlled by the electrical system herein called the S-II and S-IVB Engine Servicing Platform Subsystem.

The platform is raised and lowered by four winches, figure 3-66. Each winch is driven by one squirrel-cage induction motor which is totally enclosed and fan-cooled. This 3-horsepower motor is operated by 3 phase, 480 volt, 60 cps power. Full load current at 3500 rpm will not exceed 4.5 amperes. It is equipped with one 480 volt single-phase space heater that is capable of maintaining motor temperature at 5 to 8 degrees above ambient. Limit switches are provided to override motor operation when reel-in or reel-out limits are reached.

The electrical junction box attached to each motor/winch assembly has a Nema type 4 watertight enclosure. It is equipped with three overload elements (one for each phase) and a non-fused ac reversing starter. The starter is a commercial Nema size 0, open-type, with contacts for carrying 400 volts, 60 cycles. Its 120 volt, 60-cycle magnet coil is easily accessible for quick change.

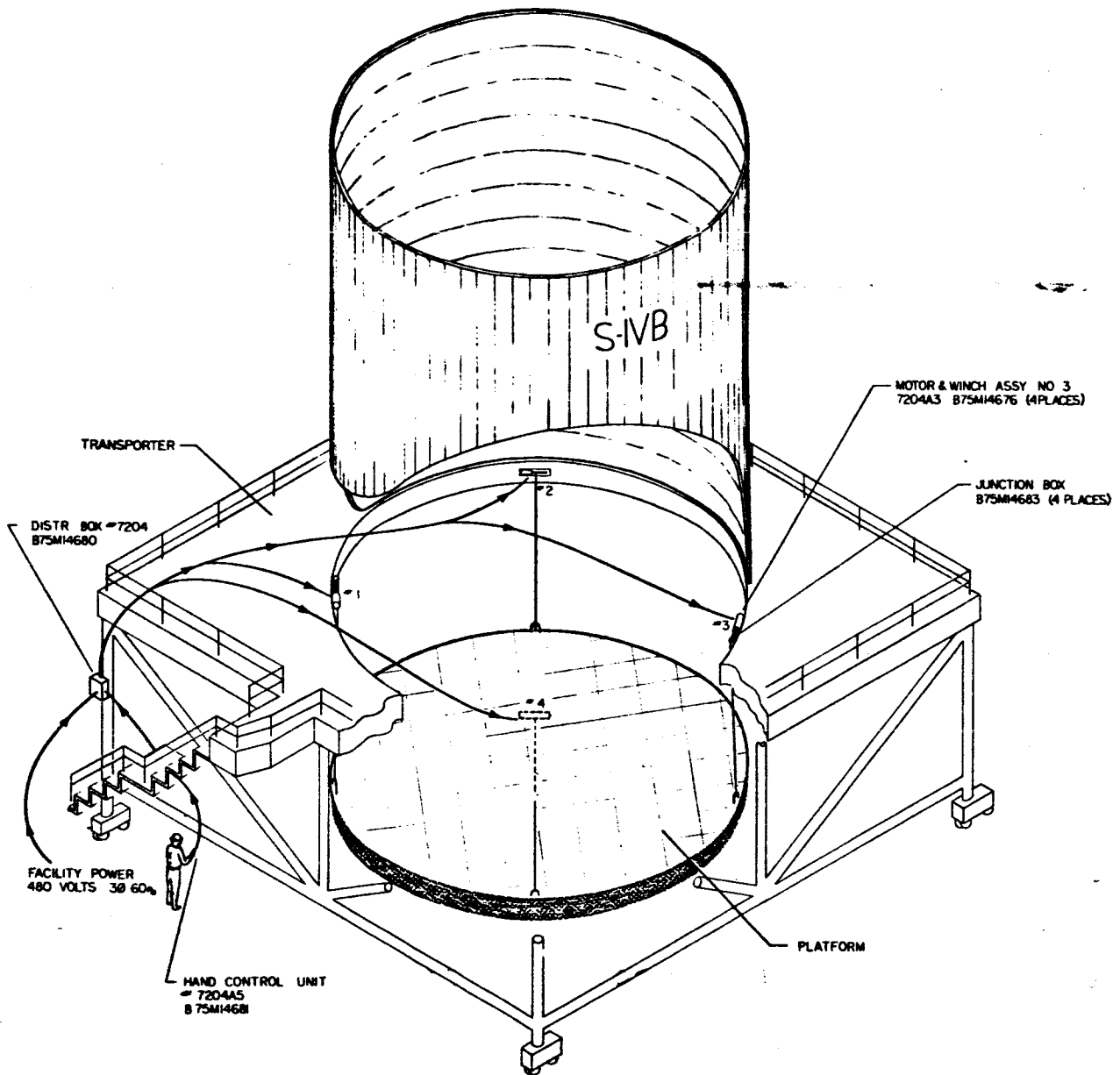


Figure 3-65 S-II and S-IVB Engine Servicing Platform Subsystem

Voltage for this coil is supplied by a step-down transformer in Distribution Box #7204. The 0.5 kva, single phase transformer receives 480 volt facility power and has two 115 volt outputs for the motor heaters and motor-starter coils. Distribution Box #7204, a Nema type 4 watertight enclosure, also has terminal blocks for distributing power to the transformer, motors, and hand control unit.

Operation of the four winches is actuated and controlled by means of hand control unit No. 7204A5. This unit has a separate select switch for each winch so that any combination of switches can be operated simultaneously. REEL-IN and REEL-OUT commands are actuated by pushbutton switches. All switches are magnetic reed type. The unit weighs approximately 10 pounds. See figure 3-66.

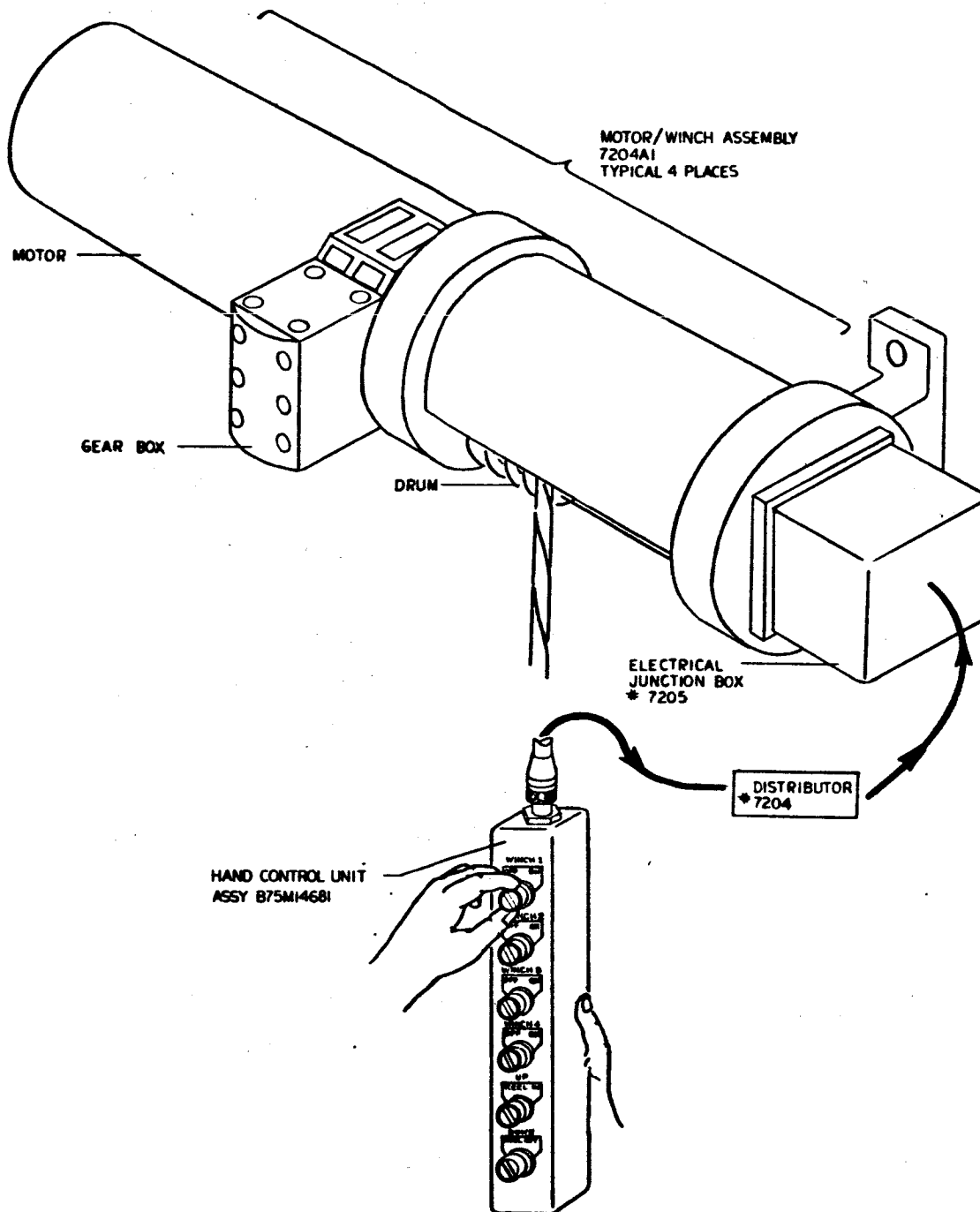


Figure 3-66 Motor/Winch Assembly B75M14676

**3-59. LAUNCH EQUIPMENT CONTROL AREA, LCC**

In the Launch Control Center sixteen consoles (figure 3-67) are assigned for control, test, and monitor of eight of the twelve ESE subsystems. These eight subsystems and their corresponding control panels are:

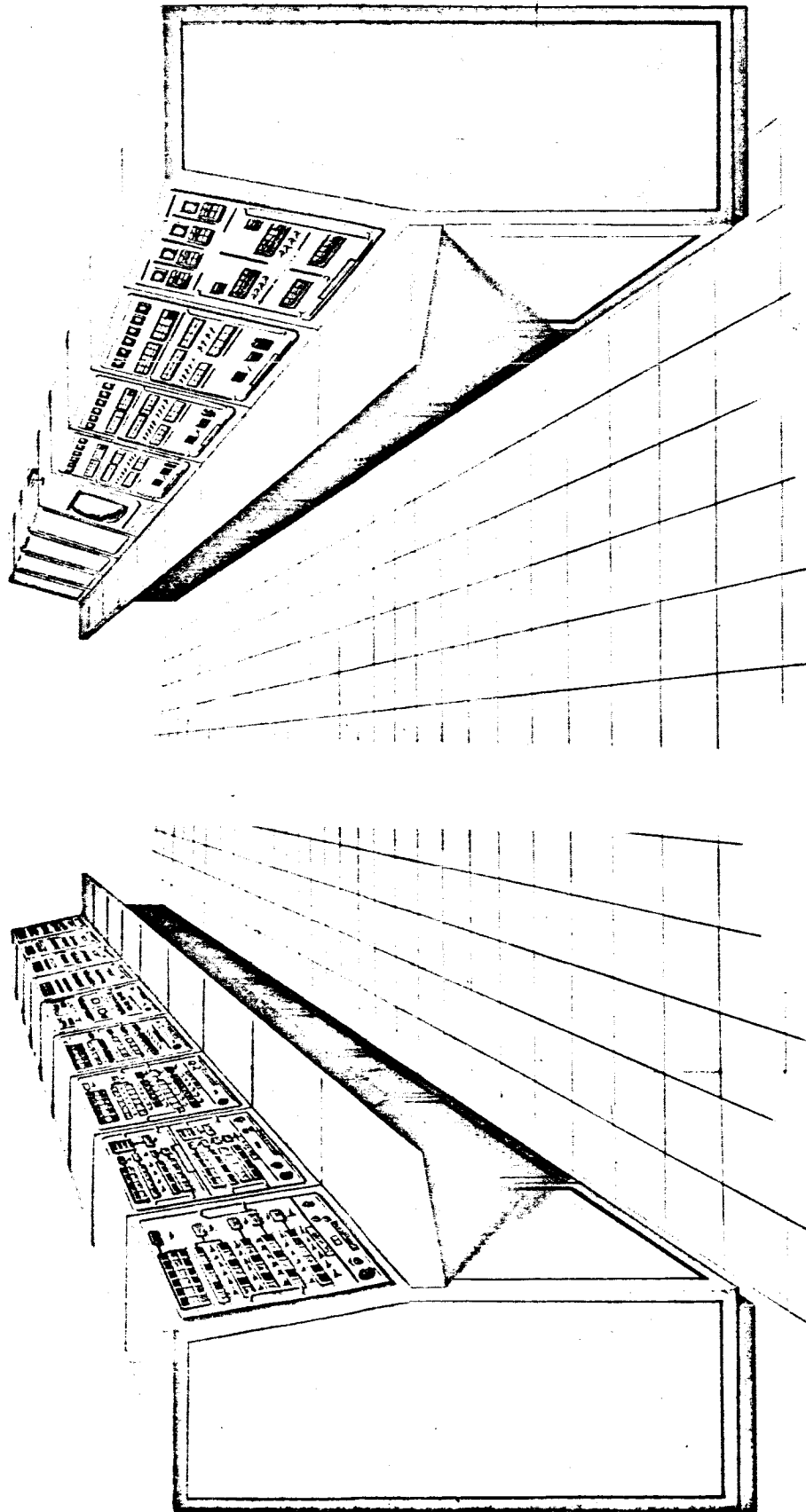
- |   |   |
|---|---|
| a. Service Arms Subsystem                       | S-IC Intertank thru Service Module (eight panels).            |
| b. Apollo Access Arm Subsystem                  | Command Module  |
| c. Pneumatic Control and Distribution Subsystem | Q-Ball and Pneumatic Distribution                             |
| d. Q-Ball Cover Removal Subsystem               | Q-Ball and Pneumatic Distribution                             |
| e. Hydraulic Charging Unit Subsystem            | Hydraulic Charging Unit                                       |
| f. Tail Service Masts Subsystem                 | TSM 1-2, TSM 3-2, TSM 3-4                                     |
| g. Launch Equipment Firing Circuits Subsystem   | Holddown Arms & Purge Valves and Service Arm Control Switches |
| h. ESE Power Subsystem                          | Launch Equipment Power Panel (Astrionics)                     |

The Status Panel provides monitoring for the eight Service Arms, the Command Module, Pneumatic Distribution, the Hydraulic Charging Unit, the three Tail Service Masts, the Holddown Arms and the Service Arm Control Switches.

Control, test, and monitor signals for these subsystems are transmitted between the LCC and Mobile Launcher via the DDAS, Mobile Launcher/LCC computers, and hardwire installations.

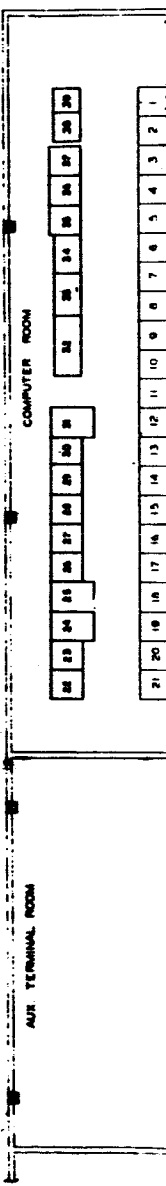
The basic enclosure for these LCC consoles is described in paragraph 4-102.





1. The computer room is located in the basement of the building. It is a large room with a tiled floor and a wall with horizontal lines. The room contains two long rows of computer cabinets. Each cabinet has a control panel with numerous buttons and switches, and a large monitor screen on top. The room is well-lit and appears to be a modern computer room for its time.

CORRIDOR



3-88-1



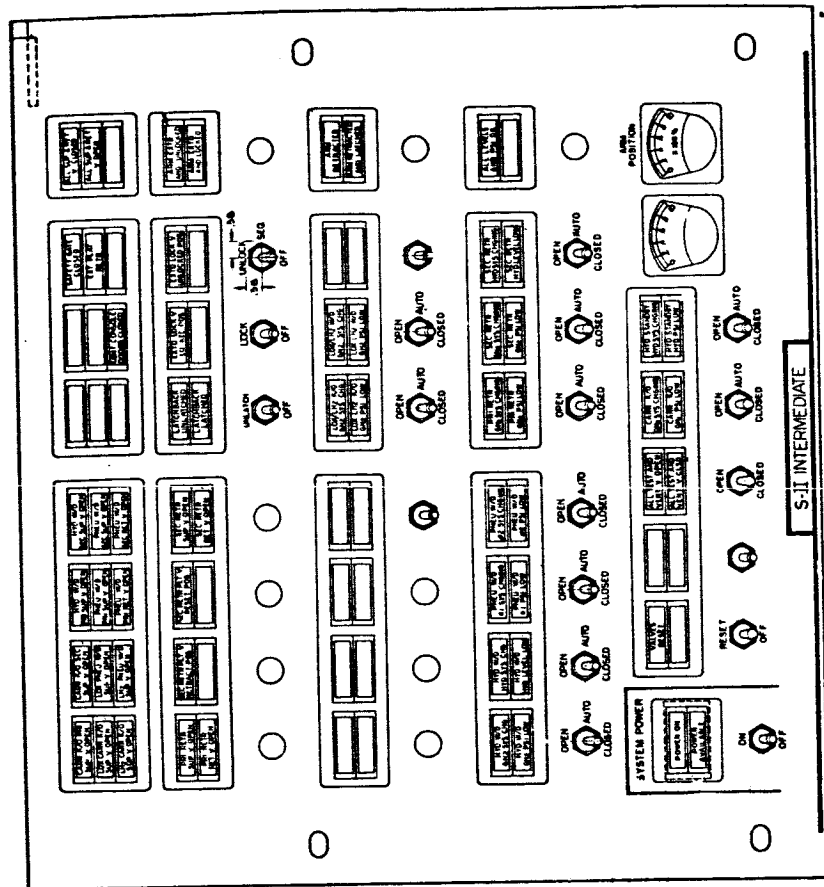
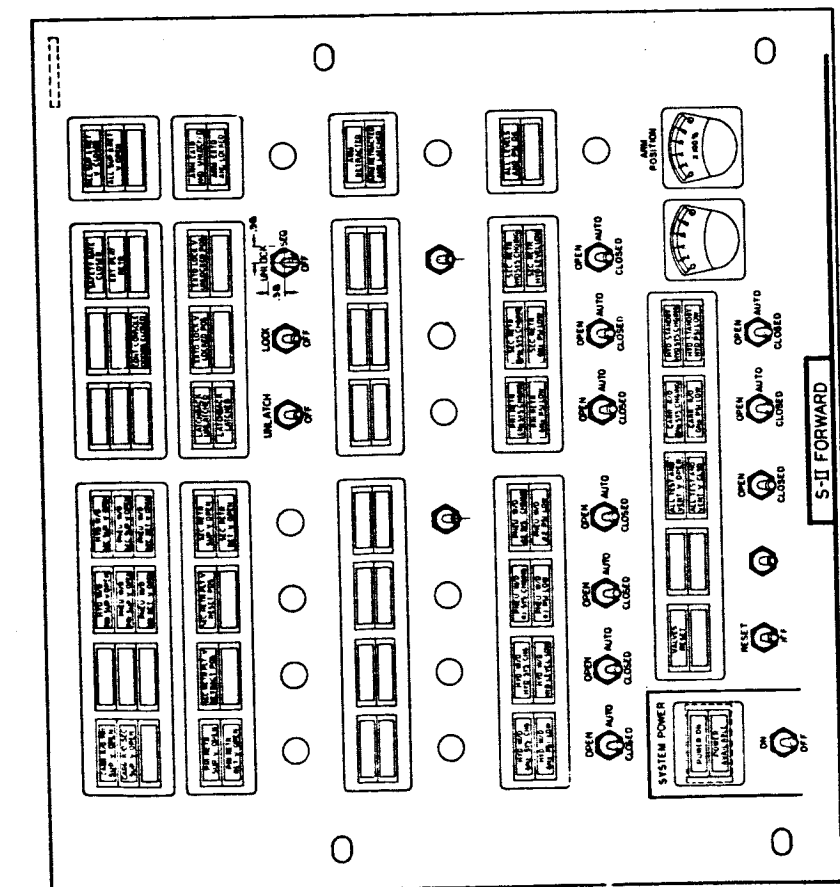
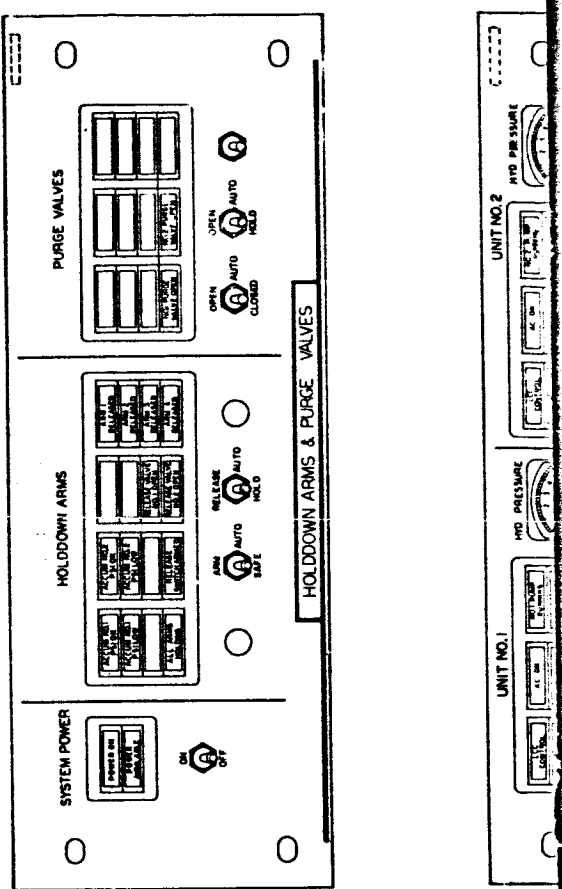
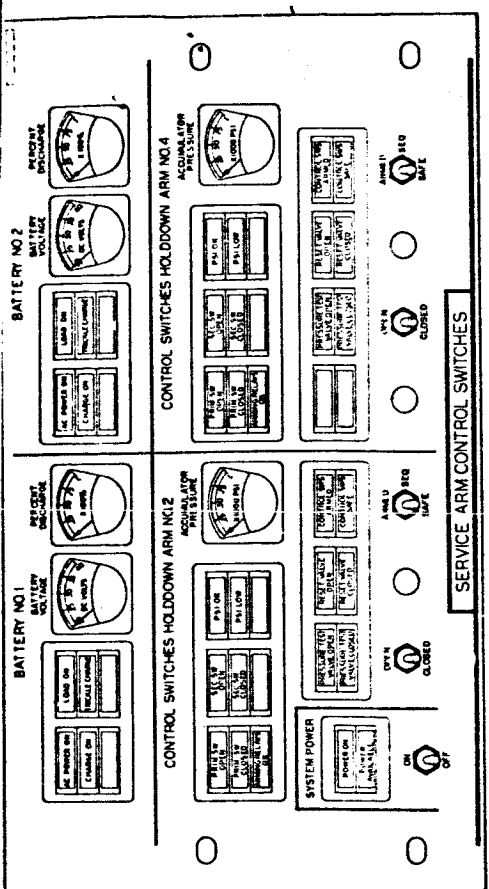
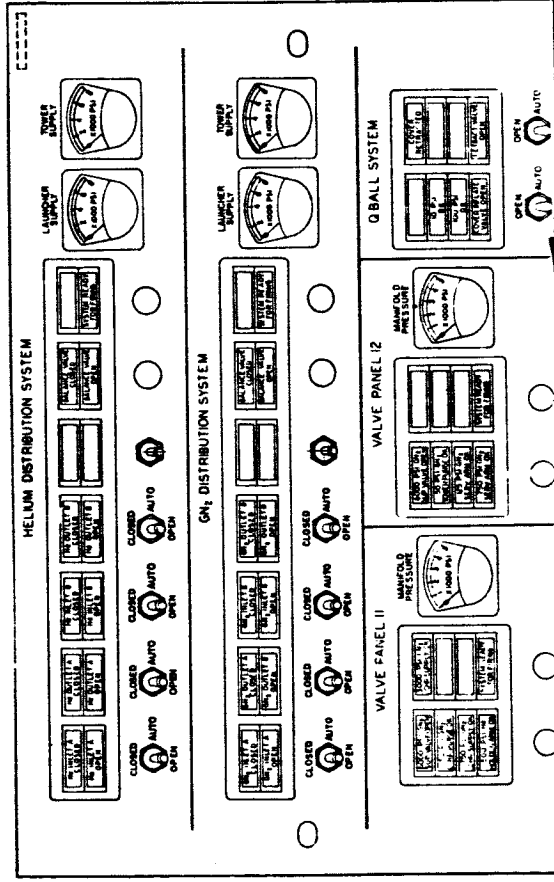
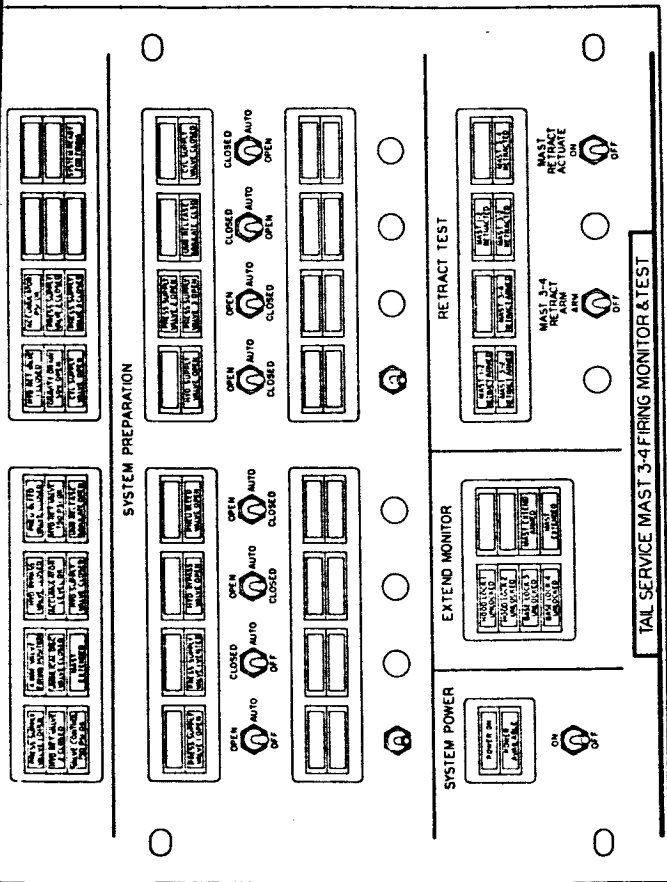


Figure 3-67

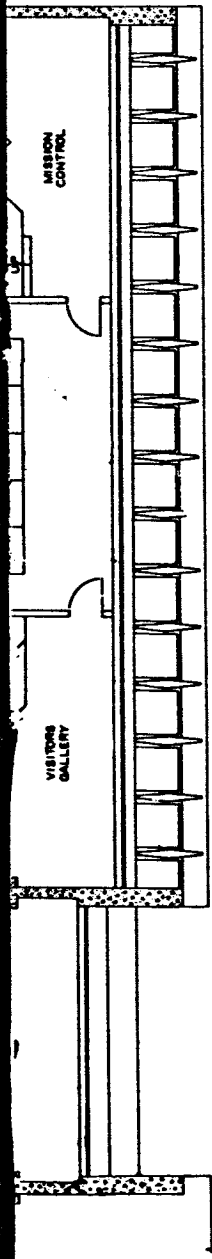
Launch Equipment Control Area, LCC



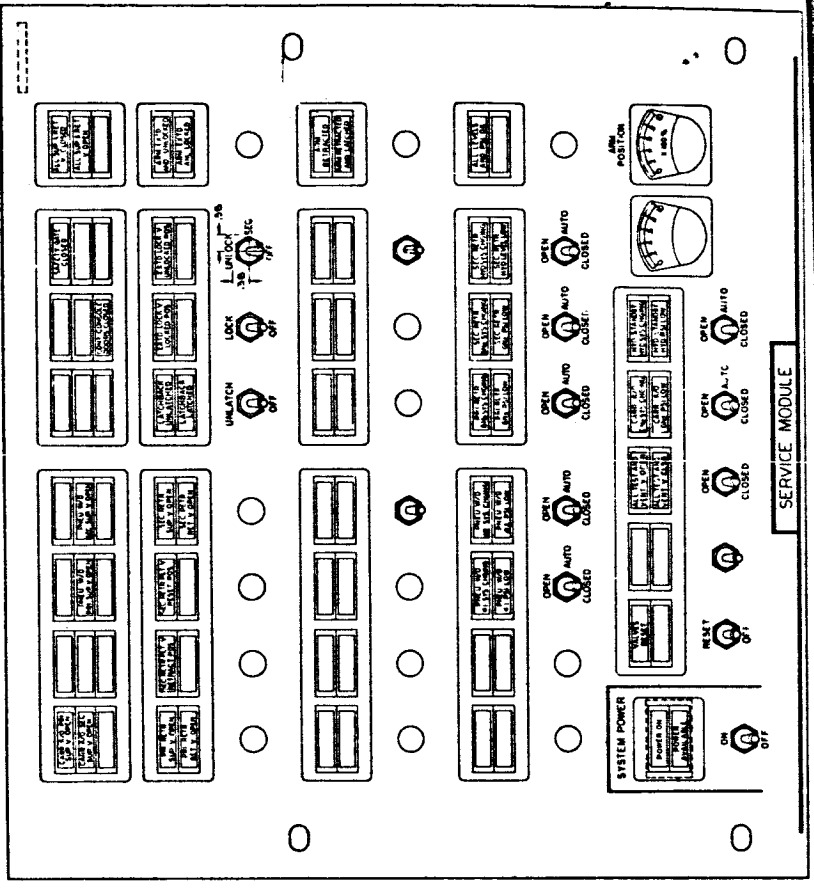
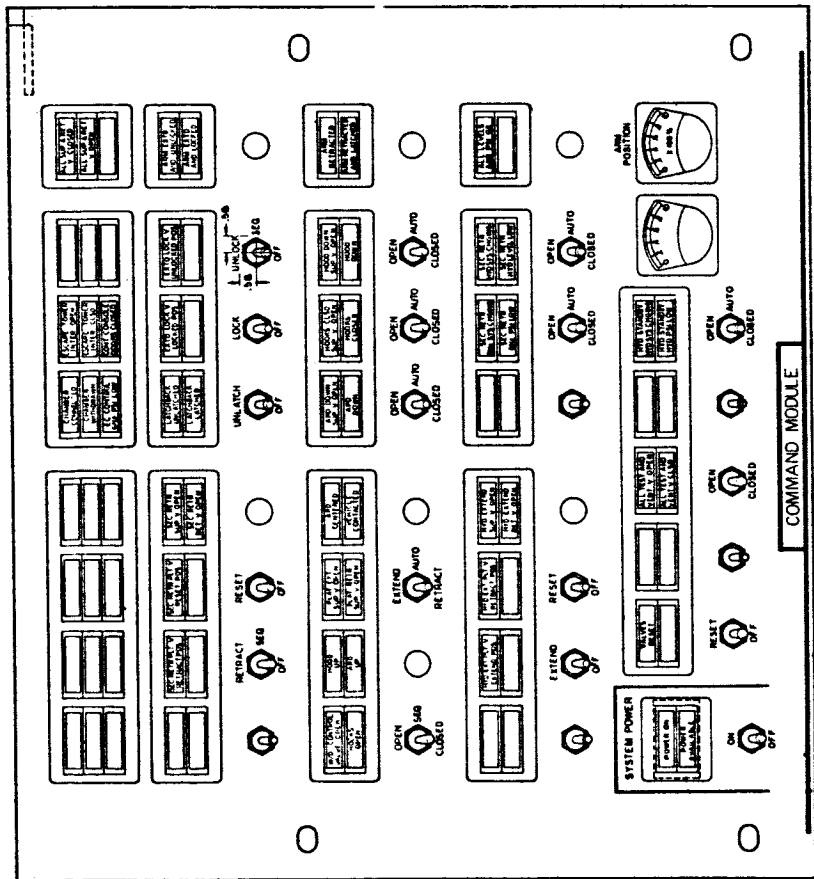




IF PUBLIC AFFAIRS OFFICER  
IS NECESSARY COMMANDER



FIRING ROOM NO. 1  
LAUNCH CONTROL CENTER LC-39



285-2







SECTION IV  
ESE STANDARD COMPONENTS

4-1. GENERAL.

The standard Launch Equipment Branch assemblies shown in figure 4-1 are individually delineated in paragraphs 4-2 through 4-116 following.

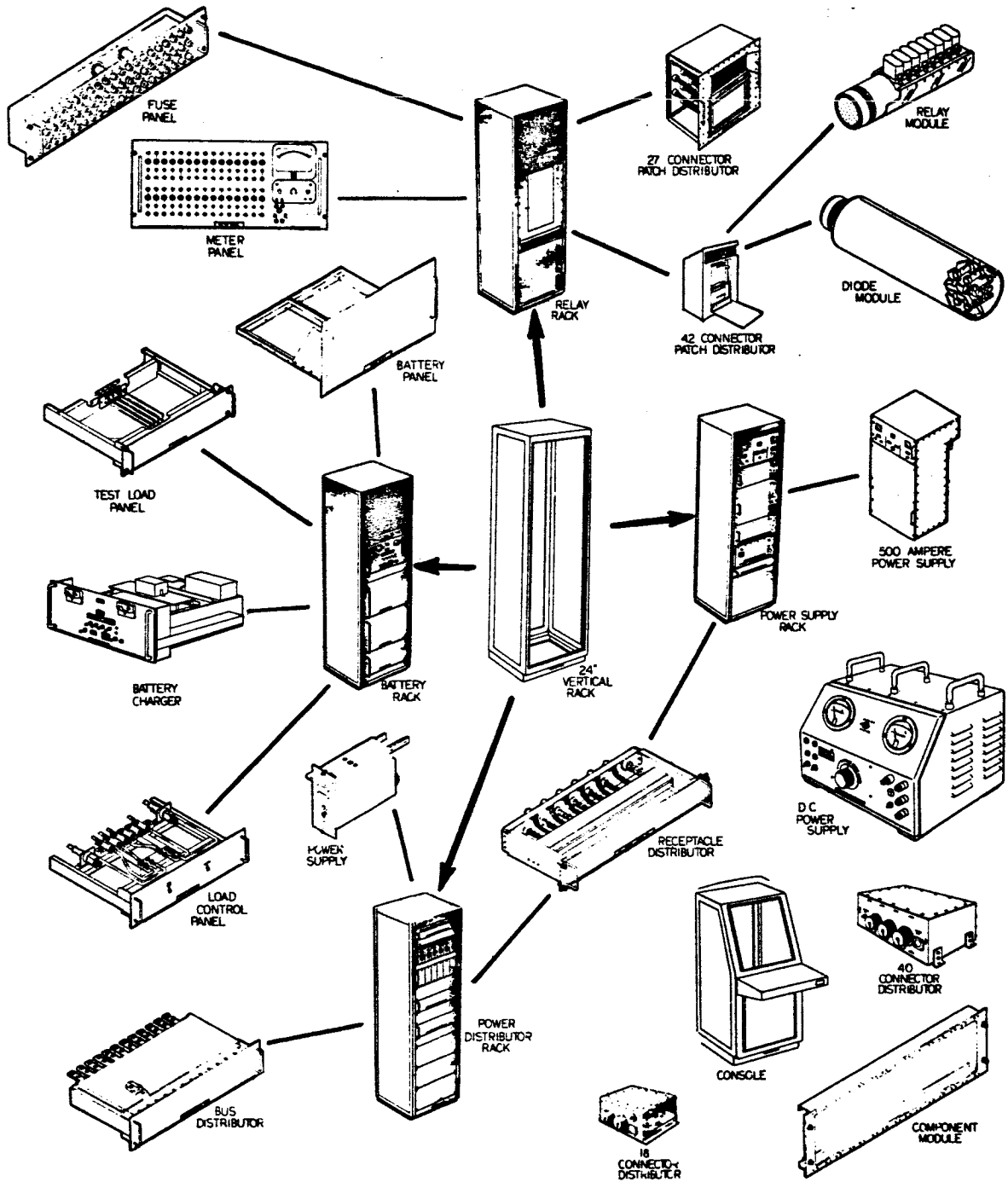


Figure 4-1 Standard Assemblies for ESE

4-2. STANDARD ELECTRICAL DISTRIBUTOR E75M07835.

4-3. Description of Distributor. Standard Electrical Distributor E75M07835 (figure 4-2) is a pressurizable enclosure 7.5 inches high, 18 inches wide, and 12 inches deep. It contains forty 7-pin connectors (Cannon BFR16S-1S or equal), one 5-pin connector (Bendix 71-323940-75P or equal) and two 61-pin connectors (Bendix 71-323940-73P or equal). All connectors are wired to terminal boards using standard solder terminals (Cambion X1782-D or equal), and patching between terminal boards is required to complete the interface between input and output connectors.

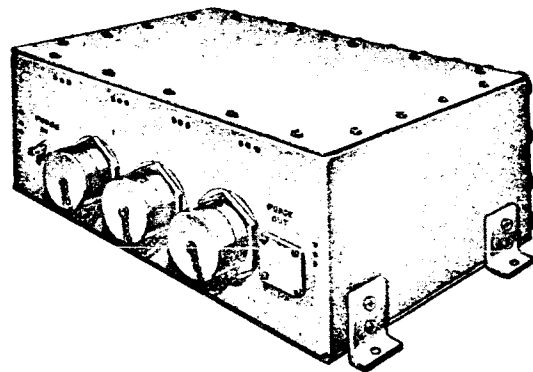
This basic distributor is normally mounted with angle brackets, but is also available for mounting in a standard rack panel.

4-4. Use of Distributor. Distributor J75M07835 is used for electrical control of the Mobile Launcher service arms, tail service masts, high pressure pneumatics system, and holddown arms. At each location it gathers electrical lines from individual components and consolidates them into two cables for further transmission. The distributor also contains positive and negative power buses for supplying power to relays, solenoid-operated valves, and other dc components within the service arms and masts.

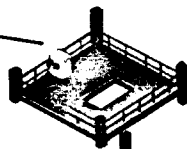
4-5. Locations of Distributor. Distributor J75M07835 appears in approximately 28 places in the Launch Equipment Branch ESE. The distributors, although not precisely located in the illustration, are:

- |  |   |
|--|---|
| a. Control Distributors 1, 2, 3                    | Service Arms Subsystem<br>See figures 3-2 and 3-3                   |
| b. Tail Service Mast Distributor                   | Tail Service Masts Subsystem,<br>See figure 3-31                    |
| c. Holddown Arm Control Distributor                | Launch Equipment Firing<br>Circuits, Room 12-A<br>See figure 3-22   |
| d. High Pressure Pneumatics<br>Control Distributor | Pneumatics Control and<br>Distribution Subsystem<br>See figure 3-37 |

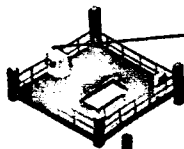
4-6. Design Organization. Distributor J75M07835 was designed by the Electrical Section, Launch Equipment Branch, KSC.



CONTROL DISTR.  
1, 2  
LEVEL 320'



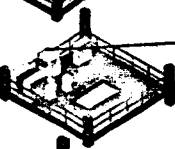
CONTROL DISTR.  
1, 3  
LEVEL 140'



CONTROL DISTR.  
1, 2, 3  
LEVEL 300'



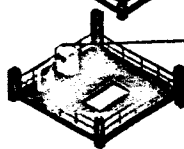
CONTROL DISTR.  
1, 3  
LEVEL 120'



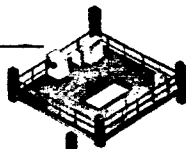
CONTROL DISTR.  
1, 2, 3  
LEVEL 260'



CONTROL DISTR.  
1, 3  
LEVEL 60'

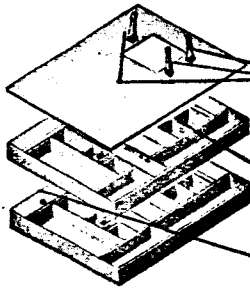
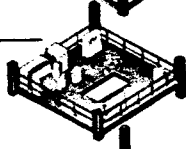


CONTROL DISTR.  
1, 2, 3  
LEVEL 220'



TSM CONTROL DISTR.

CONTROL DISTR.  
1, 2, 3  
LEVEL 200'



HOLDDOWN ARM  
CONTROL DISTR.  
HIGH PRESSURE PNEUMATICS  
CONTROL DISTR. 6407

CONTROL DISTR.  
1, 2, 3  
LEVEL 160'



Figure 4-2 Standard Electrical Distributor J75M07835

4-7. STANDARD ELECTRICAL DISTRIBUTOR E75M09808.

4-8. Description of Distributor. Standard Electrical Distributor E75M09808 (figure 4-3) is a pressurizable stainless steel unit 7.5 inches high, 11.5 inches wide, and 12.75 inches deep. On the back are eighteen 7-pin connectors (Cannon BFR16S-1S or equal). On the front are a 5-pin connector (Bendix 71-323940-75P or equal) and a 61-pin connector (Bendix 71-323940-73P or equal). All connectors are wired to terminal boards using standard solder terminals (Cambion X1782-P or equal), and patching between terminal boards is required to complete the interface between input connectors.

This distributor is outfitted with anchor nuts and angle brackets for cabinet or panel mounting.

4-9. Use of Distributor. Distributor E75M09808 is used in the Water Glycol Control System, Service Arm Control Switch Test System, Q-Ball and Deluge Purge, Valve Panel No. 11 and Valve Panel No. 12. At each location it gathers electrical lines from individual components and consolidates them into one cable for further transmission. The distributor also contains positive and negative power buses for supplying power to relays, solenoid-operated valves, and other dc components within the systems it serves.

4-10. Locations of Distributor. Distributor E75M09808 appears in approximately six places in the Launch Equipment Branch ESE. See figure 4-3.

- |   |  |
|---|--|
| a. Valve Panel 11 Distributor                     | Pneumatics Control and Distribution.<br>See figure 3-38  |
| b. Valve Panel 12 Distributor                     | Pneumatics Control and Distribution.<br>See figure 3-40. |
| c. Service Arm Control Switch<br>Test Distributor | Firing Circuits.<br>See figure 3-26.                     |
| d. Water Glycol Control Distributor               |  |
| e. Q-Ball Purge Control Distributor               | Q-Ball Cover Removal Subsystem.<br>See figure 3-42.      |
| f. Deluxe Purge Control Distributor               | Pneumatics Control and Distribution.<br>See figure 3-35. |

4-11. Design Organization. Distributor E75M09808 was designed by the Electrical Section, Launch Equipment Branch, KSC

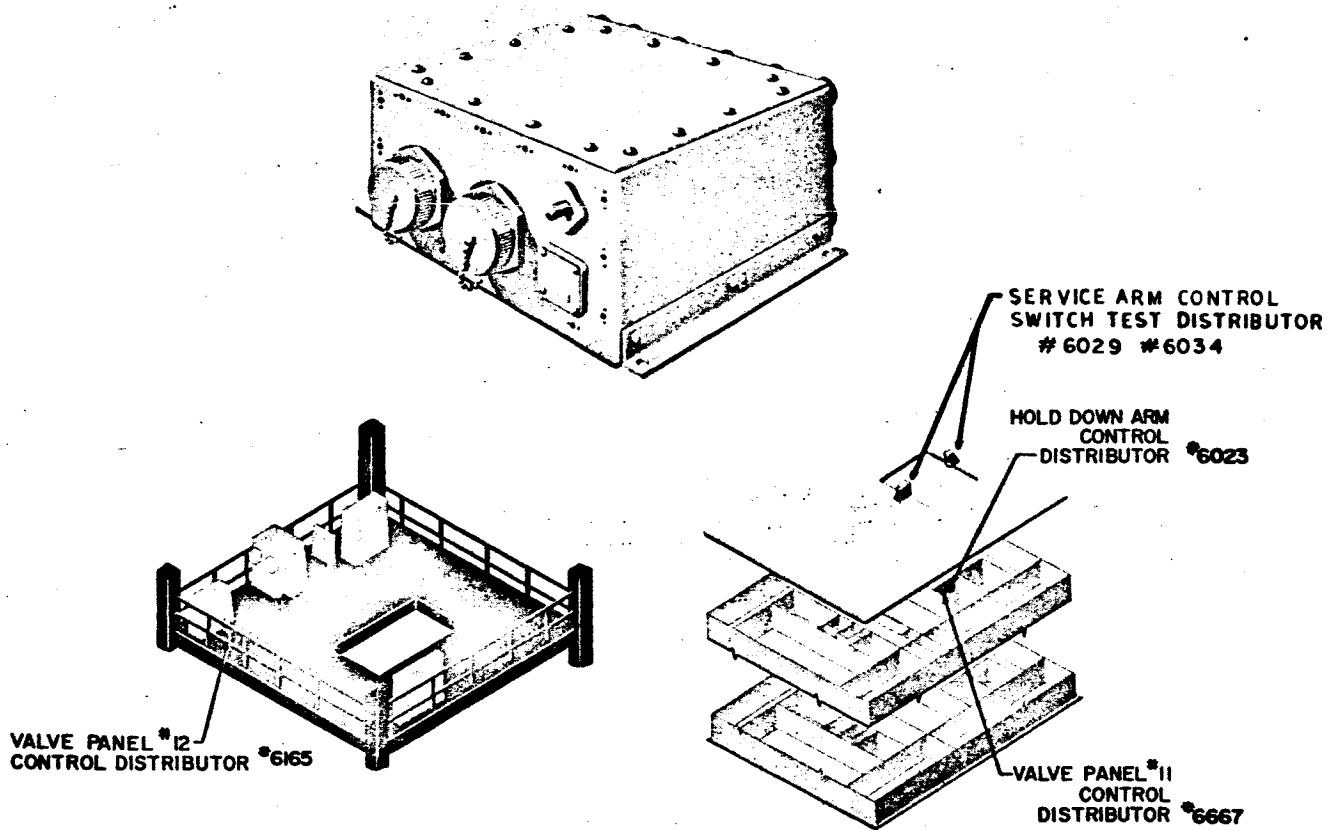


Figure 4-3 Standard Electrical Distributor J75M09808

4-12. PORTABLE DC POWER SUPPLY D75M51116.

4-13. Description of Supply. Portable Power Supply D75M51116 (figure 4-4) is packaged in a hand-carry case 13 inches high, 14 inches wide, and 13 inches deep. It provides unregulated dc in the ranges 0-18 volts or 18-36 volts, controlled by a range select switch and rheostat. Maximum output load is 15 amperes. Both voltage and load are displayed on front-panel meters.

Output power is supplied from front and rear binding posts, or from a rear-mounted cable connector (MS3102E-22-2S). Input power, 120 volts, 60 cps, is supplied through a rear-mounted standard 3-pin ac panel connector.

A positive and negative ground-detector circuit is actuated by a switch on the front panel.

4-14. Use of Supply. Power Supply D75M51116 can be easily hand carried about the Mobile Launcher to provide dc power where requirements do not exceed 36 volts, 15 amperes and 1 volt ripple/noise. It is used to provide power to the Tail Service Mast Test Set, Launcher Accessories Test Set No. 1, Launcher Accessories Test Set No. 2, and Tower Test Set.

4-15. Locations of Supply. See figures 3-22, 3-26, 3-34, 3-37, 3-38, 3-40.

4-16. Design Organization. Power Supply D75M51116 was designed by the Electrical Section, Launch Equipment Branch, KSC.

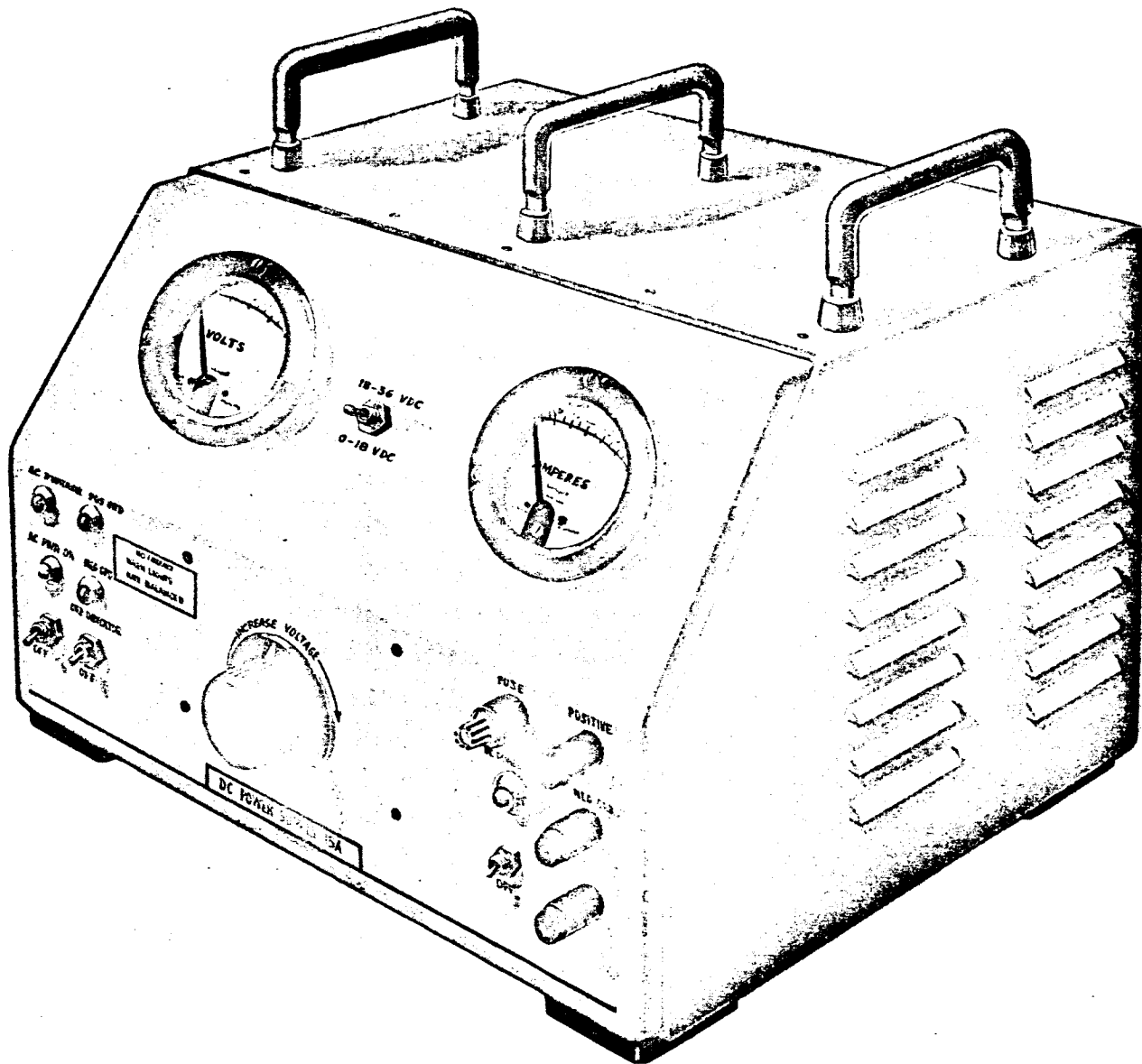


Figure 4-4 Portable DC Power Supply D75M51116



4-17. STANDARD 24 INCH VERTICAL RACK E64-KN-F-260-10.

4-18. Description of the Vertical Rack. Standard 24 Inch Vertical Rack E64-KN-F-260-10 is an R. F. I. shielded enclosure 86.25 inches high, 26.06 inches wide and 30.56 inches deep. It is designed to house standard 24 inch component panels. See figure 4-5.

Panels may be mounted either front or rear, and may be individually shielded. The rack base, side panels, and cable entry (at the enclosure bottom) are removable.

4-19. Use of the Vertical Rack. Standard 24 inch racks for the ESE house seven types of assemblies:

- a. Nine service arm relay racks, figure 3-53.
  - b. Two TSM & launcher accessories relay racks containing an internal fuse panel in lieu of Fuse Panel J75M10163.
  - c. Four standby battery racks and firing batteries, figure 3-52.
  - d. Three power distribution racks, figure 3-51.
  - e. Two power supply racks (one spare), figure 3-50.
  - f. One arming panel rack, figure 3-54.
  - g. Four integration racks and two cable entrance racks.
- Spare racks will be assigned as required.

4-20. Locations of Vertical Rack. The ESE employs approximately twenty-four racks in Mobile Launcher room 8-A and three in the LCC. See figure 6-2.

4-21. Design Organization. The Standard 24 Inch Vertical Rack is commercial equipment purchased to KSC/MSFC specifications.

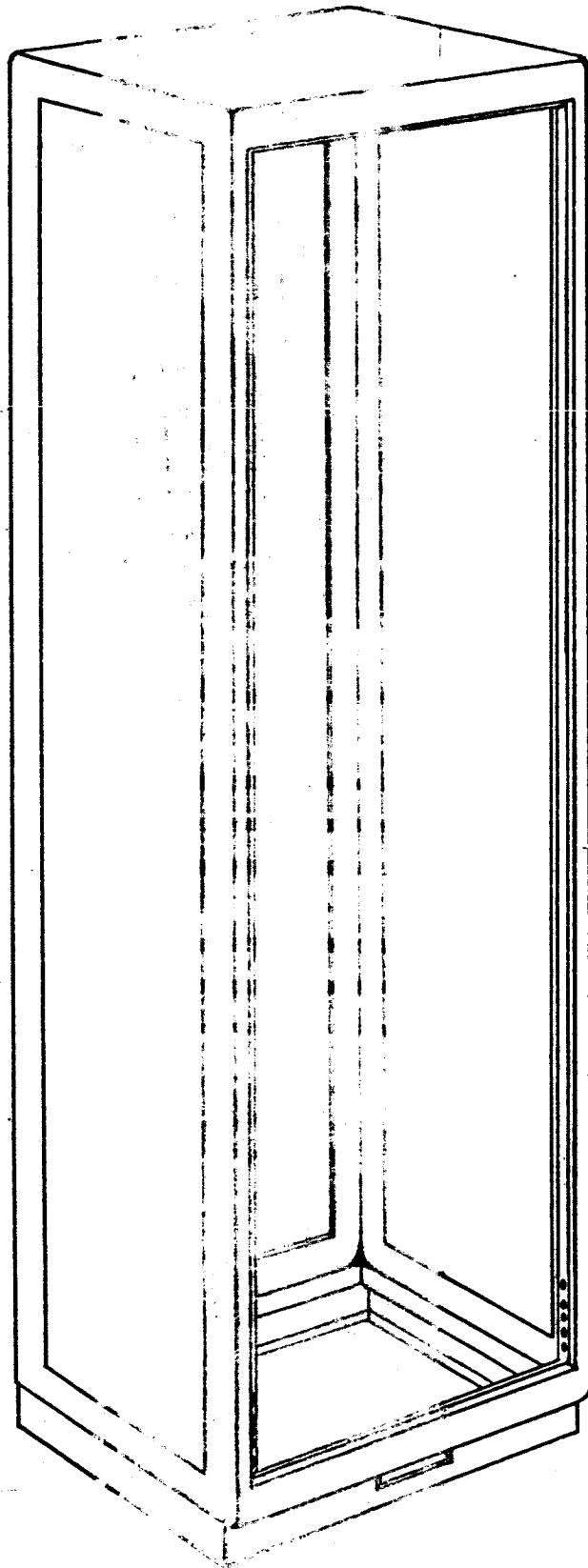


Figure 4-5 Stancore 24-Inch Vertical Rack E64-KN-F-260-10

4-22. STANDARD PATCH DISTRIBUTOR F75M04681 (42 CONNECTOR).

4-23. Description of the Distributor. Standard Patch Distributor F75M04681 is an enclosure 12.5 inches deep, 36.73 inches high and 24 inches wide. It may be mounted in a standard 24 inch rack. On the back of the enclosure, mounted on two hinged plates, are forty-two 61 pin connectors (Bendix PT0SE-24-61P or equal).

Patching between these connectors is provided by the following patchboard assembly, which is accessible through the front panel door:

- a. Frame & Spring Assembly (Anderson Electric Co. BT 2560 or equal)
- b. Removable 2560-pin patchboard (Anderson Electric Co. IPB2560S or equal)

Above the front panel door is a row of 84 taper-pin blocks. These are two types, mounted alternately; 42 blocks (AMP 480107-6 or equal) and 42 blocks (AMP 480107-9 or equal). Each taper-pin block has 30 female test jacks wired to the patchboard.

4-24. Use of the Patch Distributor. The standard patch distributor is employed in various 24-inch relay racks used in the ESE. In these locations, the distributor provides monitoring and flexible distribution of signals. It also accommodates standard relay modules (paragraph 4-37) and standard diode modules (paragraph 4-42), which may be patched into the ESE systems as required.

4-25. Locations of Patch Distributor. The ESE employs approximately eleven standard patch distributors located in the relay racks, room 8-A. See figure 4-6.

4-26. Design Organization. Patch Distributor F75M04681 was designed by MSFC-Astrionics and adopted by KSC-Launch Equipment Branch.

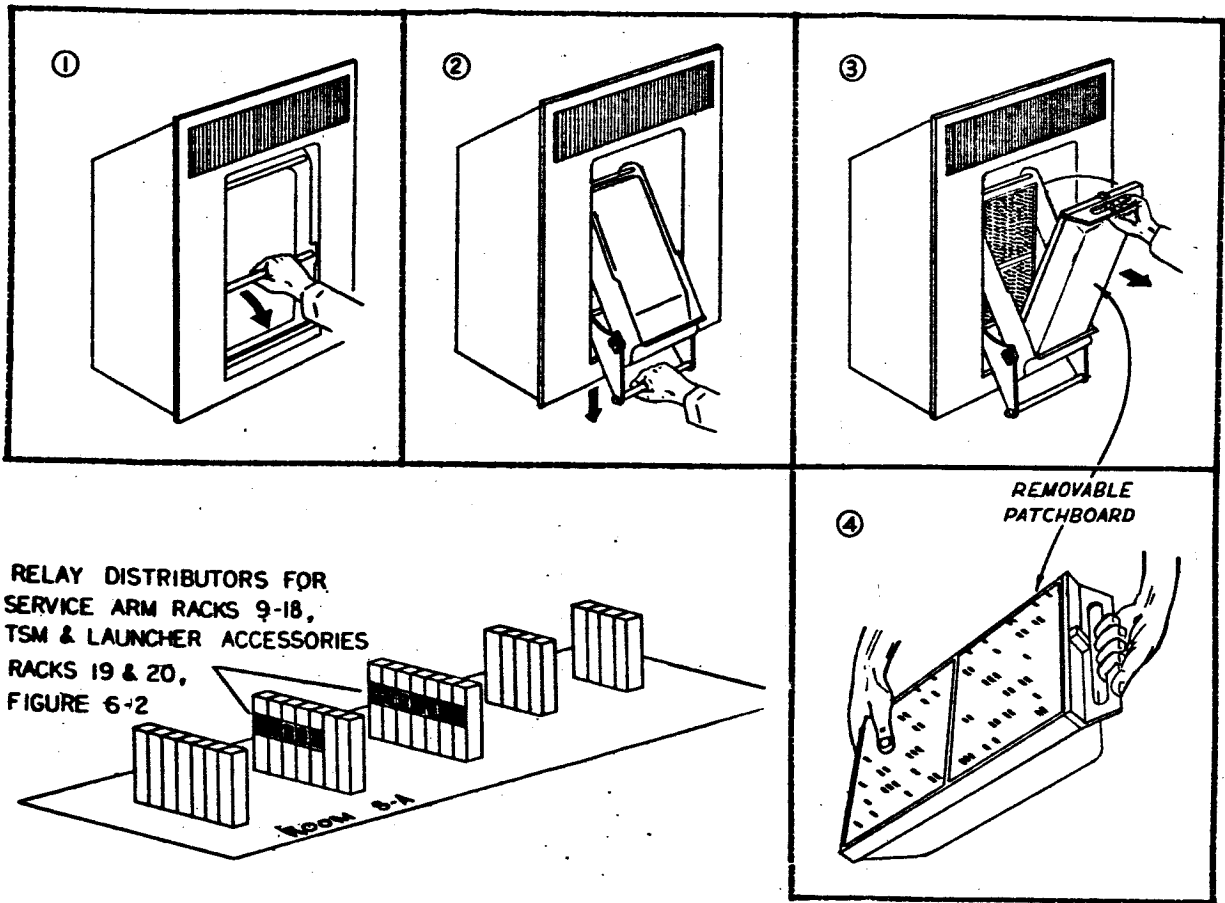
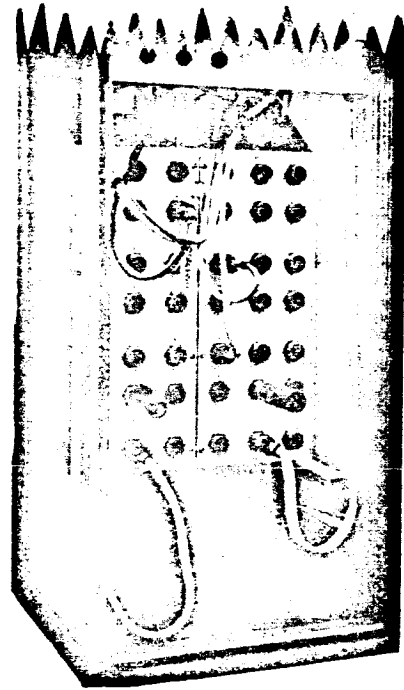
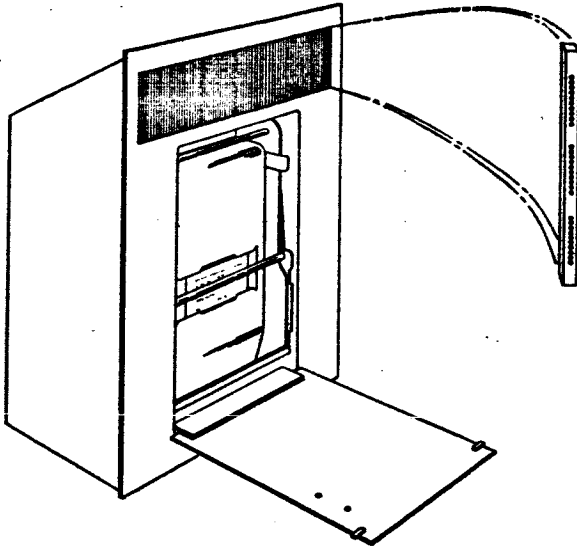


Figure 4-6 Standard Patch Distributor (42 Connector) F75M04681

4-27. STANDARD PATCH DISTRIBUTOR F75M04681 (54 CONNECTOR).

4-28. Description of the Distributor. Standard Patch Distributor F75M04681 is an enclosure 15.12 inches deep, 48.47 inches high and 24 inches wide. It may be mounted in a standard 24 inch rack. On the back of the enclosure, mounted on two hinged plates, are fifty-four 61 pin connectors (Bendix PT07SE-24-61P or equal).

Patching between these connectors is provided by the following patchboard assembly, which is accessible through the front panel door:

a. Frame & Spring Assembly (AMP Inc. 420050-2 or equal)

b. Removable 3254-pin patchboard (AMP Inc. 420051-2 or equal)

Above the front panel door are 2 rows of 54 taper-pin blocks each. These are two types, mounted alternately; 27 blocks (AMP 480107-6 or equal) and 27 blocks (AMP 480107-9 or equal) in each row. Each taper-pin block has 30 female test jacks wired to the patchboard. See figure 4-7.

4-29. Use of the Patch Distributor. The standard patch distributor is employed in various 24-inch relay racks used in the ESE. In these locations, the distributor provides monitoring and flexible distribution of signals. It also accommodates standard relay modules (paragraph 4-37) and standard diode modules (paragraph 4-42), which may be patched into the ESE systems as required.

4-30. Locations of Patch Distributors. The ESE employs approximately four standard patch distributors; two are located in the Integration Racks, room 8-A and one each is located in the Integration Racks in the LCC.

4-31. Design Organization. Patch Distributor F75M04681 was designed by MSFC-Astrionics and adopted by KSC-Launch Equipment Branch.

TO BE PROVIDED

Figure 4-7 Standard Patch Distributor (54 Connector)

**4-32. STANDARD PATCH DISTRIBUTOR D75M09313 (27 CONNECTOR).**

**4-33. Description of the Distributor.** Standard Patch Distributor D75M09313 (figure 4-8) has an enclosure 22.5 inches high, 21.0 inches wide, and 13.06 inches deep. On the back of this enclosure are twenty-seven 61-pin connectors (Bendix PTO7SE-24-61P or equal). Patching between these connectors is provided by the following patchboard assembly:

- a. Frame & Spring Assembly (AMP, Inc. 420048-4 or equal)
- b. Patchboard (AMP, Inc. 420049-3 or equal)

Above the front opening is a row of 54 taper-pin blocks, mounted alternately: 27 blocks (AMP 480107-6 or equal) and 27 blocks (AMP 480107-9 or equal). Each taper-pin block has 30 female test jacks wired to the patchboard.

**4-34. Use of the Patch Distributor.** This patch distributor provides monitoring and flexible distribution of signals. It also accommodates standard relay modules (paragraph 4-37) and standard diode modules (paragraph 4-42), which may be patched into any system using the distributor. It is presently used in the Portable Arm Control Console.

**4-35. Locations of Patch Distributor.** See Portable Arm Control Console, figure 5-12.

**4-36. Design Organization.** Patch Distributor D75M09313 was designed by the Electrical Section, Launch Equipment Branch, KSC.

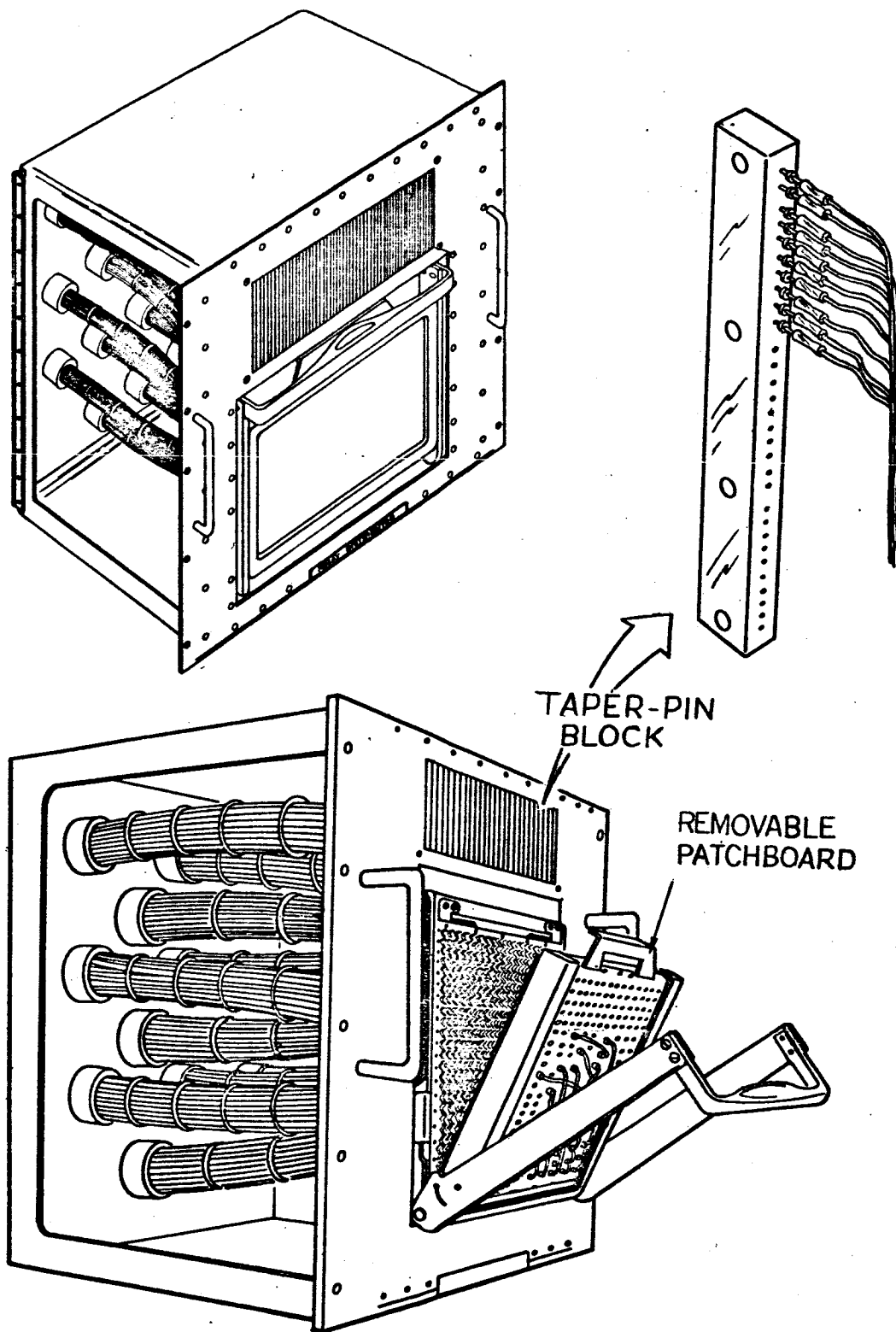


Figure 4-8 Standard Patch Distributor (27 Connector) D75M09313



4-37. STANDARD RELAY MODULE D75M50348.

4-38. Description of the Relay Module. Standard Relay Module D75M50348

(figure 4-9) is an assembly 2.43 inches high and 5.87 inches long, and weighs 14 ounces. It contains:

- a. 8 relays      Per MSFC-SPEC-339/53 (DPDT 28 volts dc relay, coil resistance 600 ohms, contact rating 2.0 amperes at 28 volts dc resistive load)
- b. 8 bases      Burndy Engineering Co. MT9R-2 or equal  
72 base contacts (Burndy RC16Y-1-F59 or equal)

Each relay contact is wired to one pin in the 61-pin connector (Bendix PTO6P-24-61S or equal). For four of the relays, each side of the dc coil is independently wired to a connector pin. In the remaining four relays, one side of each dc coil is wired to a common bus.

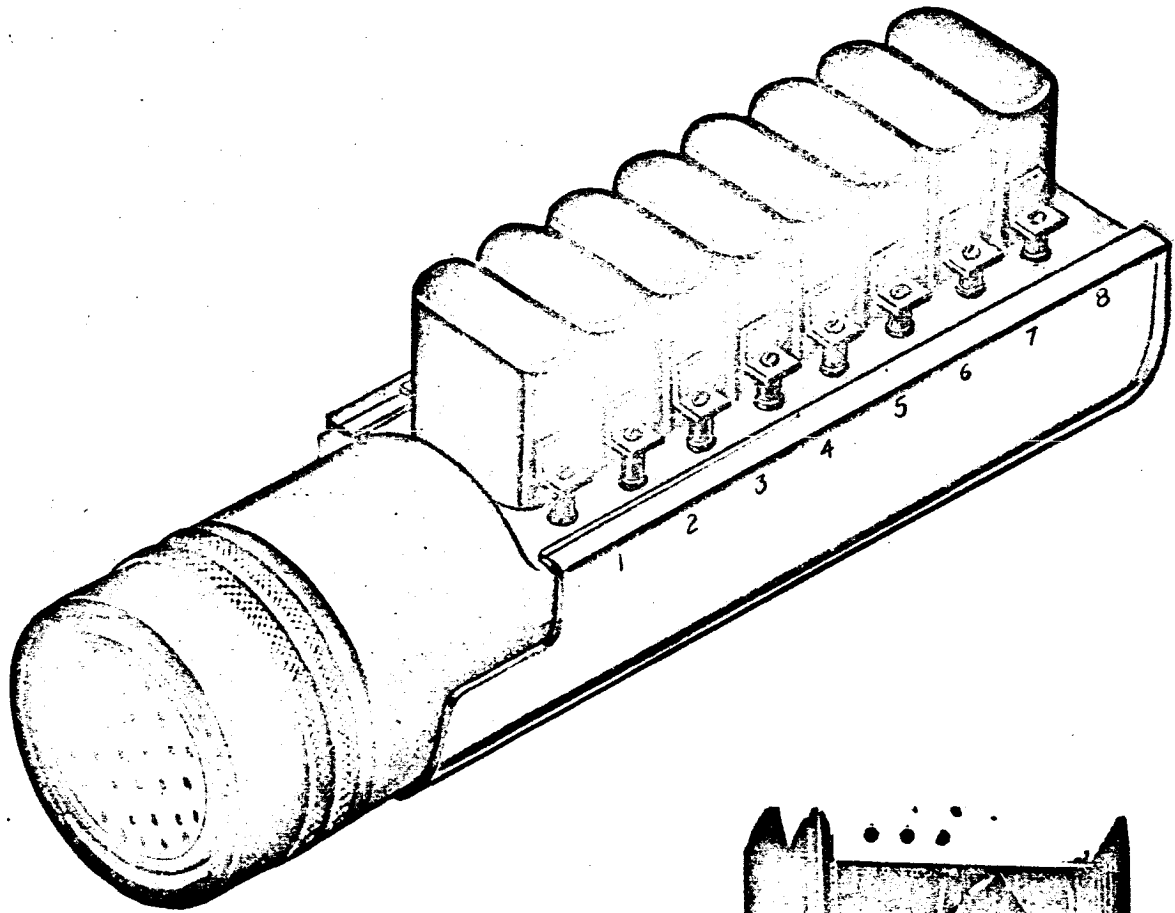
4-39. Use of the Relay Module. The 2-ampere relay contacts in this module are adequate both for logic circuits and for control, since no solenoid valve in the ESE control systems required more than 1.5 amperes.

In the ESE, these relays are used as logic and control elements in the control systems for the service arms, tail service masts, firing circuits and others. They are also used as transfer elements for interface separation so that no control subsystem uses dc power from another.

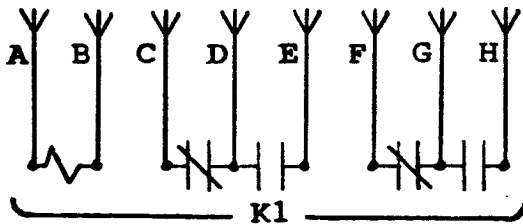
In the Launcher Ground Equipment Test Set (Launcher GETS), standard relay modules simulate switches and relays in the ESE. They also close indicator lamp circuits within the GETS.

4-40. Locations of Relay Module. In the ESE, about 150 relay modules are connected to various patch distributors in the relay racks in room 8-A and in the Launcher GETS.

4-41. Design Organization. Relay Module D75M50348 was designed by MSFC-Astrionics and adopted by KSC-Launch Equipment Branch.



TYPICAL OF FOUR



TYPICAL OF FOUR

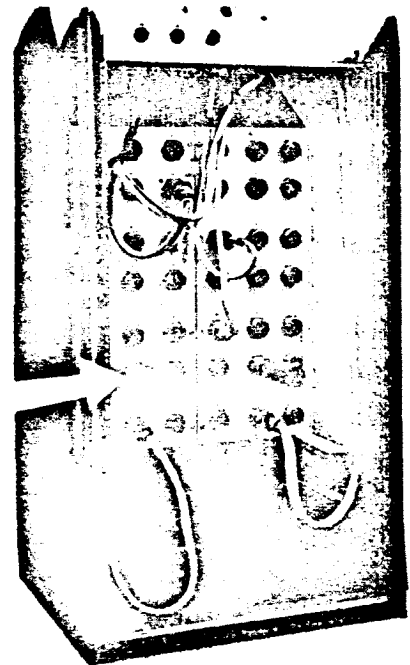
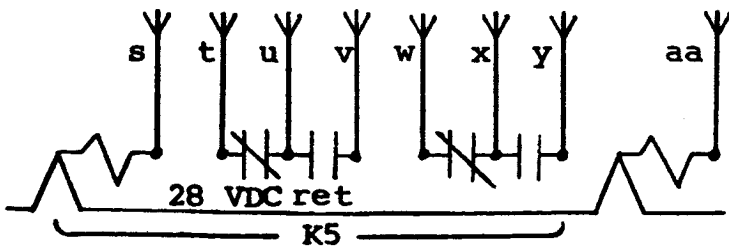


Figure 4-9 Standard Relay Module D75M50348

4-42. STANDARD DIODE MODULE D75M50364.

4-43. Description of the Diode Module. Standard Diode Module D75M50364 is 2.75 inches in diameter and 10.25 inches long, including connector. It weighs 2.25 pounds.

Inside are thirty diode-fuse pairs, each series-wired to two pins in the 61-pin connector (Bendix PT06P-24-61S or equal). The diodes are type 1N540 (Texas Instruments or equal); the fuses are Bussman GLD-1 or equal.

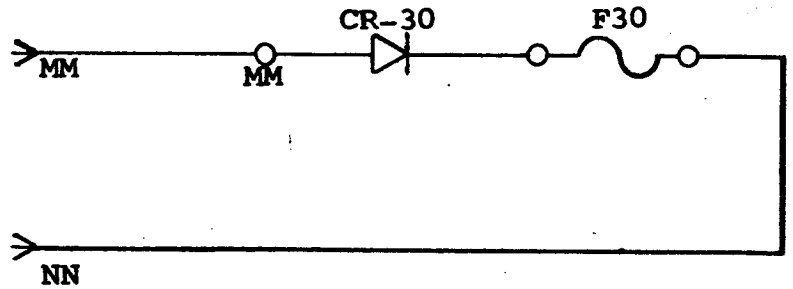
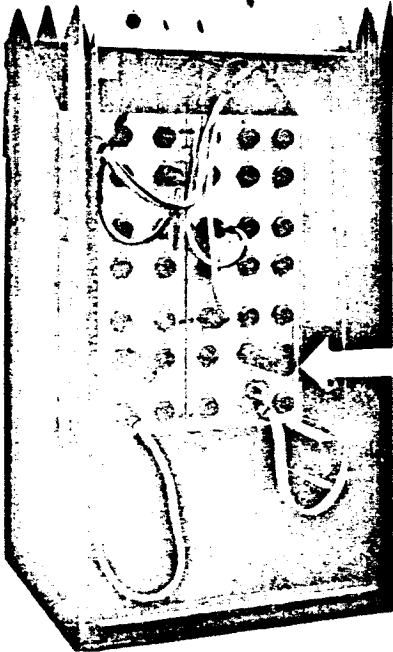
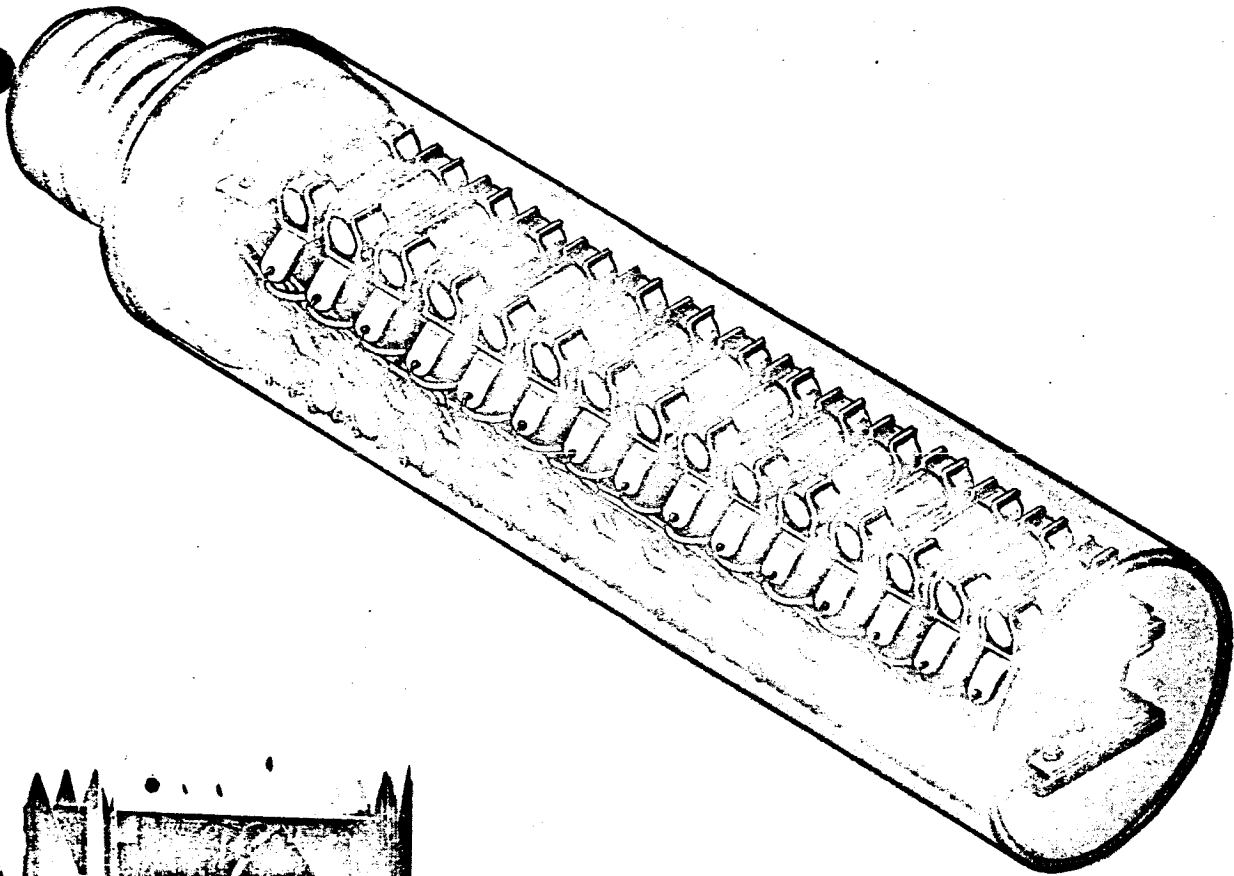
All fuse clips are mounted together on two fuse boards, and all diodes are mounted on a single board in between. All wiring at the adapter end is potted.

4-44. Use of the Diode Module. Standard Diode Module D75M50364 can be used with any electrical distributor mating with the connectors named above, and is presently used with the Standard Patch Distributors. See figure 4-10.

In the ESE, the diodes in this module are used to suppress arcing between relay contacts.

4-45. Locations of Diode Module. About 50 diode modules are used in Electrical Support Equipment, including the Launcher GETS and the relay racks in room 8-A.

4-46. Design Organization. Diode Module D75M50364 was designed by the Electrical Section, Launch Equipment Branch.



TYPICAL CIRCUIT

Figure 4-10 Standard Diode Module D75M50364

4-47. STANDARD METER PANEL J75M10165.

4-48. Description of the Meter Panel. Standard Meter Panel J75M10165 (figure 4-11) is designed for placement in the standard 24-inch rack. It is 10.5 inches high and approximately 3 inches deep. The panel has:

- |                 |  |
|-----------------|--|
| a. 30 fuses     | Bussmann GLD-1 or equal<br>Bussman Fuseholder HLD or equal           |
| b. 30 switches  | Toggle T-2150, Control Co. of America or equal                       |
| c. 60 jacks     | E. F. Johnson Co. or equal<br>108-901 thru 108-913 (assorted colors) |
| d. 1 multimeter | Simpson Model 270 or equal   |

Interconnections of these components are shown on the front panel silkscreen.

Note that a shorting switch is added to the multimeter input for zeroing the ohmmeter scale.

4-49. Use of the Meter Panel. In the ESE, this meter panel is employed in the relay and power racks in room 8-A. In each case, test leads will normally be connected to test pin blocks in a patch distributor. See figure 4-8.

Standard Meter Panel J75M10165 is normally used to measure resistance and dc voltage.

4-50. Locations of Meter Panel. Approximately seven standard meter panels are used in the Mobile Launcher relay racks and arming panel racks, room 8-A.

4-51. Design Organization. Standard Meter Panel J75M10165 was designed by the Electrical Section, Launch Equipment Branch.

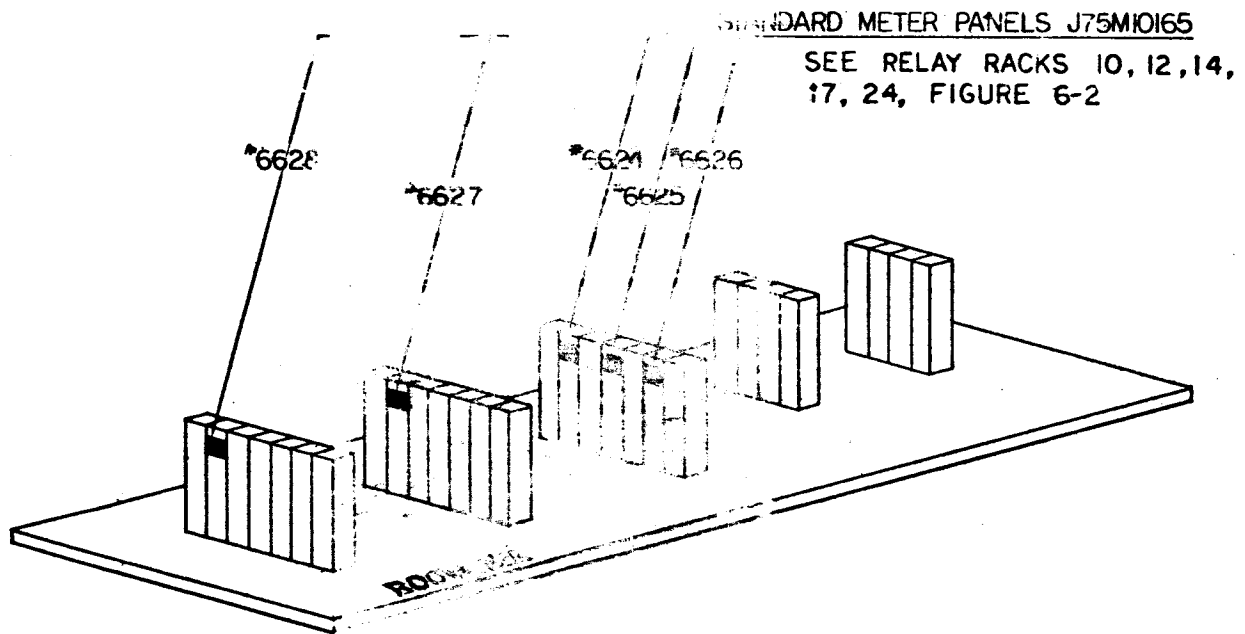
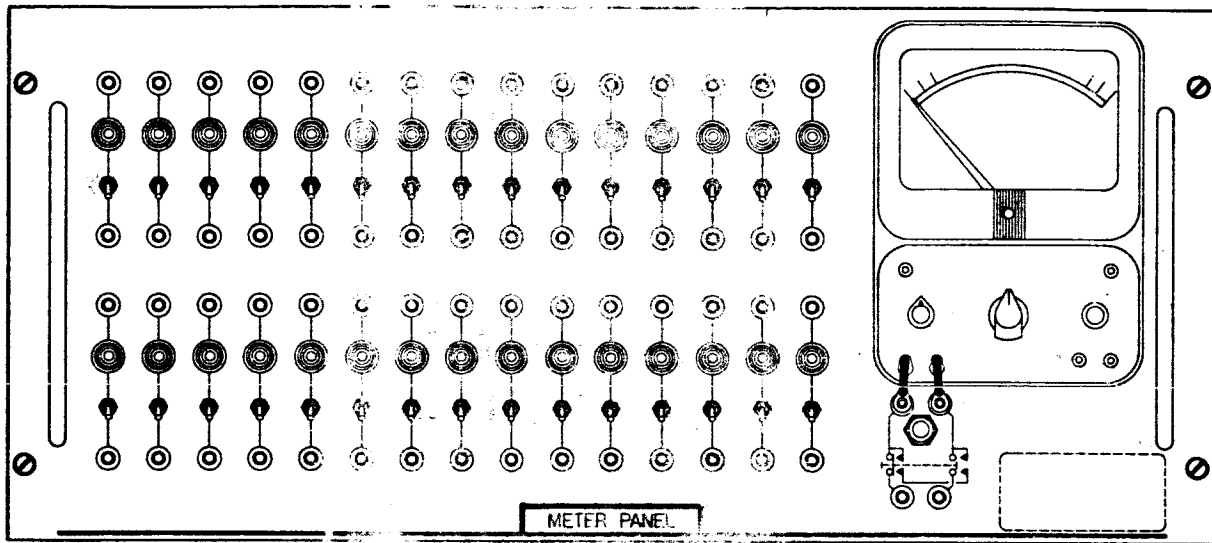


Figure 4-11 Standard Meter Panel J75M10165

4-52. STANDARD FUSE PANEL J75M10163.

4-53. Description of the Fuse Panel. Standard Fuse Panel J75M10163 (figure 4-12) is designed for placement in the standard 24 inch rack. It is 5.25 inches high and 6 inches deep.

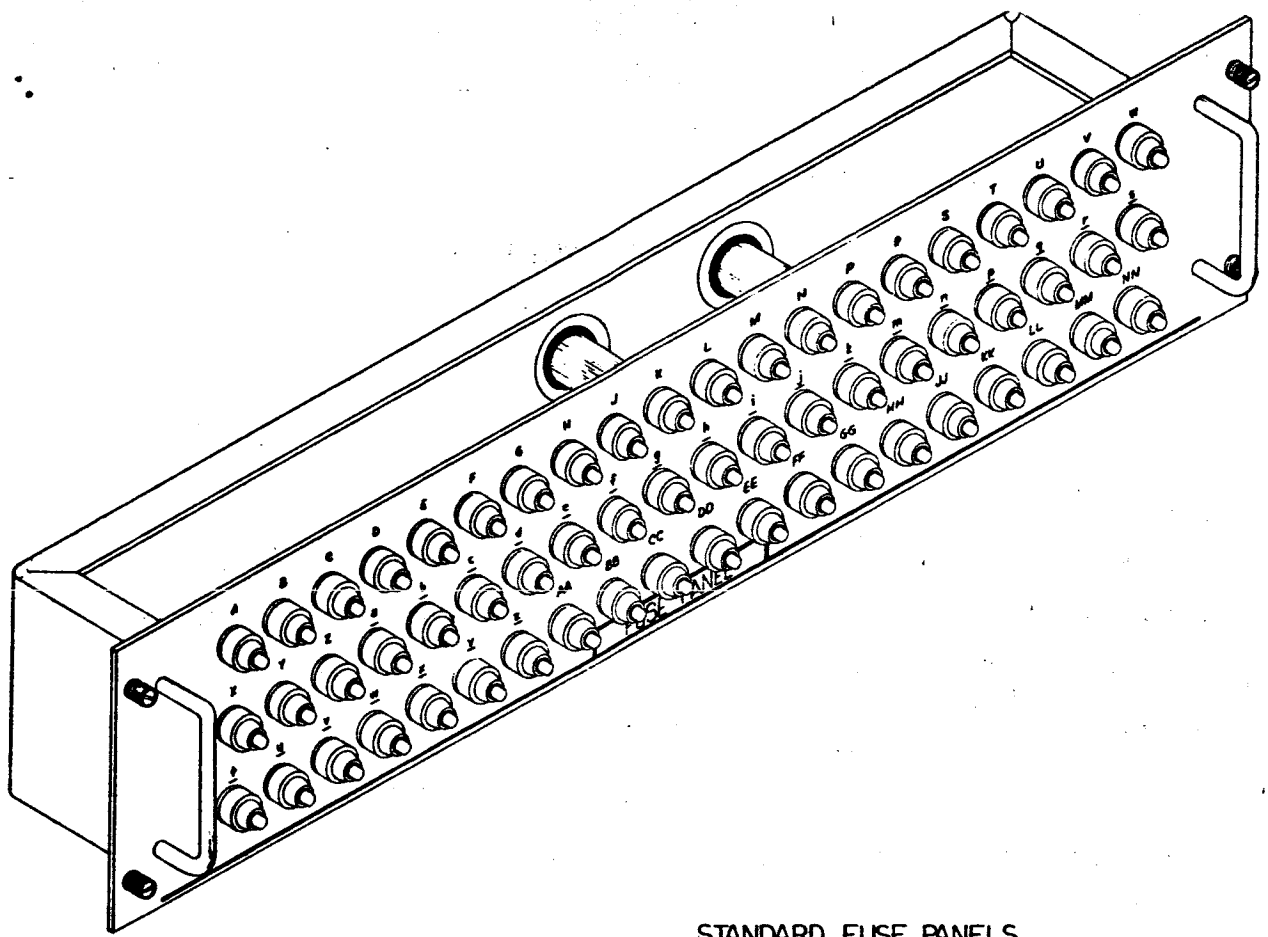
Sixty 3-ampere indicator fuses (Bussman GBA-3 or equal) are connected to two 61-pin connectors (Bendix PTO7SE-24-61P or equal). All signals enter one connector and leave through the other.

4-54. Use of the Fuse Panel. From one to three fuse panels are used in each relay rack in room 8-A. Signals passing through the relay racks are routed through this fuse panel, so that cabling is protected during tests and exercises.

Prior to launch, the fuse panel is bypassed by disconnecting its two cable connectors and mating them to each other.

4-55. Locations of Fuse Panel. Approximately 15 fuse panels are used in the relay racks and GETS, as in figure 4-12.

4-56. Design Organization. Standard Fuse Panel J75M10163 was designed by the Electrical Section, Launch Equipment Branch.



STANDARD FUSE PANELS  
SEE RELAY RACKS 9-16, FIGURE 6-2

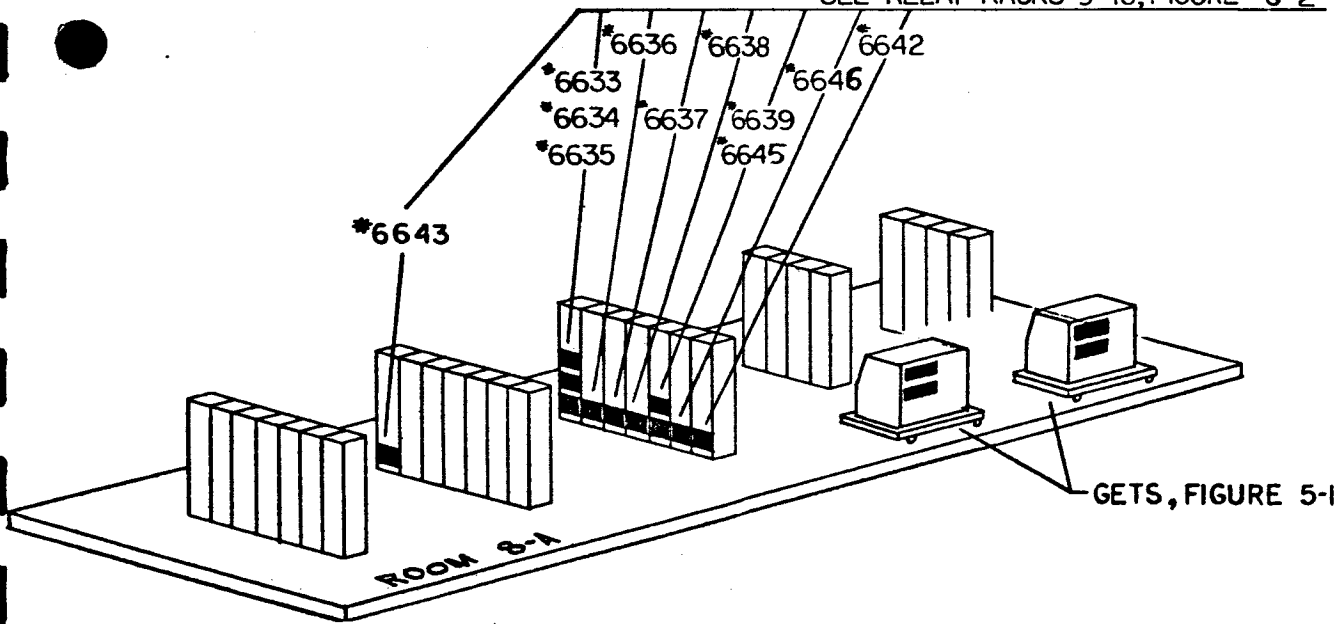


Figure 4-12 Standard Fuse Panel J75M10163



4-57. STANDARD MODULE FRAME J75M10432.

4-58. Description of the Module Frame. Standard Module Frame J75M10432 (figure 4-13) is designed for placement in the back of the standard 24-inch vertical rack. It is approximately 7 inches high and 2 inches deep.

4-59. Use of the Module Frame. The module frame accommodates a variety of components to be connected to the standard patch distributor, as illustrated in figure 4-13. These are normally components other than diodes or relays, since standard relay and diode modules are already available.

A typical use of the frame is shown in figure 4-13, where three electronic timers are mounted in a component module assembly D75M10433, which is mounted to the Standard Module Frame.

4-60. Locations of Module Frame. At present the module frame is not used in the ESE.

4-61. Design Organization. The Standard Module Frame J75M10432 was designed by MSFC-Astrionics and adopted by KSC-Launch Equipment Branch.

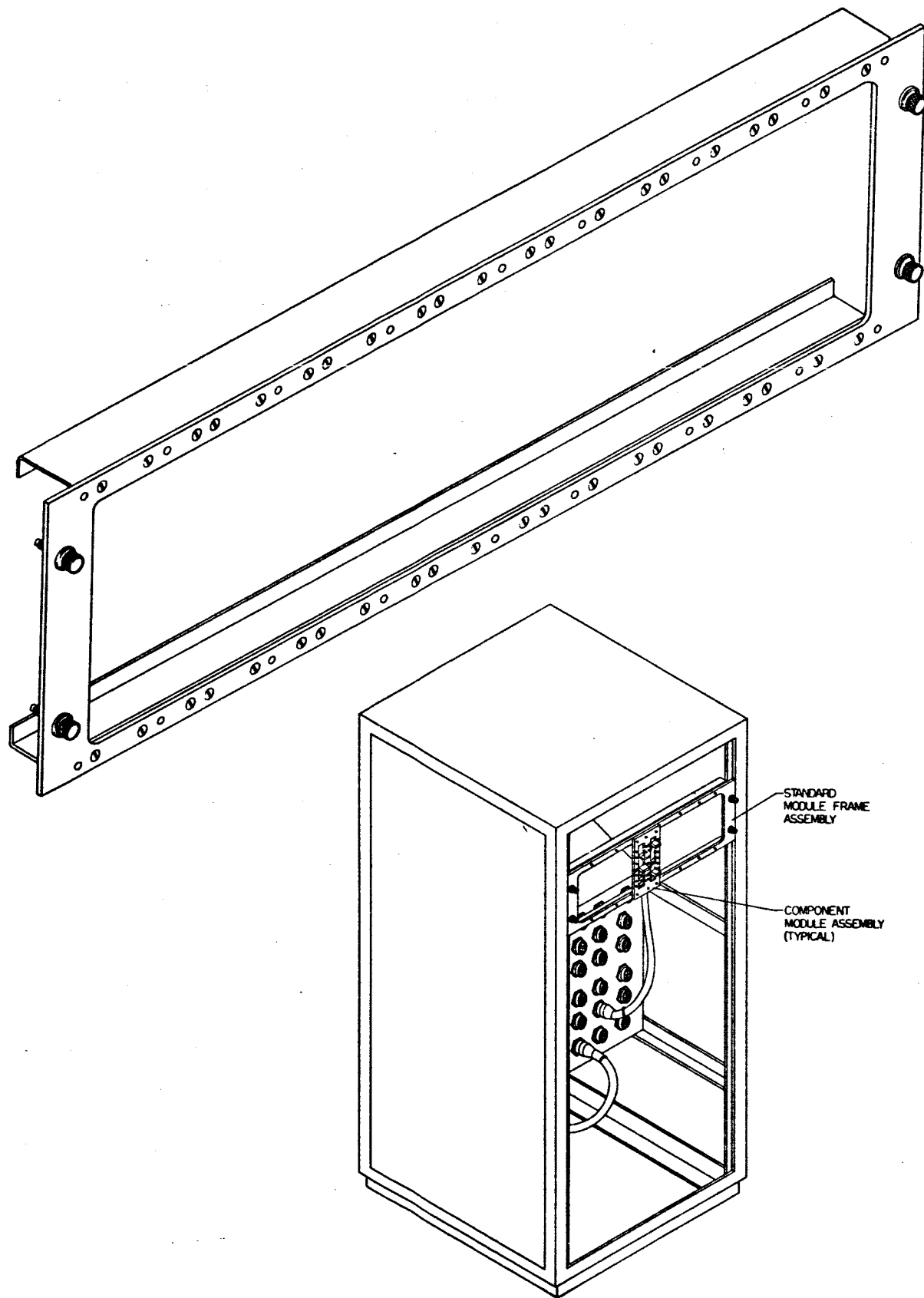


Figure 4-13 Standard Module Frame J75M10432

4-62. STANDARD 500 AMPERE POWER SUPPLY.

4-63. Description of the Power Supply. The 500 Ampere Power Supply (figure 4-14) is designed for placement in a standard 24-inch rack. It is 47.25 inches high and 24 inches deep. Electrical connections to this supply are provided as follows:

- |                             |                                      |
|-----------------------------|--------------------------------------|
| a. AC Input Connector       | Bendix 323340-57P or equal           |
| b. Remote Sensing Connector | Bendix PT07SE-20-39S or equal        |
| c. DC Output                | Bus bars extending from rear of unit |

The front panel of the supply contains:

- Panel meters for dc output voltage and current
- Switches POWER ON and START-STOP
- Control for voltage output level
- Meter indicating elapsed time

This supply is designed for a maximum output current of 500 amperes and a nominal output voltage of 28 volts dc, adjustable from 24 volts to 36 volts.

Output ripple is less than 0.1 volts peak to peak.

The input power required is 220/440 volts, 3-phase, 60 cycle. A stepdown transformer with a three-phase, full-wave semiconductor bridge is used for the ac to dc conversion. The output voltage level is controlled by silicon control rectifiers.

4-64. Use of the Power Supply. The standard 500-ampere power supply is the prime source of 28 volt dc power for all electrical functions on the service arms other than battery-operated devices. It is also used to provide 28 volt power for the Tail Service Masts, Launcher Accessories, and other ESE Subsystems.

4-65. Location of Power Supply. Two 500 ampere power supplies are located in room 8-A of the Mobile Launcher. One is currently a spare.

4-66. Design Organization. The standard 500 ampere power supply was designed by MSFC-Astrionics. For LC-39, it is furnished by MSFC and used by KSC-Launch Equipment Branch.

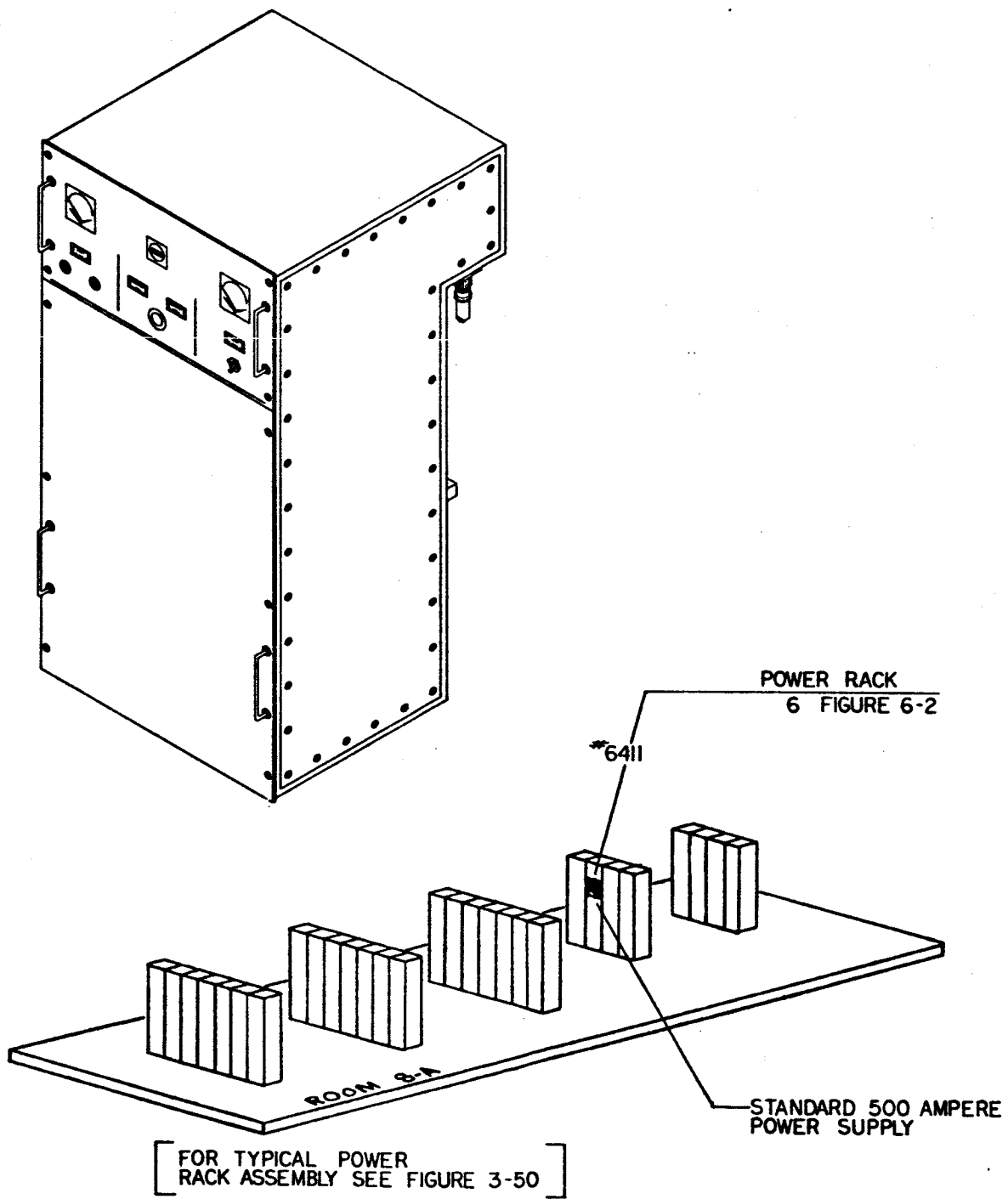


Figure 4-14 Standard 500 Amp Power Supply A40M03709

4-67. STANDARD TEST LOAD E40M03971.

4-68. Description of the Test Load. Test Load Assembly E40M03971 is designed for placement in the standard 24-inch rack. It is 5.15 inches high and 18 inches deep.

It contains four resistance elements (Chromalox 4-43209 or equal, 30 volts, 300 watts) connected in parallel by bus bars at the rear of the chassis. The buses are connected to input cables by terminal lugs.

4-69. Use of the Test Load. Test Load Assembly E40M03971 is a part of the standby battery racks and firing battery racks. It serves as a test load for the batteries and may be connected to the batteries by the Load Control Panel, figure 4-16.

4-70. Locations of Test Load. See figure 4-15.

4-71. Design Organization. Standard Test Load E40M03971 is designed by MSFC-Astrionics. For LC-39, it is furnished by MSFC and used by KSC-Launch Equipment Branch.

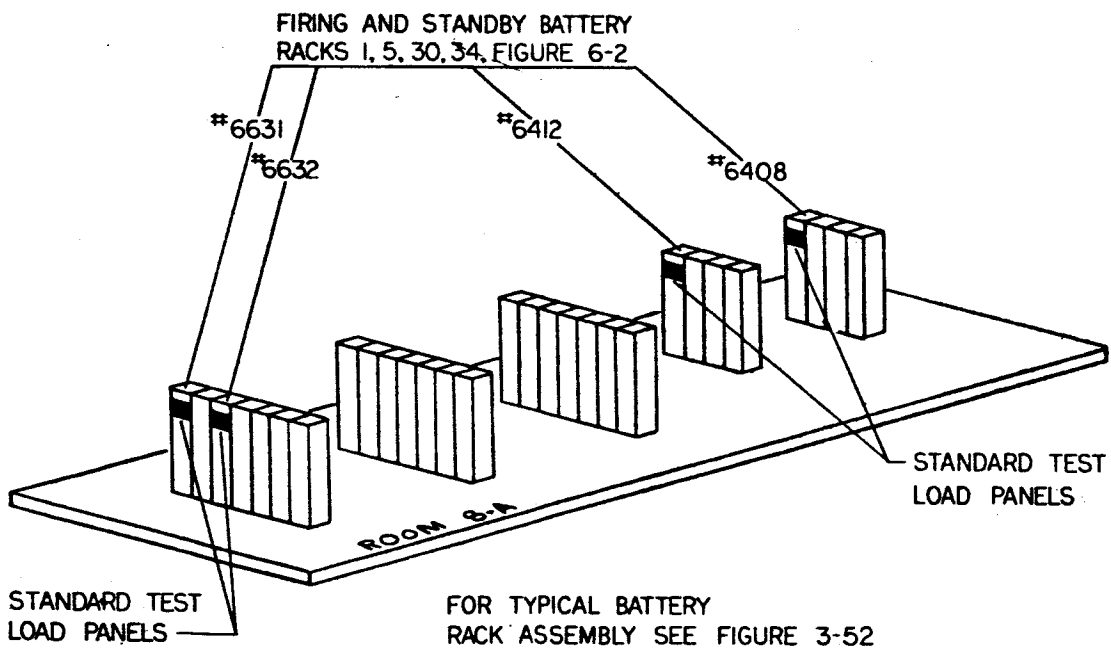
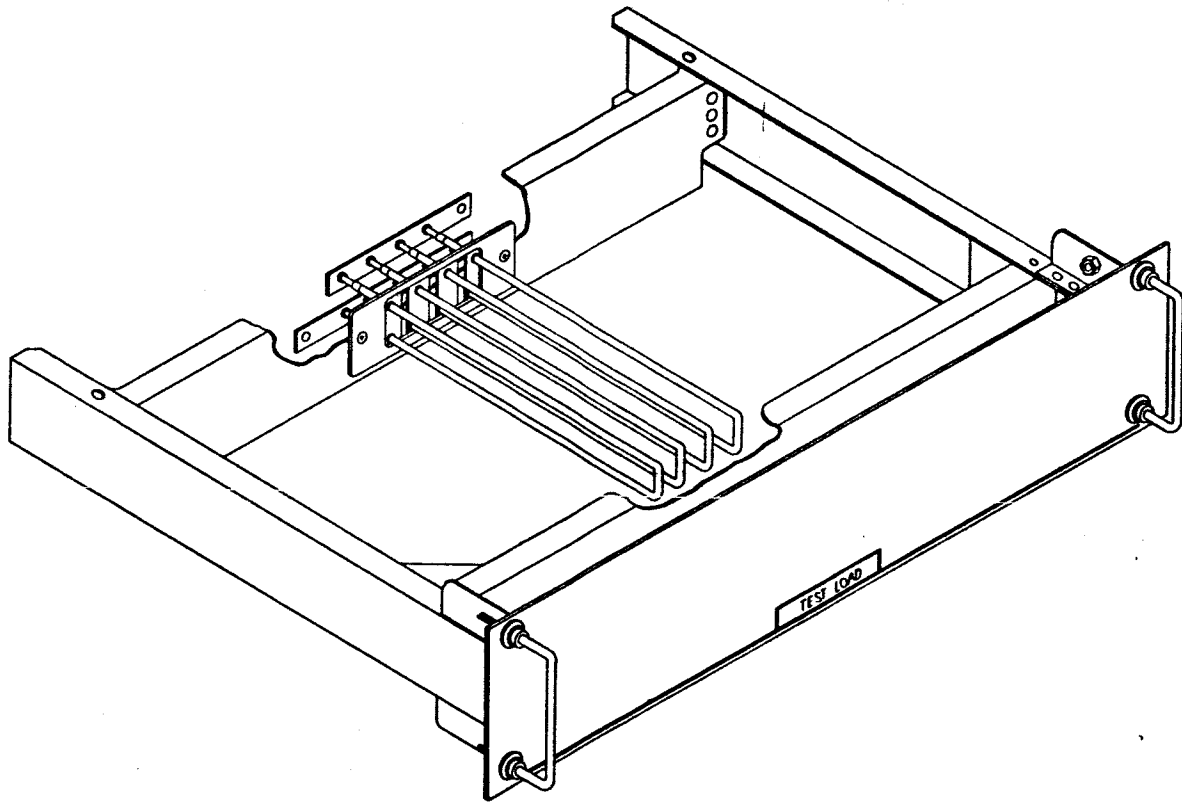


Figure 4-15 Standard Test Load E40M03971

4-72. STANDARD LOAD CONTROL PANEL E40M03535.

4-73. Description of the Load Control Panel. Load Control Panel E40M03535 is designed for placement in the standard 24 inch rack. It is 7 inches high and 20.25 inches deep.

It contains a DPDT contactor (Hartman A-848K or equal) controlled by a front-panel switch. The contactor is actuated by a 28-volt dc coil. The normally open contacts are rated at 200 amperes, and the normally closed contacts at 300 amperes.

The dc power controlled by the contactor is supplied through two single-pin connectors (Superior RS250GB, RS250GR or equal) and is routed through contacts to a group of bus bars.

4-74. Use of the Load Control Panel. In the ESE, a Load Control Panel is located in each standby battery rack and firing battery rack. It is used to disconnect the normal circuits from the batteries in these racks and to connect the standard test load, figure 4-15.

4-75. Locations of Load Control Panels. Four Load Control Panels are used in room 8-A as shown in figure 4-16.

4-76. Design Organization. The Standard Load Control Panel E40M03535 is designed by MSFC-Astrionics. It is furnished by MSFC and used by KSC-Launch Equipment Branch.

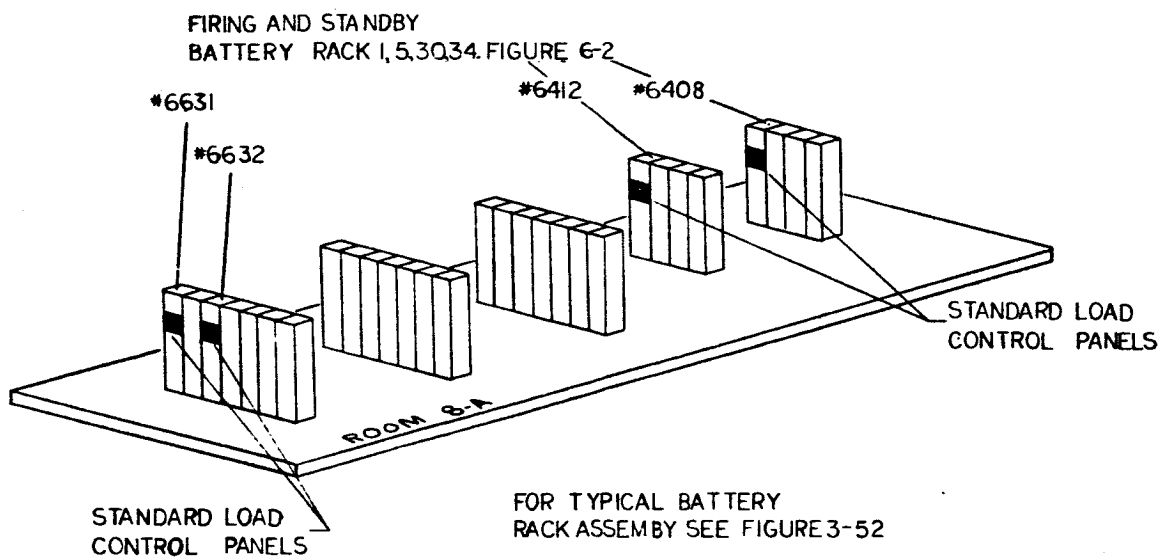
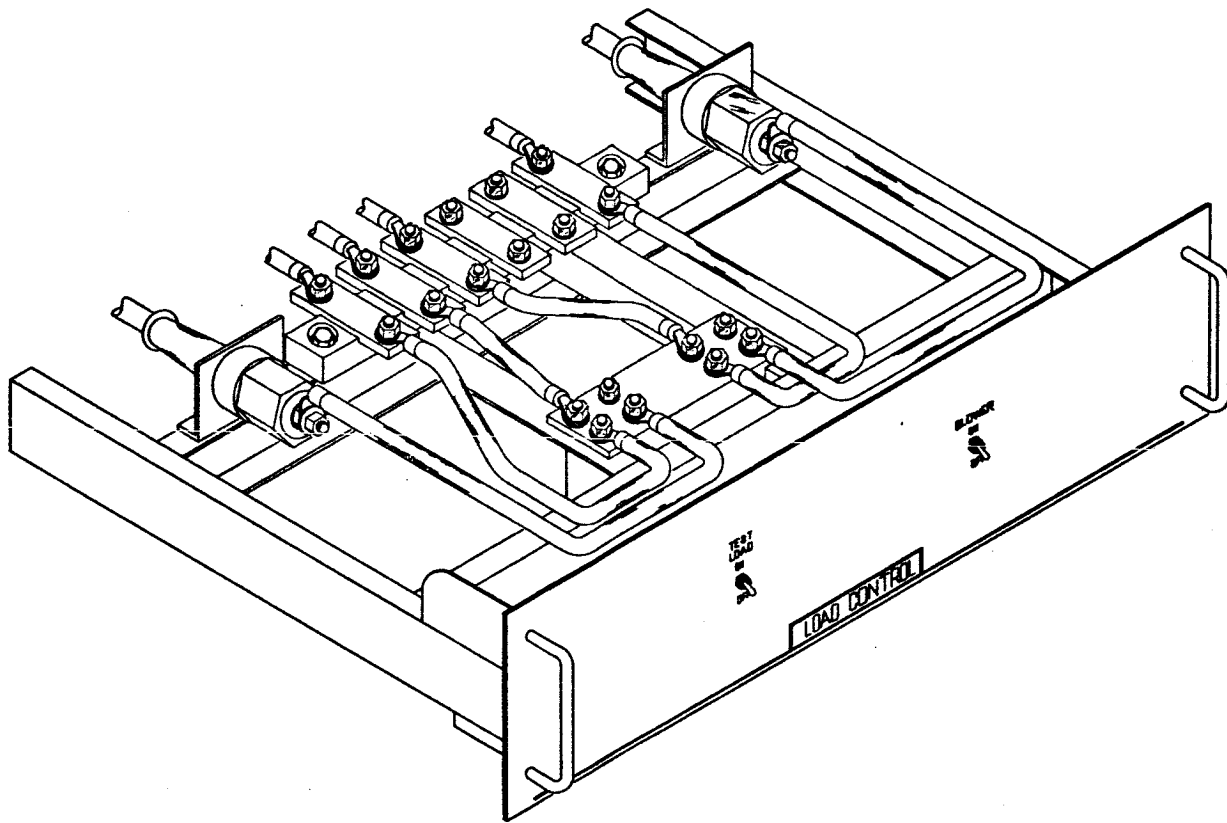


Figure 4-16 Standard Load Control Panel E40M03535



**4-77. STANDARD BATTERY CHARGER.**

**4-78. Description of the Battery Charger.** The Standard Battery Charger (figure 4-17) is designed for placement in a standard 24-inch rack. It is 12.25 inches high and 24 inches deep.

Input power to the charger is 220/440 volts, 3-phase 60-cycle. A terminal strip is provided for this input. A connector is provided for remote-readout and control-function cabling. DC output cables to the battery load are connected by terminal lugs fixed to bus bars.

The front panel of the charger contains:

- a. Switch-lights which indicate Power ON-OFF, Load Charge, Reset, Temperature Alarm, and Cells Charging
- b. Panel meters showing charger dc voltage and current
- c. Digital readouts for percent discharge, peak discharge, and charge cycle.

The dc power for battery charging is obtained by transforming and rectifying the three-phase line voltage. Semiconductors are used for rectifying, some of which are Silicon Control Rectifiers (SCR) which also control load current. The SCR's are gated by magnetic amplifiers whose input is a comparison of battery voltage to a Zener reference.

**4-79. Use of the Battery Charger.** To maintain battery charge, one battery charger is used in each ESE battery rack. These are:

- a. Two firing battery racks used in the service arms firing circuit, paragraph 3-25.
- b. Two standby battery racks which provide power supply backup.

**4-80. Locations of Battery Chargers.** The ESE uses four Standard Battery Chargers as in figure 4-17.

**4-81. Design Organization.** The Standard Battery Charger was designed by MSFC-Astrionics. It is furnished by MSFC and used by KSC-Launch Equipment Branch.

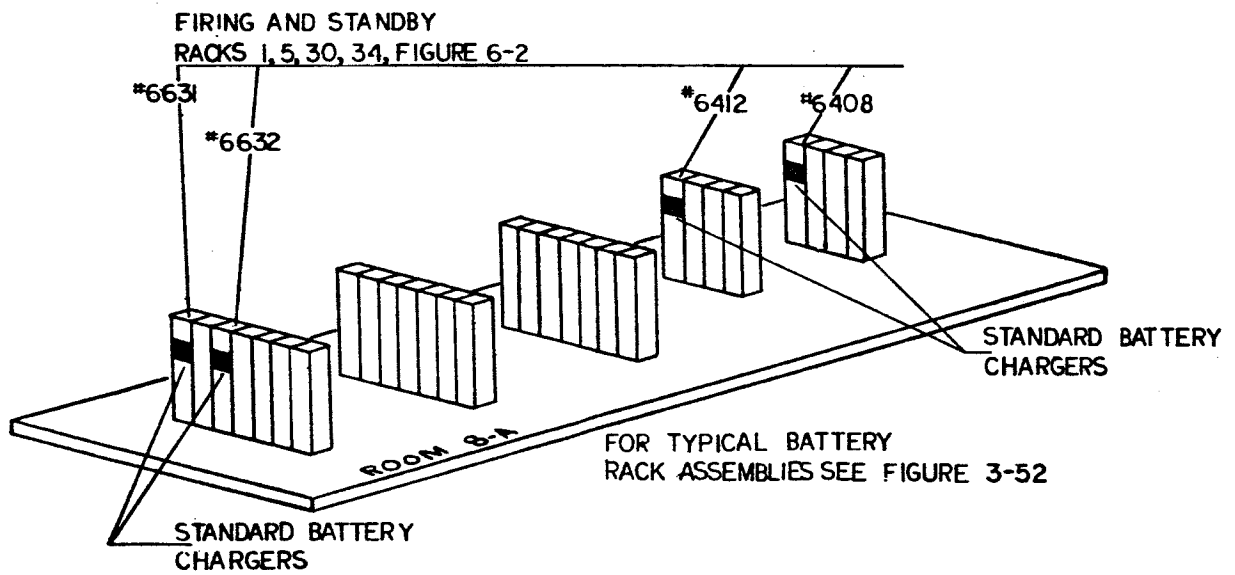
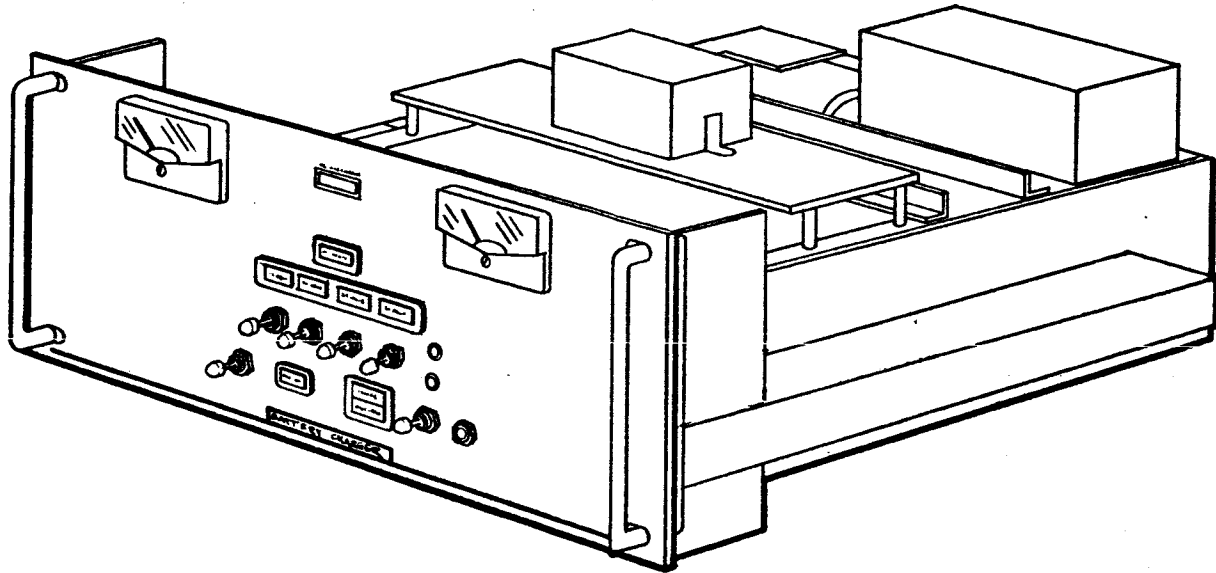


Figure 4-17 Standard Battery Charger

4-82. STANDARD BATTERY PANEL J40M03605.

4-83. Description of the Battery Panel. Standard Battery Panel J40M03605 (figure 4-18) is designed for placement in the standard 24-inch rack. It is 15.72 inches high and approximately 19 inches deep, excluding front handles.

The panel is designed for a maximum load of 240 pounds (120 pounds per battery module, 2 modules maximum).

4-84. Use of the Battery Panel. Standard Battery Panel J40M03605 accommodates batteries for the Firing Batteries and Standby Battery Racks.

4-85. Locations of Battery Panels. Eight battery panels are used for the ESE in room 8-A.

4-86. Design Organization. Standard Battery Panel J40M03605 was designed by the Electrical Section, Launch Equipment Branch.

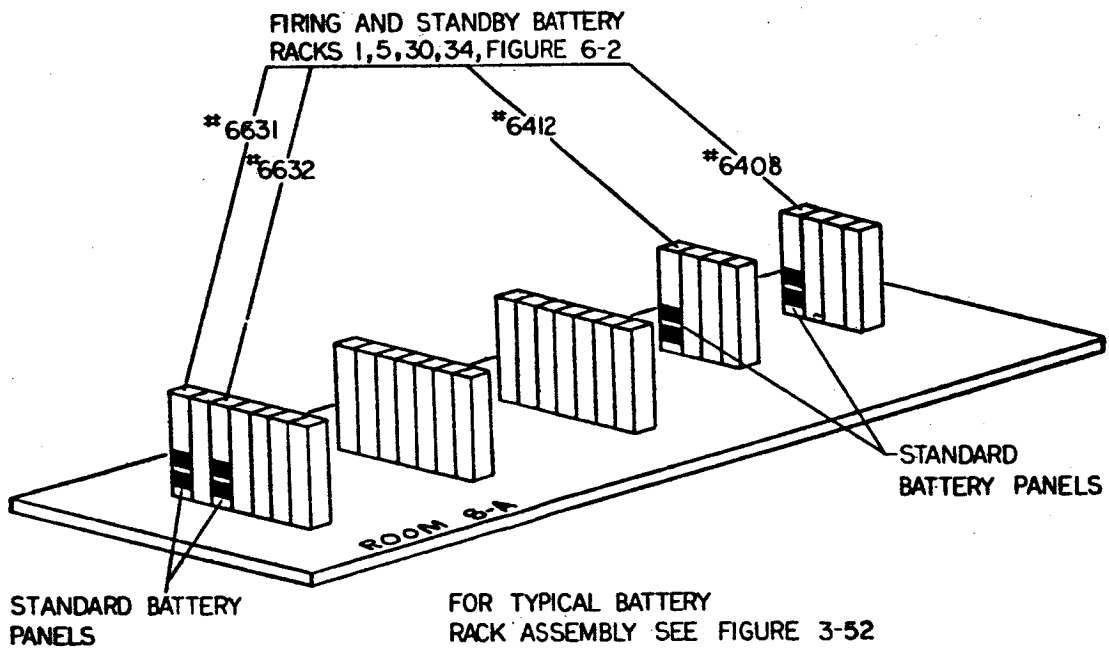
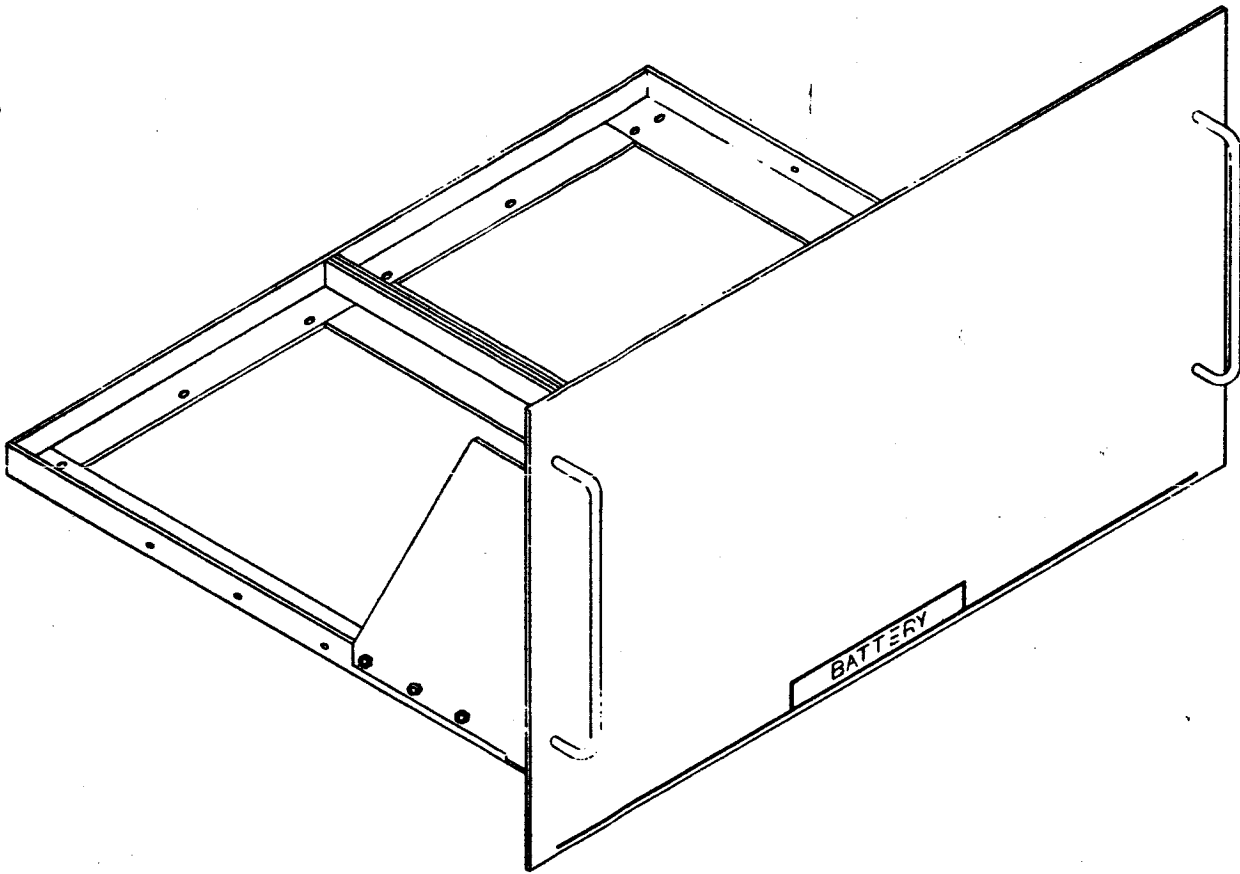


Figure 4-18 Standard Battery Panel J40M03605

4-87. STANDARD POWER MODULE TYPE IIA J75M12295.

4-88. Description of the Power Module. Standard Power Module J75M12295 (figure 4-19) is 8.5 inches high, 4 inches wide, and 14 inches deep. These modules are mounted in a Standard DC Power Frame Type II E75M12301, which in turn mounts in the standard 24-inch rack. Each power module contains:

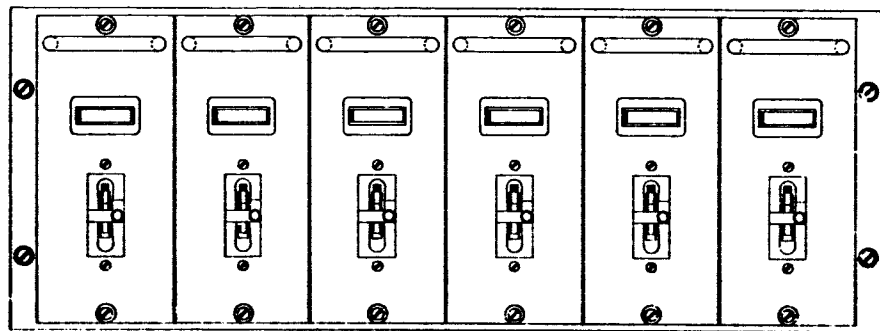
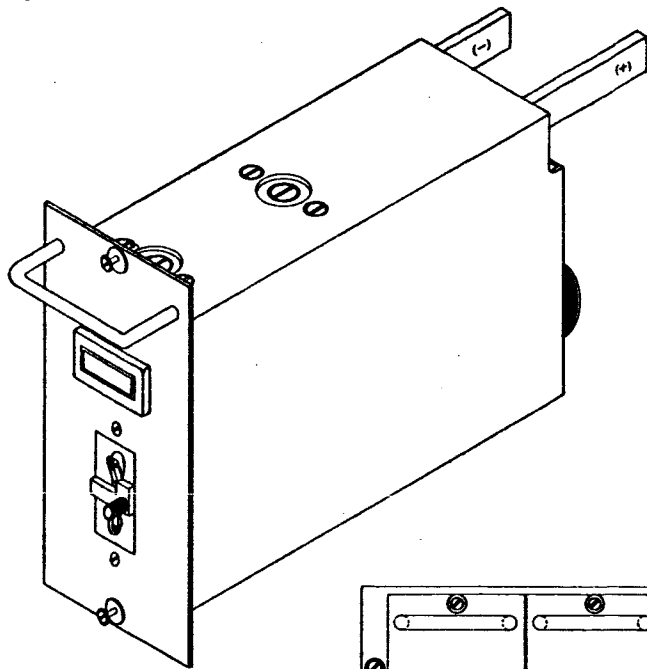
- |                       |   |
|-----------------------|---|
| a. 1 4-pin connector  | Bendix HUS-KEY 10-323332-17S<br>or equal                                |
| b. 1 10-pin connector | Bendix PT07SE-12-10S or equal   |
| c. 1 circuit breaker  | Heinemann Electric Co.<br>AM1510-70 Single-pole,<br>70 amperes or equal |
| d. 1 contactor        | Cutler-Hammer 6042-H119<br>or equal                                     |
| e. 1 shunt            | Empro Mfg. Co. 1242-11-50<br>75 amperes or equal                        |
| f. 1 indicating light |   |

A circuit breaker and contactor are series-wired to provide both overload protection and remote control capability. A shunt is provided for current measurement.

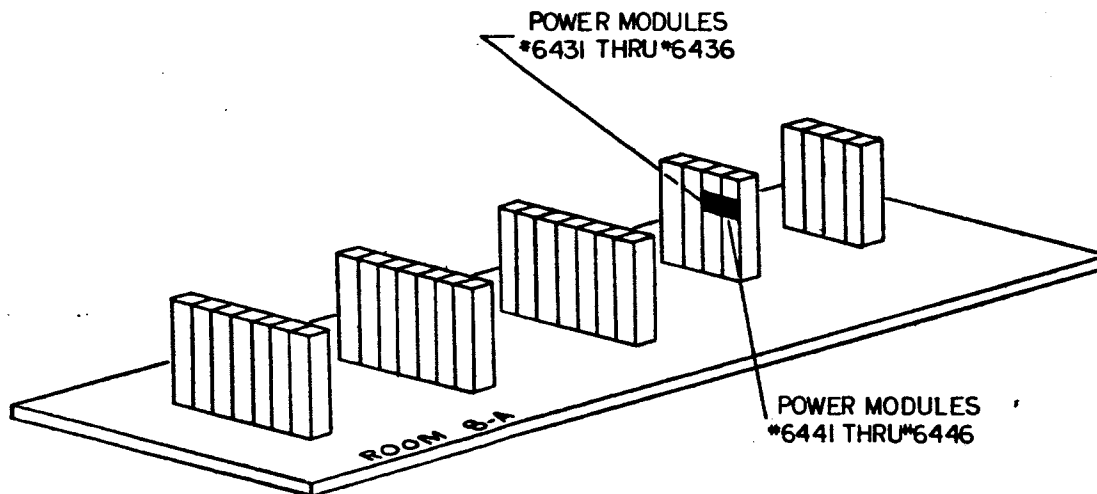
4-89. Use of the Power Module. All power modules are mounted in the ESE power distributor racks. A power module is provided for each of the nine Service Arms, the Hydraulic Charging Unit, Computer Command Functions and the Service Arm Control Switch Assembly (liftoff switch pneumatics). One module is used for the three Tail Service Masts and another for the Pneumatics Control and Distribution Subsystem.

4-90. Locations of Power Modules. There are 18 power modules used in the ESE. Power Distribution Racks #6413, #6414 and #6719 contain six each. All are in room 8-A of the Mobile Launcher base.

4-91. Design Organization. Standard Power Module Type IIA was designed by MSFC-Astrionics and is used by KSC-Launch Equipment Branch.



FOR LIST OF COMPONENTS SEE  
POWER DISTRIBUTION RACKS  
3,7,8 FIGURE 6-2



FOR TYPICAL POWER DISTRIBUTION  
RACK ASSEMBLY SEE FIGURE 3-51

**Figure 4-19** Standard Power Module Type IIA J75M12295  
and Standard Frame Assembly E75M12301

4-92. STANDARD RECEPTACLE DISTRIBUTOR TYPE I E75M12316.

4-93. Description of the Receptacle Distributor. Standard Receptacle Distributor Type I E75M12316 is designed for placement in a standard 24-inch rack. It is 3.6 inches high and 12 inches deep (see figure 4-20).

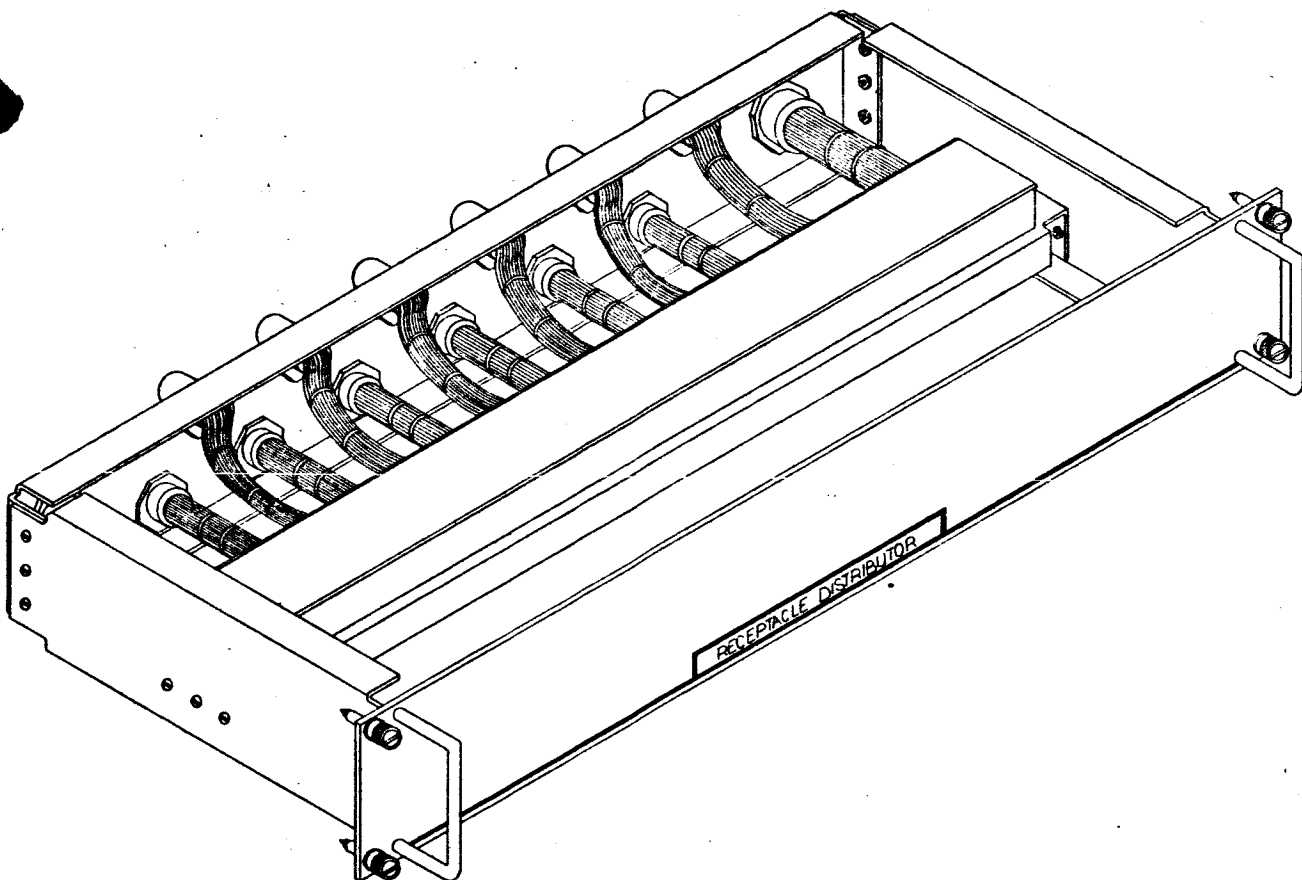
It contains twelve 10-pin connectors (Bendix PT07SE-12-10P or equal) and two 61-pin connectors (Bendix PT07SE-24-61S or equal). One of the 61-pin connectors is wired to six of the 10-pin connectors, and the other 61-pin connector is wired to the remaining 10-pin connectors.

4-94. Use of the Receptacle Distributor. The Receptacle Distributor Type I is used in the Power Distribution Racks (figure 3-51) to route the 28 volt control and monitoring signals from individual power modules to the relay racks. It combines the twelve 10-conductor cables from the power modules into two 61 conductor cables.

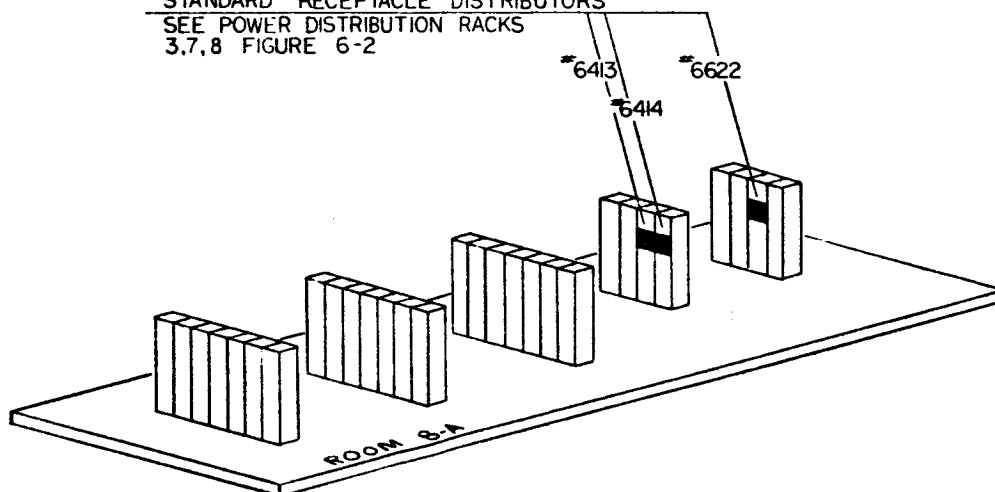
4-95. Locations of Receptacle Distributors. Three type I receptacle distributors are used in the ESE: one each in Power Distribution Racks #6413, #6414, and #6719.

All are in room 8-A of the Mobile Launcher base.

4-96. Design Organization. Standard Receptacle Distributor Type I E75M12316 was designed by MSFC-Astrionics and used by KSC-Launch Equipment Branch.



STANDARD RECEPTACLE DISTRIBUTORS  
 SEE POWER DISTRIBUTION RACKS  
 3.7.8 FIGURE 6-2



[ FOR TYPICAL POWER DISTRIBUTION  
 RACK ASSEMBLY SEE FIGURE 3-51 ]

Figure 4-20 Receptacle Distributor E75M12316



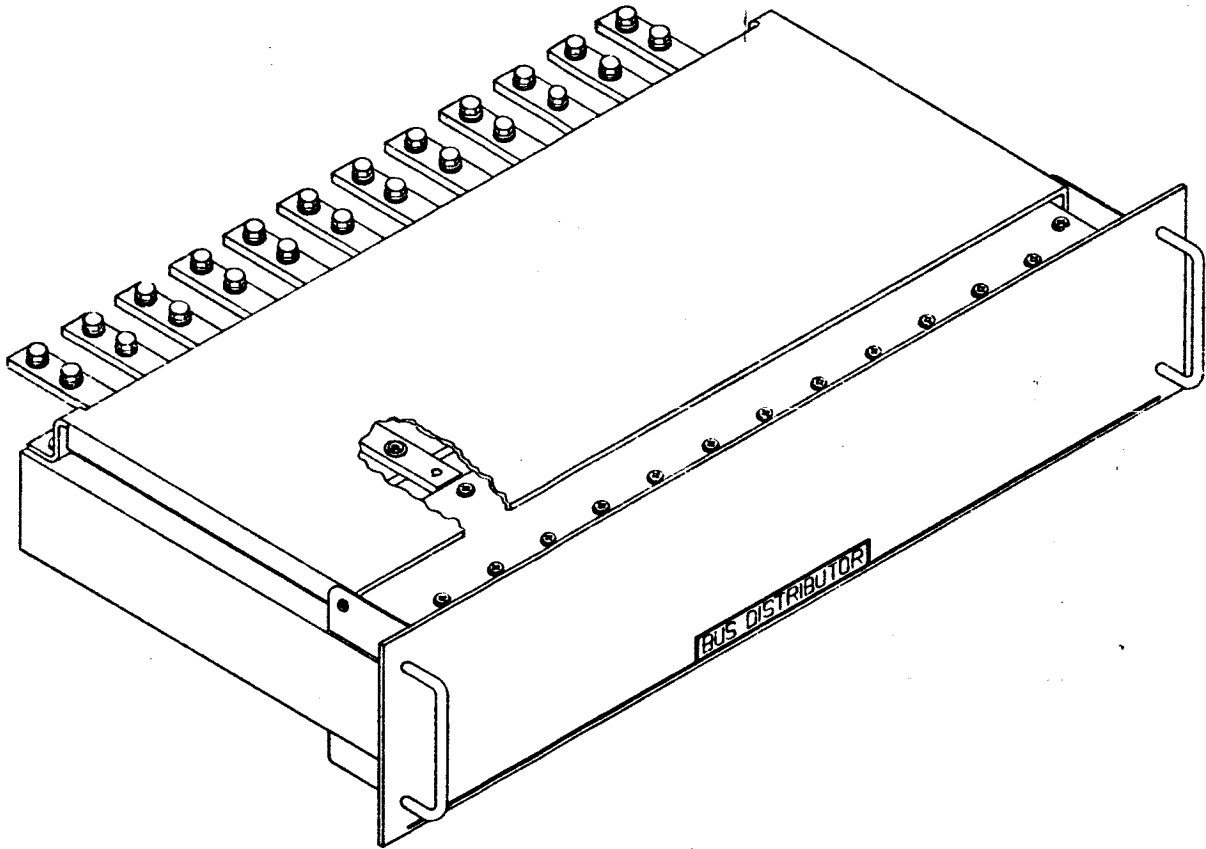
4-97. STANDARD BUS DISTRIBUTOR E75M12323.

4-98. Description of the Bus Distributor. Standard Bus Distributor E75M12323 (figure 4-21) is designed for placement in the standard 24 inch rack. It is 5.25 inches high and 14 inches deep. This distributor contains twelve 0.25 inch x 0.75 inch bus bars, each designed for terminating one #4 conductor. The twelve bus bars are divided into six adjacent pairs, with each pair connected to one 61-pin connector (Bendix PT07SE-24-61S or equal).

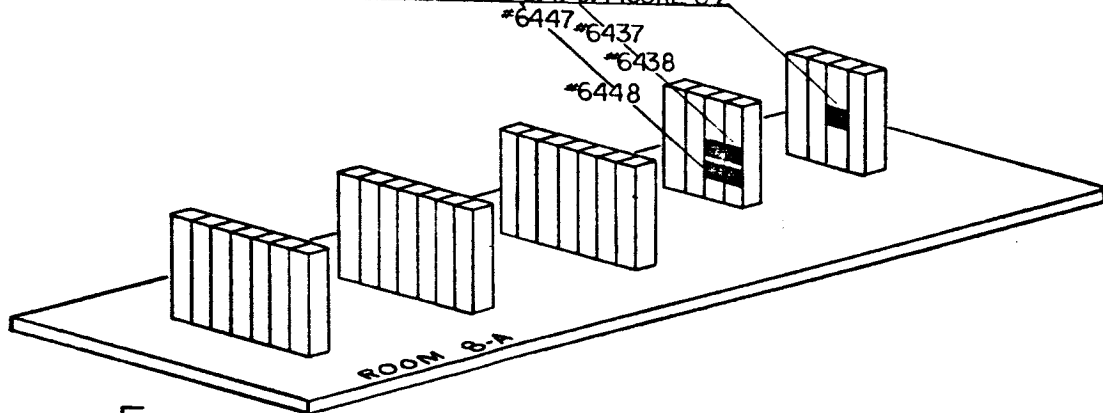
4-99 Use of the Bus Distributor. The Bus Distributor is used in power distribution racks to distribute dc power from individual power modules to individual 61-pin connectors. For ESE use both bus bars of a given pair have the same polarity, so that all pins of any one 61-pin connector are of the same polarity. It is through these connectors that power is distributed to the ESE systems.

4-100. Locations of Bus Distributors. The Bus Distributors are in Power Distribution Racks #6413, #6414, and #6719, located in room 8-A, (see figure 4-21).

4-101. Design Organization. Bus Distributor E75M12323 was designed by MSFC-Astrionics and is used by KSC-Launch Equipment Branch.



STANDARD BUS DISTRIBUTORS  
 SEE POWER DISTRIBUTION RACKS 3, 7, 8, FIGURE 6-2



FOR TYPICAL POWER DISTRIBUTION  
 RACKS ASSEMBLY SEE FIGURE 3-51

Figure 4-21 Standard Bus Distributor E75M12323

4-102. STANDARD CONSOLE E64-KN-F-261-4.

4-103. Description of the Console. Standard Console E64-KN-F-261-4 (figure 4-22) is a radio frequency interference (R. F. I.) shielded enclosure 53 inches high, 26.06 inches wide, and 30.56 inches deep, with a 19 degree sloping front. The console is designed to house 19-inch component panels on its upper front.

The enclosure base, side panels, and cable entry (enclosure bottom) are removable.

4-104. Use of the Console. These consoles house the LCC control panels which provide remote control, test, and monitor functions for the ESE.

4-105. Location of Consoles. The consoles are in the firing rooms, third floor, Launch Control Center.

4-106. Design Organization. Standard Console E64-KN-F-261-4 is commercial equipment purchased to KSC/MSFC specifications.

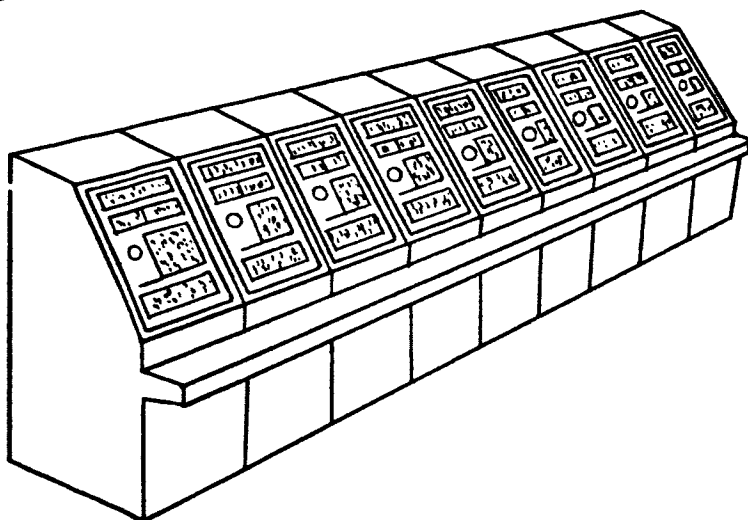
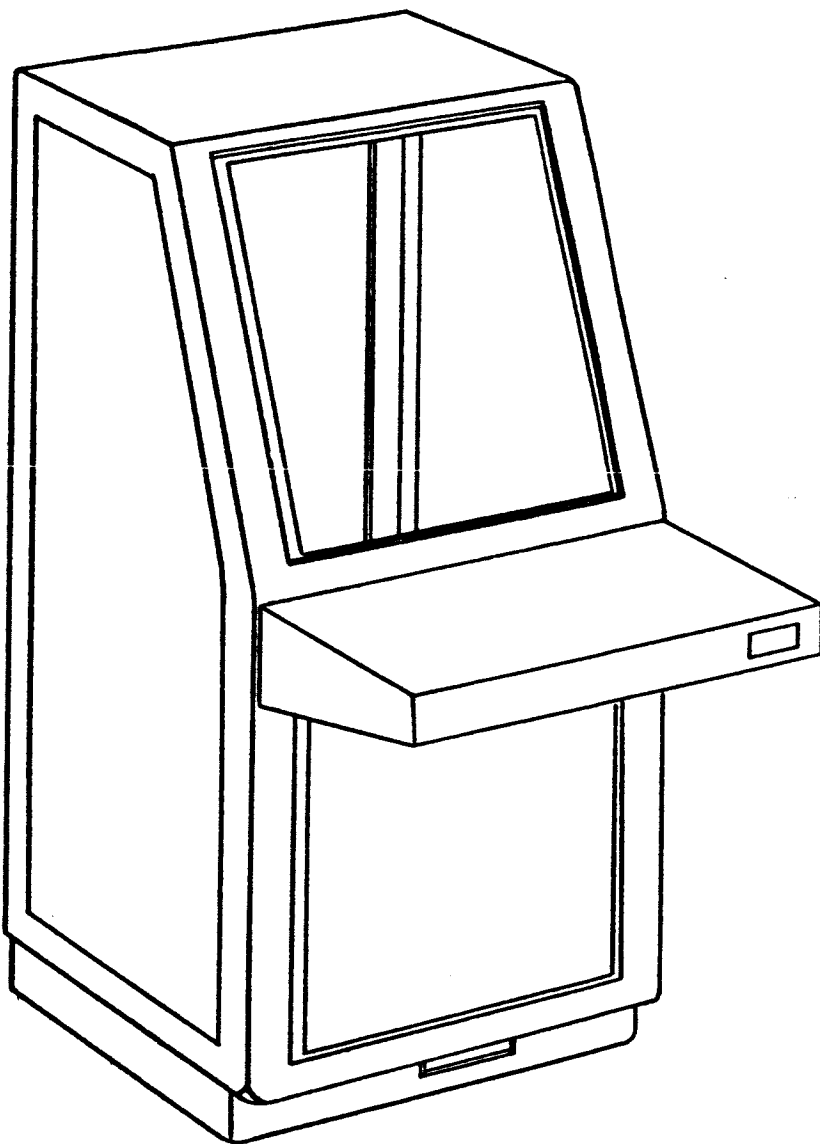


Figure 4-22 Standard Console E64-KN-F-261-4

4-107. INTEGRATION RACK.

4-108. Description of the Integration Rack. The integration rack basically consists of a standard 24-inch vertical rack (Para. 4-17) installed with a standard 54-connector patch distributor (Para. 4-27). Since the base entrance plate to the standard rack can accommodate a maximum of 25 connector entrances, it is necessary to incorporate a side entrance plate to meet the total capacity of the 54-connector distributor.

To provide side cable access, an entrance rack is placed between two integration racks. It will also be used to mount ancillary equipment such as meter panels and communication panels. The entrance rack is a standard 24-inch vertical rack with the side panels removed. (See figure 4-23).

4-109. Use of the Integration Rack. The integration rack is used as an interface between various relay distributors, communication links and control consoles. Its purpose is to gather related signal functions (i. e., commands, analogs, discrete indications, etc.) which emanate from various sources and, by means of patching, route these functions to specific cables which interface with communications hardware. In this way, the total number of interface cables can be reduced by maximum use of available circuits in each cable.

4-110. Locations of Integration Racks. There is an integration rack set located both in room 8-A of the mobile launcher and in each firing room of the launch control center. A set consists of two basic integration distributor racks and one entrance rack.

4-111. Design Organization. The integration rack was designed by the Electrical Section, Launch Equipment Branch.

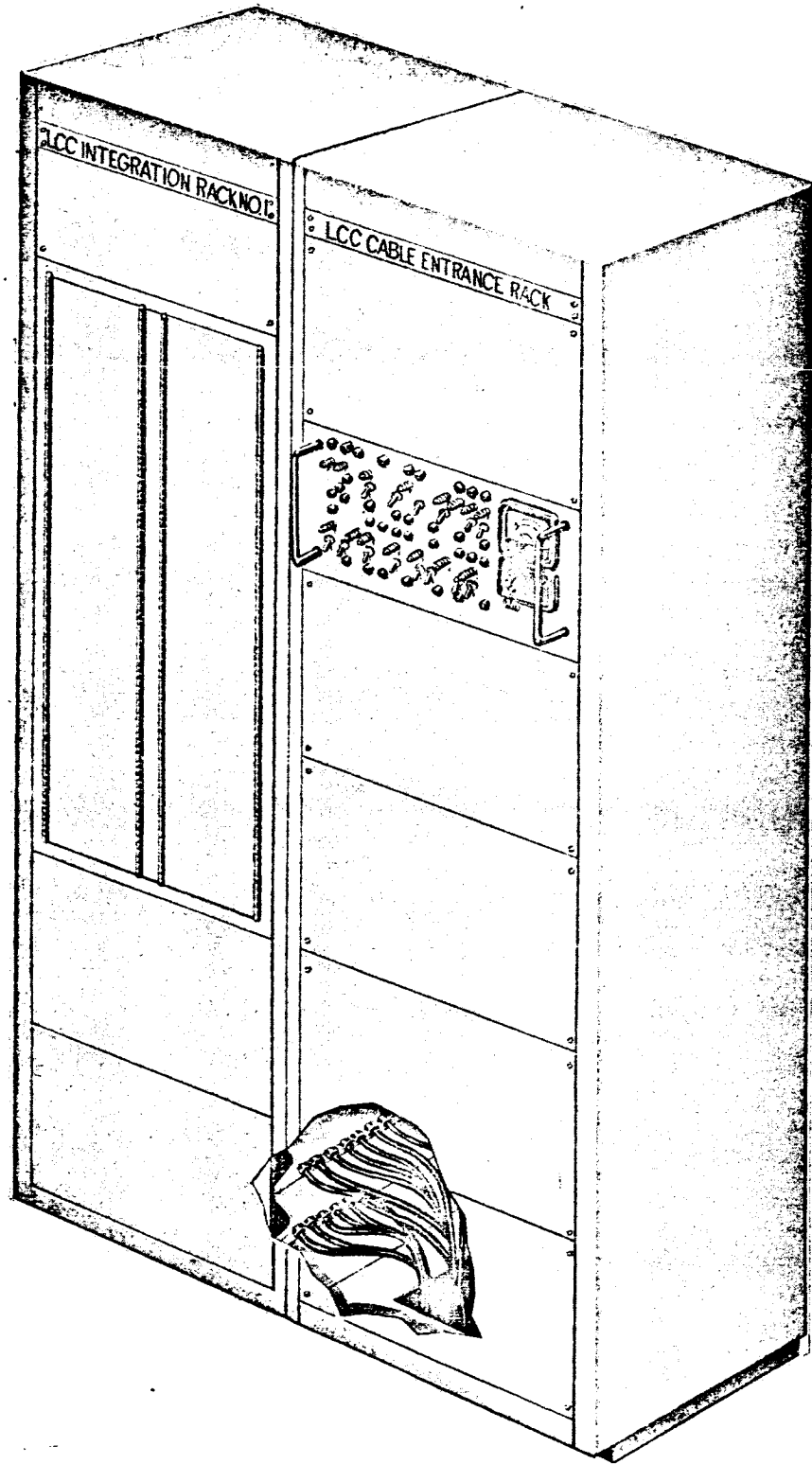


Figure 4-23 Integration and Cable Entrance Racks



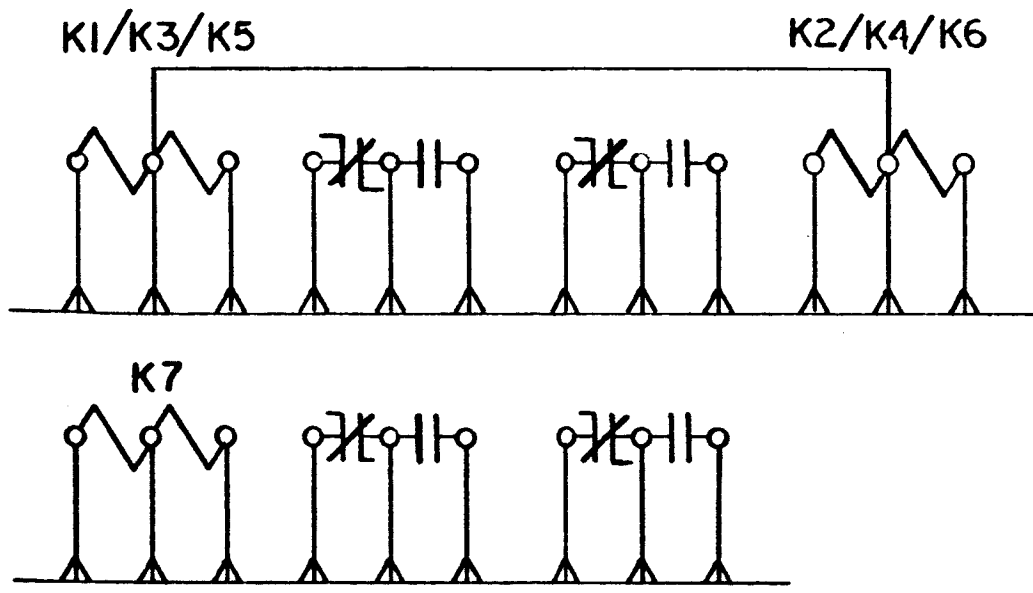
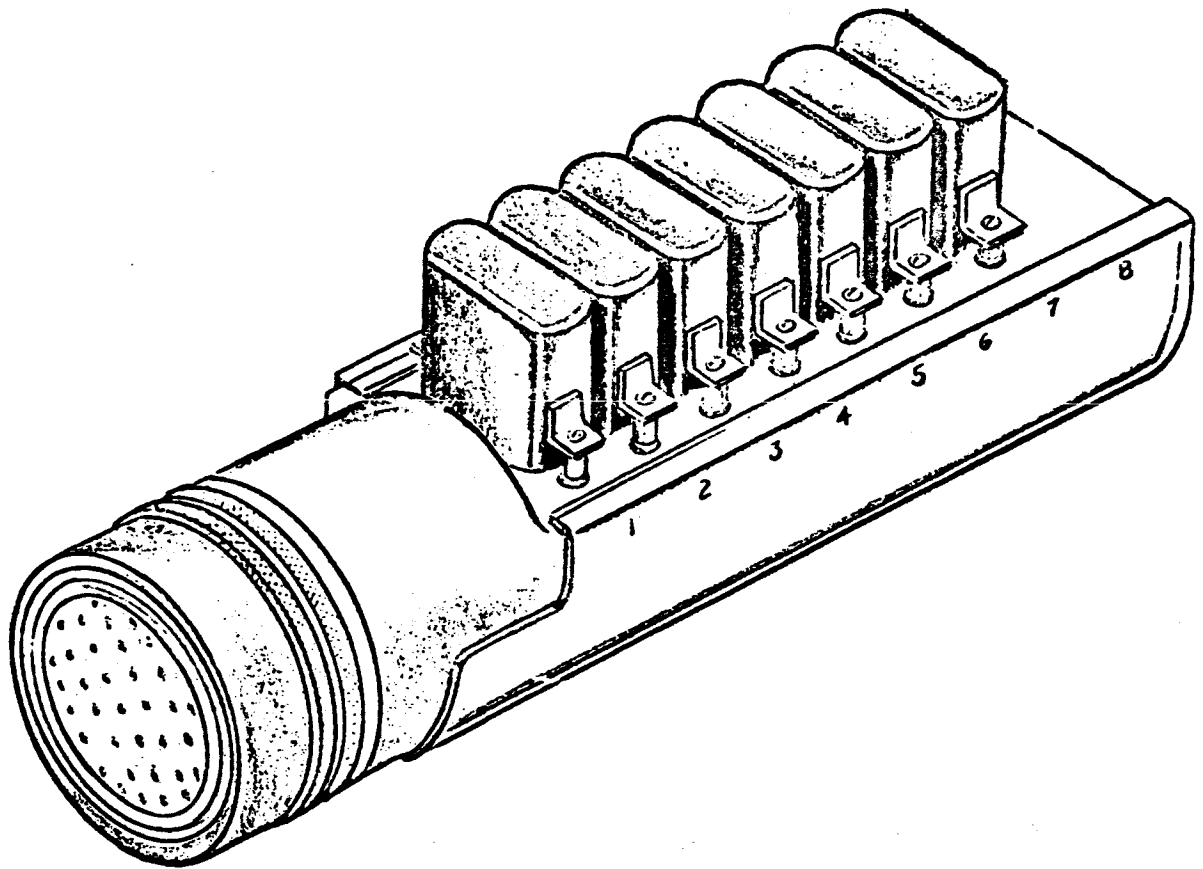


Figure 4-24 Standard Latching Relay Module D75M05895



SECTION V  
TEST EQUIPMENT  
ELECTRICAL SUPPORT EQUIPMENT

5-1. GENERAL.

Section V describes the four basic test units which are used to test or simulate the ESE. Among these units, three are patch-programmable assemblies which satisfy several test functions, and can be readily programmed for testing other similar systems.

Also included in this section is a description of the swing arm and tail service mast qualification test site at MSFC, where certain of the test equipments will be first employed.

5-2. LAUNCHER GROUND EQUIPMENT TEST SET E75M13286.

5-3. Description of the Test Set. The Launcher Ground Equipment Test Set (GETS) (figure 5-1) is a roll-around portable test console approximately 57 inches high, 53 inches wide, and 28 inches deep. It contains:

- a. 1 standard 60-connector patch distributor
- b. 1 standard 40-connector patch distributor
- c. 250 indicator lamp modules (JAY-EL P/N10182 or equal)
- d. 90 meters Triplett model .5E or equal
- e. 90 analog potentiometers
- f. 300 switches (Cutler Hammer 8867K1 or equal)
- g. 2 standard fuse panels
- h. 1 standard communication panel
- i. 1 power monitor panel

This test set has provision for 20 function-cable connections and four power-cable connections. These cables enter the console through a hinged door in the console base. Two doors on the front of the console provide access for connecting cables, diode modules, and relay modules to the patch distributor.

Two drawers with writing surfaces are provided for the operator.

5-4. Use of the Launcher GETS. For the purpose of checkout and test of the ESE subsystems, two Launcher GETS are employed as in figure 5-3. One is programmed as a control and monitor unit which generates all signals normally provided to the ESE subsystems from the LCC. At the same time, it simultaneously displays all discrete and analog signals arising in the particular subsystem under test. See figures 5-4 and 5-5. As a control and monitor unit, the Launcher GETS uses up to 10 diode modules (paragraph 4-42) in conjunction with the components listed above.

The other Launcher GETS is connected to system cables coming from the LCC. It is thus a simulator, and is programmed to simulate the electrical-mechanical control and monitor components within a given ESE subsystem. These are mainly solenoids, pressure transducers, and pressure switches in the service arms, tail service masts, and Mobile Launcher base. Programmed as a

LEFT BAY FRONT PANEL

RIGHT BAY FRONT PANEL

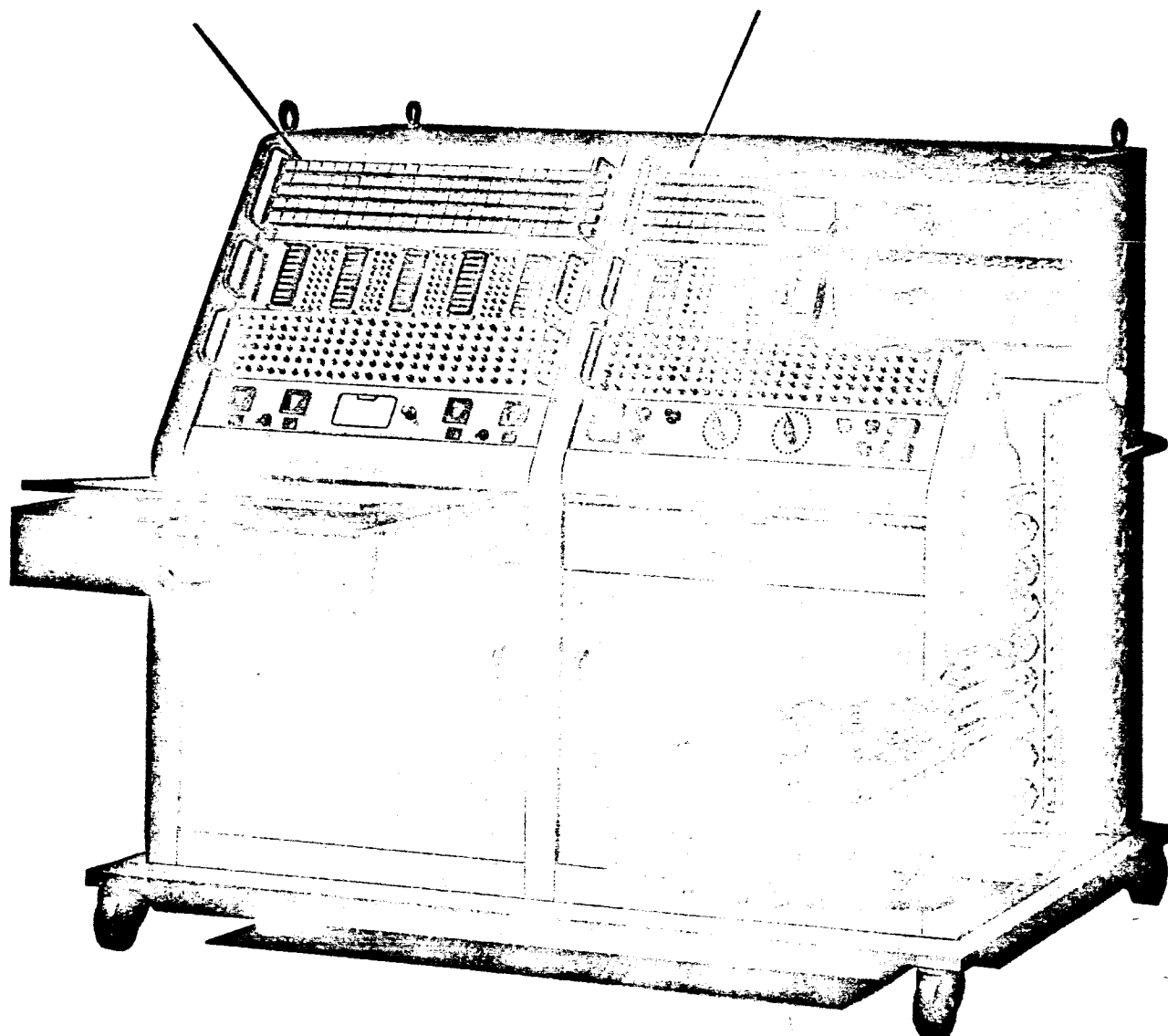


Figure 5-1 Ground Equipment Test Set E75M13286

simulator the Launcher GETS uses up to 36 relay modules (paragraph 4-37) along with its other test components.

Both the GETS control unit and the simulator unit interface with the ESE subsystems at the relay racks in room 8-A. The control unit is connected to the relay racks in place of systems cables from the LCC; the monitor unit is connected to the LCC cables themselves.

For these purposes, pre-installed test cables from the relay racks are routed beneath the deck to a deck plate in room 8-A. During tests the GETS is placed near the deck plate and connected to appropriate test cables.

5-5. Location of the Launcher GETS. The Launcher GETS is used in room 8-A.

5-6. Design Organization. The Launcher GETS was designed by the Electrical Section, Launch Equipment Branch.

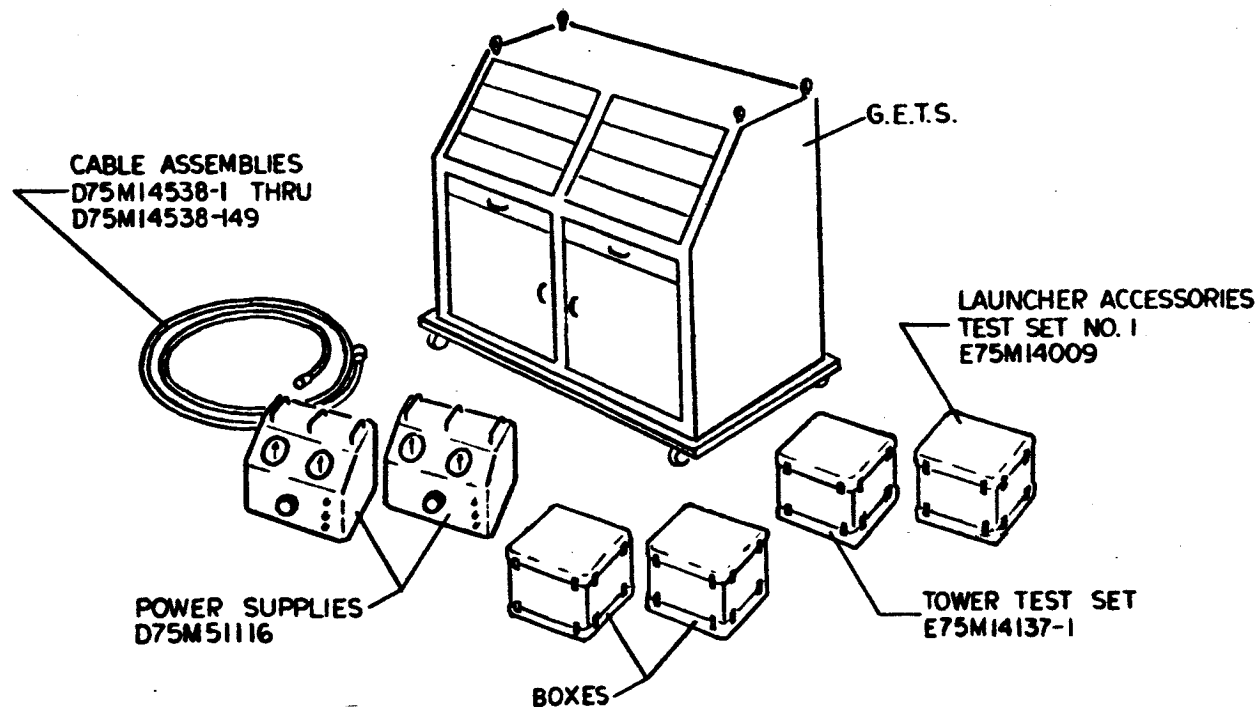


Figure 5-2 GETS Auxiliary Equipment

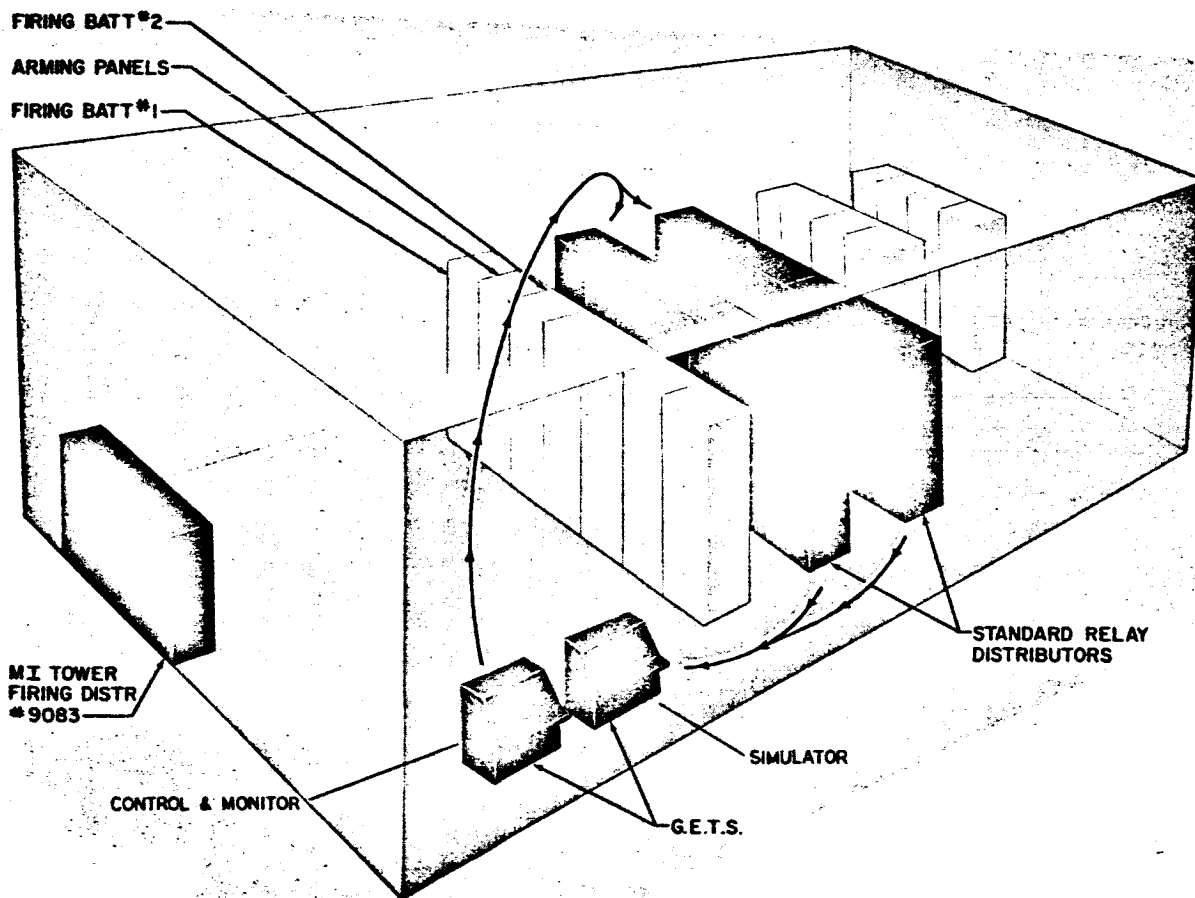
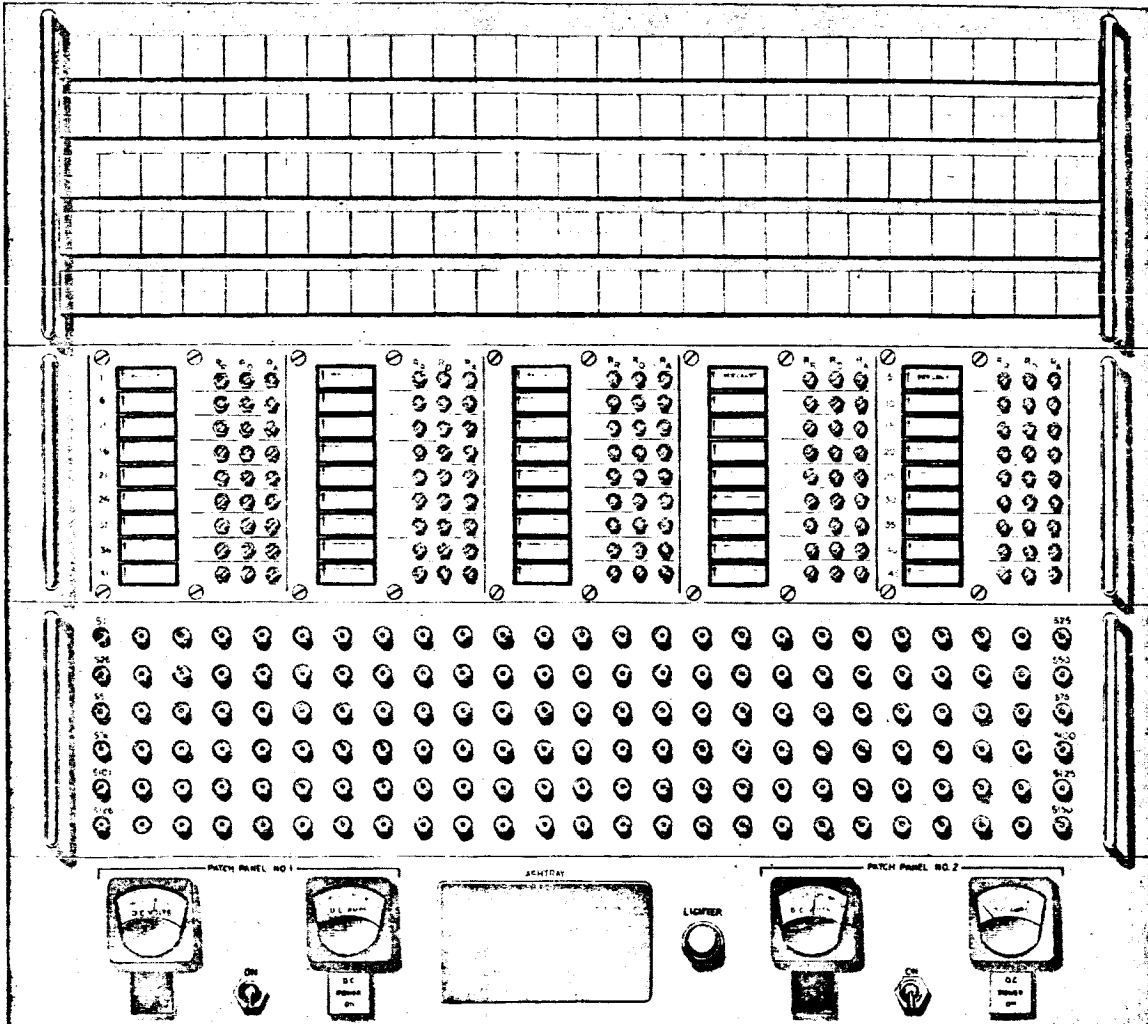
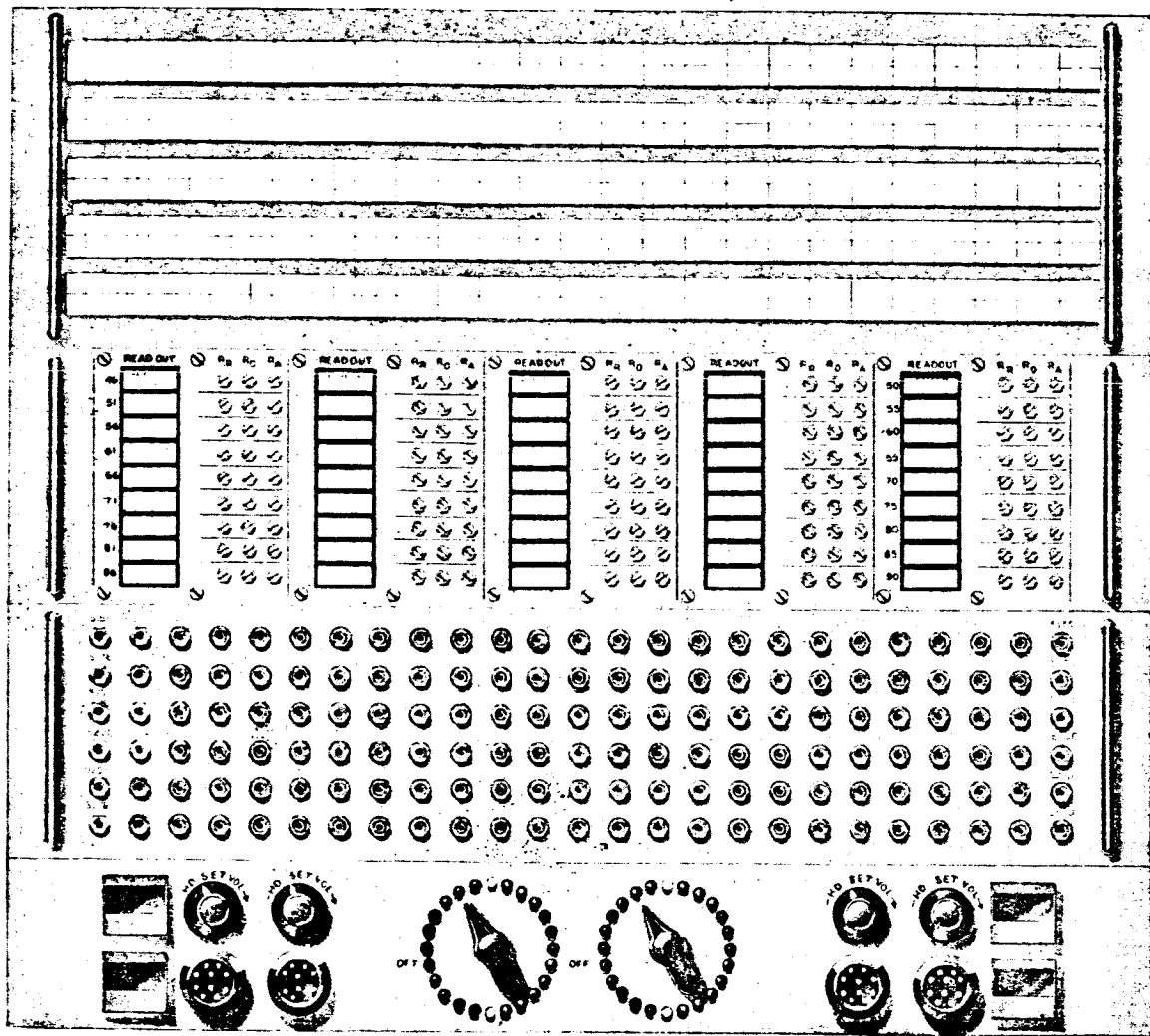


Figure 5-3 GETS Control, Monitor and Simulation



1	LIGHT PANEL ASSY.	E75MI3292
2	ANALOG PANEL ASSY.	E75MI3294
3	SWITCH PANEL ASSY.	E75MI3296
4	POWER MONITOR & CONTROL PANEL ASSY.	E75MI3298

Figure 5-4 Left Bay Front Panel, Launcher GETS



1	LIGHT PANEL ASSY.	E75MI3293
2	ANALOG PANEL ASSY.	E75MI3295
3	SWITCH PANEL ASSY.	E75MI3297
4	COMMUNICATIONS PANEL ASSY	E75MI3299

Figure 5-5 Right Bay Front Panel, Launcher GETS

5-7. MINIATURIZED UNIVERSAL TEST SET (MUTS) E75M13761.

5-8. Description of the MUTS. The Miniaturized Universal Test Set (figure 5-6) is a portable test unit 8 inches high x 13.5 inches wide x 11.5 inches deep, and weighs approximately 35 pounds. It contains these components:

- |                        |  |
|------------------------|--|
| a. 3 meters            | Weston Model 3911, 50 microamperes or equal                      |
| b. 24 Twist-Lites      | Series 10/E; horizontally split display DPDT switch combinations |
| c. 16 relay sockets    | Burndy MTR-92 or equal   |
| d. 19 taper pin blocks | AMP 53 series, 30 pins or equal                                  |
| e. 16 relays           | MSFC-SPEC-339/53   |

The program of the MUTS is carried to the controlled unit by one or two 61-conductor cables using a pigmy connector (Bendix PT07P-24-61S or equal).

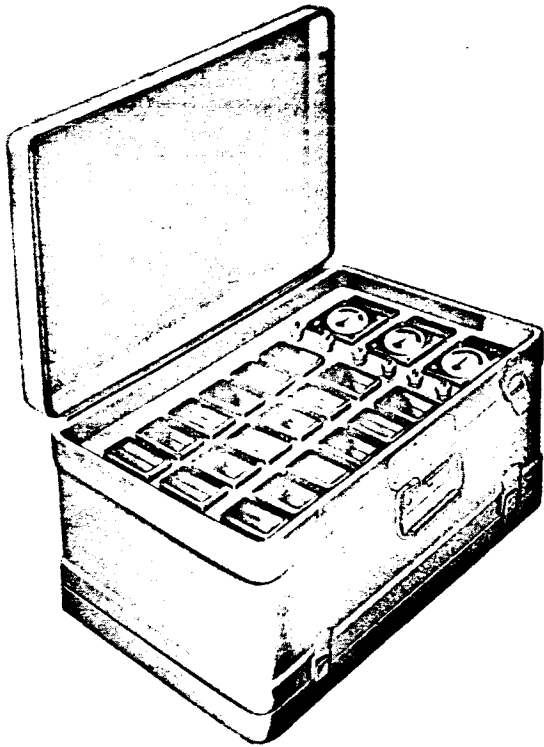
Input 28 volts dc is supplied through one 3-pin connector (Cannon BFR22-2P or equal). For self-checks, the MUTS contains a transistorized ground indicator circuit, a POWER AVAILABLE indicator, and a POWER ON indicator.

5-9. Use of the MUTS. The MUTS can be patched to check pneumatic pressures and the operation or position status of valves and relays, and in addition can be hand carried about the Mobile Launcher by test personnel. It is presently used as the Tower Test Set for the Q-Ball Cover Removal Subsystem and for Valve Panel #12.

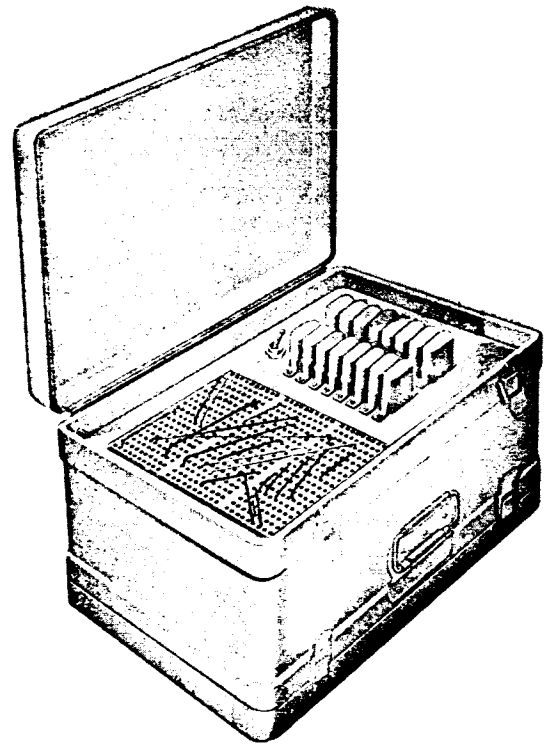
5-10. Locations of MUTS. The MUTS tests the Q-Ball cover mechanism on level 360 (figure 3-43), and Valve Panel 12 on level 160 (figure 3-40).

5-11. Design Organization. The MUTS was designed by the Electrical Section, Launch Equipment Branch.





TOP VIEW



BOTTOM VIEW

Figure 5-6 Miniaturized Universal Test Set E75M13761

5-12. PORTABLE ARM CONTROL CONSOLE J75M07542.

5-13. Description of the Console. The Portable Arm Control Console (figure 5-7) is a roll-around portable test console approximately 50 inches high, 45 inches wide, and 28 inches deep. It contains its own dc power supply, and so can use either 28 volts dc or 115 volts ac input. The console's arm control panel has the following components:

- a. 103 indicator lamps                      Lamps Control of America (Series L3000) or equal
- b. 1 meter                                      Minneapolis-Honeywell HS2F, 0-50 ua. or equal
- c. 39 switches                                Cutler Hammer 8833-K4 & 8821-K6 or equal

Its charging control panel has these components:

- a. 108 indicator lamps                      Lamps Control of America (series L3000) or equal
- b. 54 switches                                Cutler Hammer 8825-K6 or equal

The console also contains a 27-connector patch distributor (paragraph 4-32), plus the following connectors for power and control-function connections:

- a. 24 Cannon Connectors                      TBFR 16S-1PS or equal
- b. 7 Cannon Connectors                      TBFR 24-10PS or equal
- c. 1 Cannon Connector                      TBFR 22-2PS or equal
- d. 5 Cannon Connectors                      TBFR 22-22PS or equal
- e. 1 Bendix Connector                      10-248454-75PS or equal
- f. 6 Bendix Connectors                      10-248454-73PS or equal

5-14. Use of the Console. The Portable Arm Control Console may be programmed to control, test, and monitor the Apollo access arm and any of the eight service arms. Test programs are altered by changing patchboards in its patch distributor. The console is positioned on the tower level associated with the arm under test, and its test cables are connected to distributors and electro-mechanical components as required.

5-15. Locations of Console. The Portable Arm Control Console is used on tower levels 60, 120, 140, 160, 200, 220, 260, and 300.

5-16. Design Organization. The Portable Arm Control Console was designed by the Electrical Section, Launch Equipment Branch.

CHARGING CONTROL PANEL  
J75M07546-1

ARM CONTROL PANEL  
J75M07546-3

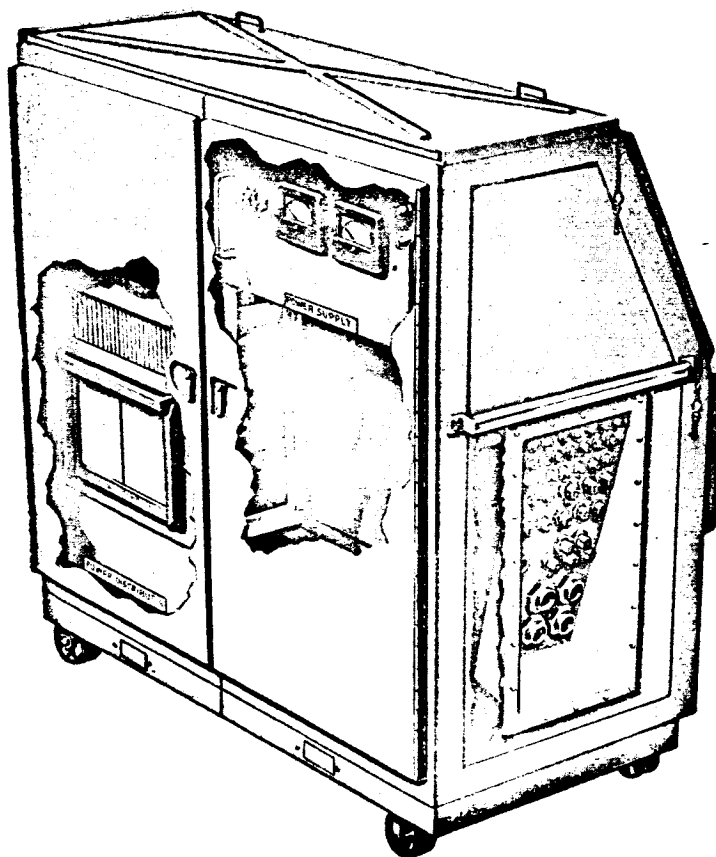
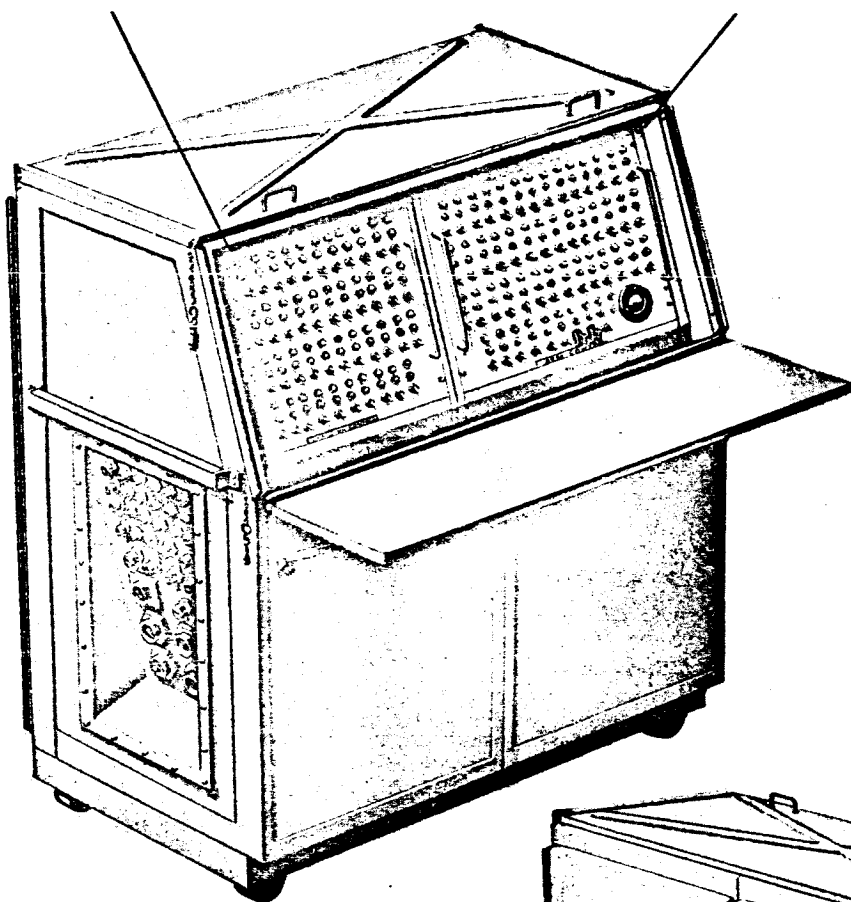


Figure 5-7 Portable Arm Control Console J75M07542-1

5-17. UNIVERSAL PATCHBOARD TEST SET E75M09032.

5-18. Description of the Test Set. The Universal Patchboard Test Set (figure 5-8) E75M09032 is a portable test unit approximately 16 inches x 16 inches x 14.9 inches high, weighing about 50 pounds. It contains these test components:

- |                       |  |
|-----------------------|--|
| a. 6 Meters           | Weston Model 3911, 50 microampere dc or equal  |
| b. 54 Lights          | Dialco Socket 101-8430W-975; lamps ASA 327 or equal                                  |
| c. 27 Switches        | Cutler-Hammer 8867K6 or equal  |
| d. 28 Relay Sockets   | Burndy MTR-92 or equal   |
| e. 1 Patchboard Assy. | AMP P816D, 695081-3 Frame & Spring or equal<br>AMP 595005-2 Removable Board or equal |

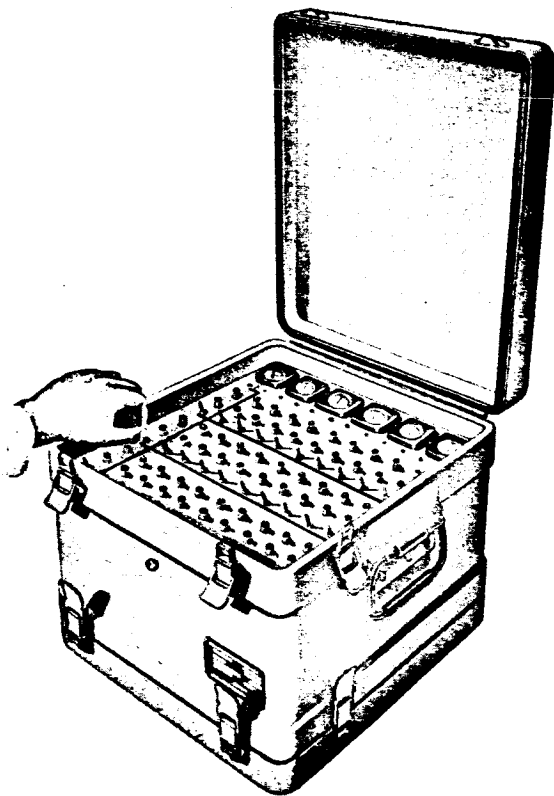
The program from the Universal Patchboard Test Set is carried to the controlled unit by one or two 61-conductor cables using pygmy connectors (Bendix PT07P-24-61S or equal). Input 28 volts dc is supplied through one 3-pin connector (Cannon BFR22-2P or equal). For self-checks, the test set contains a ground indicator circuit, a lamp filament test switch, POWER ON and POWER AVAILABLE indicators, and a test socket and switch for individual lamp tests.

5-19. Use of the Test Set. The Universal Patchboard Test Set can be programmed to measure pneumatic pressures and check the operation or position status of various valves and relays. It can be hand carried about the Mobile Launcher by test personnel. At present, the test set has three separate programmings:

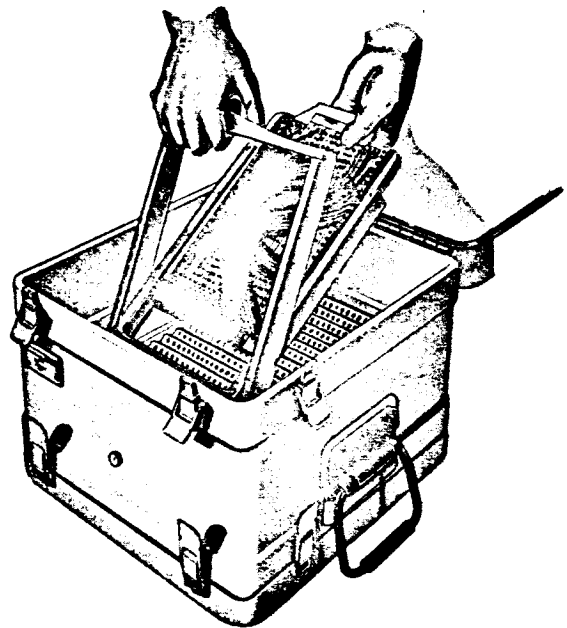
- a. Tail Service Mast Test Set, figures 3-33 and 3-34
- b. Launcher Accessories Test Set No. 1, figures 3-37 and 3-38
- c. Launcher Accessories Test Set No. 2, figures 3-22 and 3-26

5-20. Locations of Test Set. The Universal Patchboard Test Set is used on levels 0, A, and B.

5-21. Design Organization. The Universal Patchboard Test Set was designed by the Electrical Section, Launch Equipment Branch.



TOP VIEW



BOTTOM VIEW

Figure 5-8 Universal Patchboard Test Set E75M09032

5-22. MSFC SERVICE ARM/TAIL SERVICE MAST TEST SITE.

5-23. Purpose of the Test Site. The MSFC Test Site is used to functionally test and qualify for operational use each service arm and tail service mast in conjunction with its associated control equipment. The tests simulate launch conditions and are performed on the equipment prior to delivery to KSC.

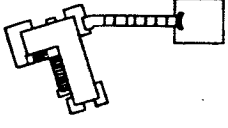
5-24. Equipment for the Test Site. The Test Site at Marshall Space Flight Center (figure 5-9) is a test facility comprised of ten test stands, a tail service mast pad, and a block house.

5-25. Service Arm Test Stands. There are nine swing arm test stands, four of which contain preflight vehicle simulators and five of which contain vehicle liftoff simulators. All swing arm test stands contain power and signal distributors, mounting facilities for equipment tested and closed-circuit television for visual monitoring of tests. Four of the test stands also share cryogenic handling equipment and nine stands share an engine exhaust pressure simulator.

On each swing arm test stand there is one service arm which is to be functionally tested before it is installed on a Mobile Launcher at LC-39. Also available for test are the swing arm related control cabinets No.1 and No. 2, and hydraulic/pneumatic actuators. This equipment will be provided with the swing arm on delivery to LC-39.

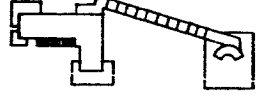
Certain electrical distributors used on the Mobile Launcher for service arm control are facility end items; on the test stands they are replaced by special permanently installed distributors. Hence, each Umbilical Instrumentation and Control Distributor is replaced by a test site Tower Terminal Box. Each Mineral Insulated Firing Distributor is replaced by a special test site Firing Distributor. In addition, a test site Vehicle Receptacle Box on each inflight test stand connects the liftoff switches to the arm control system.

SUPPORT  
BLDG NO.5  
4642



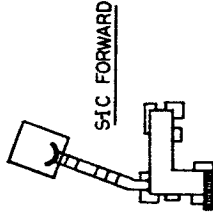
S-II AFT

COMMAND  
MODULE



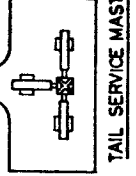
SERVICE  
MODULE

PUMP  
HOUSE  
4645

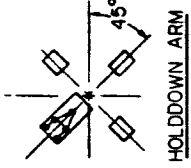


S1C  
FORWARD

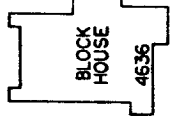
SUPPORT  
BLDG NO.1  
4638



TAIL  
SERVICE  
MAST

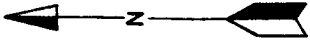


HOLDDOWN  
ARM

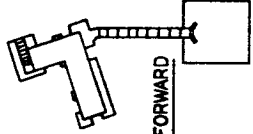


BLOCK  
HOUSE  
4636

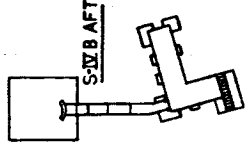
SUPPORT  
BLDG NO.2  
4639



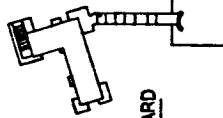
SUPPORT  
BLDG NO.4  
4641



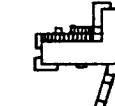
S1VB  
FORWARD



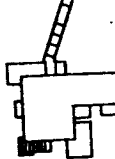
S1VB  
AFT



S-II  
FORWARD



S1I  
INTERMEDIATE



S1C  
INTERTANK

SUPPORT  
BLDG NO.3  
4640

Figure 5-9 MSFC Service Arm/Tail Service Mast Test Site

Some of the electrical equipment needed for control and monitoring of the service arm is shown in figure 5-11. Equipment supplied by the Electrical Section, Launch Equipment Branch, includes:

- a. Blockhouse control rack assemblies (4)
- b. Blockhouse distribution rack assemblies (5)
- c. Firing distributors (5)
- d. Tower terminal boxes (9)
- e. Vehicle receptacle boxes (5)
- f. Integration rack (1)

5-26. Electrical Control for the Service Arms. Primary control and monitoring of the service arm under test is conducted from the second floor of the blockhouse. Four control racks (figures 5-11 and 5-12) provide test functions shown on the typical control rack overlay (figure 5-10). There is one overlay for each of the nine arms, one for the liftoff switches, and one for the hydraulic charging unit and holddown arm control.

The control panel overlay shows that the electrical control system can be programmed to allow the swing arm to perform as a complete, automatic system. Alternately, many discrete mechanical functions can be tested separately.



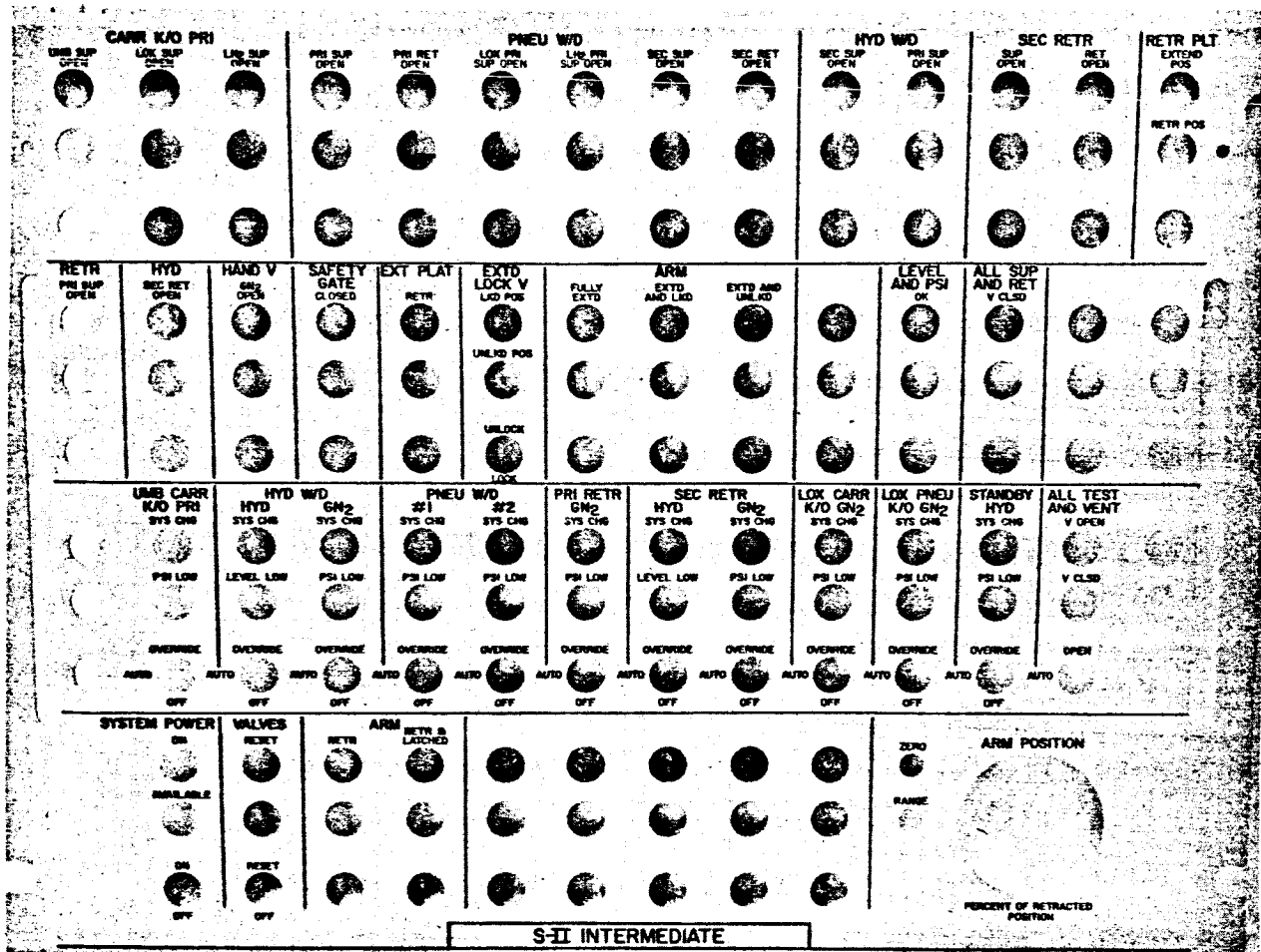


Figure 5-10 Typical Control Panel, Service Arms Test Site

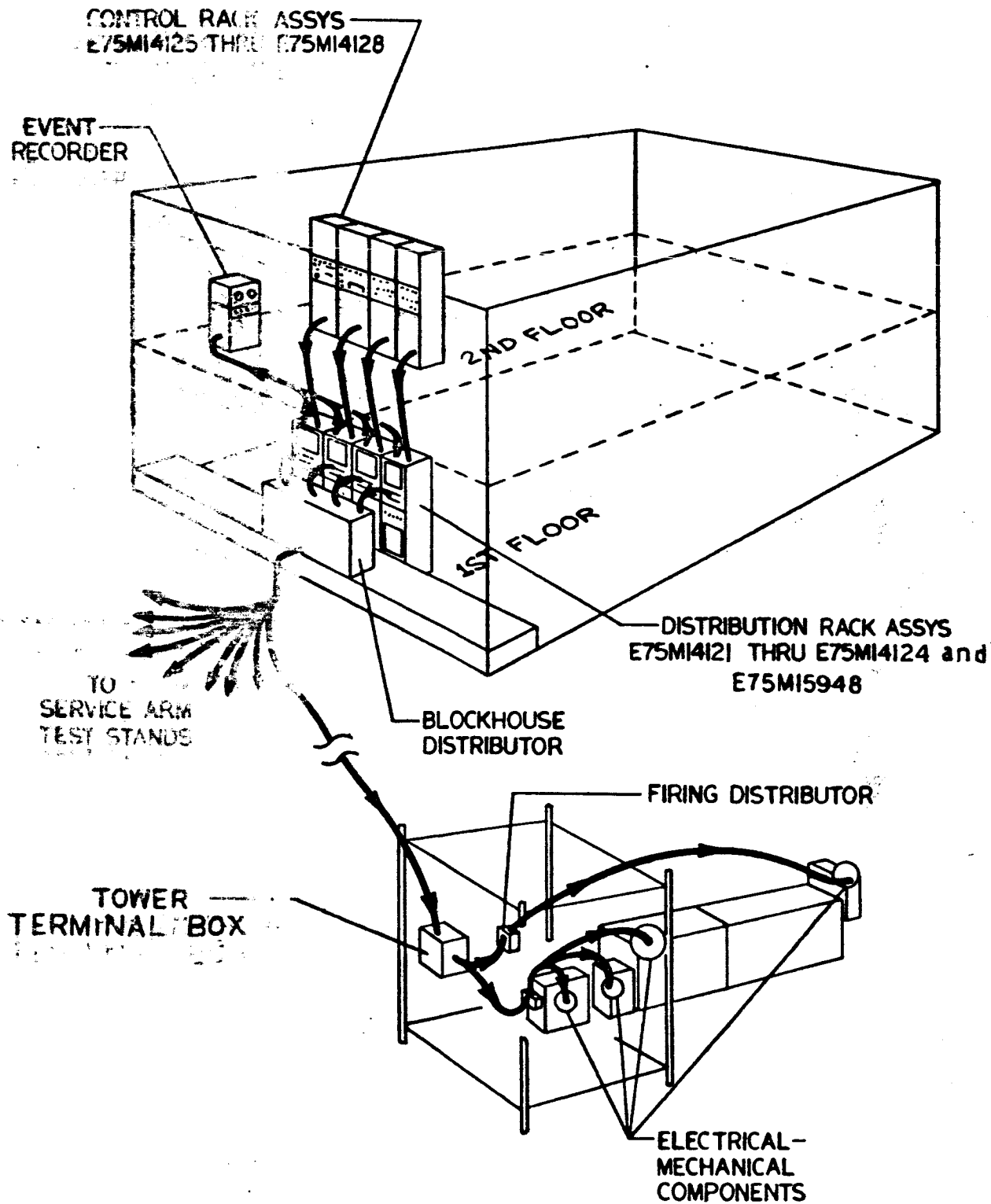


Figure 5-11 ESE Equipment, Service Arms Test Site

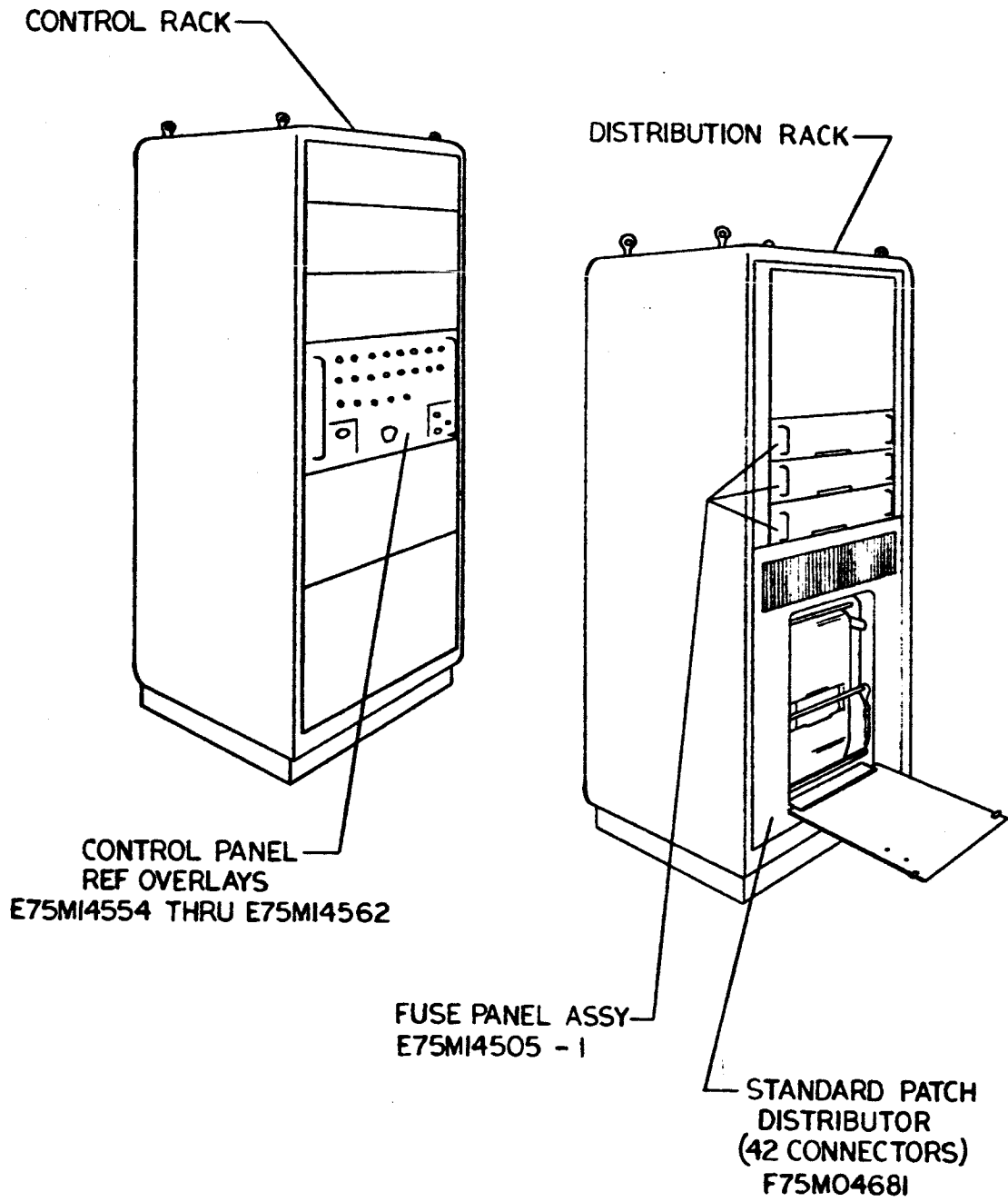


Figure 5-12 Control and Distribution Racks, Service Arms Test Site

5-27. Tail Service Masts Test Pad. The tail service masts (TSM) test pad (figure 5-13) has one to three tail service masts, hydraulic and pneumatic systems for the masts, a  $\text{GN}_2$  storage tank, a vehicle liftoff simulator, and a control room separate from the swing arm control blockhouse.

Testing of the tail service mast is controlled primarily from the Firing Monitor and Test Panels in the control room. These panels are functionally similar to the TSM control panels in the Launch Control Center at LC-39. They provide control and monitor functions for that portion of the tail service mast system which is furnished by the Electrical Section, Launch Equipment Branch. Other equipment provided by the Electrical Section includes:

- a. Tail Service Mast 1-2 #6006 (Internal Cables)
- b. TSM 1-2 Control Distributor #6009
- c. Tail Service Mast 3-2 #6007 (Internal Cables)
- d. TSM 3-2 Control Distributor #6010
- e. Tail Service Mast 3-4 #6005 (Internal Cables)
- f. TSM 3-4 Control Distributor #6008
- g. Tail Service Mast Distribution Rack #T6650
- h. Firing Monitor and Test Panel 1-2
- i. Firing Monitor and Test Panel 3-2
- j. Firing Monitor and Test Panel 3-4

Control power (28 volts dc) and an event recorder are furnished by the MSFC test facility.

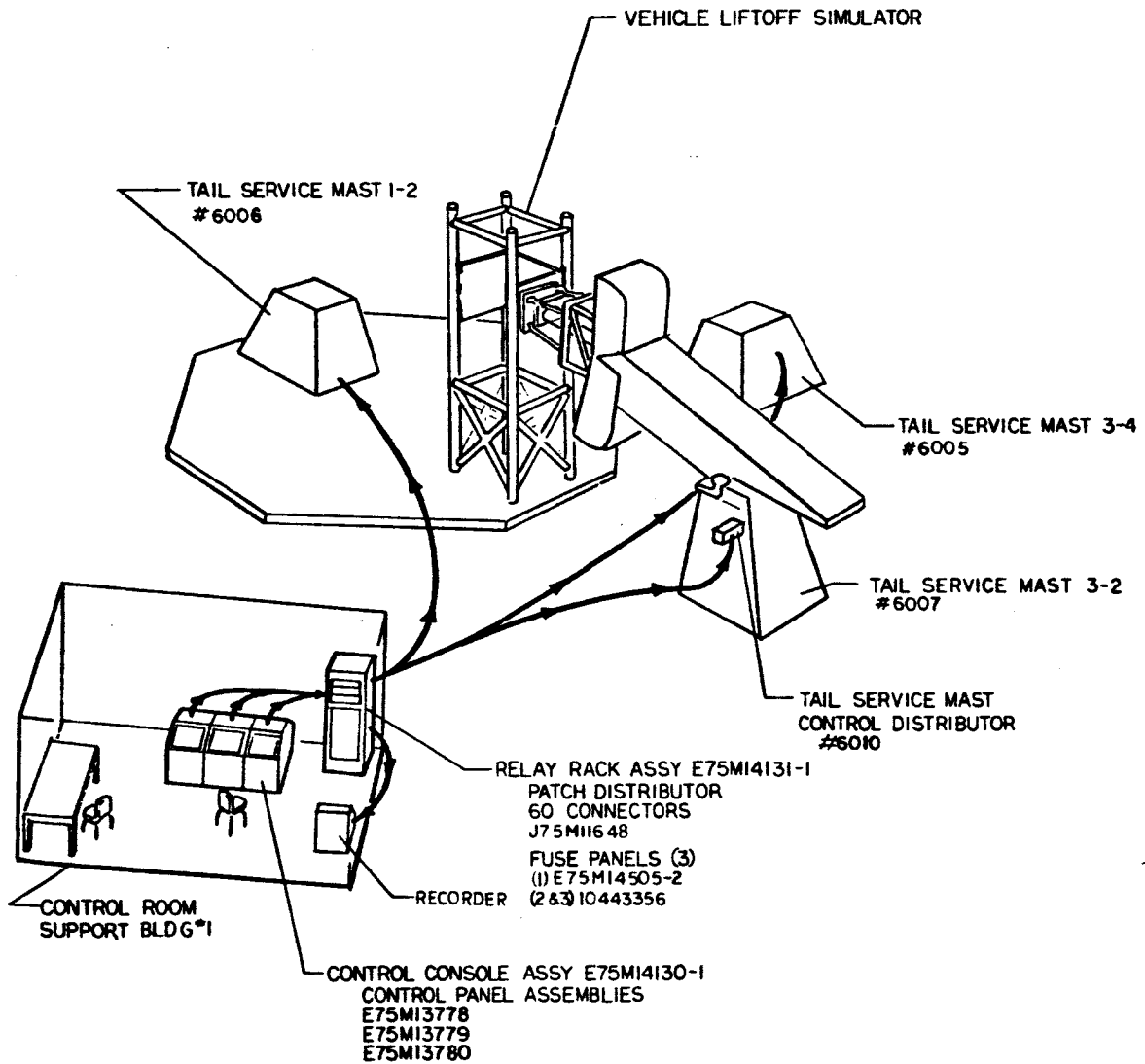


Figure 5-13 Tail Service Masts Test Site, MSFC

5-28. Electrical Control for the Tail Service Masts. Before a tail service mast is tested, pneumatic and hydraulic accumulators for its actuators are charged and vented, and appropriate lines are bled. Some of these preparations require adjusting various pneumatic and hydraulic valves by hand. Other valves in the system are actuated by solenoids, and are therefore electrically controlled and monitored from the Firing Monitor and Test Panel, figure 5-14. When the monitor lamps and meters on this panel indicate that all conditions are ready, the mast extension operation can be started.

By the use of the MAST LOCK RELEASE and MAST EXTEND switches on the tail service mast, an operator now gradually extends the mast toward the vehicle liftoff simulator. He is in communication with a control room technician who monitors lamps and meters on the Firing Monitor and Test Panel. Final connection of the mast's umbilical carrier to the vehicle liftoff simulator is completed manually.

The mast retract test is then conducted from the Firing Monitor and Test Panel. Closure of the MAST RETRACT PRESET and MAST RETRACT ACTUATE switches retracts any one mast or any combination of the three, as required. Switch closure provides a 28 volt dc signal to relay rack #T6650, which in turn performs all necessary relay logic and transmits control signals to appropriate components in the mast. All such events are recorded by instrumentation equipment.

Alternately, the mast can be armed for retract and actuated by motion of the vehicle liftoff simulator. The simulator itself is MSFC test facility equipment, and is not controlled by the KSC Firing Monitor and Test Panel.

As the mast retracts, all major movements of the mast and its actuators are detected by transducers, limit switches, and potentiometers. Signals from these components return to relay distributor #T6650 and subsequently to lamps on the control panel. All such events are also recorded by instrumentation equipment.

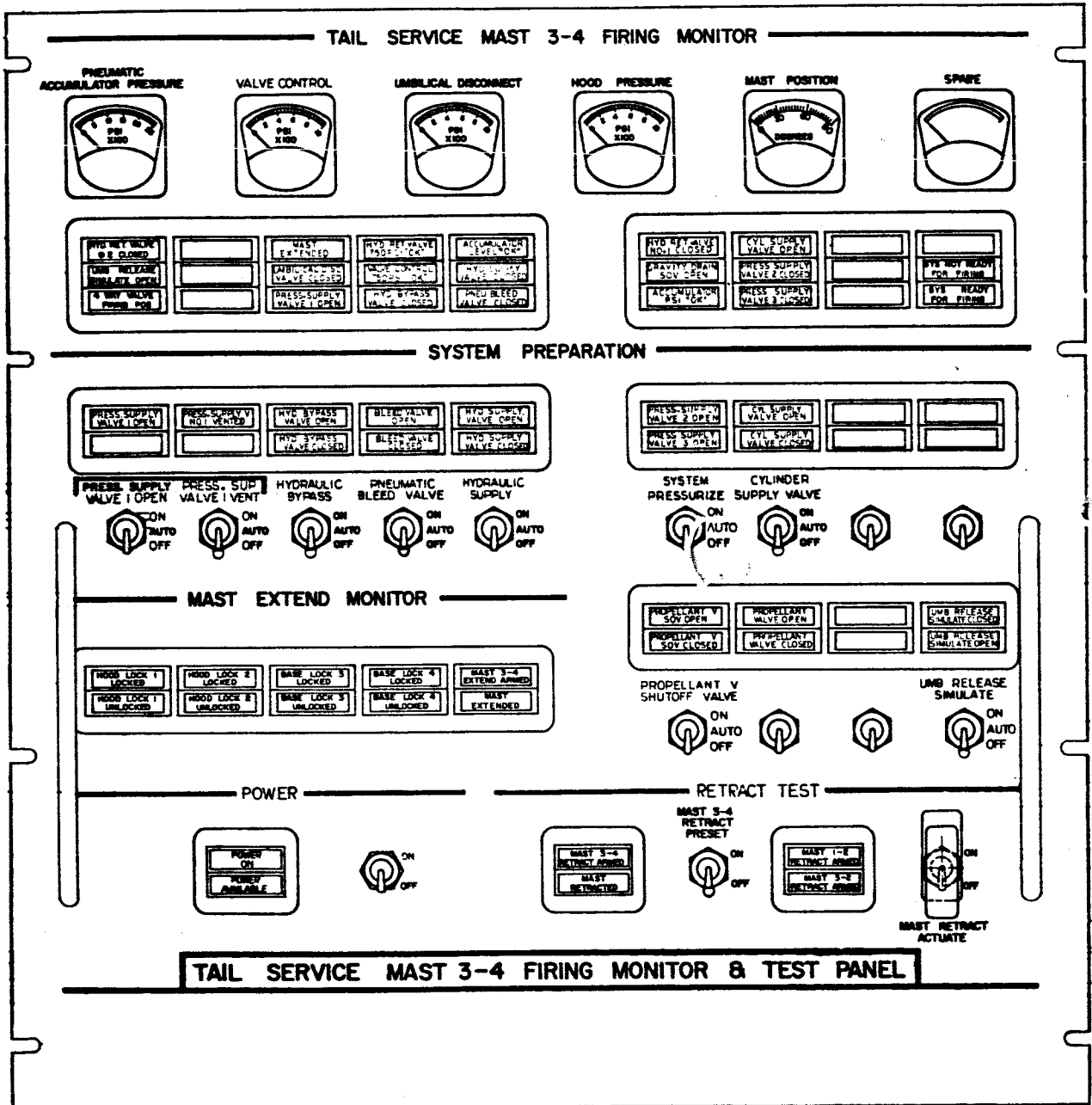


Figure 5-14 TSM Test Site Control Panel E75M13780

SECTION VI  
REFERENCE INFORMATION  
ELECTRICAL SUPPORT EQUIPMENT

6-1. ESE FAMILY TREE OF DRAWINGS.

The ESE Family Tree of Drawings illustrates, by generation breakdown, that Electrical Launch Support Equipment (ELSE) which is the responsibility of the Launch Equipment Branch, EDV-15. The tree contains the following branches:

LC-39 Ground Support Equipment	J75M05970
Off LUT Installation LUT-1	J75M05971
Launcher Installation LUT-1	J75M05972
Launcher Deck Installation LUT-1	J75M05973
Tower Supply Installation LUT-1	J75M05974
Tower Platforms Installation LUT-1	J75M05975
LC-39 GSE Test Equipment	—————
F-1 Engine Servicing Equipment	J75M13482

The ESE Family Tree of Drawings may be found in Volume II of this ESE Electrical Reference Handbook.



6-2. REFERENCE SPECIFICATIONS.

The documents listed in Table 6-1 will be used by contractors for Launch Equipment Branch fabrication and installation. Table 6-2 contains substitutions and deletions of specifications and procedures. Abbreviations for cable specifications are listed in Table 6-3.

Table 6-1 REFERENCE SPECIFICATIONS

DOCUMENT NUMBER	DESCRIPTION
<b>KENNEDY SPACE CENTER DOCUMENTS</b>	
KSC-S-101	Semiconductor Devices, General Specification
KSC-S-101/1	Diode, Zener, Silicon, Power, Type K1N2970B
KSC-S-101/2	Transistor, NPN Switching, Type K2N914
KSC-S-101/3	Transistor, NPN, General Purpose, Type K2N2192A
KSC-S-101/4	Transistor, PNP Switching, Type K2N2905A
KSC-S-101/5	Transistor, Power, NPN Silicon, Type K2N1724
KSC-S-101/6	Transistor, NPN, Dual, Silicon, .K2N2060
KSC-S-101/7	Diode, Silicon, High Speed Switching, Type K1N4153
KSC-S-101/8	Diode, Silicon, General Purpose, Type K1N645
KSC-S-101/9	Diode, Zener, Silicon, Type K1N967B
KSC-S-101/10	Diode, Silicon, Power Rectifier, Type K1N250C
KSC-S-101/11	Transistor, NPN, Switching, Type K2N2846
KSC-S-101/12	Transistor, NPN, Silicon, Power, Type ZN2811
KSC-S-101/13	Diode, Zener, Silicon, Power, Type K1N2814A, K1N2827A, K1N2828A
KSC-S-101/14	Transistor, NPN, Silicon, Power, Type K2N3265
KSC-S-101/15	Diode, Rectifier, Silicon, Power, Fast Recovery, Type K1N3902
KSC-S-101/16	Transistor, PNP Silicon, Power, Type K2N3026
KSC-S-101/17	Transistor, PNP, Silicon, Power, Type K2N3026
KSC-S-101/18	Transistor, PNP, Silicon, Type K3516
KSC-S-101/19	Transistor, NPN, Silicon, Power, Type K2N2658
KSC-S-101/20	Transistor, PN, Silicon, Unijunction, Type K2N493
KSC-S-101/21	Transistor, PNP, Silicon, Type K2N2276
KSC-S-101/22	Transistor, NPN, Silicon, High Power, Type K2N2746
KSC-S-101/23	Transistor, NPN, Silicon, Type K2N2432
KSC-S-101/24	Diode, Temperature Compensated Reference, Sub-miniature, Type K1N825A
KSC-R-102	Resistors, Fixed, Film (High Stability) (Styles KRN 55/60/65/70C)

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>KENNEDY SPACE CENTER DOCUMENTS (Continued)</b>	
KSC-R-103	Resistors, Fixed, Film (High Stability) (Styles KRL 07/20/32/42)
KSC-R-104	Resistor, Fixed, Power, Wirewound (Chassis Mount)
KSC-R-105	(Style KRE 65/70/75)
KSC-C-106	Resistors, Fixed, Wirewound (Insulated) (Style KRW 67/68/69)
KSC-C-107	Capacitor, Fixed, Glass Dielectric, High Reliability
KSC-C-108	(Styles CYER 10/15/20/30)
KSC-C-109	Capacitors, Fixed, Tantalum (Polarized, Etched Foil)
KSC-C-110	(Style KCL 21)
KSC-C-111	Capacitors, Fixed, Tantalum Non-Polar Etched Foil
KSC-C-112	(Style KCL 23)
KSC-C-113	Capacitors, Fixed, Tantalum (Polarized, Plain Foil)
KSC-C-114	(Style KCL 31)
KSC-C-115	Capacitors, Fixed, Tantalum (Nonpolarized, Plain Foil)
KSC-C-116	(Style KCL 33)
KSC-C-117	Capacitors, Fixed, Solid, Tantalum (Style KCS13)
KSC-C-118	Capacitors, Fixed, Plastic Dielectric, (Nonmetallic Case)
KSC-C-119	(Style KCTM)
KSC-C-120	Capacitors, Fixed, Polarized, Tantalum Foil (Style KCL 51
KSC-C-121	and KCL 53)
KSC-C-122	Capacitors, Fixed, Paper or Paper-Plastic (Style KCVP09)
KSC-P-116	Packaging and Marking for Cables and Harnesses, Procedure for
KSC-STD-132	Potting and Molding Cable Assembly Termination
KSC-W-151	Solderless Wrap Process, Electrical Connections, Specification for
KSC-STD-152-1	Graphical Symbols for Drawings, Part 1 - Electrical and Logic
KSC-STD-152-2	Symbols
KSC-E-153	Graphical Symbols for Drawings, Parts, Mechanical Symbols
KSC-STD-164	Enclosures, Modular, Radio Frequency Interference Shielded,
KSC-E-165	Specification for
KSC-W-167	Environmental Test Methods for Ground Support Equipment
KSC-STD-169	Installations at John F. Kennedy Space Center
SP-4-28-D	Electrical Ground Support Equipment Fabrication
	Wiring Programming System Patchboards, Procedure for
	Marking of Ground Support Equipment
	Design Test Data, Mechanical and Electro-Mechanical
	Components

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>KENNEDY SPACE CENTER DOCUMENTS (Continued)</b>	
SP-4-38-D	Shock and Vibration Environments and Test Specification Levels, Ground Support Equipment, Launch Complex 39
SP-80-D	Guide for Environmental Protection When Using Electrical Ground Equipment Within the Areas of Saturn Complexes Where Hazardous Areas Exist
SP-82-D	Electrical and Electro-Mechanical Components for Saturn Launch Complexes
10M01671	Cleanliness Levels, Cleaning, Protection and Inspection Procedures for Parts, Assys., Sub-systems, and Systems for Pneumatic Use in Support Equipment, Specification for Sealing Compound
A10430105	Pneumatic and Mechanical Components, Electrical Spec for Ring Nut (for Cannon 40 Shell Size Connectors)
A75M00100	Cable Plug Assy (2-3/8 Thd)
B75M012383	Plug Hull (for Cannon 40 Shell Size Connectors)
B75M02122	Nut (for Cannon and Bendix 40 Shell Size Connectors)
B75M02123	Boot (for Banana Plug)
B75M02124	Ring Nut (for Bendix 40 Shell Size Connectors)
B75M02177	Adapter
B75M02183	Protection Cap (Umbilical)
C75M02196	Boot
C75M02197	Cable Plug Assy (2-5/16 Thd)
C75M02197-1	Plug Hull (for Bendix 40 Shell Size Connectors)
B75M02943	Application of Heat and Blast Resistant Coating of Cables, Proc for
B75M02944	Adapter (Connector to Conduit) (Refer to dwg. for dash no.)
A75M03589	Cast Chassis, Complex 39
B75M04097-X	Solderless Electrical Connections, Proc for
F75M05218	Fabrication and Installation of Tube Assemblies and Installation of Fitting and Fitting Assemblies, Spec for
A75M05668	MI Cable Terminations, Proc for
A75M05875	Adapter
A75M07450	Electrical Cable Assembly, Spec for
B75M07830	Splicing and Serving Lifting Cable, Procedure for
A75M09397	Hydraulic System Components and Hydraulic Fluids for
A75M09465	Ground Support Equipment (GSE) Cleaning, Testing, and Handling Procedure
A75M09467	Sealing of Electrical Components, Spec for
A75M09468	

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>KENNEDY SPACE CENTER DOCUMENTS (Continued)</b>	
A75M09470	Welding Stainless Steel and Invar Pipe, Specification for
A75M09946	Welding of Aluminum Alloy Pipe, Specification for
A75M11300	Tape, Marker (Green -1, Red -2)
A75M12110	Heat and Blast Protective Coating
A75M13302	Connector Inspection Specifications
C75M13303	Molding Adapter General Purpose Connector
C75M13304	Potting Adapter General Purpose Connector
C75M13676	Shield Rings
A75M13965	Flash Evaporation Method of Determining Non-Volatile
	Residue Content in Freon and Trichloroethylene Solvents,
	Procedure for
B75M50703	Adapter (Flexible Armored Cable)
A75M51074	Electrical Spec. Ground Support Equipment Installation
A75M10180	6 Cond. 16 AWG OS 6 UNSH
A75M10181	60 Cond. 16 AWG OS 60 UNSH
A75M10182	60 Cond. 16 AWG OS 60 SS
A75M10183	60 Cond. 16 AWG OS 30 PTSI
A75M10184	60 Cond. 16 AWG OS 20 TTSI
A75M10185	60 Cond. 16 AWG OS 15 QTSI
A75M10186	60 Cond. 16 AWG OS 60 SSI
A75M10187	60 Cond. 14 AWG OS 2TTSI, 3PTSI, 48UNSH
A75M10188	60 Cond. 14 AWG OS 3TTSI, 3PTSI, 45SSI
A75M10189	7 Cond. 8 AWG OS 7 UNSH
A75M10190	60 Cond. 20 AWG OS 60 UNSH
A75M10191	60 Cond. 14 AWG OS 60 UNSH
A75M10192	60 Cond. 14 AWG OS 60SS
A75M10193	60 Cond. 14 AWG OS 30 PTSI
A75M10194	60 Cond. 14 AWG OS 20 TTSI
A75M10195	60 Cond. 14 AWG OS 15 QTSI
A75M10196	10 Cond. 16 AWG OS 10 UNSH
A75M10197	6 Cond. 8 AWG OS 6 UNSH
A75M10198	4 Cond. -- OS 3#8, 1#16
A75M10199	5 Cond. 12 AWG -- 5 UNSH
A75M10200	4 Cond. 0 OS 4 UNSH
A75M10201	61 Cond. 20 AWG -- 61 UNSH
A75M10202	32 Cond. 20 AWG OS 32 UNSH
A75M10203	3 Cond. 8 AWG -- 3 UNSH
A75M10204	6 Cond. -- OS 4RG214/U, 1RG62A/U, 1#12

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION				
KENNEDY SPACE CENTER DOCUMENTS (Continued)					
A75M11003	40 Cond.	--	OS	5 QTSI#10, 20UNSH#14	
A75M11004	52 Cond.	--	OS	4PTSI#7, (11TTSI, 5PTSI, 1SSI#14)	
A75M11005	60 Cond.	14 AWG	OS	5PTSI, 50SSI	
A75M11006	33 Cond.	--	OS	3#0, 3#4, 13PTSI#14, ISSI#14	
A75M11007	29 Cond.	--	OS	4#2, 9#6, 16#14	
A75M11008	40 Cond.	14 AWG	OS	20 PTSI	
A75M11009	60 Cond.	14 AWG	OS	4TTSI, 12PTSI, 24SSI, ABLATIVE SHEATH	
A75M11010	60 Cond.	14 AWG	OS	5PTSI, 50 UNSH	AB
A75M11011	60 Cond.	14 AWG	OS	2TTSI, 3PTSI, 48 UNSH	AB
A75M11012	60 Cond.	14 AWG	OS	6TTSI, 17PTSI, 8SSI	AB
A75M11013	60 Cond.	14 AWG	OS	15 OTSI	AB
A75M11014	60 Cond.	14 AWG	OS	20 TTSI	AB
A75M11015	10 Cond.	16 AWG	OS	10 UNSH	AB
A75M11016	60 Cond.	16 AWG	OS	60 UNSH	AB
A75M11017	6 Cond.	16 AWG	OS	6 UNSH	AB
A75M11018	4 Cond.	0	OS	4 UNSH	AB
A75M11019	29 Cond.	--	OS	4#2, 9#6, 16#14	AB
A75M11020	40 Cond.	--	OS	5QTSI#10, 20 UNSH#14	AB
A75M11021	52 Cond.	--	OS	4PTSI#7, (11TTSI, 5PTSI, 1SSI#14)	AB
A75M11022	60 Cond.	14 AWG	OS	5PTSI, 50SSI	AB
A75M11023	33 Cond.	--	OS	3#0, 3#4, 13PTSI#14, 1SSI#14	AB
A75M11024	60 Cond.	14 AWG	OS	60 SSI	AB
A75M11025	60 Cond.	14 AWG	OS	30 PTSI	AB
A75M11026	60 Cond.	14 AWG	OS	60 UNSH	AB
A75M11027	6 Cond.	--	OS	4RG214/U, 1RG62A/U, 1#12	AB
A75M14073	60 Cond.	14 AWG	OS	6TTSI, 17PTSI, 8SSI	
A75M14074	7 Cond.	--	OS	7RG180B/U	
A75M14075	60 Cond.	16 AWG	--	60 UNSH	
A75M14076	60 Cond.	14 AWG	OS	5PTSI, 50UNSH	
A75M14077	21 Cond.	8 AWG	OS	21 UNSH	
A75M14078					
A75M14079					
A75M14080					
A75M14081					

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>KENNEDY SPACE CENTER DOCUMENTS (Continued)</b>	
<p>A75M14082  A75M14083  A75M14084  A75M14085  A75M14086  A75M14087  A75M14088  A75M14089  A75M14090  A75M14091  A75M14092</p>	<p>6 Cond.      8 AWG      OS      6 UNSH      AB</p>
<b>MARSHALL SPACE FLIGHT CENTER DOCUMENTS</b>	
MSFC-STD-105	Synthetic Rubber, Age Control of, Std for
MSFC-SPEC-106	Testing Compatibility of LOX, Spec for
MSFC-SPEC-119	Connectors, Receptacle, Electrical, Spec for
MSFC-SPEC-130	Certification of Welding Machine Operators and Welders, Spec for
MSFC-SPEC-143	Fittings, Flared Tube (Premium Quality) Pressure Connections, Spec for
MSFC-STD-154	Printed Circuit Design & Construction, Std for
MSFC-STD-156	Riveting, Fabrication and Inspection, Std for
MSFC-SPEC-164	Cleanliness of Components for use in Oxygen, Fuel, and Pneumatic Systems, Spec for
MSFC-PROC-195	Cleanliness Level Requirements and Inspection Methods for Determining Cleanliness Level of Gas Bearing Gas Supply and Slash Measuring Systems, Proc for
MSFC-SPEC-202	Compound, Potting and Molding, Elastomeric, Spec for
MSFC-SPEC-222	Resins Systems, Elec and Environmental, Insulation, Epoxy, Spec for
MSFC-PROC-238	Fittings, Female Threads (Prem. Qual.), Preparation and Lubrication, Proc for
MSFC-SPEC-249	Bonding and Grounding, Electrical, Spec for
MSFC-PROC-257	Conformal Coating of Printed Circuit Assys, Proc for
MSFC-PROC-273	Tubing, Heat Reactive, Instl. of, Proc for
MSFC-PROC-274	Terminals, Instl. of, Proc for
MSFC-SPEC-276	Tubing, Heat Reactive, Spec for
MSFC-SPEC-278	Terminals, Bifurcated and Turret, Swage Type;

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>MARSHALL SPACE FLIGHT CENTER DOCUMENTS (Continued)</b>	
MSFC-SPEC-278 (Continued)	Terminals, Bifurcated and Turret, Standoff, Insulated Screw Type, Spec for
MSFC-PROC-310	Potting of Electrical Distributors, Proc for
MSFC-SPEC-331	Enclosures, Modular Shielded, RFI, Spec for
MSFC-SPEC-332	Cables, Electrical General Spec for (Slash Sheets not applicable)
MSFC-QPL-332	Qualified Products List for MSFC-SPEC-332
MSFC-SPEC-338	Semiconductor Devices, General Spec for (Slash Sheets/1 through/119 may apply)
MSFC-SPEC-339	Relays, DC, Hermetically Sealed, for Space Vehicles and GSE, General Spec for
MSFC-SPEC-339/1 thru/94	Individual Relay Specs
MSFC-STD-350	Abbreviations for use on Drawings, Std for
NPC 200-2	Quality Program Provisions for Space System Contractors
NPC 200-3	Inspection System Provisions for Supplies of Space Materials, Parts, Components, and Services
NPC 200-4	Quality Requirements for Hand Soldering of Electrical Connections
D10507600	Flared Tube Fittings Concentricity Gage
A10509301	Riveting, Fabrication, and Inspection, Standard for
A10509308	Welding, Carbon, Low Alloy and Stainless Steel, (Manual or Auto) Spec for
A10M01671	Cleanliness Levels, Cleaning, Protection and Inspection Procedures for Parts, Assemblies, Subsystems, and Systems, for Pneumatic use in Support Equipment, Spec for
A50M01161	Aluminum GSE (Except Equipment Racks) Paint Finishing of, Proc for
A50M01162	Aluminum Equipment Racks, GSE, Paint Finishing of, Proc for
A50M01165	Aluminum Electrically Conductive Chromate Coating of, Proc for
<b>MILITARY STANDARDS AND SPECIFICATIONS</b>	
MIL-STD-28	Drawing Titles, Approved Method For Assignment of
MIL-STD-108	Definitions of and Basic Requirements for Enclosures For Electric and Electronic Equipment
MIL-STD-129	Marking for Shipment and Storage
MIL-STD-163	Steel Mill Products Preparation for Shipment and Storage
MIL-STD-171	Finishing of Metal and Wood Surfaces

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>MILITARY STANDARDS AND SPECIFICATIONS (Continued)</b>	
MIL-STD-183	Canceled (See FED-STD-183)
MIL-STD-429	Printed Circuit Terms & Definitions
MIL-STD-810	Environmental Test Methods for Aerospace and Ground Equipment
MIL-A-1154	Adhesive, Bonding, Vulcanized Synthetic Rubber to Steel
MIL-A-6091	Alcohol, Ethyl, Specially Denatured, Aircraft
MIL-A-8625	Anodic Coatings, for Aluminum and Aluminum Alloys
MIL-A-25994	Aluminum Alloy Angles, Channels I and Z Beams, Extruded or Rolled Structural Shapes
MIL-B-117	Bags, Interior Packaging
MIL-B-5087	Bonding, Electrical (For Aircraft) ASG#
MIL-B-7883	Brazing of Steels, Copper, Copper Alloys, and Nickel Alloys
MIL-B-22205	Bags, Transparent, Flexible, Heat Sealable for Packaging Applications
MIL-B-43014	Boxes, Water Resistant Paper Board, Folding, Set-up and Metal-Stayed
MIL-C-17	Cables, Radio Frequency, Coaxial, Dual Coaxial, Twin Conductor, Twin Lead
MIL-C-104	Crates, Wood, Lumber and Plywood Sheathed, Nailed and Bolted
MIL-C-5015	Connectors, Electric, AN, Type
MIL-C-7769	Cushioning Material, Bound Fiber
MIL-C-12000	Cable, Cord, and Wire, Electric Packages of Connectors. Electrical, Waterproof, Quick Disconnect, Heavy Duty Type
MIL-C-26482	Connectors, Electric, Circular, Miniature, Quick Disconnect (Navy)
MIL-C-26636	Contacts, Crimp Type, For Electrical Connectors
MIL-C-40091	Crimping Tool, Terminal, Hand
MIL-D-3464	Desiccants, Activated, Bagged, Packaging use and Static Dehumidification
MIL-E-463	Ethyl Alcohol (for Ordnance Use)
MIL-E-5272	Environmental Testing, Aeronautical and Associated Equipment, General Spec.
MIL-E-15090	Enamel, Equipment, Light Gray, Formula No. (11)
MIL-F-21608	Ferrule, Shield Grounding, Insulated, Crimpt Style, Brass (ASG)
MIL-G-613	Grapnels, Marine, Trip Wire, and Crash Truck



Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>MILITARY STANDARDS AND SPECIFICATIONS (Continued)</b>	
MIL-G-5510	Packing, Preformed, Straight, Thread Tube Fitting Boss
MIL-G-45204	Gold Plating - Electrodeposited
MIL-I-631	Insulation, Electrical, Synthetic - Resin Composition; Nonrigid
MIL-I-3190	Insulation Sleeving, Electrical, Flexible Treated
MIL-I-6181	Interference Control Requirements, Aircraft Equipment
MIL-I-7444	Insulation Sleeving, Electrical, Flexible
MIL-I-8660	Insulating and Sealing Compound, Electrical
MIL-I-10428	Isopropyl Alcohol, Technical
MIL-I-22129	Insulation Tubing, Electrical Polytetrafluorethylene Resin Nonrigid
MIL-P-78	Plastic-Material-Laminated, Thermosetting For Designation Plates
MIL-P-116	Preservation, Methods of
MIL-P-130	Paper, Wrapping, Laminated and Creped
MIL-P-8585	Primer Coating, Zinc Chromate, Low-Moisture-Sensitivity
MIL-P-11414	Primer, Lacquer, Rust-inhibiting
MIL-P-13949	Plastic Sheet, Laminated, Copper Clad
MIL-P-18177	Plastic Sheet, Laminated, Thermosetting, Glass Fiber Base, Epoxy-Resin
MIL-R-26	Resistors, Fixed, Wirewound (Power Type)
MIL-R-93	Resistors, Fixed, Wirewound (Accurate) General Spec for
MIL-R-94	Resistors, Variable, Composition, General Spec for
MIL-R-6855	Rubber, Synthetic, Sheet, Molded, and Extruded for Aircraft Applications
MIL-R-10509	Resistors, Fixed Film, High Stability General Spec for
MIL-R-55182	Resistors, Fixed, Film, Established Reliability, General Spec for
MIL-S-4461	Sealing Machines, Heat, Bench, and Portable (Temperature, Pressure and Time Controlled).
MIL-S-5002	Surface Treatments (Except Priming and Painting) for Metal and Metal Parts in Aircraft
MIL-S-7952	Steel - Sheet and Strip, Uncoated, Carbon 1020 and 1025 (Aircraft Quality)
MIL-S-18729	Steel-Plate, Sheet and Strip, Alloy, 4130 Aircraft Quality
MIL-S-19500	Semiconductor Devices, General Spec for
MIL-T-713	Twine and Tape, Lacing and Typing (for use in Electrical and Electronic Equipment)

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>MILITARY STANDARDS AND SPECIFICATIONS (Continued)</b>	
MIL-T-5021	Tests, Aircraft and Missile Welding Operations Qualifications ASG#
MIL-T-7928	Terminals, Lug and Splice, Crimp Style, Copper (ASG)
MIL-T-10727	Tin Plating, Electrode Deposited or Hot-Dipped, For Ferrous and Non-Ferrous Metals
MIL-T-21595	Tape, Pressure-Sensitive Adhesive, Paper, Masking-Non-Staining
MIL-V-173	Varnish, Moisture-and-Fungus-Resistant (For the Treatment of Communications, Electronic, and Associated Electrical Equip)
MIL-W-5086	Wire, Electrical, 600 Volt, Copper, Aircraft
MIL-W-5088	Wiring, Aircraft, Installation of (ASG) Three-Phase For the Production of Aircraft Quality Welds
MIL-W-7973	Welding Machines, Electrical Resistance, Spot, Press Type
MIL-W-8604	Welding of Aluminum Alloys - Process of
MIL-W-8611	Welding, Metal Arc and Gas, Steels, and Corrosion and Heat Resistant Alloys, Process for
MIL-W-16878	Wire, Electrical, Insulated, High Temperature (Navy)
MIL-Y-1140	Yarn, Cord, Sleeving, Cloth, and Type-Glass
<b>FEDERAL STANDARDS AND SPECIFICATIONS</b>	
FED-STD-102	Preservation, Packaging, and Packing Levels
FED-STD-183	Identification Markings for Iron and Steel Proc
FED-STD-595	Colors
BB-N-411	Nitrogen
J-C-98	Cable and Wire, Insulated, Methods of Sampling and Testing
L-P-391	Plastic-Methacrylate, Sheets, Rods and Tubes-Cast
L-P-590	Plastic Compounds, Molding and Extrusion, Polyethylene
O-E-760	Ethyl Alcohol Ethanol, Denatured Alcohol, and Proprietary Solvent
O-T-634	Trichloroethylene, Technical
P-S-661	Solvent-Dry-Cleaning
PPP-T-60	Tape, Pressure Sensitive Adhesive, Water Proof for Packaging and Sealing
PPP-B-601	Boxes, Wood, Cleated, Plywood
PPP-B-621	Boxes, Wood, Nailed and Lock Corner
PPP-B-636	Box Fiber Board

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>FEDERAL STANDARDS AND SPECIFICATIONS (Continued)</b>	
PPP-B-640	Boxes, Fiber Board, Corrugated Triple Wall
QQ-A-200/9	Aluminum Alloy Bar, Rod, and Structural and Special Shaped Sections Extruded 6063
QQ-A-250/8	Aluminum Alloy 5052, Plate & Sheet
QQ-A-250/11	Aluminum Alloy 6061, Plate & Sheet
QQ-A-274	Aluminum Alloy Bars, Rods, and Shapes Extruded 6063
QQ-A-282	Aluminum Alloy Bars, Rods, Wire, and Special Shapes, Rolled, Drawn, or Cold Finished 7075
QQ-A-362	Aluminum Alloy Plate and Sheet, Alclad 2024
QQ-B-613	Brass, Leaded and Non-Leaded, Plate Rolled, Bar Sheet and Strip
QQ-P-416	Plating, Cadmium, Electrodeposited
QQ-S-571	Solder, Lead Alloy, Tin Lead Alloy, and Tin Alloy, Flux Cored Ribbon and Wire and Solid Form
QQ-S-633	Steel Bars, Carbon, Cold Finished and Hot Rolled (General Purpose)
QQ-W-343	Wire, Electrical and Non-Electrical, Copper, (Uninsulated)
TT-C-490	Cleaning Methods and Pretreatment of Ferrous Surfaces for Organic Coatings
TT-E-489	Enamel, Alkyd, Gloss, (For exterior and interior surfaces)
TT-E-529	Enamel, Alkyd, Lustreless
TT-I-558	Ink, Marking Stencil, Opaque, for nonporous Surfaces, Metals, Glass, etc.
TT-M-261	Methyl-Ethyl-Ketone for use in Organic Coatings
TT-P-636	Primer Coatings, Synthetic, Wood and Ferrous Metal
TT-P-662	Primer Surfacer, Sanding, Lacquer and Enamel Type
TT-P-666	Primer Coating, Zinc Yellow, for Aluminum and Magnesium Surfaces
TT-X-9-6	Xylene for use in Organic Coatings
UU-T-81	Tags, Shipping and Stock
WW-T-799	Tubing, Copper, Seamless for use with Solder-Joint or Flared to be Fitting
Z-O-358	Oil-Peanut
<b>MILITARY STANDARDS</b>	
MS 20426	Rivet, Solid, Countersunk 100 Degrees, Precision Head Aluminum
MS 20470	Rivet, Solid-Universal Head, Aluminum and Aluminum Alloy

Table 6-1 REFERENCE SPECIFICATIONS (Continued)

DOCUMENT NUMBER	DESCRIPTION
<b>MILITARY STANDARDS (Continued)</b>	
MS 20600	Rivet, Blind, Structural Pull Stem Self Plugging, Protruding Head, Type II, Class I
MS 20601	Rivet, Blind, Self Plugging, 100 deg flush Head, Type II, Class II
MS 20602	Rivet, Blind, Chemically Expanded, Protruding Head, Type I, Class I, Styles A and B
MS 20603	Rivet, Blind, Chemically Expanded, 100 degree flush head, Type I, Class II, Styles A and B
MS 20659	Terminal, Lug, Crimp Style, Copper, Uninsulated, Class I (ASG)
MS 25036	Terminal, Lug, Crimp Style, Copper Insulated, Class I (ASG)
MS 25037	Crimping Tool, Hand, For Copper Insulated Terminal
MS 25042	Cap, Electrical Connector Plug, Dust
MS 25043	Cap, Electrical Connector, Receptacle, Dust
MS 25083	Jumper - Assemblies Bonding and Current Return (ASG)
MS 25274	Cap, Wire End (Class I)
MS 25311	Ferrule, Shield Grounding, one piece, Insulated, Class I, for Coaxial and Shielded Cable
MS 25312	Tool - Crimping, Hand, For Insulated Shield, Ground (ASG)
MS 33584	Tubing End, Standard Dimensions for Flared
<b>MISCELLANEOUS SPECIFICATIONS</b>	
AN 735	Clamp, Loop Type Bonding
AN 742	Clamp, Plain, Support, Loop Type, Aircraft
AND 10387	Drill Sizes and Drilled Hole Tolerances, Twist
NAS 523	Code, Rivet

Table 6-2 DOCUMENTS DISCONTINUED FROM USE BY LAUNCH  
EQUIPMENT BRANCH, ELECTRICAL SECTION

DISCONTINUED DOCUMENTS	SUPERCEDING DOCUMENTS
ABMA-PD-W-45	MIL-W-8604
ABMA-PD-R-187	MSFC-SPEC-339
ABMA-STD-428	MSFC-STD-154
A10338552	KSC-E-165
A10419905	KSC-STD-169
A10423785	KSC-STD-169
A10M01071	SP-80-D
A75M01260	KSC-STD-132 (In Process)
A75M50222	KSC Dwgs. A75M10180 thru 10204, 11003 thru
A75M50223	11027, 14073 thru 14092, and MSFC-SPEC 332
A75M50497	KSC-E-165
A75M50741	KSC-E-165
A75M51073	KSC-E-165
MIL-STD-12	MSFC-STD-350
MIL-STD-130	KSC-STD-169 and KSC-P-116
MIL-C-13777	MSFC-SPEC 332
MIL-I-7444	MSFC-SPEC-276
MIL-S-8484	A75M09468
QQ-A-318	QQ-A-250/8 (5052)
	QQ-A-250/11 (6061)
QQ-A-327	QQ-A-200/9 Extrusions
MSFC-STD-110	Use only for guideline without use of applicable documents
MSFC-STD-158	MSFC-PROC-256
MSFC-STD-163	Discontinued
MSFC-PROC-186	MSFC-PROC-256
MSFC-PROC-196	MSFC-PROC-256
MSFC-PROC-256	NPC 200-4 and KSC-E-165

Table 6-3 ABBREVIATIONS, CABLE SPECIFICATIONS

ABBREVIATION	DESCRIPTION
C	Conductors
P	Pairs
S	Single
T	Triplets
Q	Quads
(Q)	Quints
SH	Shield
SS	Single conductor shielded
SSI	Single conductor shielded and insulated
PTS	Pairs twisted and shielded
PTSI	Pairs twisted, shielded, and insulated
TTS	Triplets twisted and shielded
TTSI	Triplets twisted, shielded, and insulated
QTS	Quads twisted and shielded
QTSI	Quads twisted, shielded, and insulated
(Q)TS	Quints twisted and shielded
UNSH	Unshielded
OS	Overall shield
OSDB	Overall shield double braid

**6-3. ESE EQUIPMENT LOCATIONS.**

ESE electrical enclosures on the Mobile Launcher are shown in at least one of the figures 6-2, 6-3, or 6-4. Enclosures for the following subsystems are largely absent:

- a. S-IC Engine Servicing Platforms Subsystem
- b. Platform Transporter Subsystem
- c. S-II and S-IVB Engine Servicing Platform Subsystem

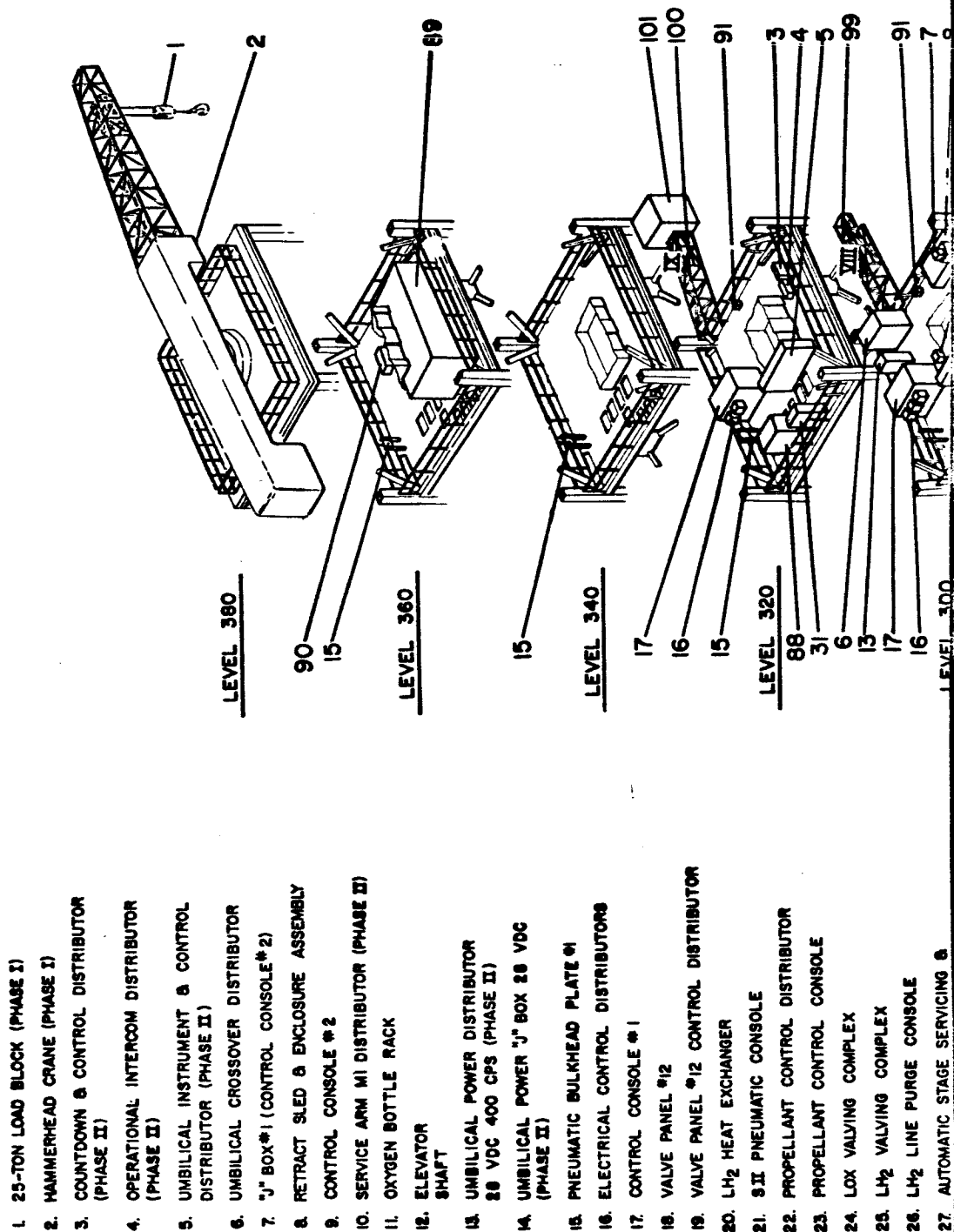


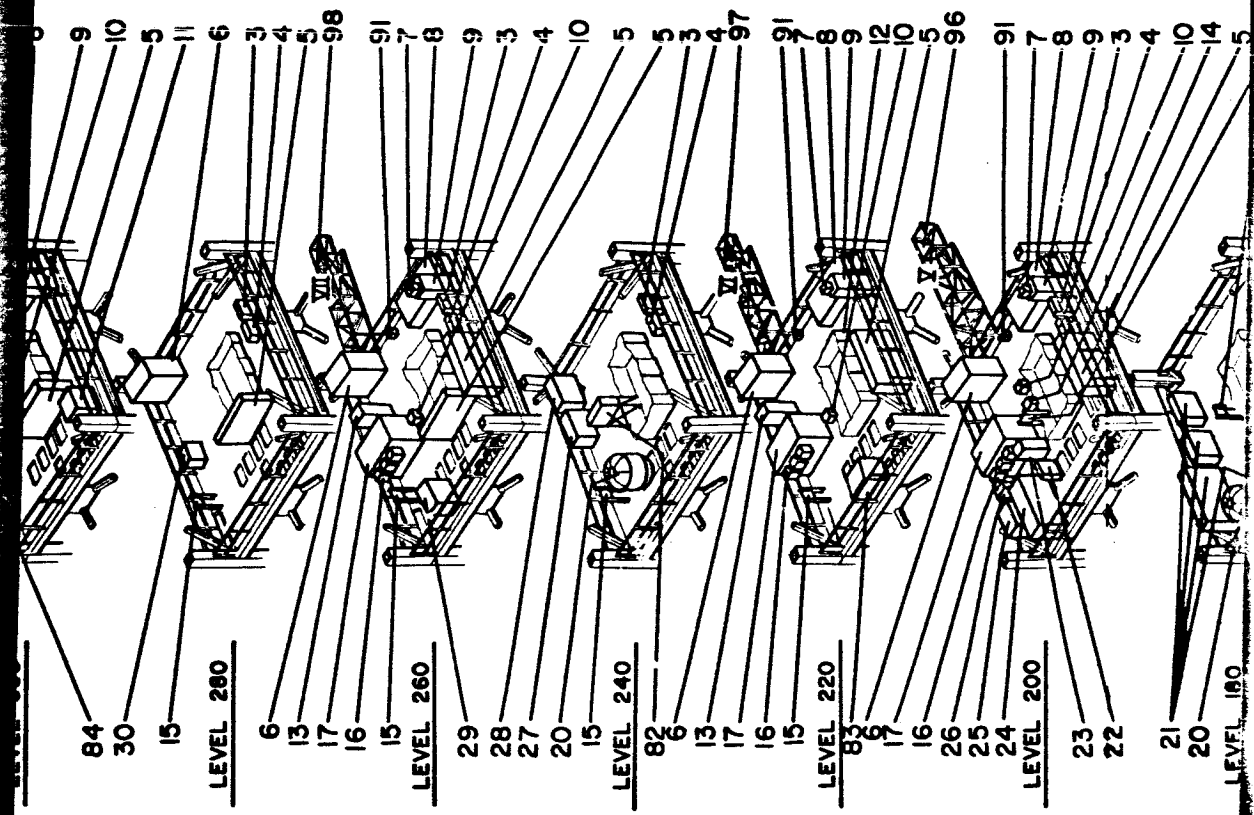
Figure 6-3 Equipment Location, Mobile Launcher



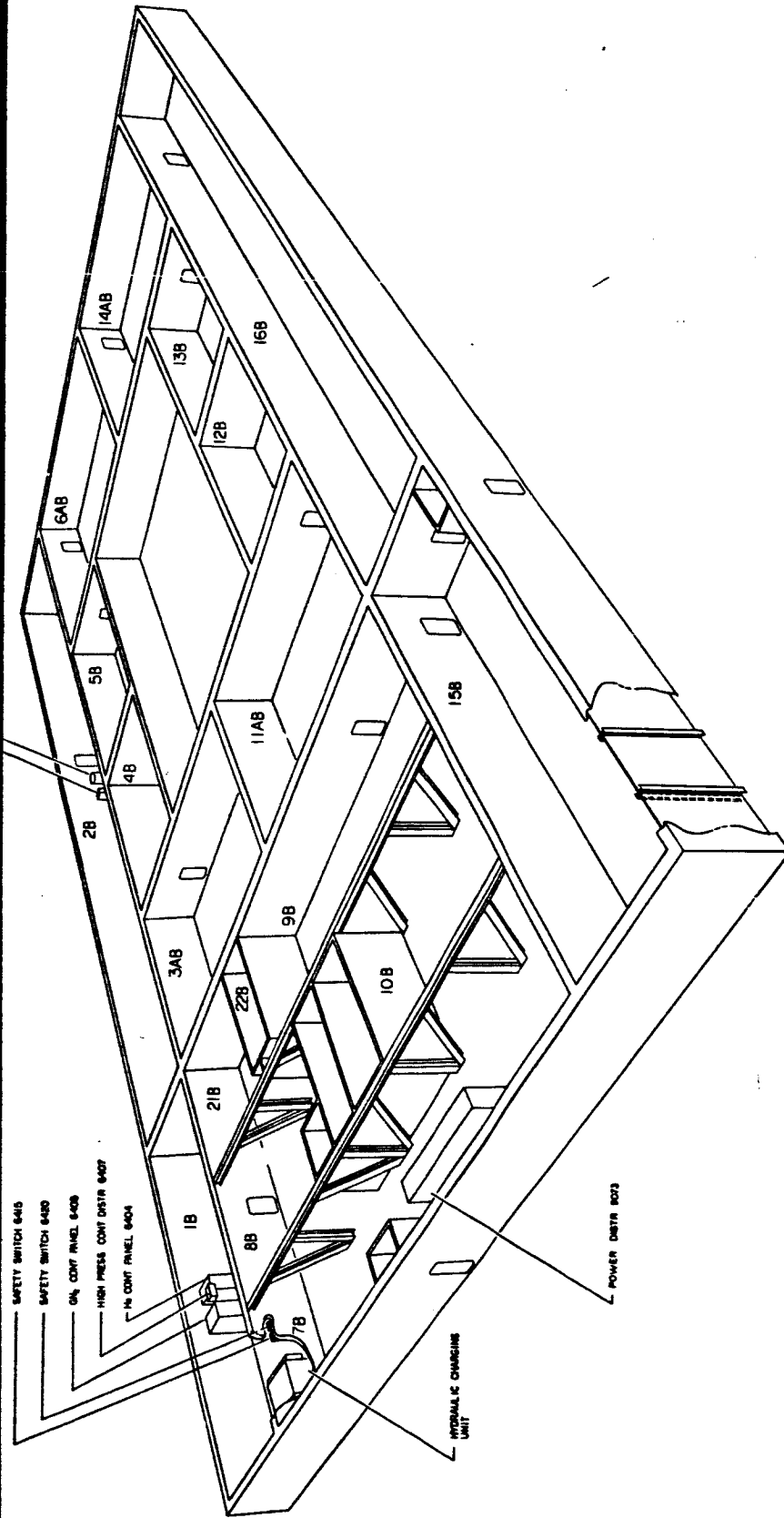


CHECKOUT PNEUMATIC CONSOLE 'B'

- 28. AUTOMATIC STAGE SERVICING CHECKOUT PNEUMATIC CONSOLE 'A'
- 29. WATER METHANOL UNIT
- 30. INSTRUMENT UNIT PNEUMATIC CONSOLE
- 31. HYDRAULIC ACCUMULATOR
- 32. HOLDDOWN ARM #I
- 33. TAIL SERVICE MAST 1-2
- 34. TAIL SERVICE MAST 3-2
- 35. HOLDDOWN ARM #II
- 36. TAIL SERVICE MAST 3-4
- 37. ENGINE SERVICING DISTRIBUTOR
- 38. HOLDDOWN ARM #III
- 39. HOLDDOWN ARM #II
- 40. ENGINE SERVICING PLATFORM RELAY DISTRIBUTOR
- 41. INSTRUMENTATION & CONTROL DISTRIBUTOR 9092
- 42. HOLDDOWN ARM CONTROL DISTRIBUTOR
- 43. SIC PNEUMATIC CONSOLE RACKS
- 44. INSTRUMENTATION & CONTROL DISTRIBUTOR 9091 (PHASE II)
- 45. VALVE PANEL #11
- 46. VALVE PANEL #11 CONTROL DISTRIBUTOR
- 47. SIC PNEUMATIC CHECKOUT RACKS
- 48. STAGE SERVICING ARM POWER RACKS
- 49. MEASURING RACKS
- 50. HYDRAULIC CHARGING UNIT
- 51. G<sub>N</sub>2 CONTROL PANEL
- 52. HIGH PRESSURE CONTROL DISTRIBUTOR
- 53. COMPUTER ELECTRICAL EQUIPMENT RACKS (PHASE III)
- 54. H<sub>2</sub> VALVE PANEL
- 55. INERT PREFILL RESERVOIR SKID
- 56. INSTRUMENTATION & CONTROL DISTRIBUTOR 9082 (PHASE II)



6-164-2

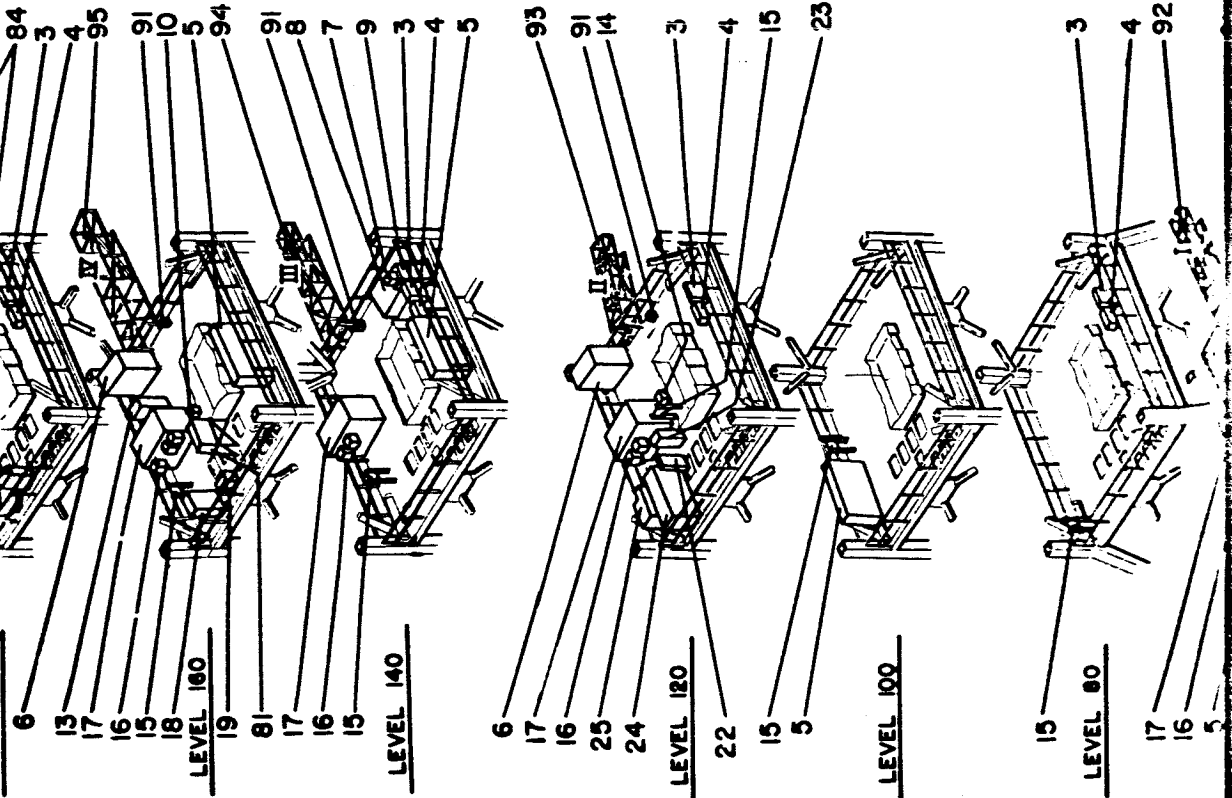


**LEVEL 'B'**

COMPONENT LIST

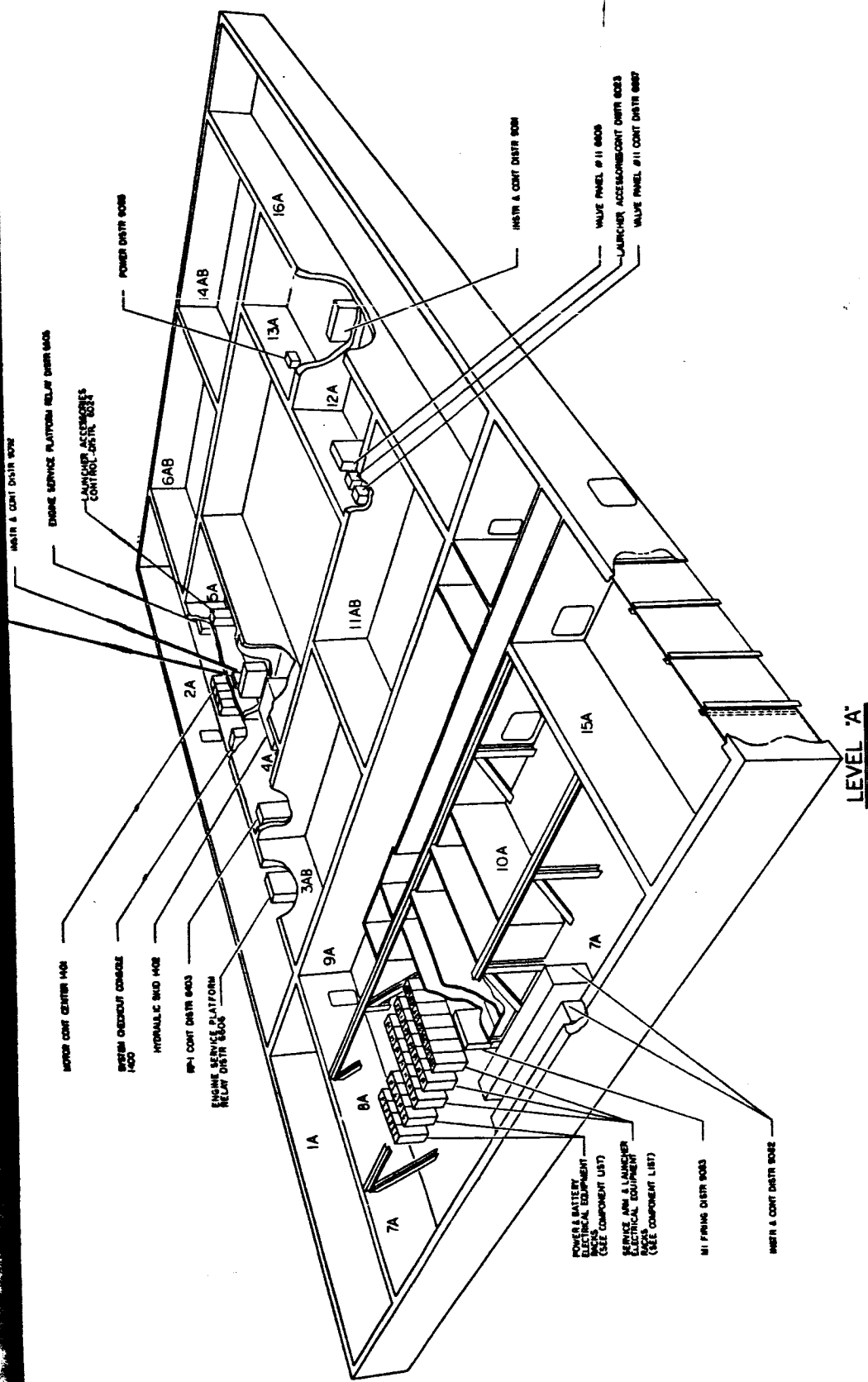
- 1. 8-D ESE
- 2. 8-D ESE
- 3. 8-D ESE
- 4. 8-D ESE
- 5. STANDBY BATTERY RACK #  
CABLE PANEL  
LOAD CONTROL PANEL  
REC. DISTR. TYPE 1A12
- 6. POWER SUPPLY RACK #  
METER PANEL  
POWER SUPPLY
- 7. POWER DISTR. RACK NO.1  
REC. DISTR. TYPE 1A13  
POWER METER PANEL
- 8. POWER DISTR. RACK NO.2  
REC. DISTR. TYPE 1A12

6-1622



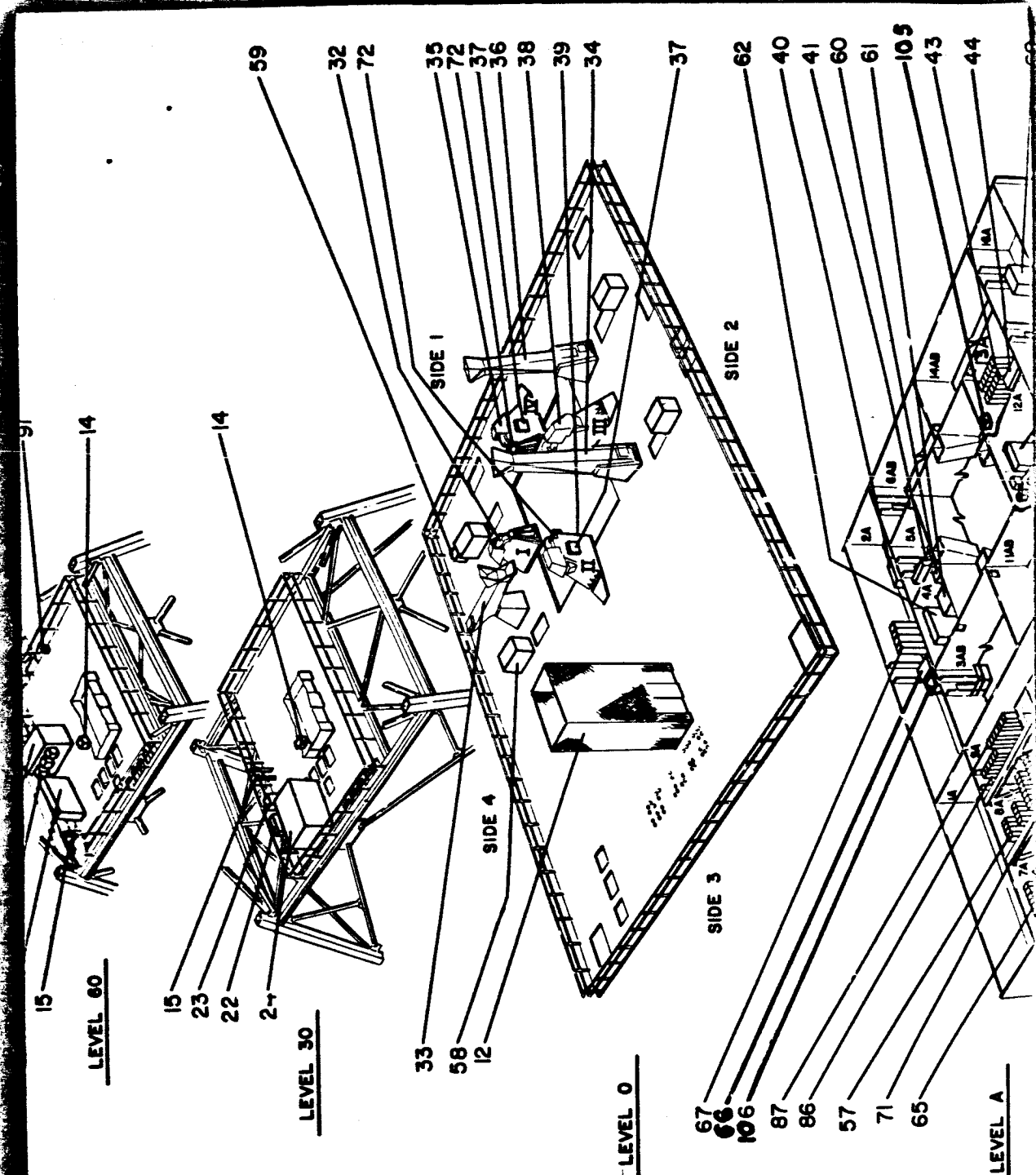
- 57 SERVICE ARM & LAUNCHER ELECTRICAL EQUIPMENT RACKS
- 58 ENGINE SERVICING PLATFORM WINCH (4 PLACES)
- 59 ENGINE SERVICING PLATFORM "J" BOX DISTRIBUTOR
- 60 FUEL VALVE PANEL
- 61 RP-1 SYSTEM
- 62 ENGINE GIMBAL HYDRAULIC UNIT
- 63 INSTRUMENTATION & CONTROL DISTRIBUTOR 9070
- 64 AIR HANDLING UNIT
- 65 STAGE E SE RACKS
- 66 ENGINE GIMBAL HYDRAULIC CHECKOUT CONSOLE
- 67 ENGINE GIMBAL MOTOR CONTROL CENTER
- 68 ENGINE GIMBAL HYDRAULIC SKID
- 69 INSTRUMENTATION UNIT SUBSTATION
- 70 INDUSTRIAL UNIT SUBSTATION
- 71 MI TOWER FIRING DISTRIBUTOR 9063
- 72 SERVICE ARM CONTROL SWITCH
- 73 GROUND MEASURING & DDAS RACKS
- 74 CARD PUNCHER
- 75 CARD READER
- 76 LINE PRINTER
- 77 OIS & OTV RACKS
- 78 PAGING RACK
- 79 S-IC INERT PREFILL CHECKOUT CONSOLE
- 80 DDAS & COMMAND RACKS
- 81 LOX BLANKET PURGE PANEL
- 82 LH<sub>2</sub> BLANKET PURGE PANEL
- 83 S-IB AP SYSTEM PNEUMATIC CONSOLE
- 84 PNEUMATIC BULKHEAD PLATE #2
- 85 WATER VALVE CONTROL DISTRIBUTOR 9084
- 86 ENGINE AND LAUNCHER RACKS

L-164-3



1401 MOTOR CONT CENTER  
 1400 ENGINE OVERCUT CONSOLE  
 HYDRAULIC WIND MOTOR  
 1402 LAUNCHER ACCESSORIES CONTROL CENTER  
 POWER & BATTERY ELECTRICAL EQUIPMENT (SEE COMPONENT LIST)  
 SERVICE AND LAUNCHER ELECTRICAL EQUIPMENT (SEE COMPONENT LIST)  
 M1 FILING DISTRIBUTION BOX  
 MASTER & COAST DISTRIBUTION BOX

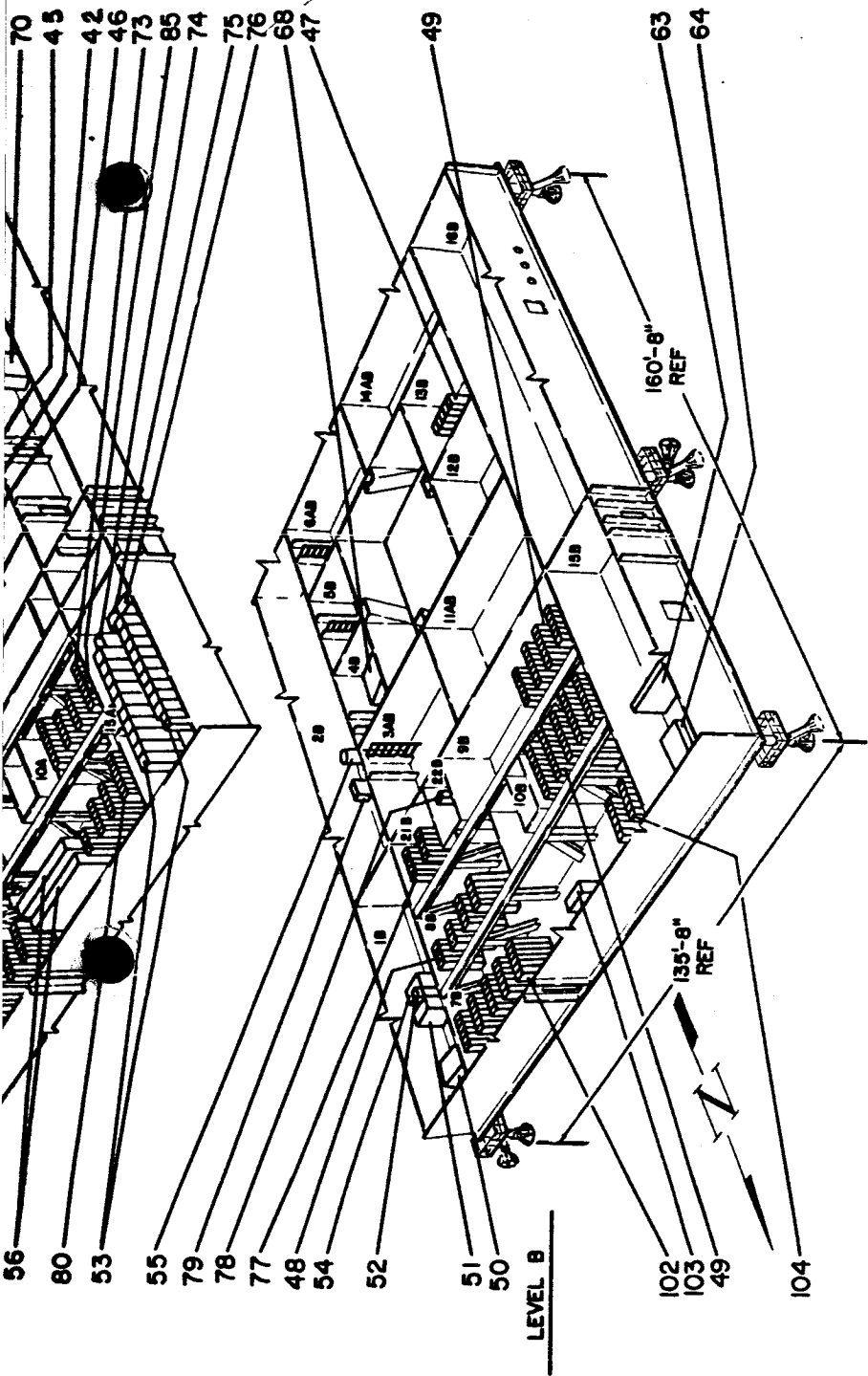
6-163-2



- 67. TANKING & COMPUTING RACKS
- 68. AUTOMATIC CHECKOUT EQUIPMENT ROOM (FACE)
- 69. ELEVATOR MACHINE EQUIPMENT ROOM
- 90. G BALL RETRACT SYSTEM
- 91. H<sub>2</sub>O QUENCH VALVE
- 92. S-IC INTERTANK SERVICE ARM
- 93. S-IC FORWARD SERVICE ARM
- 94. S-II AFT SERVICE ARM
- 95. S-II INTERMEDIATE SERVICE ARM
- 96. S-II FORWARD SERVICE ARM ASSEMBLY
- 97. S-II B AFT SERVICE ARM
- 98. S-II B FORWARD SERVICE ARM
- 99. SERVICE MODULE ARM
- 100. COMMAND MODULE SERVICE ARM
- 101. ENVIRONMENTAL CHAMBER
- 102. STAGE & AUXILIARY POWER RACKS
- 103. 28 VDC 400 CPS POWER DISTRIBUTOR 9073
- 104. M S C POWER & ESE RACKS
- 105. POWER DISTR 9093
- 106. ENGINE SERVICE PLATFORM RELAY DISTR 6606

*6-164-4*





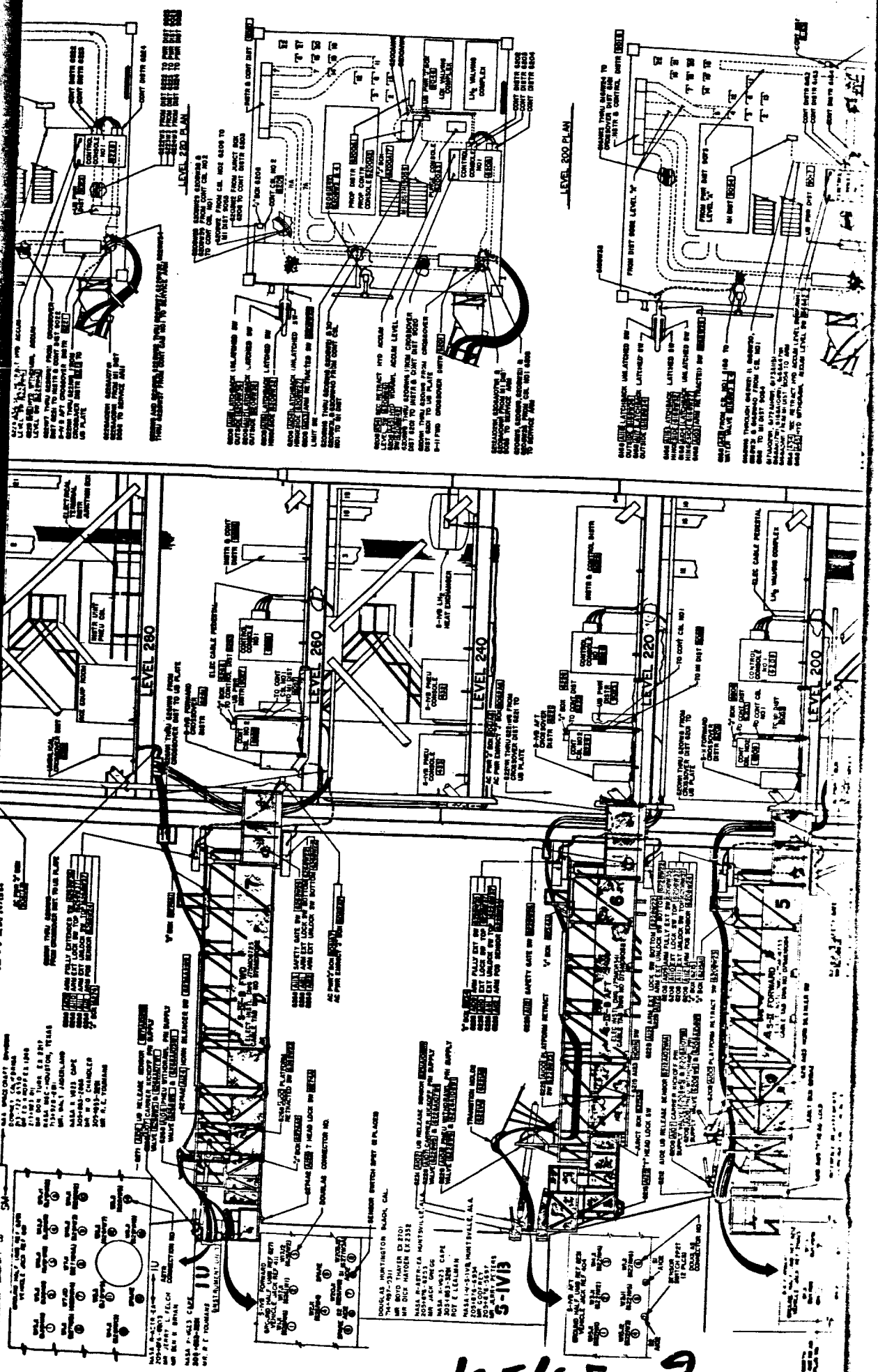
JULY 1, 1964

6-164-5

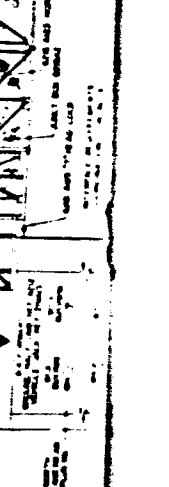
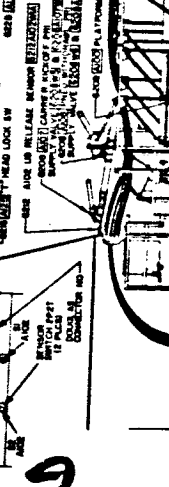
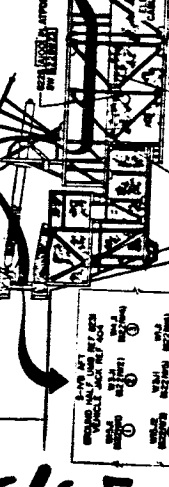
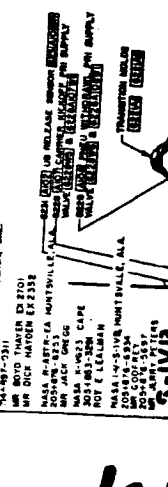
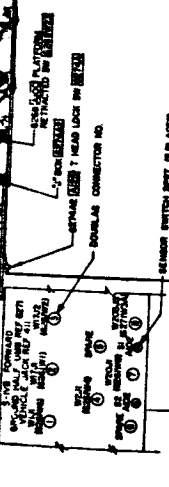
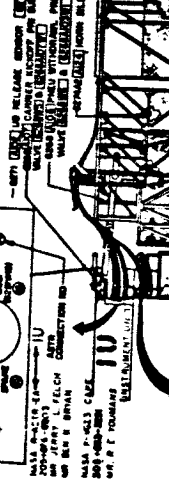








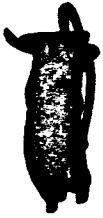
515 5-4 5-10 5-11 5-12 5-13 5-14 5-15 5-16 5-17 5-18 5-19 5-20 5-21 5-22 5-23 5-24 5-25 5-26 5-27 5-28 5-29 5-30 5-31 5-32 5-33 5-34 5-35 5-36 5-37 5-38 5-39 5-40 5-41 5-42 5-43 5-44 5-45 5-46 5-47 5-48 5-49 5-50 5-51 5-52 5-53 5-54 5-55 5-56 5-57 5-58 5-59 5-60 5-61 5-62 5-63 5-64 5-65 5-66 5-67 5-68 5-69 5-70 5-71 5-72 5-73 5-74 5-75 5-76 5-77 5-78 5-79 5-80 5-81 5-82 5-83 5-84 5-85 5-86 5-87 5-88 5-89 5-90 5-91 5-92 5-93 5-94 5-95 5-96 5-97 5-98 5-99 5-100



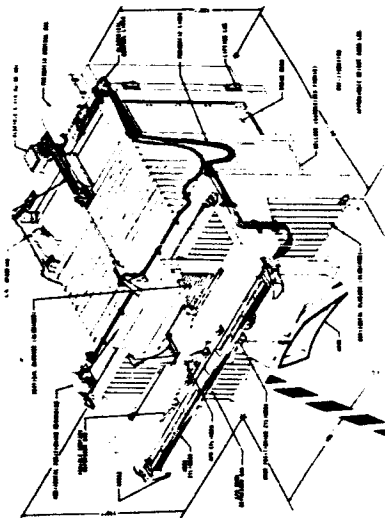
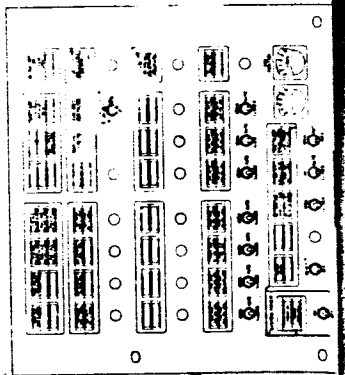
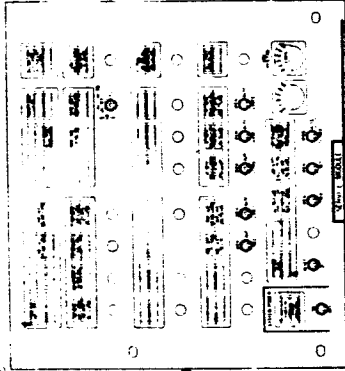
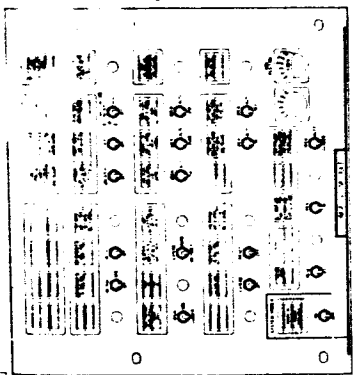
6-195-8







167-1



ENVIRONMENTAL CHAMBER

COMMAND MODULE (REF)

SERVICE MODULE (REF)

J-BOT NO 1 (REF)

WHEEL MOUNTED AIR/FAC SYSTEM (REF)

SERVICE AND CONTROL JACKET (REF)

MEDICAL OXYGEN DISTRIBUTOR (REF)

J-BOT NO 1 (REF)

3000.275

3748.555

3004.555

SM

LIEM



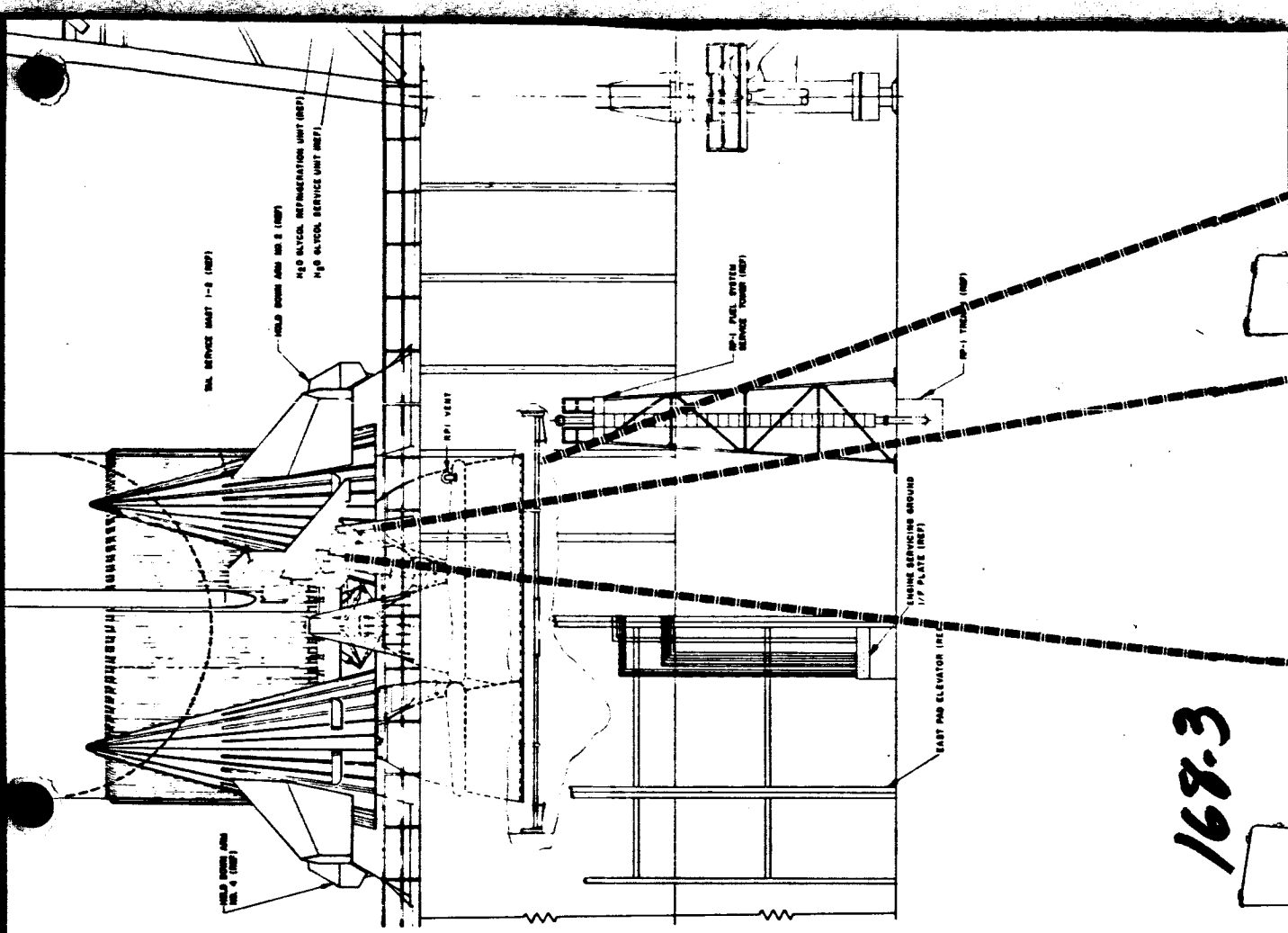
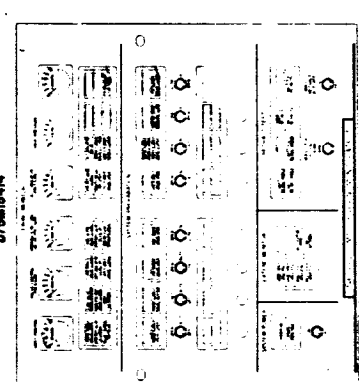
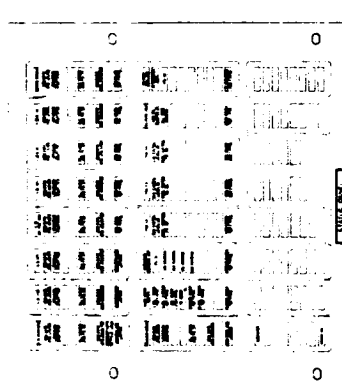
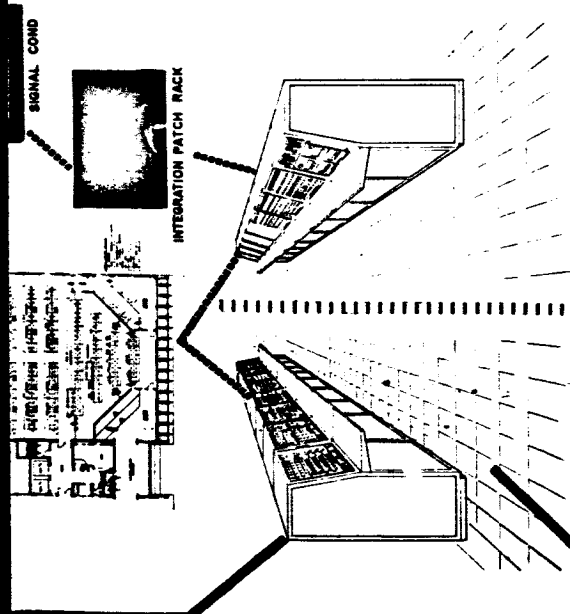












168.3

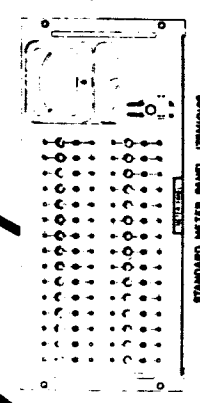
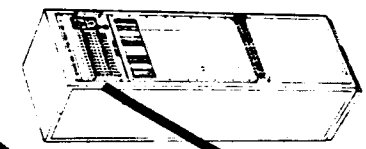
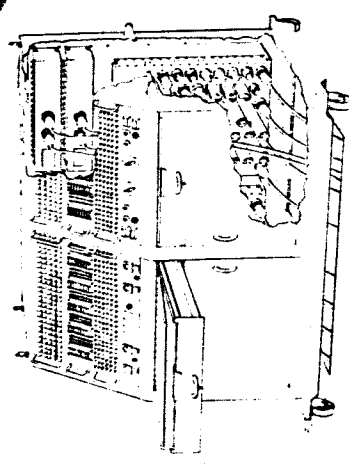


168-4  
h-291

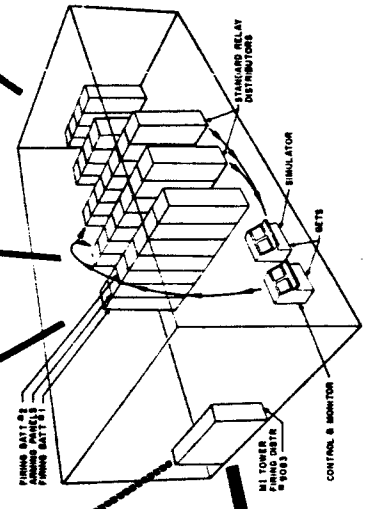
GROUND EQUIPMENT TEST SET  
E75M13286

STANDARD METER PANEL J75M1018

HYDRAULIC CHARGING UNIT (IB)  
REF ELECT  
INSTALLATION  
D7507087



COMPARTMENT BA  
75M13125



STANDARD 800-AMP SUPPLY A40M03708

HI LOWER  
POWER BATT'S

CONTROL & MONITOR

SIMULATOR

STANDARD RELAY  
DETECTOR



SIGNAL COND.

COMPUTER

LEVEL 30

ALTIMETER  
(S-4E115) (REF)

6000 PSI 80G (REF)

6000 PSI 40G (REF)

30 PSI 80G (REF)

45 PSI 40G (REF)

6000 PSI 40G (REF)

HYDRAULIC NETWORK (REF)

HYDRAULIC NETWORK  
REF INSTALLATION

LEVEL 0

DYNAMIC  
INSTALLATION

LVG FILL (REF)

LVG VENT (REF)

LEVEL A (REF)

LEVEL B (REF)

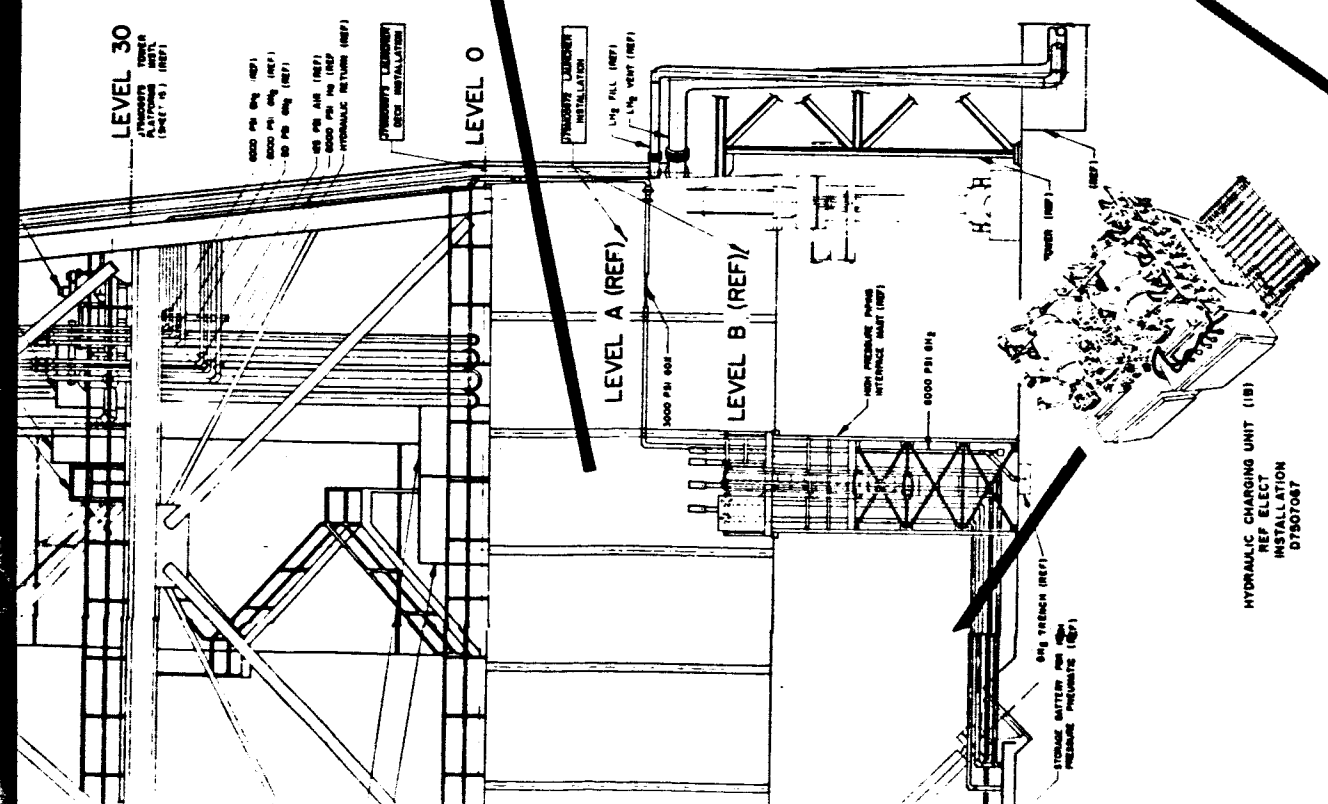
6000 PSI 60G

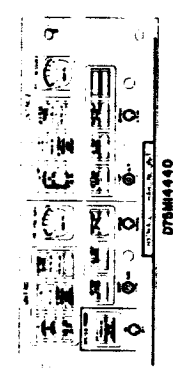
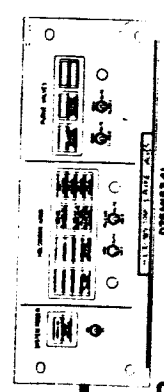
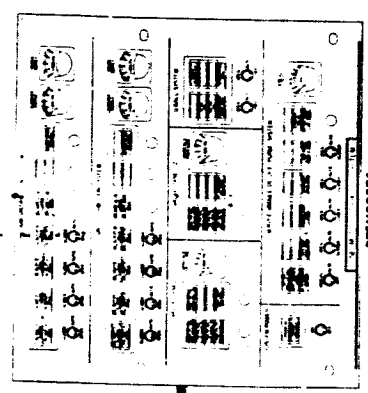
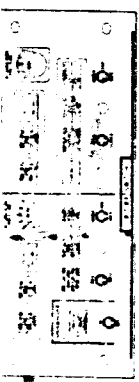
6000 PSI 80G

HIGH PRESSURE PUMP  
INTERFACE UNIT (REF)

84G TRENCH (REF)

STORAGE BATTERY FOR HIGH  
PRESSURE PRELUARTE (REF)

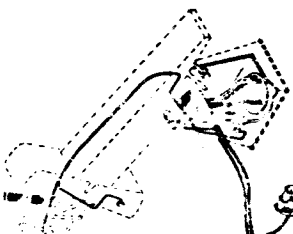
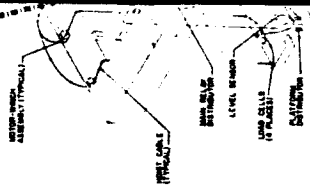




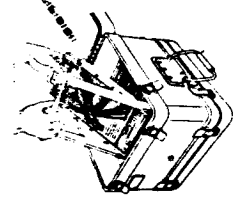
LAUNCHER TEST SET  
E75M4008



FOR ENGINE SERVICE PLATFORM  
E75M13570

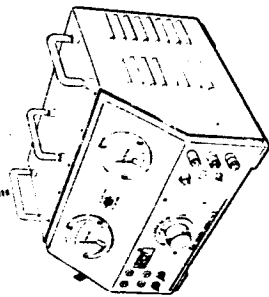


LAUNCHER TEST SET  
E75M4008



BOTTOM VIEW  
UNIVERSAL PATCH BOARD  
TEST SET  
E75M9032

TYPICAL INSTALLATION  
TEST SET  
E75M0768



PORTABLE DC POWER SUPPLY  
E75M8318

Figure 6-5 Mobile Launcher ESE Details

168-5



6-4. ESE CABLE INTERCONNECT DIAGRAMS.

Cable interconnect diagrams for Electrical Support Equipment are on pages 6-170 through 6-215 following. These diagrams were recently revised and are scheduled for further revision before the next issue of this document.

Also included here are four pages of Cable Interconnect and Electrical Schematic Criteria (6-216 through 6-219).







































FIGURE 6-6  
CABLE INDEX

FIGURE 6-6  
CABLE INDEX  
CABLE LAUNCH  
EQUIPMENT, ICSB

NOTES:  
FOR APPLICABLE REFER TO FIG. 2.

REF DESCRIPTION	CABLE ASSEMBLY	CABLE ASSEMBLY	BULK CABLE	CONNECTION	CONNECTION	CONNECTION	REF DESCRIPTION	CABLE ASSEMBLY	CABLE ASSEMBLY	BULK CABLE	CONNECTION	CONNECTION	CONNECTION
662187	D75M1007-48	D75M1007-48	D75M1007-48				662187	D75M1007-48	D75M1007-48	D75M1007-48			
662188							662188						
662189							662189						
662190							662190						
662191							662191						
662192							662192						
662193							662193						
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662195							662195						
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662197							662197						
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662200							662200						
662201							662201						
662202							662202						
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662250							662250						
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662299							662299						
662300							662300						

Figure 6-6 Cable Interconnect Diagrams (Continued)











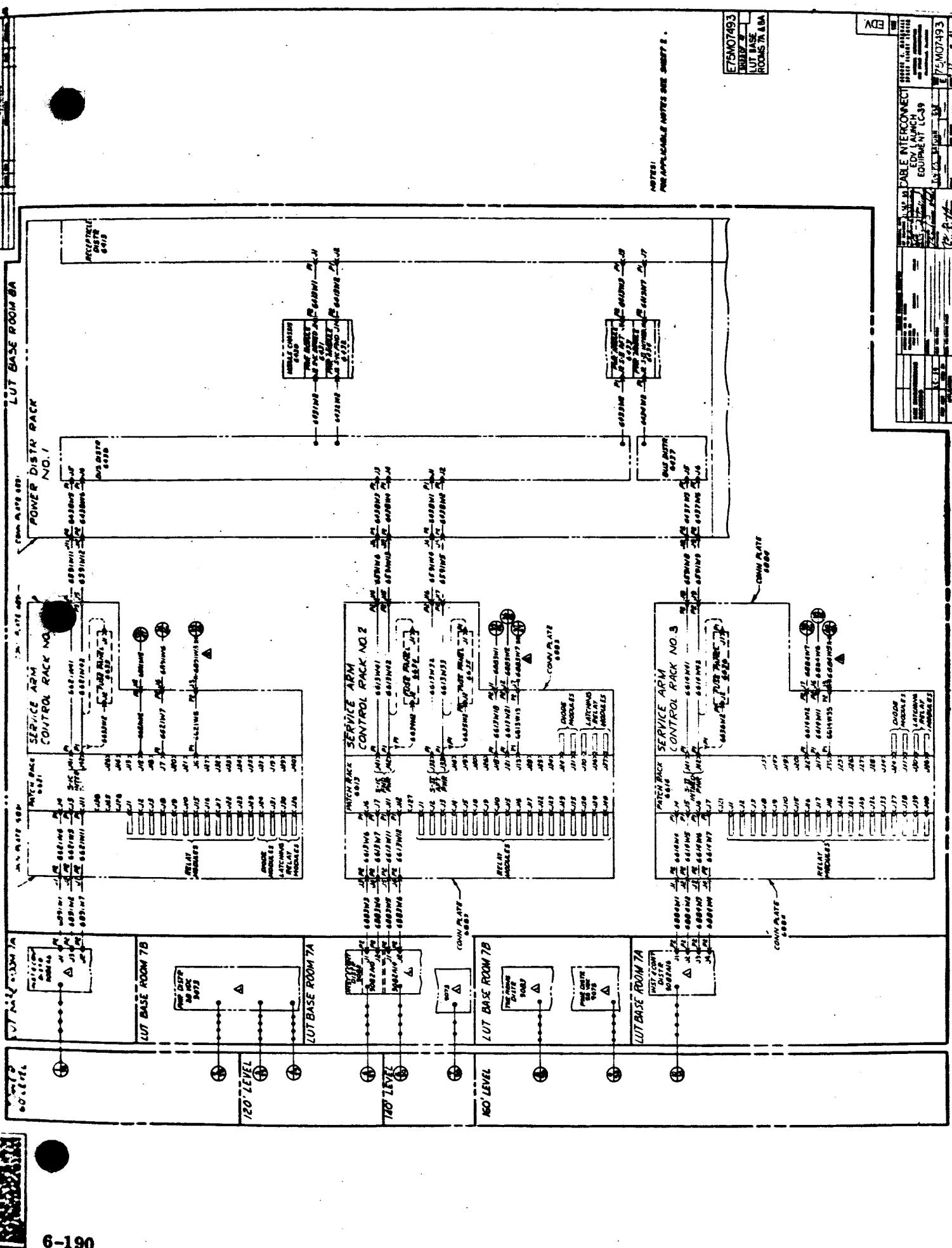


Figure 6-6 Cable Interconnect Diagrams (Continued)

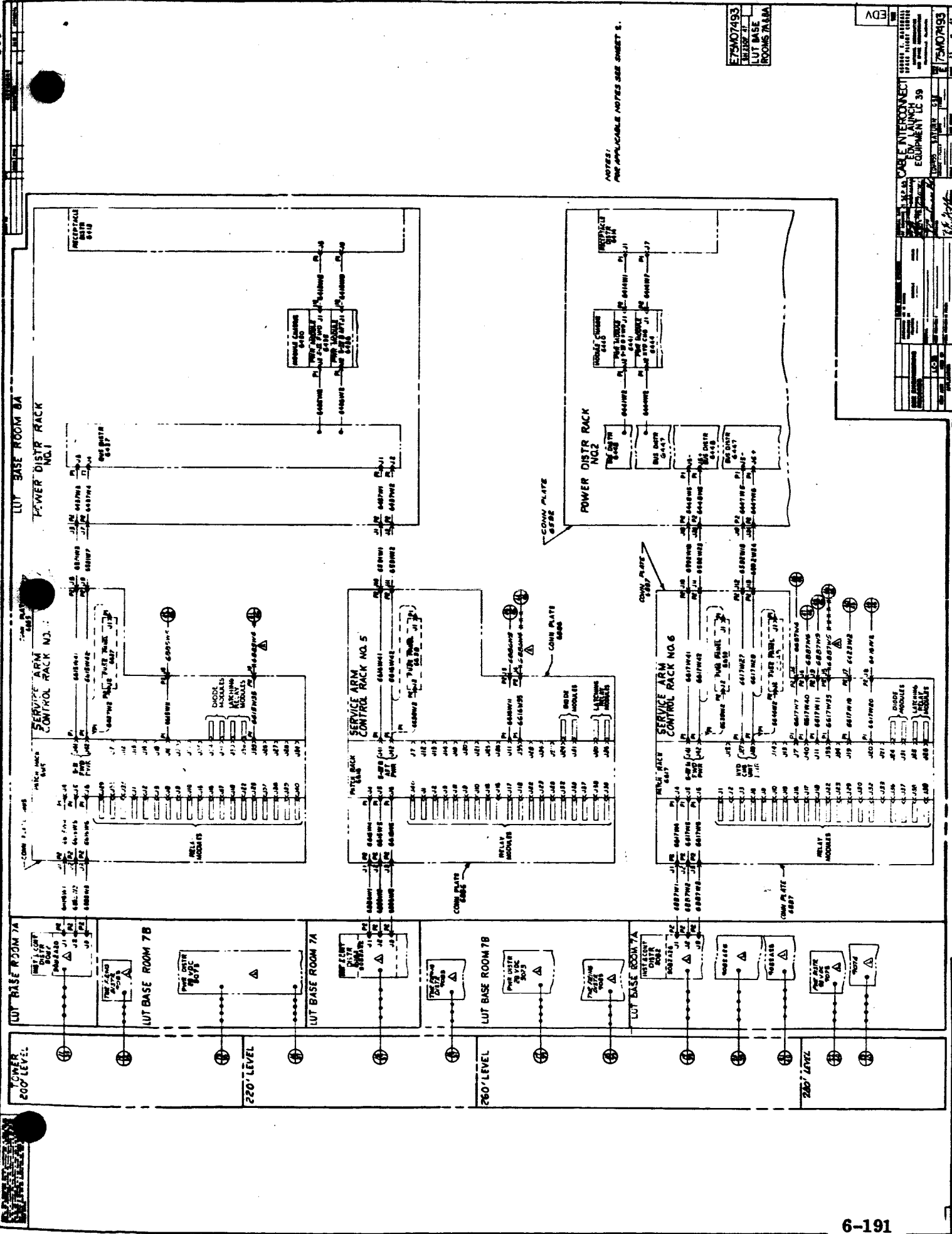
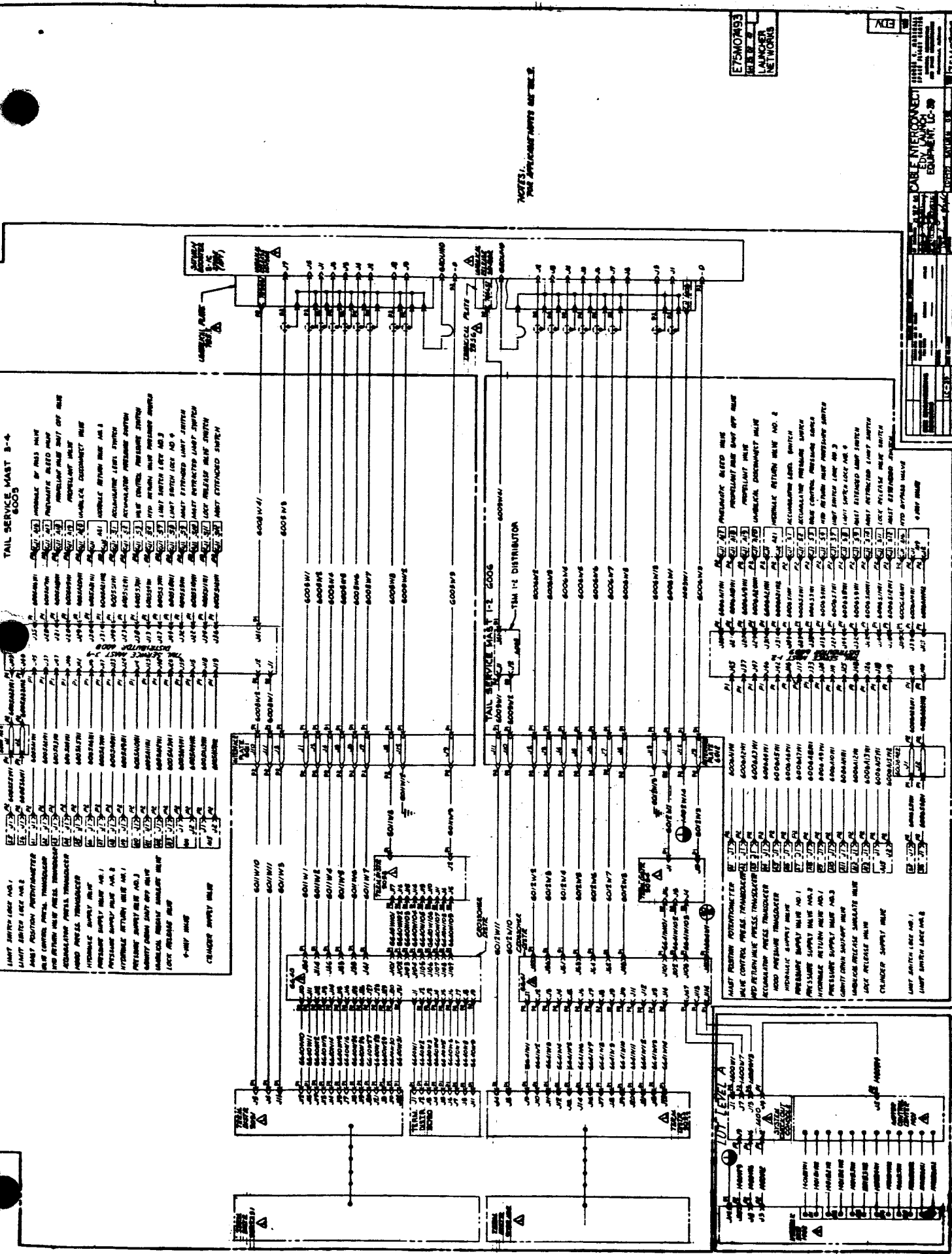


Figure 6-6 Cable Interconnect Diagrams (Continued)





E75MOW93  
LAUNCHER NETWORKS

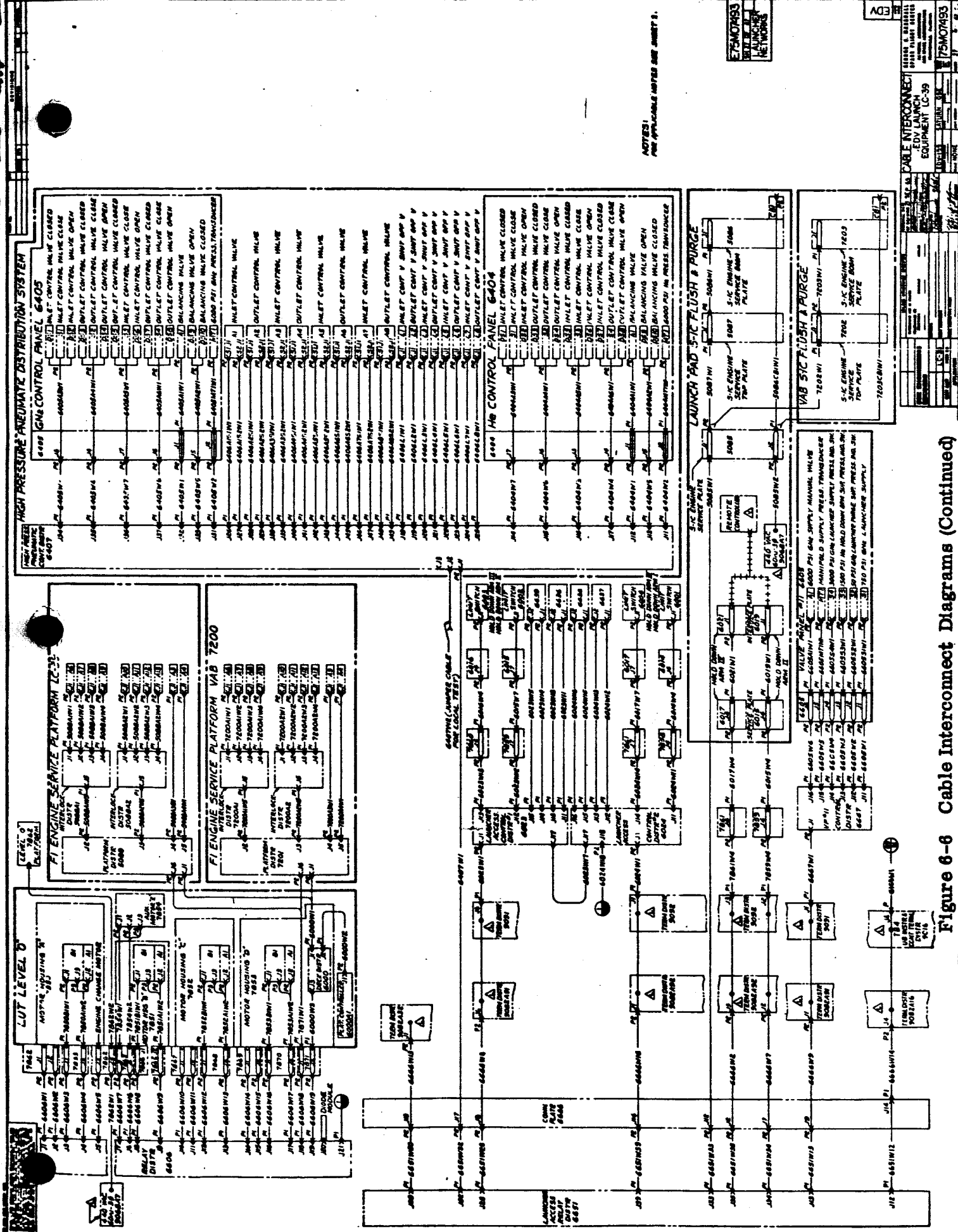
EDV  
CABLE INTERCONNECT EQUIPMENT CO.  
E75MOW93

NOTES:  
THE APPLICATIONS ARE SET.

Figure 6-6 Cable Interconnect Diagrams (Continued)







NOTES:  
FOR APPLICABLE NOTES SEE SHEET 1.

EDV LAUNCH EQUIPMENT LC-99

EDV	LAUNCH EQUIPMENT LC-99	DATE	BY
75M07493			

Figure 6-6 Cable Interconnect Diagrams (Continued)











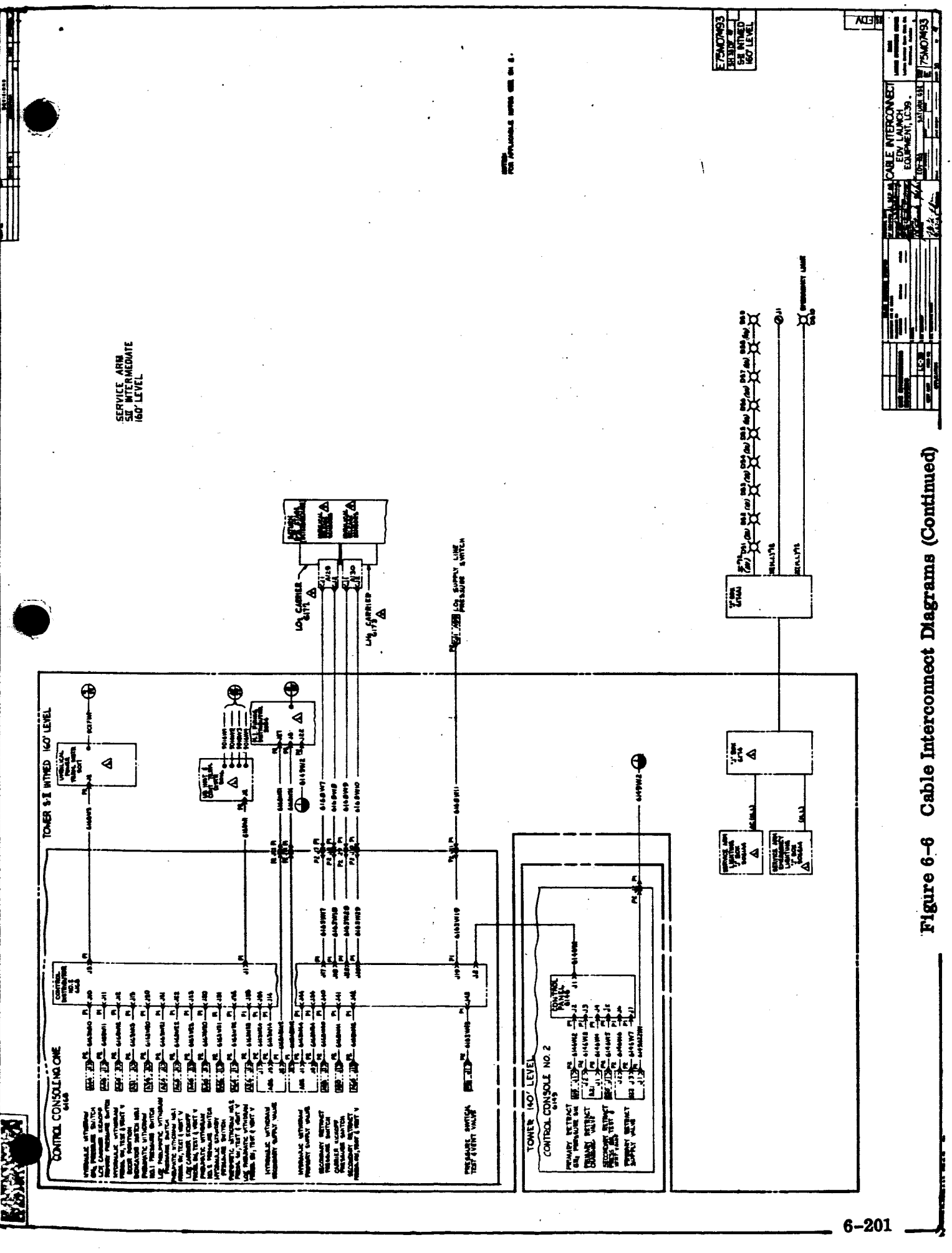


Figure 6-6 Cable Interconnect Diagrams (Continued)

E75M07953		EDV	
REVISION	DATE	BY	CHKD
58	1/60		
160 LEVEL			
CABLE INTERCONNECT EDV LAUNCH EQUIPMENT, LC99			
NO. OF SHEETS	TOTAL SHEETS	DATE	BY
1	1		
E75M07953			





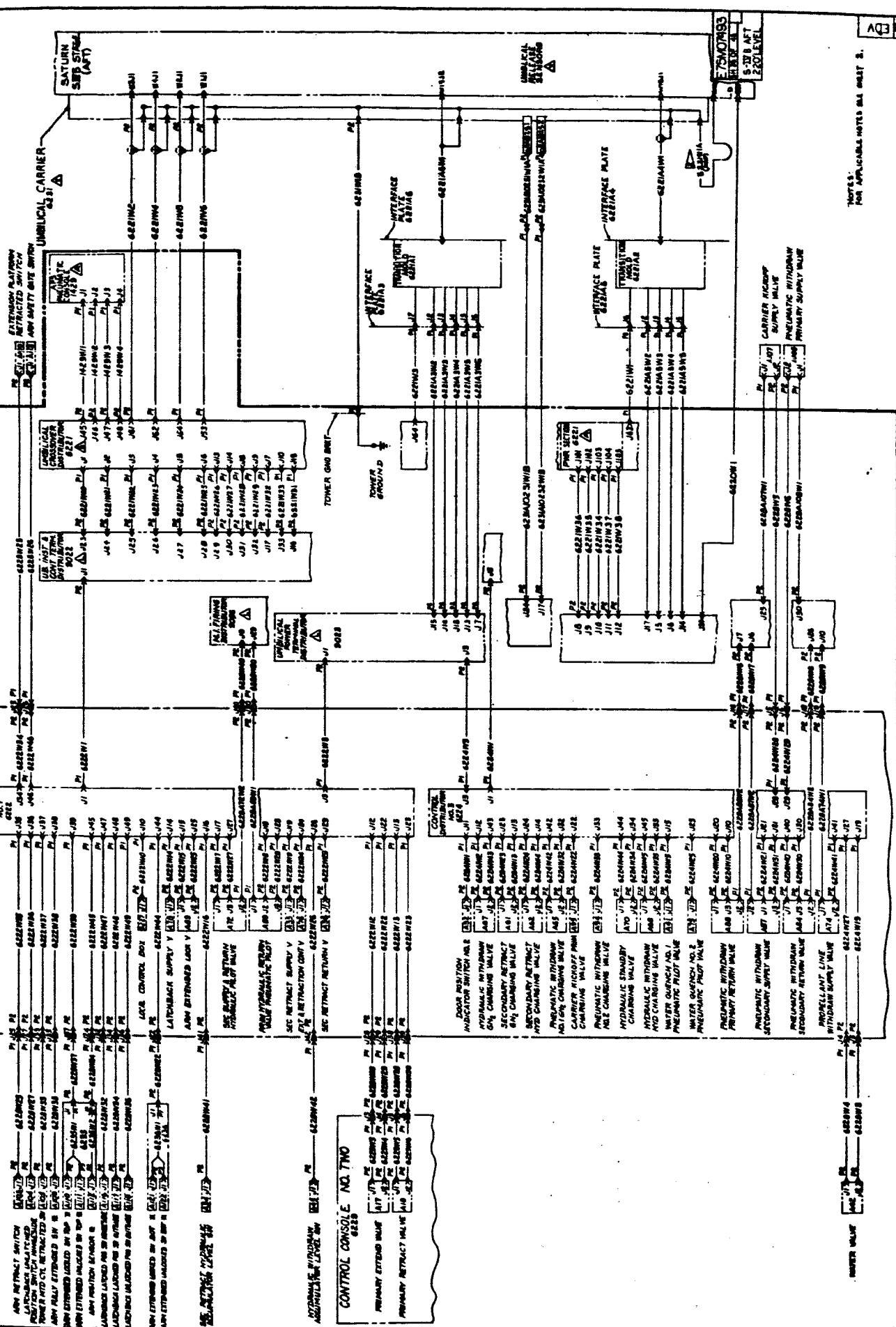


SERVICE ARM  
S-12B AFT  
220' LEVEL

TOWER S-12B AFT  
220' LEVEL

CONTROL CONSOLE NO. ONE  
4228

CONTROL CONSOLE NO. TWO  
4228



NOTES:  
FOR APPLICABLE NOTES SEE SHEET 2.

Figure 6-6 Cable Interconnect Diagrams (Continued)















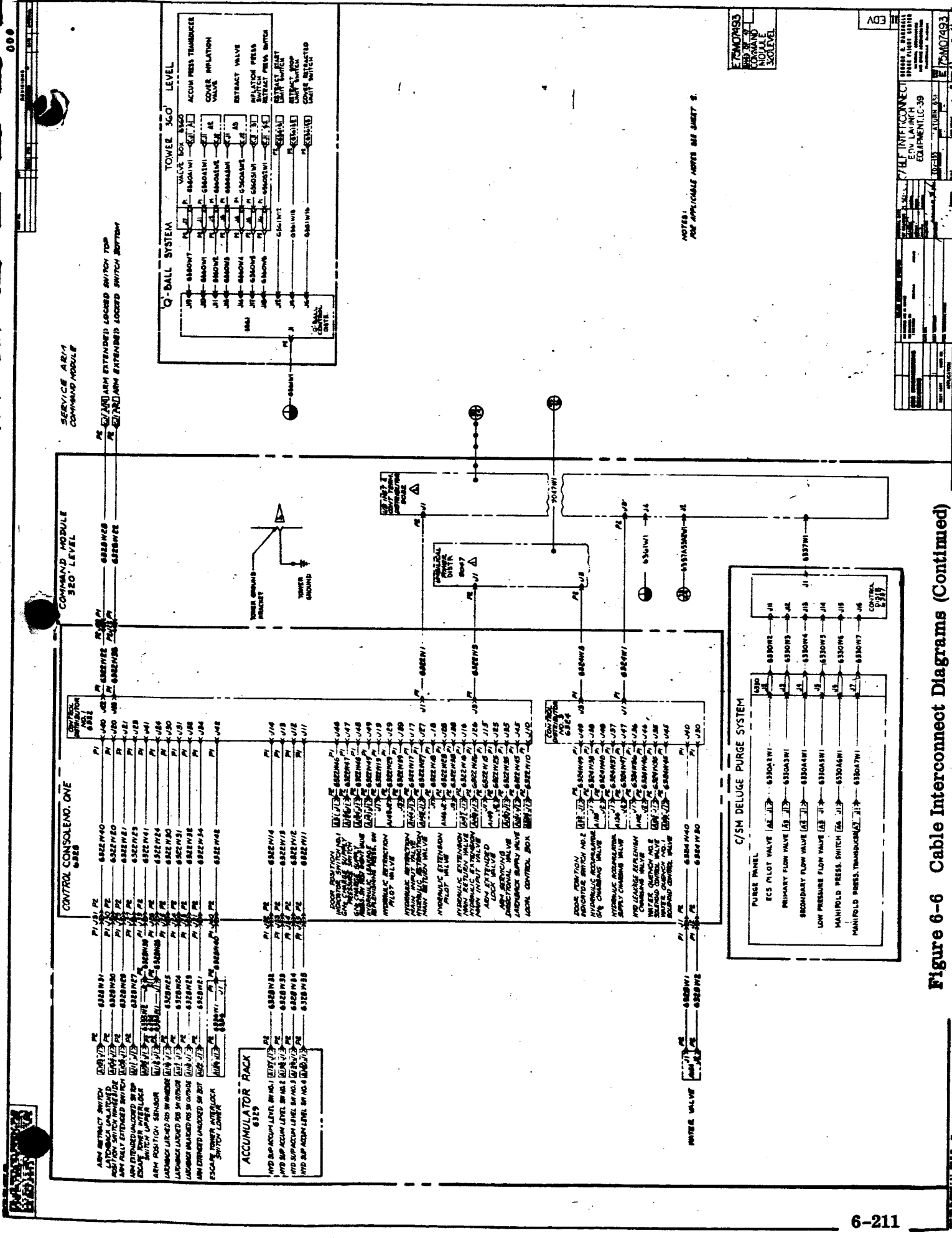


Figure 6-6 Cable Interconnect Diagrams (Continued)



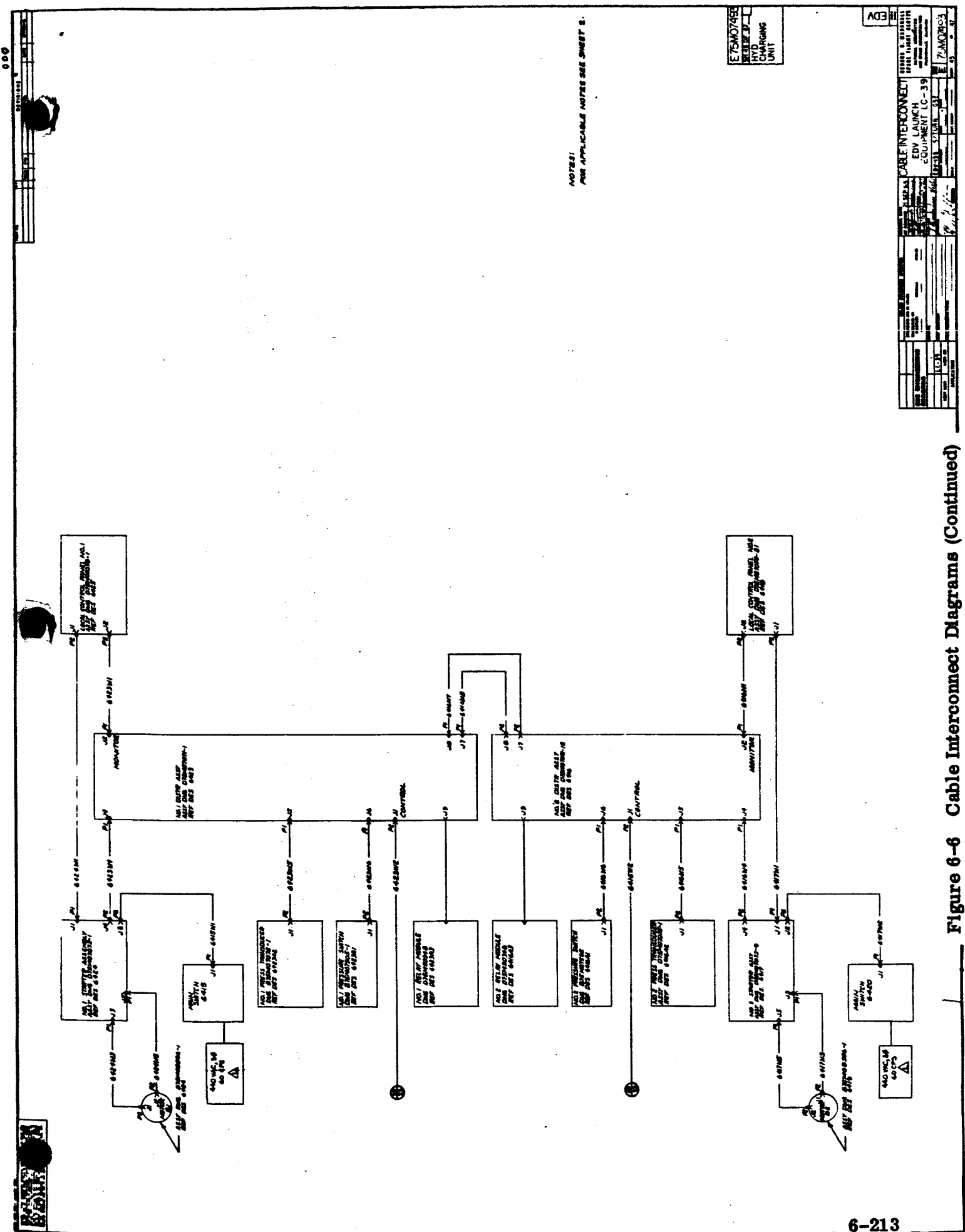


Figure 6-6 Cable Interconnect Diagrams (Continued)

E75M07495  
 HYD  
 CHARGING  
 UNIT

EDV	EDV LAUNCH EQUIPMENT LC-39	75M07495
REV	REV	REV
1	1	1
2	2	2
3	3	3
4	4	4
5	5	5
6	6	6
7	7	7
8	8	8
9	9	9
10	10	10
11	11	11
12	12	12
13	13	13
14	14	14
15	15	15
16	16	16
17	17	17
18	18	18
19	19	19
20	20	20
21	21	21
22	22	22
23	23	23
24	24	24
25	25	25
26	26	26
27	27	27
28	28	28
29	29	29
30	30	30
31	31	31
32	32	32
33	33	33
34	34	34
35	35	35
36	36	36
37	37	37
38	38	38
39	39	39
40	40	40
41	41	41
42	42	42
43	43	43
44	44	44
45	45	45
46	46	46
47	47	47
48	48	48
49	49	49
50	50	50
51	51	51
52	52	52
53	53	53
54	54	54
55	55	55
56	56	56
57	57	57
58	58	58
59	59	59
60	60	60
61	61	61
62	62	62
63	63	63
64	64	64
65	65	65
66	66	66
67	67	67
68	68	68
69	69	69
70	70	70
71	71	71
72	72	72
73	73	73
74	74	74
75	75	75
76	76	76
77	77	77
78	78	78
79	79	79
80	80	80
81	81	81
82	82	82
83	83	83
84	84	84
85	85	85
86	86	86
87	87	87
88	88	88
89	89	89
90	90	90
91	91	91
92	92	92
93	93	93
94	94	94
95	95	95
96	96	96
97	97	97
98	98	98
99	99	99
100	100	100

**LUT Cable Interconnect**

**Information**

**to be**

**Supplied**

**LCC Cable Interconnect**

**Information**

**to be**

**Supplied**

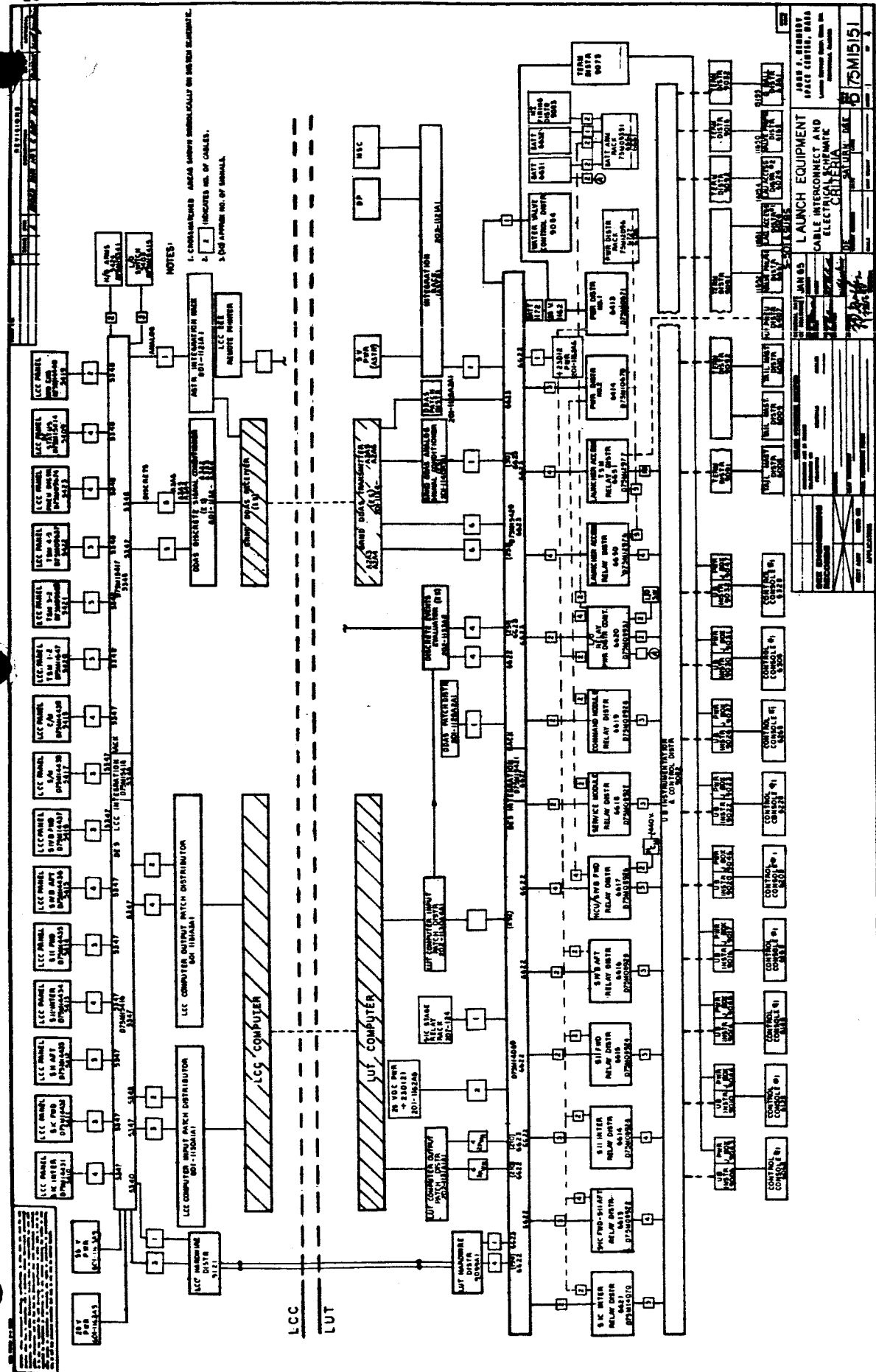


Figure 6-6 Cable Interconnect Diagrams (Continued)

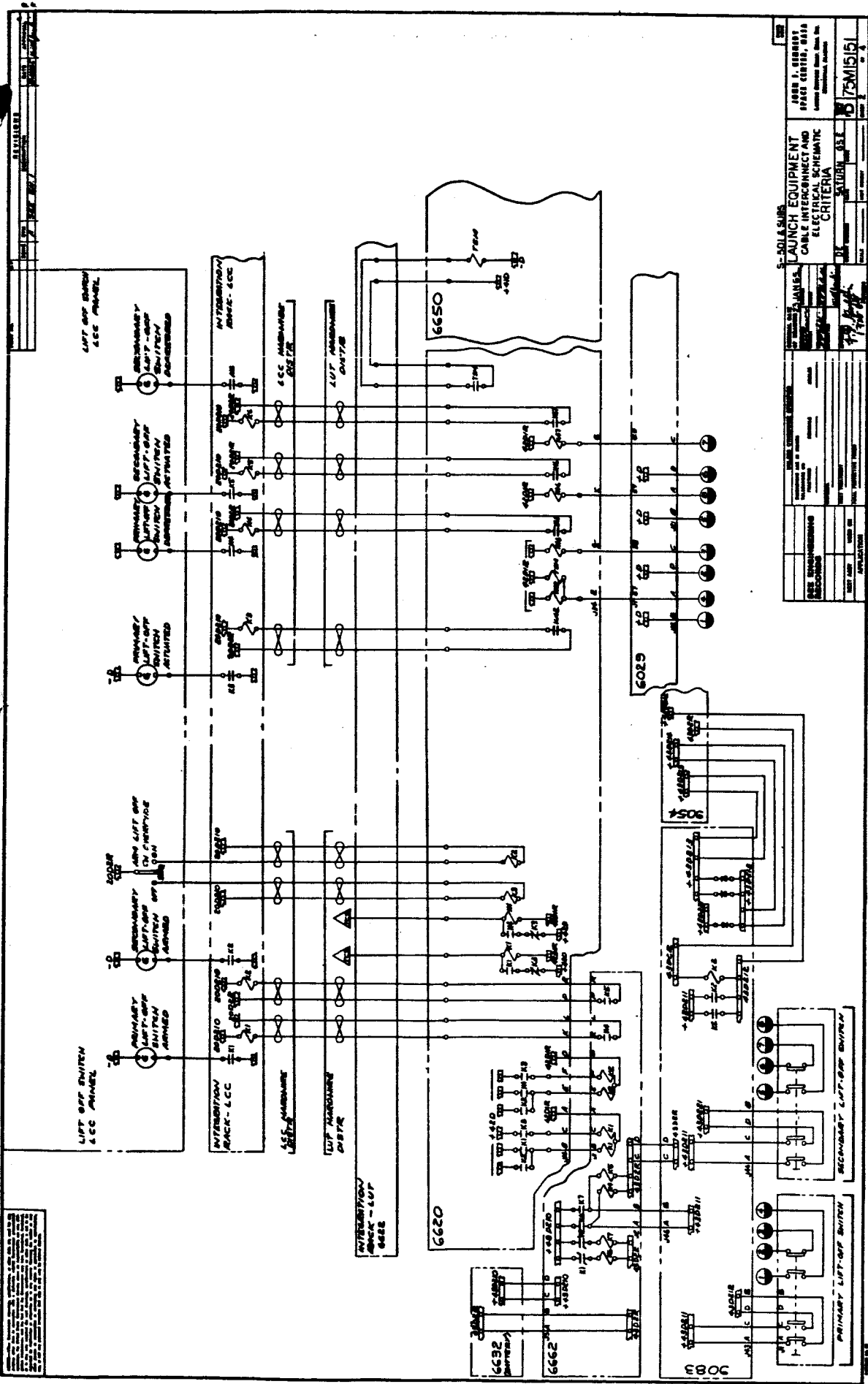


Figure 6-6 Cable Interconnect Diagrams (Continued)



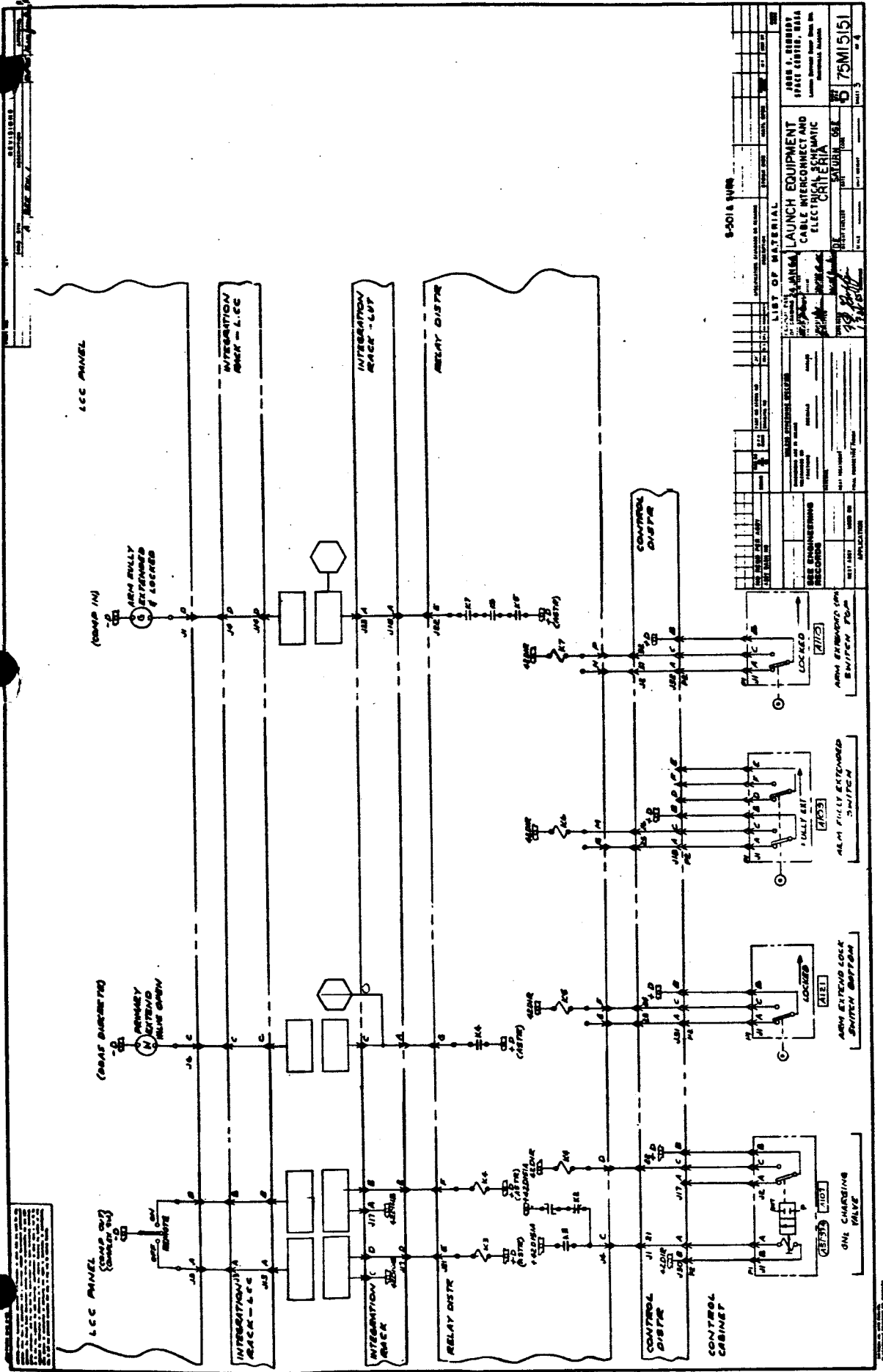
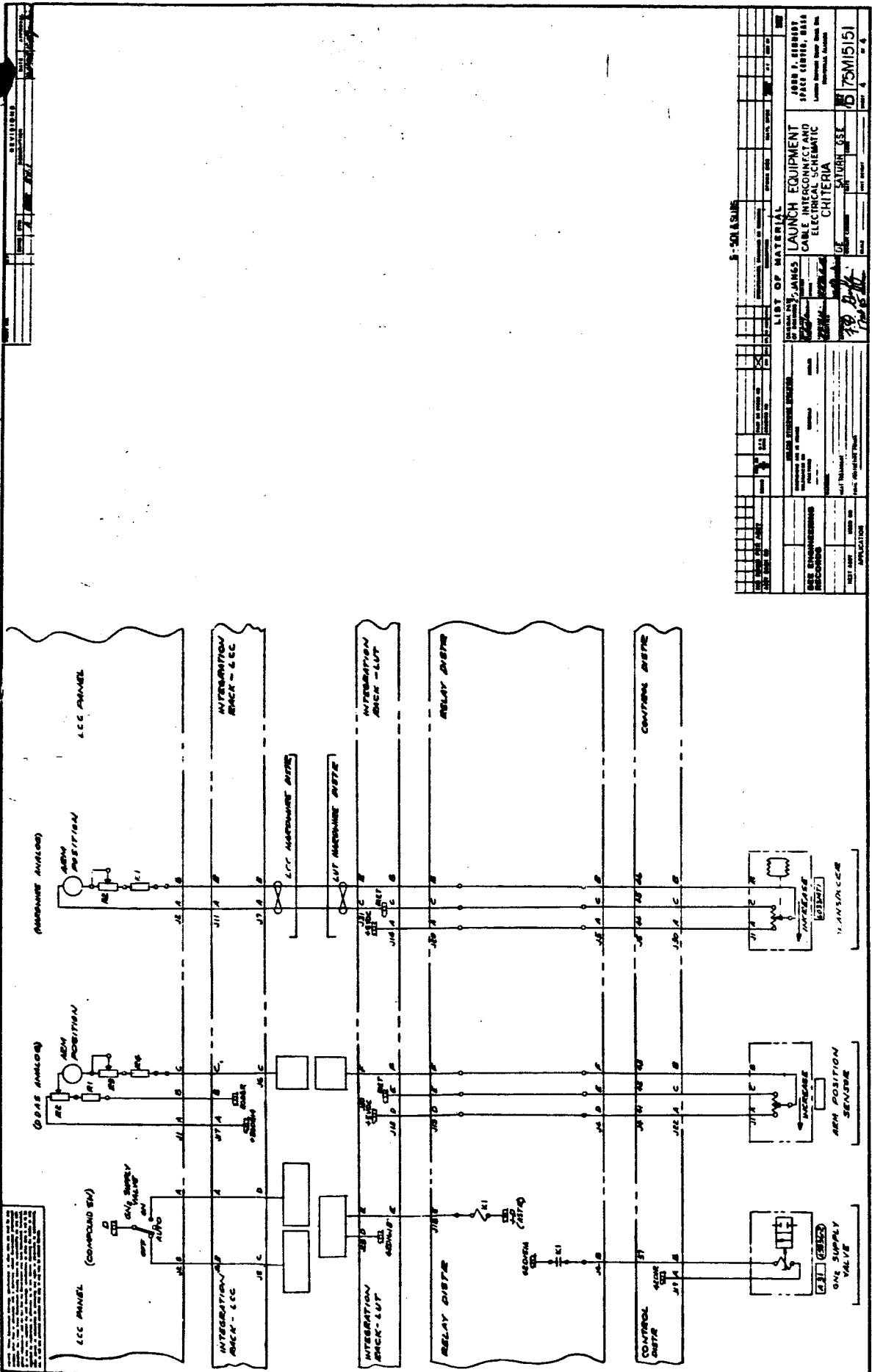


Figure 6-6 Cable Interconnect Diagrams (Continued)



6-501515

NO.	DESCRIPTION	QUANTITY	UNIT	DATE	BY	CHKD
1	...	...	...	...	...	...
2	...	...	...	...	...	...
3	...	...	...	...	...	...
4	...	...	...	...	...	...
5	...	...	...	...	...	...
6	...	...	...	...	...	...
7	...	...	...	...	...	...
8	...	...	...	...	...	...
9	...	...	...	...	...	...
10	...	...	...	...	...	...
11	...	...	...	...	...	...
12	...	...	...	...	...	...
13	...	...	...	...	...	...
14	...	...	...	...	...	...
15	...	...	...	...	...	...
16	...	...	...	...	...	...
17	...	...	...	...	...	...
18	...	...	...	...	...	...
19	...	...	...	...	...	...
20	...	...	...	...	...	...
21	...	...	...	...	...	...
22	...	...	...	...	...	...
23	...	...	...	...	...	...
24	...	...	...	...	...	...
25	...	...	...	...	...	...
26	...	...	...	...	...	...
27	...	...	...	...	...	...
28	...	...	...	...	...	...
29	...	...	...	...	...	...
30	...	...	...	...	...	...
31	...	...	...	...	...	...
32	...	...	...	...	...	...
33	...	...	...	...	...	...
34	...	...	...	...	...	...
35	...	...	...	...	...	...
36	...	...	...	...	...	...
37	...	...	...	...	...	...
38	...	...	...	...	...	...
39	...	...	...	...	...	...
40	...	...	...	...	...	...
41	...	...	...	...	...	...
42	...	...	...	...	...	...
43	...	...	...	...	...	...
44	...	...	...	...	...	...
45	...	...	...	...	...	...
46	...	...	...	...	...	...
47	...	...	...	...	...	...
48	...	...	...	...	...	...
49	...	...	...	...	...	...
50	...	...	...	...	...	...
51	...	...	...	...	...	...
52	...	...	...	...	...	...
53	...	...	...	...	...	...
54	...	...	...	...	...	...
55	...	...	...	...	...	...
56	...	...	...	...	...	...
57	...	...	...	...	...	...
58	...	...	...	...	...	...
59	...	...	...	...	...	...
60	...	...	...	...	...	...
61	...	...	...	...	...	...
62	...	...	...	...	...	...
63	...	...	...	...	...	...
64	...	...	...	...	...	...
65	...	...	...	...	...	...
66	...	...	...	...	...	...
67	...	...	...	...	...	...
68	...	...	...	...	...	...
69	...	...	...	...	...	...
70	...	...	...	...	...	...
71	...	...	...	...	...	...
72	...	...	...	...	...	...
73	...	...	...	...	...	...
74	...	...	...	...	...	...
75	...	...	...	...	...	...
76	...	...	...	...	...	...
77	...	...	...	...	...	...
78	...	...	...	...	...	...
79	...	...	...	...	...	...
80	...	...	...	...	...	...
81	...	...	...	...	...	...
82	...	...	...	...	...	...
83	...	...	...	...	...	...
84	...	...	...	...	...	...
85	...	...	...	...	...	...
86	...	...	...	...	...	...
87	...	...	...	...	...	...
88	...	...	...	...	...	...
89	...	...	...	...	...	...
90	...	...	...	...	...	...
91	...	...	...	...	...	...
92	...	...	...	...	...	...
93	...	...	...	...	...	...
94	...	...	...	...	...	...
95	...	...	...	...	...	...
96	...	...	...	...	...	...
97	...	...	...	...	...	...
98	...	...	...	...	...	...
99	...	...	...	...	...	...
100	...	...	...	...	...	...

LIST OF MATERIAL  
 LAUNCH EQUIPMENT  
 CABLE INTERCONNECT AND  
 ELECTRONIC  
 CRITERIA

0007 J. BIRNEY  
 1967 08 01  
 15 75M15151

Figure 6-6 Cable Interconnect Diagrams (Continued)

5.

REFERENCE DESIGNATION NUMBERS VS. ASSY DRAWING NUMBERS.

The electrical reference designation numbers in Table 6-4 are assigned to ESE assemblies or to assemblies carrying ESE cable. The drawing given for each assembly should call out other drawings showing any assembly or installation information required.

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS  
ASSY DRAWING NUMBERS**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
5085	S-IC Engine Servicing Plate	J75M09312
5086	S-IC Engine Servicing Boom Plate	J75M09312-3A
5087	S-IC Engine Servicing Top Plate	J75M09312-1A
5088	S-IC Engine Servicing Platform (Pad)	*E75M51238
5089	F-1 Engine Skirt Instl. Distr.	E75M13209
5347	LCC Integration Rack Assy #1	D75M15416
5348	LCC Integration Rack Assy #2	D75M15417
5349	LCC Cable Assy Installation	D75M15419
6000	Deck Distributor, F-1 Engine Servicing Platform	E75M13211
6000A1	Platform Controller, F-1 Engine Servicing Platform	D75M13570
6005	Tail Service Mast 3-4	*J75M12019
6006	Tail Service Mast 1-2	*J75M12023
6007	Tail Service Mast 3-2	*J75M12027
6008	Control Distributor, TSM 3-4	F75M12017
6009	Control Distributor, TSM 1-2	F75M12025
6010	Control Distributor, TSM 3-2	F75M12029
6011	Interface Plate, TSM 3-4	J75M11882
6012	Interface Plate, TSM 1-2	J75M11902
6013	Interface Plate, TSM 3-2	J75M11932
6014	Interface Plate, Holddown Arm #1	J75M05972-677
6015	Interface Plate, Holddown Arm #2	J75M05972-681
6016	Interface Plate, Holddown Arm #3	J75M05972-678
6017	Interface Plate, Holddown Arm #4	J75M05972-682
6019	Service Plate, Holddown Arm #2	E75M06443
6021	Service Plate, Holddown Arm #4	E75M06443
6023	Launcher Accessories Control Distributor #1	F75M14254
	* Electrical Installation Drawing	

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6024	Launcher Accessories Control Distributor #2	F75M11651
6026	Primary Liftoff Switch, Holddown Arm #2	J75M12231
6027	Secondary Liftoff Switch, Holddown Arm #2	J75M12231
6028	Liftoff Switch Pneumatic Assy, Holddown Arm #2	*J75M11078
6029	Liftoff Switch Distr. Assy, Holddown Arm #2	D75M11083
6031	Primary Liftoff Switch, Holddown Arm #4	J75M12231
6032	Secondary Liftoff Switch, Holddown Arm #4	J75M12231
6033	Liftoff Switch Pneumatic Assy, Holddown Arm #4	*J75M11078
6034	Liftoff Switch Distributor Assy, Holddown Arm #4	D75M11084
6062	Control Distr. #1	Assy. J75M10030 Wiring A75M10558
6063	Control Distr. #2	Assy. J75M10030 Wiring A75M10557
6064	Control Distr. #3	Assy. J75M10030-3 Wiring A75M10561
6067	Local Control Unit, S-IC Int.	Assy. J75M10898 Instl. J75M07578-1
6068	Control Console No. 1, S-IC Intertank	*J75M08751
6070	Reconnect Mech. S-IC Intertank	J75M08754
6071	Umb. Carrier, S-IC Intertank	Cable Assy. D75M09071
6072	LOX Carrier, S-IC Intertank	Cable Assy. D75M09071
6073	LH <sub>2</sub> Carrier, S-IC Intertank	Cable Assy. D75M09071
6074	Service Arm Power "J" Box	Cable Assy. D75M09071
	*Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6075 6076 6122	Hinge Upper, S-IC Int. Hinge Lower, S-IC Int. Control Distr. #1	J75M08260-1 J75M08261-1 Assy.
6124	Control Distr. #3	J75M10030-5 Wiring A75M10559 Assy. J75M10030-7 Wiring A75M10562 Assy.
6127	Local Control Unit, S-IC FWD	J75M10898 *J75M07578-3
6128 6132	Control Console No. 1 Umb. Carrier, S-IC Fwd	*J75M08746 Cable Assy. D75M09074
6133	LOX Carrier, S-IC Fwd	Cable Assy. D75M09074
6134	LH <sub>2</sub> Carrier, S-IC Fwd	Cable Assy. D75M09074
6135	Service Arm, Power "J" Box	Cable Assy. D75M09074
6136 6137 6142	Hinge Upper, S-IC Fwd Hinge Lower, S-IC Fwd Control Distr. #1	J75M08260-1 J75M08261-1 Assy.
6144	Control Distr. #3	J75M10030-9 Wiring A75M10560 Assy. J75M10030-11 Wiring A75M10563 Assy.
6146	Console No. 2 "J" Box	J75M07975 Harness D75M11336
* Electrical Installation Drawing		

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6147	Local Control Unit, S-II Aft	Assy. J75M10898
6148	Control Console No. 1	*J75M07578-5
6149	Control Console No. 2	*J75M08741
6151	Umb. Carrier, S-II Aft	*J75M07976 Cable Assy. D75M09077
6152	LOX Carrier, S-II Aft	Cable Assy. D75M09077
6153	LH <sub>2</sub> Carrier, S-II Aft	Cable Assy. D75M09077
6154	Service Arm Power "J" Box	Cable Assy. D75M09077
6155	Hinge Upper, S-II Aft	J75M08260-1
6156	Hinge Lower, S-II Aft	J75M08261-1
6162	Control Distr. #1	Assy. J75M10030-13 Wiring A75M10546
6163	Control Distr. #2	Assy. J75M10030-15 Wiring A75M10550
6164	Control Distr. #3	Assy. J75M10030-17 Wiring A75M10554
6165	Valve Panel No. 12 Control Distr.	Assy. F75M11650 Wiring A75M11644
6167	Valve Panel #12	J75M05958
6168	Control Console #1	J75M08735
6170	Local Control Unit, S-II Int.	Assy. J75M10898 *J75M07578-7

\* Electrical Installation Drawing

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6171	Umb. Carrier, S-II Int.	Cable Assy. D75M09080
6172	LOX Carrier, S-II Int.	Cable Assy. D75M09080
6173	LH <sub>2</sub> Carrier, S-II Int.	Cable Assy. D75M09080
6174	Service Arm Power "J" Box	Cable Assy D75M09080
6175	Hinge Upper, S-II Int.	J75M08260-1
6176	Hinge Lower, S-II Int.	J75M08261-1
6202	Control Distr. #1	Assy. J75M10030-19
6203	Control Distr. #2	Wiring A75M10547 Assy. J75M10030-21
6204	Control Distr. #3	Wiring A75M10551 Assy. J75M10030-23
6206	Console No. 2 "J" Box	Wiring A75M10555 Assy. J75M07975
6207	Local Control Unit, S-II Fwd	Harness D75M11366 Assy. J75M10898
6208	Control Console #1	*J75M07578-9
6209	Control Console #2	*J75M08728
6212	Umb. Carrier, S-II Fwd	*J75M07977 Cable Assy. D75M09084
6213	LOX Carrier, S-II Fwd	Cable Assy. D75M09084
6214	LH <sub>2</sub> Carrier, S-II Fwd	Cable Assy. D75M09084

\* Electrical Installation Drawing

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6215	Service Arm Power "J" Box	Cable Assy. D75M09084
6216	Hinge Upper, S-II Fwd	J75M08260-1
6217	Hinge Lower, S-II Fwd	J75M08261-1
6222	Control Distr. #1	Assy. J75M10030-25 Wiring A75M09346
6223	Control Distr. #2	Assy. J75M10030-27 Wiring A75M09347
6224	Control Distr. #3	Assy. J75M10030-29 Wiring A75M09348
6226	Console #2 "J" Box	Assy J75M07975 Harness D75M11366
6227	Local Control Unit, S-IVB Aft	Assy. J75M10898 *J75M07578-11
6228	Control Console #1	*J75M08726
6229	Control Console #1	J75M07978
6231	Umb. Carrier, S-IVB Aft	Cable Assy. D75M09088
6232	LOX Carrier, S-IVB Aft	Cable Assy. D75M09088
6233	LH <sub>2</sub> Carrier, S-IVB Aft	Cable Assy. D75M09088
6234	Service Arm Power "J" Box	Cable Assy. D75M09088
6235	Hinge Upper, S-IVB Aft	J75M08260-1
6236	Hinge Lower, S-IVB Aft	J75M08261-1
* Electrical Installation Drawing		



**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6262	Control Distr. #1	Assy. J75M10030-31 Wiring
6263	Control Distr. #2	A75M10548 Assy. J75M10030-33 Wiring
6264	Control Distr. #3	A75M10552 Assy. J75M10030-35 Wiring
6266	Console #2 "J" Box	A75M10556 Assy. J75M07975 Harness
6267	Local Control Unit, S-IVB Fwd	D75M11366 Assy. J75M10898
6268	Control Console #1	*J75M07578-13
6269	Control Console #2	*J75M08721
6271	Umb. Carrier, S-IVB Fwd	*J75M07979
6272	LOX Carrier, S-IVB Fwd	Cable Assy. D75M09092
6273	LH <sub>2</sub> Carrier, S-IVB Fwd	Cable Assy. D75M09092
6274	Service Arm Power "J" Box	Cable Assy. D75M09092
6275	Hinge Upper, S/M	J75M08260-1
6276	Hinge Lower, S/M	J75M08261-1
6291	Umb. Plate, Inst. Unit S-IVB Fwd	Cable Assy. D75M09092
* Electrical Installation Drawing		

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6302	Control Distr. #1	Assy. J75M10030-37 Wiring A75M10549
6303	Control Distr. #2	Assy. J75M10030-39 Wiring A75M10553
6304	Control Distr. #3	Assy. J75M10030-41 Wiring A75M10557
6306 6307	Console #2 "J" Box Local Control Unit, S/M	*J75M07975 Assy. J75M10898 *J75M07578-15
6308	Control Console #1	*J75M09055
6309	Control Console #2	*J75M09052
6311	Umb. Carrier, S/M	Cable Assy. D75M09096
6312	LOX Carrier, S/M	Cable Assy. D75M09096
6313	LH <sub>2</sub> Carrier, S/M	Cable Assy. D75M09096
6314	Service Arm Power "J" Box	Cable Assy. D75M09096
6315	Hinge Upper, C/M	J75M12139-1
6316	Hinge Lower, C/M	J75M12139-3
6322	Control Distr. #1	Assy. J75M10030-49 Wiring A75M12674
6324	Control Distr. #3  * Electrical Installation Drawing	Assy. J75M10030-51 Wiring A75M12675

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6325	Local Cont. Unit, C/M	Assy. J75M10898 *J75M07578-17
6326	Oxygen Conditioning Console	J75M11608
6327	Service Arm Lighting "J" Box	Cable Assy. D75M09100
6328	Control Console #1	*J75M08722
6329	Accumulator Bottles	J75M05974
6357	Environmental Chamber	*J75M07104
6360	Q-Ball Control Panel	D75M06150
6361	Q-Ball Cover Control Distr.	F75M13155
6401	Flush & Purge Boom	delete
6402	Flush & Purge Distr.	delete
6404	He Distribution Panel	J75M07022
6405	GN <sub>2</sub> Distribution Panel	J75M07021
6406	High Press Pneu. Distr. System	delete
6407	High Pressure Pneumatics Control Distr.	J75M11649
6413	Power Distr. S/A	J75M10671
6414	Power Distr. S/A	J75M10670
6415	Safety Sw.	D75M12973
6416	Distr. Assy Hyd Unit #2	D75M07071
6417	Starter Assy Hyd Unit #2	D75M07073
6418	Control Pnl Hyd Unit #2	D75M07070
6419	Motor Assy Hyd Unit #2	B75M08506
6420	Safety Sw. Hyd Unit #2	D75M12973
6421	Pwr Supply Launcher Acc.	delete
6422	Cable Installation Rack	J75M12096
6423	Distr. Assy Hyd Unit #1	D75M07071
6424	Starter Assy Hyd Unit #1	D75M07073
6425	Cont. Pnl. Assy Hyd Unit #1	D75M07070
6426	Motor Assy Hyd Unit #1	B75M08506
6427	Fuse Panel	delete
6428	Fuse Panel	delete
6429	Interface Plate, Launcher Pwr Distr	delete

\*Electrical Installation Drawing

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6430	Pwr Module, Launcher Pwr Distr. 2	delete
6431	Pwr Module Launcher Pwr Distr. 1	delete
6432	Pwr Module, Spare	delete
6433	Power Transfer Module Chassis	delete
6434	Pwr Transfer Unit Batt. Rack	delete
6604	Deluge Purge Panel	delete
6605	Valve Panel #11	J75M07433
6606	Engine Service Platform Relay Distr.	E75M13471
6611	Launch Aux. Relay Patch Distr.	
6612	Flush & Purge Power Supply	
6613	Service Arm Control Rack #2	D75M08922
6614	Service Arm Control Rack #3	D75M08923
6615	Service Arm Control Rack #4	D75M08924
6616	Service Arm Control Rack #5	D75M08925
6617	Service Arm Control Rack #6	D75M08926
6618	Service Arm Control Rack #7	D75M08927
6619	Service Arm Control Rack #9	D75M08928
6620	Service Arm Control Rack #8	D75M08931
6621	Relay Distr. (Spare)	F75M04681
6622	LUT Integration Rack #1	D75M14069
6623	LUT Integration Rack #2	D75M15420
6629	Meter Panel, Integration Rack	J75M10165
6636	Purge Valve #1	B75M14551-1
6637	Purge Valve #2	B75M14551-1
6638	Purge Valve #3	B75M14551-1
6639	Purge Valve #4	B75M14551-1
6640	Crossover Distr. S-IC, Room 13-A	delete
6641	Crossover Distr. S-IC, Room 5-A	delete
6650	Launcher Access. Relay Rack #1	J75M12976
6651	Launcher Access. Relay Rack #2	J75M12977
6652	Hydraulic Charging Unit Electric Distr. #2	D75M07071
6653	Hydraulic Charging Unit Starter Panel #2	D75M07073
6654	Hydraulic Charging Unit Control Panel #2	D75M07070
6655	Hydraulic Charging Unit Pump Motor #2	Cable Assy. D75M07072
	* Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6656	Connector Plate, Relay Rack	Assy. . J75M09711
6657	Connector Plate, Relay Rack	Assy. J75M09711
6658	Connector Plate, Relay Rack	Assy. J75M09711
6659	Connector Plate, Relay Rack	Assy. J75M09711
6660	Connector Plate, Relay Rack	Assy. J75M09711
6661	Connector Plate, Relay Rack	Assy. J75M09711
6662	Connector Plate, Relay Rack	Assy. J75M09711
6663	Connector Plate, Relay Rack	Assy. J75M09711
6664	T. S. Mast Relay Distr. Int. Plate	Assy. J75M09711
6665	Launcher Access Distr. Int. Plate	Assy. J75M09711
6667	Valve Panel #11 Control Distr.	F75M11652
6700	Interface Plate, Launcher Power Rack	Assy. J75M09711-1
6701	Fuse Panel Assy	*J75M12096 Assy. J75M10163-1
6712	Power Module Frame Assy	*J75M12096
6713	Power Module R/ASTP Integration	D75M12301 Assy.
6714	Power Module, Launcher Accessories Distr. #1	*J75M12096 Assy.
6715	Power Module, Launcher Accessories Distr. #2	*J75M12096 Assy.
6716	Power Module Spare	*J75M12096 Assy. *J75M12096

\* Electrical Installation Drawing

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6717	Power Module GETS #1	Assy. *J75M12096
6718	Power Module GETS #2	Assy. *J75M12096
6726	Bus Distributor	*J75M12096
6727	Bus Distributor	*J75M12096
7200	F1 Engine Change Platform (VAB)	*J75M51342
7201	F1 ESP Platform Distr.	E75M13209
7202	S-IC Engine Service Top Plate (VAB)	J75M09312-5A
7203	S-IC Engine Service Boom Plate (VAB)	J75M09312-3A
7204	Distribution Box, S-II Engine Servicing Platform	B75M14680
7205	"J" Box Winch #1	B75M14683
7206	"J" Box Winch #2	B75M14683
7207	"J" Box Winch #3	B75M14683
7208	"J" Box Winch #4	B75M14683
7601	Tail Mast Test Set	E75M07658
7602	Launcher Accessories Test Set No. 2	J75M11087
7603	Portable Arm Control Console	J75M07542
7604	Portable Control Unit Distr.	delete
7605	Portable Control Unit Power Supply	
7606	Tower Test Set	E75M14137
7607	Portable Power Supply (Pneumatic)	
7650	Launcher Acc. Test Set No. 1	E75M14009
7656	GETS	E75M13286
7657	GETS Simulator	E75M13286
7658	Launcher Test Set	
7850	Platform Motor & Brake Assy #1	*J75M51413
7851	Platform Motor & Brake Assy #2	*J75M51413
7852	Platform Motor & Brake Assy #3	*J75M51413
7853	Platform Motor & Brake Assy #4	*J75M51413
7854	Platform Motor & Brake Assy #5	*J75M13064
7855	Umbilical Plate, TSM 3-4	J65B80001
7856	Umbilical Plate, TSM 1-2	J65B80002
7857	Umbilical Plate, TSM 3-2	J65B80003
	* Electrical Installation Drawing	

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
7858	Haunch Plate, Holddown Arm #1	J75M05972
7859	Haunch Plate, Holddown Arm #2	J75M05972
7860	Haunch Plate, Holddown Arm #3	J75M05972
7861	Haunch Plate, Holddown Arm #4	J75M05972
7862	LUT Level Interface Plate Aux Motor "A"	*J75M51413
7863	LUT Level Interface Plate Motor Brake "A"	*J75M51413
7865	LUT Level Interface Plate Aux Motor "B"	*J75M51413
7866	LUT Level Interface Plate Motor & Brake "B"	*J75M51413
7867	LUT Level Interface Plate Aux Motor "C"	*J75M51413
7868	LUT Level Interface Plate Motor & Brake "C"	*J75M51413
7869	LUT Level Interface Plate Aux Motor "D"	*J75M51413
7870	LUT Level Interface Plate Motor & Brake "D"	*J75M51413
7872	LUT Level "0" Platform Elect. System	*J75M51639
7950	Distributor Box #1, Platform Transporter Control North	B75M14679-1
7951	Distributor Box #2, Platform Transporter Control North	B75M14679-2
7952	Portable Controller, Platform Transporter Control	B75M14682
7953	Distributor Box #1, Platform Transporter Control South	B75M14679-3
7954	Distributor Box #2, Platform Transporter Control South	B75M14679-4
7955	Portable Controller, Platform Transporter Control South	B75M14682
7956	Motor/Winch Bulkhead, Platform Transporter Control #1	J75M14891
7957	Motor/Winch Bulkhead, Platform Transporter Control #2	J75M14891
9001	LUT - Pad Interface, Power	F75M12270
9002	LUT - Crawler Interface, PWR	F75M05121
9006	UB. Instr. & Control Term Distr	F75M05121
9008	UB. Instr. & Control Term Distr-80' (MSC)	F75M12270
9009	UB. Power Term Distr-80' (MSC)	F75M12270
9010	UB. Instr. & Control Term Distr-100'	F75M05121
	*Electrical Installation Drawing	

**Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)**

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
9013	UB. Power Term Distr-120' (S-IC FWD)	F75M05121
9014	UB. Instr & Control Term Distr - 140'	F75M05121
9016	UB. Instr & Control Term Distr - 160'	F75M05121
9017	UB. Power Term Distr - 160'	F75M05121
9020	UB. Instr & Control Term Distr-200'	F75M05121
9022	UB. Instr & Control Term Distr-220'	F75M05121
9023	UB. Power Term Distr-220'	F75M05121
9025	UB. Instr & Control Term Distr-260'	F75M05121
9026	UB. Instr & Control Term Distr-260'	F75M05121
9027	UB. Power Term Distr-260'	F75M05121
9028	UB. Instr & Control Term Distr-280'	F75M05121
9029	UB. Power Term Distr-280'	F75M05121
9030	UB. Instr & Control Term Distr-300'	F75M05121
9031	UB. Power Term Distr-300'	F75M05121
9032	UB. Instr & Control Term Distr-320'	F75M05121
9033	Prop. Loading A. C. PWR "J" Box-30'	F75M05121
9034	Prop. Loading A. C. PWR "J" Box-120'	F75M05121
9035	Prop. Loading A. C. PWR "J" Box-200'	F75M05121
9038	Crane	F75M05120
9042	UB. Power "J" Box 30' Tower Level	F75M05121
9043	UB. Power "J" Box 60' Tower Level	F75M05121
9044	UB. Power "J" Box 120' Tower Level	F75M05121
9045	UB. Power "J" Box 140' Tower Level	F75M05121
9046	UB. Power "J" Box 200' Tower Level	F75M05121
9047	UB. Power "J" Box 320' Tower Level	F75M05121
9048	OTV Cables, Phase III	F75M12270
9049	Measuring Acquisition Term Distrs.	F75M05121
9050	Measuring Coax Term. Distrs.	F75M05121
9051	UB. Instr & Control Distr-300' (MSC)	F75M12270
9052	Service Arm MI Term Distr-120' Level	F75M05121
9054	Service Arm MI Term Distr-160' Level	F75M05121
9055	Service Arm MI Term Distr-200' Level	F75M05121
9056	Service Arm MI Term Distr-220' Level	F75M05121
9057	Service Arm MI Term Distr-260' Level	F75M05121

\* Electrical Installation Drawing



Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY  
DRAWING NUMBERS (Continued)

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
9059	Service Arm MI Term Distr-300' Level	F75M05121
9060	Coax Distr - MSC. 300' Level	F75M12270
9061	Service Arm Lighting Junction Boxes	F75M05121
9062	Service Arm Emergency PWR. "J" Boxes	
9063	"J" Box, S-IC FWD Foamflex (120')	F75M12270
9064	"J" Box, S-II FWD Foamflex (200')	F75M12270
9065	"J" Box, S-IVB FWD Foamflex (260')	F75M12270
9068	Power Panel ("A" Suffixes)	F75M05121
9069	Lighting Panel ("A" Suffixes)	F75M05121
9070	Term. Rm. Instr & Control Distr.	F75M05121
9072	Rm. 1-A Instr & Control Distr.	
9073	Rm. 7-B Power Term Distr.	F75M05121
9074	MSC PWR Distr Rm. 1-A	F75M12270
9075	Anem & Film Camera Distr 9-B	F75M12270
9076	Coax Cable, Meas, 10B Rack 42	
9077	Pair Cables, Meas, 10B Rack 43	
9078	S-IC Eng. Ign. Term. Distr, Rm. 7-B	
9081	Rm. 8-A Instr & Control Distr.	
9082	Rm. 7-A Instr & Control Distr.	F75M05121
9083	Firing Distributor	F75M05121
9084	ML Water Control Cabinet	F75M05121
9085	Prop Loading AC PWR "J" Box, Rm. 7-A	
9086	Inst Gnd Plates, MSC, Rm. 1-A	
9087	Computer Interface, Rm. 10-A	
9088	Computer Interface, Rm. 7-A	
9089	Computer Interface, Rm. 7-A	
9090	No. 1 Tail Service Mast Instr & Control	
9091	No. 1 Tail Service Mast Instr & Control, Rm. 13-B	
	Term Distr Rm. 13-A	
9092	No. 2 Tail Service Mast Instr & Control Term Distr	
	Rm. 5-A	
9093	S-IC AFT PWR. Distr No. 1 Rm. 13-A	
9094	S-IC AFT PWR. Distr No. 2 Rm. 5-A	
9095	Prop Loading AC PWR "J" Box, Rm. 4-A	
9099	PTCR-LUT Instr & Control Interface	
	* Electrical Installation Drawing	

5000-5049	ENVIRONMENTAL CONTROL SYSTEM	6380-6399	380'
5050	LH <sub>2</sub> STORAGE AREA	6360-6379	360'
5051	LOX STORAGE AREA	6340-6359	340'
5052	RP-1 STORAGE AREA	6320-6339	320'
5053-5054	(FUTURE)	6300-6319	300'
5055-5079	PNEUMATICS SYSTEM (CCF)	6280-6299	280'
5080-5099	PAD & PERIMETER AREA	6260-6279	260'
5100-5299	PTCR (1st & 2nd FLOOR)	6240-6259	240'
5300-5999	LCC (FIRING ROOMS 1 THRU 4)	6220-6239	220'
		6200-6219	200'
6900-6999	(FUTURE)	6180-6199	180'
7000-7199	ARMING TOWER	6160-6179	160'
7200-7499	VAB	6140-6159	140'
7500-7549	MOBILE LAUNCHER ERECT. & REFURB.	6120-6139	120'
7550-7599	CRAWLER - TRANSPORTER	6100-6119	100'
7600-7799	PORTABLE TEST EQUIPMENT	6080-6099	80'
7800-7814	PNEUMATICS STORAGE BATTERY (VAB)	6060-6079	60'
7815-7829	PNEUMATICS STORAGE BATTERY (PAD)	6040-6059	40'
7830-8999	(FUTURE)	6020-6029	20'
		LEVEL "D" 6600-6899	
9000-9099	MOBILE LAUNCHER TERM. DISTR.	LEVEL "E" 6400-6599	
9100-9299	LCC TERMINAL DISTRIBUTORS		
9300-9499	VAB TERMINAL DISTRIBUTORS		
9500-9699	PTCR TERMINAL DISTRIBUTORS		
9700-9799	ARMING TOWER TERM. DISTR.		
9800-9899	OTHER LAUNCH AREA TERM. DISTR.		
9900-9999	(FUTURE)		

Figure 6-7 Reference Designation Assignment Areas, LC-39

## APPENDIX A

### GLOSSARY

C/M	Command Module, Apollo spacecraft
DDAS	Digital Data Acquisition System
ESE	Electrical Support Equipment (ESE) which is responsibility of Launch Equipment Branch
DLTR	Digital Link Transmission Repeater
Firing Batteries	See Service Arms Firing Circuits, par. 3-25
GETS	Ground Equipment Test Set
GN <sub>2</sub>	Gaseous Nitrogen
GOX	Gaseous Oxygen
GSE	Ground Support Equipment
He	Helium
H. P. Gas	High Pressure Gas
Inflight Arms	Service Arms which retract during launch
KSC	Kennedy Space Center
LCC	Launch Control Center
LC-39	Launch Complex 39
LEM	Lunar Excursion Module
LES	Launch Escape System
LH <sub>2</sub>	Liquid Hydrogen
LOX	Liquid Oxygen
MI Distributor	Terminating Mineral Insulated Cable
MI Cable	Mineral Insulated Cable
MSFC	Marshall Space Flight Center
MUTS	Miniaturized Universal Test Set

## GLOSSARY (Continued)

<b>Preflight Arms</b>	<b>Service Arms which retract prior to launch</b>
<b>Q-Ball</b>	<b>A device on the vehicle nose for measuring temperature or aerodynamic pressure</b>
<b>R. F. I.</b>	<b>Radio Frequency Interference</b>
<b>SCR</b>	<b>Silicon Control Rectifier</b>
<b>S-IC</b>	<b>Saturn V first stage</b>
<b>S-II</b>	<b>Saturn V second stage</b>
<b>S-IVB</b>	<b>Saturn V third stage</b>
<b>SM</b>	<b>Service Module, Apollo spacecraft</b>
<b>TCS</b>	<b>Terminal Countdown Sequencer</b>
<b>TSM</b>	<b>Tail Service Mast</b>
<b>UB</b>	<b>Umbilical</b>
<b>VAB</b>	<b>Vehicle Assembly Building</b>